

# **Hypersonic Shock/Boundary-Layer Interaction Database**

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# Introduction

In modern hypersonic projects such as the National Aerospace Plane (NASP), it has been recognized from the outset that Computational Fluid Dynamics (CFD) must play a major role. Indeed, the future of high-speed air and space transportation depends critically on our ability to predict solutions of those aerothermal problems which are too difficult or expensive to test in ground-based simulation facilities. Great strides have recently been made in the ability of CFD methods to do this, but it is clear that we still have a long way to go.

While not the only problem obstructing further advances in CFD, turbulence modeling is generally recognized to be the major one. A closed solution of the governing Navier-Stokes equations for turbulent flows of practical consequence is still far beyond our grasp. At the same time, the simplified models of turbulence which are used to achieve closure of the Navier-Stokes equations are known to be rigorously incorrect. While these models serve a definite purpose, they are inadequate for the general prediction of hypersonic viscous/inviscid interactions, mixing problems, transition, chemical nonequilibria, and a range of other phenomena which we must be able to predict in order to design a hypersonic vehicle computationally. For this reason turbulence modeling is a key issue in NASA's continued efforts to push forward the boundaries of knowledge of high-speed flight.

Due to the complexity of turbulence, useful new turbulence models are synthesized only when great expertise is brought to bear and considerable intellectual energy is expended. Although this process is fundamentally theoretical, crucial guidance may be gained from carefully-executed basic experiments. Following the birth of a new model, its testing and validation once again demand comparisons with data of unimpeachable quality. This report concerns these issues which arise from the experimental aspect of hypersonic turbulence modeling.

Prior to about 1970, hypersonics was a subject of considerable research in the USA and abroad. However, for a variety of social, economic, and political reasons, such research all but ceased in the USA for 15 years. This "gap" in hypersonic technology now hampers the NASP program and all other modern efforts related to hypersonic flight.

Further, during the "hypersonic gap" years, computer and laser technologies relevant to fluid-dynamic instrumentation matured considerably. Given these improvements, it is now possible to carry out far more meaningful and detailed hypersonic experiments than those of the pre-1970 period. Such efforts have begun, albeit slowly due to the difficulty of re-invigorating a line of experimental research which has lain dormant for 15 years. Thus, at the time of this writing, hypersonics suffers the following duality: "classical" hypersonics, including hypersonic flow theory and old-style data, and "modern" hypersonics, which is almost entirely computational.

With this background, a High-Speed Turbulence Modeling Workshop was held at NASA-Ames Research Center during June 7-8, 1988. This workshop had the goal of identifying ways to improve turbulence modeling for hypersonic flows, with specific applicability to the NASP Program. Both theoretical and experimental issues were discussed in detail.

In the course of this discussion, questions arose about the quantity and quality of *existing* experimental data which bear upon the issue of hypersonic turbulence modeling. Specifically, it was pointed out that existing surveys of high-speed flows (eg Ref. 1) list several hundred experiments which have been carried out at hypersonic speeds. However, some attendees of the Workshop questioned whether or not any significant number of these existing experiments could meet the high standards necessary for CFD code validation.

Since this issue could not be resolved straightforwardly at the time, one of the conclusions of the Workshop was that the need existed to review critically the database of existing hypersonic experiments for its suitability to turbulence modeling and code validation. Accordingly, an effort was begun early in 1989 at the Penn State University Gas Dynamics Laboratory to perform this critical review and to assemble the required database. The effort was sponsored by the NASP Program through NASA-Ames Research Center, and is a part of an ongoing overall task to develop compressible turbulence models. This report represents the result of the first phase of that effort.

## Database Subject Areas

In choosing the specific subject areas for this hypersonic database collection and assessment effort, some caution was exercised in favor of a few critical issues directly relevant to turbulence modeling. Our purpose in this effort is to define a database for the specific goal of the advancement of modern turbulence models, not to conduct a broad-based survey of all previous work in the field of hypersonics.

Accordingly, discussions with NASA personnel have led to the following list of specific topics for the database:

- 1) shock wave/boundary-layer interactions
- 2) supersonic shear layer mixing
- 3) high-speed attached boundary layers with pressure gradients

The first-year database collection and assessment effort has considered only topic 1) above. A brief justification of this choice follows.

Shock/boundary layer interactions are recognized as the premier pacing issue of modern CFD and turbulence modeling for high-speed flows. The reasons for this lie in the prevalence of shock/boundary layer interaction problems in both external and internal practical aerodynamics and the fundamental difficulty of such problems. For a practical example, the interaction of shock waves with boundary layers underlies the efficiency (if not the viability) of all high-speed inlets for airbreathing propulsion. For a basic example, note that these flows embody mixed hyperbolic and elliptic flow domains with boundaries not known a priori, and that the problem of turbulent boundary-layer separation (not even solved in incompressible flow) is included in shock/boundary layer interactions. For these reasons, the most advanced CFD codes have traditionally been tested against such interactions, though very little has been done so far to test CFD methods or validate codes against hypersonic shock/boundary layer interactions.

Specifically, the coverage of the present database collection and assessment effort with respect to shock/boundary layer interactions includes both supersonic ( $M \sim 3$

and above) and hypersonic data, both two-dimensional (2-D) and three-dimensional (3-D) data, and both unseparated and separated turbulent boundary layer cases (though the emphasis is on the latter). Consideration also includes not only perfect-gas behavior, but real gases and (where appropriate) chemically-reacting flows as well. It is recognized, however, that very little data of the latter two types exist within the chosen subject area.

## **Database Collection**

Our philosophy of collecting the necessary data for this study hinges around the following four strategies:

- 1) Take full advantage of pre-existing database reviews, surveys, and compilations.
- 2) Conduct machine searches to identify likely candidate studies cited in the literature.
- 3) Make use of NASA, NTIS, DTIC, AIAA, and other technical library resources to obtain data reports as necessary.
- 4) Contact investigators, both former and current, as necessary to obtain sufficient documentation of prime candidate studies.

During the initial phase of this effort we have studied a variety of prior reviews and surveys on shock/boundary-layer interactions and related subjects (eg Refs. 1-19). The library holdings of the Penn State Gas Dynamics Laboratory, which has a long-term research effort on this topic, were also thoroughly reviewed. However, the major data collection effort took the form of computerized literature searches.

We have searched the AIAA Aerospace Database, which comprises file 108 of the Dialog computerized database system. The Aerospace Database covers publications and reports since 1962 on aerospace-related subjects, and includes both International Aerospace Abstracts, compiled by the AIAA, and Scientific and Technical Aerospace Reports, compiled by NASA. Considering the strong aerospace flavor of the present subject matter, it was felt that this database was an obvious choice and that searches of other science and



engineering databases would be unlikely to turn up significant additional material of relevance.

Before beginning this search process, the NASA Thesaurus (Ref. 20) was consulted for appropriate keywords. A group of obvious references to be included in the database were called up from the Aerospace Database to determine which keywords were used. As it happens, there is no single keyword entry in use for "shock wave/boundary-layer interaction." Instead, the keywords SHOCK WAVE INTERACTION and INTERACTIONAL AERODYNAMICS are most prevalent. At the time the search was conducted, the Aerospace Database contained 3,379 references ("Set 1") indexed by one or the other of these two keyword phrases. Examination of a random sampling of these revealed a low percentage of useful entries for present purposes.

Our next step was to search for citations with one or more of the keywords BOUNDARY LAYER, TURBULENT BOUNDARY LAYER, SUPERSONIC BOUNDARY LAYER, or HYPERSONIC BOUNDARY LAYER. This subset ("Set 2") of the Aerospace Database contained 27,122 citations. The intersection of Sets 1 and 2 (998 citations) is thus the set ("Set 3") described as "shock wave/boundary-layer interactions," at least insofar as keyword descriptors are concerned. However, examination of a random sampling from Set 3 still revealed inappropriate citations for present purposes. It seems that the combination of keywords used so far is necessary but not sufficient to fully characterize the literature citations which we sought.

We next decided to narrow the range of consideration still further by requiring that, in addition to the above keywords, descriptors related to shock/boundary-layer interactions must also appear in the title or abstract of the citation. A long list of such possible descriptors was compiled and linked by Boolean "or" terms, such that the presence of any one of them would produce a "hit." Upon searching titles and abstracts of the Aerospace Database for this list, a set ("Set 4") of 815 citations was found. The intersection of Sets 3 and 4 resulted in 436 citations ("Set 5").

Examination of a random sampling of citations from Set 5 now revealed a high incidence of what appeared to be pertinent references. One final step was taken to narrow this list still further by excluding those citations in which the keywords LAMINAR,

TRANSONIC, and COMPUTATIONAL FLUID DYNAMICS appeared. This was done because these three descriptors typically characterize studies which are not pertinent for present purposes. The result of this operation on Set 5 was Set 6, containing 279 citations.

Every citation in Set 6 was scanned by abstract in order to determine its relevance. This process depended heavily on our background and experience in shock/boundary-layer interaction research in order to identify likely candidate experiments. In all cases for which a decision could not be reached from the abstract alone, a hard copy of the full document was obtained and scanned. The result of this process was the final set ("Set 7"), which consisted of 105 distinct experimental studies of shock wave interactions with turbulent boundary layers at Mach numbers of 3 or higher. Set 7 was subjected to the database assessment procedure described below.

## **Database Assessment**

This was the critical step of the study, in which the decision was made as to which of the possible candidate experiments identified above actually merit inclusion in a database to be put forth as a standard for CFD code validation and turbulence model development. Our philosophy at this stage was that we were looking only for those few experimental studies of unimpeachable quality and direct pertinence to the subject at hand. It would be a mistake to give benefit of doubt in such an assessment if that doubt might cause future turbulence modeling efforts to be misled. Also, we drew guidance from a distinguished predecessor at this task (Ref. 21), who noted that those data turn out to be most useful in which only one factor is varied at a time, and that there appears to be a certain level above which measurements can be described as being of "professional quality."

We have subjected the 105 candidate studies of Set 7 above to a test based on rigorous criteria for this purpose. The criteria are grouped in two categories: "necessary" and "desirable." Candidate experiments were required to pass all the "necessary" criteria in order to be considered further. However, even then, failure to meet any of the "desirable" criteria might result in rejection of a candidate experiment for the database, on the basis that it fails to contribute anything truly useful to the goal of the database.

It is recognized, considering that many of the candidate studies are 15 years old or older, that the rigorous application of modern code validation criteria would eliminate most or all of them. Accordingly, a second category was created to include "qualified" experiments, *ie*, the best of those which do not meet all the necessary criteria but still retain some value for code validation and turbulence modeling.

Our list of the 8 necessary criteria we applied is given below, in the hierarchical order in which they were applied.

### **1) BASELINE APPLICABILITY**

All candidate studies must be experiments involving turbulent flows in either of the supersonic or hypersonic Mach number ranges (*ie*  $M \sim 3$  or higher). Further, these studies must address the subject area of shock wave/boundary-layer interactions.

### **2) SIMPLICITY**

All candidate studies passing this criterion must involve experimental geometries sufficiently simple that they may be modeled by CFD methods without enormous difficulty. Flows through complex inlet scale-models or over the surfaces of complete 3-D flight configurations are rejected at this point, for example. Stated in other words, this criterion is a filter which passes only "building-block" experiments.

### **3) SPECIFIC APPLICABILITY**

All candidate studies passing this criterion must be capable of providing some useful test of turbulence modeling. For example, any study which provides only a surface pressure distribution over an arbitrary surface in hypersonic flow is rejected as insufficient to further the goals of turbulence modeling. To be a useful test case, such a study would at least require additional data such as flowfield profiles or heat transfer/skin friction distributions. (Some experienced judgement was called for in the application of this criterion.)

#### **4) WELL-DEFINED EXPERIMENTAL BOUNDARY CONDITIONS**

This criterion was applied in a sense similar to that of CFD studies, where a rational solution cannot be had if the boundary conditions of the problem are inadequately defined. For high-speed experiments, this criterion requires at least that all incoming conditions (especially the state of the incoming boundary-layer) be carefully documented. For turbulent incoming boundary-layers, either known upstream transition conditions (to allow a boundary-layer calculation to be made) or else the documentation of both the mean and fluctuating character of the incoming profile must be provided. Similarly, all studies claiming "2-D" flow must show data which establish the extent of spanwise flow variations.

We recognized at the outset that this criterion alone might eliminate a large proportion of all past hypersonic studies from further consideration. However, without it, the resulting database would fail to be useful for its intended purpose.

#### **5) WELL-DEFINED EXPERIMENTAL ERROR BOUNDS**

To pass this criterion, the experimenter him/her/self must have provided an analysis of the accuracy and repeatability of the data, or error bands on the data themselves. Further, such accuracy indications or error bounds must be substantiated in some rational way beyond their mere statement. (Without this criterion, a proper code validation exercise cannot be conducted with the subject data.)

#### **6) CONSISTENCY CRITERION**

If, during the consideration of a candidate study, mutually inconsistent results are discovered, said study was eliminated from further consideration for the database. This criterion amounts to a special corollary of the previous criterion.

#### **7) ADEQUATE DOCUMENTATION OF DATA**

Candidate studies were examined to determine whether or not their data are documented sufficiently to allow quantitative results to be included in the database in tabulated and machine-readable form. Those failing this criterion were eliminated. This criterion applied in particular to studies whose documentation was available only in plotted

form. If such plots were quantitatively unreadable within reasonable error bounds as mentioned above (taking note of the well-known scale distortions which often occur during publication), then the data cannot be considered useful for the stated purpose of the database. (Coles, Ref. 21, read data from plots in only two cases; we have not done so at all in the present study.)

## **8) ADEQUATE SPATIAL RESOLUTION OF DATA**

To pass this criterion, experiments must present data of sufficiently high resolution, compared with the scale of the flow in question, that the key features of the flow are clearly resolved. Failure to do so results in data which are inadequate to provide a proper example or test for turbulence modeling.

In addition to the above-listed "necessary" criteria, the following "desirable" criteria also had an influence on which candidate experiments were finally included in the database:

### **1) TURBULENCE DATA**

In addition to purely mean-flow measurements, data on fluctuating quantities such as Reynolds stresses and velocity or mass-flux fluctuations were considered highly desirable.

### **2) REALISTIC TEST CONDITIONS**

Of those flows passing the necessary criteria, special preference was given to cases with Mach numbers in the hypersonic range, non-adiabatic wall conditions, real-gas effects, or related characteristics typical of actual hypersonic flight.

### **3) NON-INTRUSIVE INSTRUMENTATION**

All other conditions being equal, preference was given to experiments wherein non-intrusive instrumentation (eg optical measurements) were employed to acquire the data. This preference is based on the automatic satisfaction of related error and boundary-condition concerns which occurs when non-intrusive measurements are made.

#### **4) REDUNDANT MEASUREMENTS**

Further preference was given to experiments in which redundant data were taken in order to establish the values of flow quantities by more than one method. This is considered to be a strong demonstration of the quality and error bounds of the data.

#### **5) FLOW STRUCTURE AND PHYSICS**

Finally, preference was also given to those experiments which, in addition to quantitative data, also reveal flow structures and physical mechanisms. The philosophy of this criterion was to allow higher-level CFD comparisons with the salient characteristics of the flows in question, rather than merely with unstructured flow profiles.

The 105 individual studies subjected to the above criteria are listed by bibliographic citation in Appendix A. Each of these studies was given a detailed assessment through examination of its source material. During the evaluation procedure a tabular evaluation matrix was kept of decisions in each assessment category for every studied considered. A form of this table is given as Appendix B. Other than an indication of whether or not each candidate met the criteria indicated, additional notes pertinent to the assessment are provided in some cases.

## Results and Conclusions

As shown in the evaluation tables of Appendix B, only a few of the "finalists" in Database Collection Set 7 passed a sufficient number of the assessment criteria to be accepted into the database. Of these, we observed the trend that far more acceptable studies fell in the supersonic Mach number range (Mach 3 to 5) than in the hypersonic range (above Mach 5). We have therefore split the Category I (accepted) studies into groups A (hypersonic) and B (supersonic).

Since the above result became clear early in the assessment phase, somewhat tougher standards were applied to supersonic than to hypersonic studies. Along the same lines, the paucity of true hypersonic data resulted in all such studies being accepted which at least met the 8 "necessary" criteria. In other words, we were in no position to be "choosy" where category IA was concerned.

The studies accepted in categories IA and IB are listed in the Tables below and tabulated data for each of them are included in Appendix C. Category II (limited acceptance) status was also extended to the axisymmetric cylinder-flare experiment of Coleman (Database References 16 and 17) and the 2-D compression corner data of Holden (Database References 41 and 42). These data have some value for code validation and turbulence modeling even though incoming boundary-layer conditions were not reported. However, we have made no attempt to tabulate data for these Category II studies.

A single *general* conclusion may be drawn from this study: high-quality data on hypersonic shock/boundary-layer interactions, suitable for use in turbulence modeling efforts, are *extremely scarce*. The existing data do not begin to satisfy the current need. Thus the authors strongly suggest that new, detailed, carefully-planned experiments be funded and carried out. Suggestions for these experiments are listed in the next section. In particular, no useful real-gas data were found in the current database assessment.

Finally, only one accepted dataset in Category I qualifies as a "new discovery" in the sense that it was not previously available in some form, *ie*, the supersonic compression-

corner data of Zheltovodov *et al.* The availability of these detailed data from the USSR is a sign of recently-renewed cooperation between US and Soviet researchers in the field of hypersonics.



## Category IA: Accepted Experiments, Hypersonic

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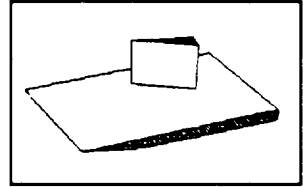
Ref.: 56, 15

Author: Law, C. H.

Geometry: 3-D Fin

Mach number: 6

Data:  $p_{wall}$ ,  $c_h$



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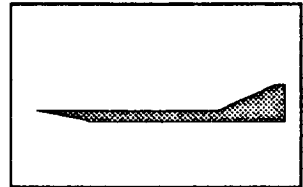
Ref.: 16, 17

Author: Coleman, G. T.

Geometry: 2-D Compression Corner

Mach number: 9

Data:  $p_{wall}$ ,  $c_h$



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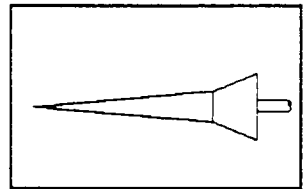
Ref.: 40, 39

Author: Holden, M. S.

Geometry: Axisymmetric Cone-Flare

Mach number: 11, 13

Data:  $p_{wall}$ ,  $c_h$



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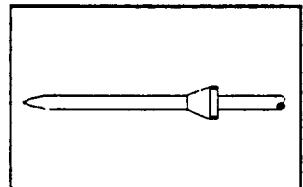
Ref.: 51

Author: Kussoy, M. I., *et al*

Geometry: Axisymmetric Ogive-Cylinder-Flare

Mach number: 7

Data:  $p_{wall}$ ,  $c_h$ , flowfield surveys



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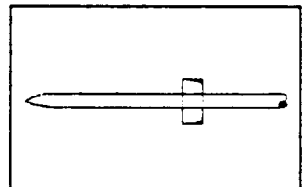
Ref.: 53

Author: Kussoy, M. I., *et al*

Geometry: Axisymmetric Impinging Shock

Mach number: 7

Data:  $p_{wall}$ ,  $c_h$ ,  $c_f$ , flowfield surveys



## Category IB: Accepted Experiments, Supersonic

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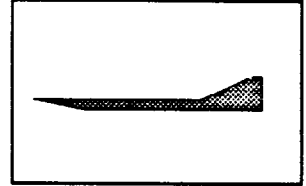
Ref.: 88, 29

Author: Smits, A. J., *et al*

Geometry: 2-D Compression Corner

Mach number: 3

Data:  $p_{wall}$ ,  $c_f$ , mean & fluctuating flowfield surveys (pitot and hot-wire anemometry)



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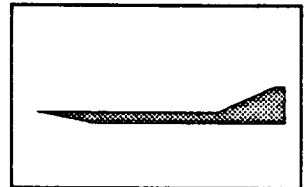
Ref.: 103, private communication

Author: Zheltovodov, A. A., *et al*

Geometry: 2-D Compression Corner

Mach number: 3

Data:  $p_{wall}$ ,  $c_h$ , mean and fluctuating flowfield surveys (pitot and hot-wire anemometry)



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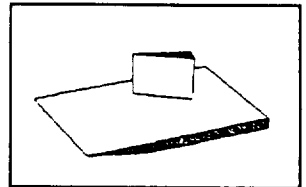
Ref.: 49

Author: Bogdonoff, S. M., *et al*

Geometry: 3-D Fin

Mach number: 3

Data:  $p_{wall}$ , mean flowfield surveys ("cobra" probe)



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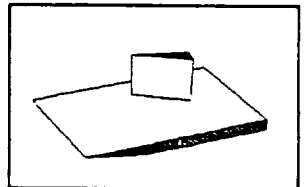
Ref.: 46

Author: Kim, K-S, *et al*

Geometry: 3-D Fin

Mach number: 3, 4

Data:  $p_{wall}$ ,  $c_f$ , surface-flow angles



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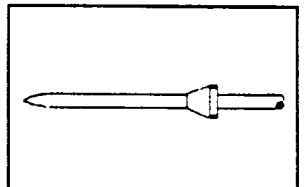
Ref.: 13, 27

Author: Dunagan, S. E., Brown, J. L., *et al*

Geometry: Axisymmetric Ogive-Cylinder-Flare

Mach number: 3

Data:  $p_{wall}$ , flowfield surveys (LDV and holographic interferometry)



## Category IB: Accepted Experiments, Supersonic (Concluded)

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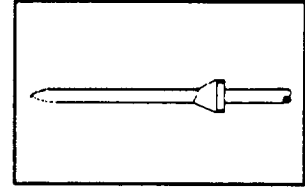
Ref.: 12

Author: Brown, J. D., *et al*

Geometry: Axisymmetric Ogive-Cylinder with Skewed Flare

Mach number: 3

Data:  $p_{\text{wall}}$ , flowfield surveys (LDV)



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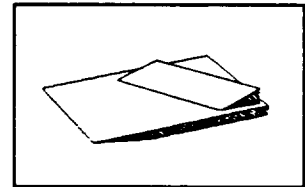
Ref.: 85

Author: McKenzie, T. M., *et al*

Geometry: 3-D Swept Compression Corner

Mach number: 3

Data:  $p_{\text{wall}}$ , mean flowfield surveys ("cobra" probe)



## Need for Further Experimentation

Based upon the results of this study, the following list conveys our recommendations for further experimentation in hypersonic shock/boundary-layer interactions.

- 1) Interactions involving real-gas effects
- 2) Flowfield turbulence data
- 3) One or more high-quality hypersonic *laminar* boundary-layer experiments for comparison purposes
- 4) Non-intrusive flowfield data (mean as well as fluctuating)
- 5) More complex types of "building-block" interactions, such as the double-fin or crossing-shock-type interaction
- 6) Emphasis on 3-D rather than 2-D interactions

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## Appendix B: Database Assessment

The following Table lists the 105 Database References which were subjected to an evaluation based on the criteria described earlier in the Database Assessment section. The Reference number given in the first column corresponds to the list of numbered Database References in Appendix A. The Table includes a brief annotation of test geometry and Mach number for each entry, followed by an evaluation field for each of 7 necessary and 5 desirable criteria as discussed earlier. Note that, by definition, each of the 105 evaluated studies met the first necessary criterion of baseline applicability. The remaining twelve criteria are indicated in the table only by numbers, the meaning of which is as follows:

### NECESSARY CRITERIA

- #2 Simplicity
- #3 Specific Applicability
- #4 Well-Defined Experimental Boundary Conditions
- #5 Well-Defined Experimental Error Bounds
- #6 Consistency Criterion
- #7 Adequate Documentation of Data
- #8 Adequate Spatial Resolution of Data

### DESIRABLE CRITERIA

- #1 Turbulence Data
- #2 Realistic Test Conditions
- #3 Non-Intrusive Instrumentation
- #4 Redundant Measurements
- #5 Flow Structure and Physics

Each of these evaluation fields contains one of three symbols:

- ✓ Acceptable
- X Not Acceptable
- ? No Determination Made

The question-mark symbol indicates that the necessary information to evaluate that category was lacking. In some cases, question marks or blanks in one or more evaluation fields also may indicate that the evaluation was terminated after the candidate study failed one or more of the "necessary" criteria. Finally, a "comments" field provides additional information on the candidate study and its evaluation.

The abbreviations used in the TEST GEOMETRY Field of the Table are as follows:

Axi.	axisymmetric
CC	compression corner
Cyl	cylinder
FF	forward-facing
IS	incident shock wave
LE	leading edge
SC	semicone or half-cone
SCC	swept compression corner
2-D	two-dimensional
3-D	three-dimensional
$\alpha$	angle of attack

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
1	3	2-D CC	✓	X	✓	?	✓	X	✓	X	X	✓	X	X	X
			Wall-pressure fluctuation data only												
2	12	2-D CC	✓	✓	X	X	✓	?	✓	X	✓	✓	X	X	X
			No incoming boundary-layer profile; boundary-layer heavily tripped												
3	15	2-D CC	✓	✓							✓	✓		X	X
			Same comments as above												
4	5	Blunt LE	✓	X	✓	?	✓	✓	✓	X	✓	✓	X	X	X
			Wall-pressure fluctuation data only												
5	3.5	2-D IS	✓	✓	✓	?	?	?	✓	X		X	X	✓	X
			Non-uniform conical nozzle flow; strange incoming boundary-layer												
6	3.7	Axi. Duct	✓	✓	X	X	✓	X	✓	X	X	✓	X	X	X
			Boundary conditions of flow not well-defined												
7	3	3-D Sharp Fin	✓	?	✓	?	?	✓	✓	X	X	X	X	✓	X
			Unseparated interaction, possibly too weak to be a useful test of turbulence modeling												
8	3	3-D Sharp Fin	✓	X	✓	?	?	✓	✓	X	X	X	X	X	X
			Surface data only												
9	3	Fin, SC, SCC	✓	✓	✓	?	✓	X	✓	X	X	X	X	✓	X
			Flowfield surveys of similar intns. due to different shock generators; data not avail.												
10	3	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	✓	✓
			Secondary reference for Princeton Mach 3 $\alpha = 20^\circ$ fin interaction (see Ref. 49)												

			NECESSARY CRITERIA								DESIRABLE CRITERIA				
REF NO	MACH NO.	TEST GEOMETRY	#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
11	3	3-D Flare	✓	✓	✓	✓	?	✓	✓	?	X	✓	X	X	X
Turbulence and shock-motion effects could not be separated															
12	3	3-D Flare	✓	✓	✓	✓	✓	✓	✓	✓	X	✓	✓	✓	✓
Primary reference for NASA-Ames Mach 3 skewed flare interaction															
13	3	Axi. Flare	✓	✓	✓	✓	✓	✓	✓	✓	X	✓	✓	✓	✓
Primary reference for NASA-Ames Mach 3 axisymmetric flare interaction (LDV)															
14	4	Cone at $\alpha$ in Axi. Channel	✓	?	X	X	?	X	✓	X	X	X	X	✓	X
The level of this experiment is not up to current code-validation standards															
15	5.85	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	X	✓
Secondary reference for ARL/AFFDL Mach 6 fin interaction (see Ref. 56)															
16	9	2-D CC, Axi. Cyl/Flare	✓	✓	✓	✓	✓	✓	✓	X	✓	X	X	X	✓
No incoming boundary-layer profile for the Axi Cyl/Flare.															
17	9	2-D CC	✓	✓	✓	✓	✓	✓	✓	X	✓	X	X	X	✓
Primary reference for Coleman/Stollery 2-D CC; could have 3-D flow @ largest CC angles															
18	5, 6.5, 9.8	Axi. IS, 3-D Fin	✓	✓	X	X	X	X	?	X	✓	✓	X	X	X
The level of this experiment is not up to current code-validation standards															
19	5, 5.5, 6	3-D Sharp & Blunt Fin	✓	✓	X	X	?	✓	✓	X	X	X	X	X	X
No incoming boundary-layer profile															
20	5	3-D Blunt LE	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure fluctuation data only															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
21	4.9	3-D Blunt Fin	✓	X	✓	?	?	✓	✓	X	X	X	X	X	X
Upstream-influence data only															
22	3	3-D Blunt Fin	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure data only															
23	3	3-D Blunt Fin	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Fluctuating wall-pressure data only															
24	3	3-D Blunt Fin	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Mean surface-pressure distributions only															
25	3	3-D Blunt Fin	✓	X	✓	✓	✓	✓	✓	X	X	X	X	X	X
Wall-pressure distributions only															
26	3	3-D Blunt Fin	✓	X	✓	✓	✓	✓	✓	X	X	X	X	X	X
Wall-pressure distributions only															
27	3	Axi Flare	✓	✓	✓	✓	✓	✓	✓	X	X	✓	✓	✓	✓
Interferometric density data supplementing LDV data on same geometry (see Ref. 13)															
28	7-9	2D CC	✓	✓	✓	?	✓	✓	✓	X	✓	✓	X	X	✓
Secondary ref. for Coleman data (see Ref. 16)															
29	3	2D CC	✓	✓	✓	✓	✓	✓	✓	✓	X	X	✓	✓	✓
Primary reference for Princeton 2-D CC hot-wire data (see also Ref. 84)															
30	2-4	Axi. Flare	✓	✓	✓	?	?	X	✓	X	X	X	X	✓	X
Data not available															
31	2-6	2-D CC	✓	✓	?	X	?	X	✓	X	✓	X	X	X	X
Data not available															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
32	3	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	X	✓
Supplementary wall-pressure data for Princeton Mach 3 Fin (see Ref. 49)															
33	5	Blunt Cyl 3-D	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Fluctuating wall-pressure data only															
34	6, 8, 10	2-D IS	✓	✓	X	X	X	?	✓	X	X	X	X	X	X
The level of this experiment is not up to current code-validation standards															
35	4	2-D IS	✓	✓	X	X	?	X	✓	X	X	✓	X	X	X
Boundary-layer not defined; evidence of transitional boundary-layer															
36	4	2-D IS	✓	✓	X	X	?	X	✓	X	X	✓	X	X	X
Wall-pressure fluctuation data only; see comments above															
37	4	2-D IS	✓	✓	X	X	?	X	✓	X	X	✓	X	X	X
See comments for Ref. 35															
38	11, 13	Axisymmetric Cone/Flare	✓	✓	✓	✓	✓	✓	✓	X	✓	✓	X	X	✓
Supplementary reference to Ref. 40 below															
39	11, 13	Axisymmetric Cone/Flare	✓	✓	✓	✓	✓	✓	✓	X	✓	✓	X	X	✓
Supplementary reference to Ref. 40 below															
40	11	Axi Cone/Flare; 3-D Fin, IS	✓	✓	✓	✓	✓	✓	✓	X	✓	X	X	X	✓
Primary ref. for Holden Axisymmetric Cone/Flare.															
41	8	2-D CC IS	✓	✓	?	X	X	✓	✓	X	✓	X	X	X	X
IS model dimensions not defined; boundary-layer not defined.															
42	6-15	2-D CC IS	✓	✓	?	X	X	✓	✓	X	✓	X	X	X	X
IS model dimensions not defined; boundary-layer not defined.															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
43	6	2-D CC, FF Step	✓	✓	X	X	?	?	✓	X	X	X	X	X	X
		Incoming boundary-layer not defined													
44	7-11	2-D CC, etc.	✓	X											X
		No new data													
45	3	2-D CC	✓	X	✓	✓	✓	✓	✓	X	X	X	X	X	X
		Upstream-influence and wall-pressure data only													
46	3, 4	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	✓	X	X	✓
		Features non-intrusive optical skin friction data													
47	3	SCC, 3-D Sharp Fin, SC	✓	✓	✓	✓	✓	X	✓	X	X	X	X	✓	X
		Data not available													
48	3	SCC	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	✓	✓
		Secondary ref. for Princeton Mach 3 - 24-40 SCC													
49	3	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	✓	✓
		Primary ref. for Princeton Mach 3 $\alpha = 20^\circ$ fin (see Ref. 10)													
50	3	2-D CC	✓	✓	?	?	?	✓	✓	✓	X	✓	X	✓	X
		Definition of boundary conditions not clear; LDA data may be inconsistent													
51	7	2-D Flare	✓	✓	✓	✓	✓	✓	✓	X	✓	X	✓	✓	✓
		Features flowfield data and redundant surface data													
52	2.9	3-D Flare	✓	✓	✓	✓	✓	✓	✓	✓	X	✓	✓	✓	✓
		Secondary reference for NASA-Ames skewed flare experiment (see Ref. 12)													
53	7.2	Axi IS	✓	✓	✓	✓	✓	✓	✓	X	✓	X	X	✓	✓
		Includes both flowfield and surface data													



REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
54	7.5, 10.5	2-D IS	✓	X	?	X	?	?	X	X	✓	X	X	✓	X
Boundary-layer possibly transitional; only upstream and downstream surveys taken															
55	3	2-D IS, CC	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure distributions only															
56	6	3-D Sharp Fin	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	X	✓
Primary reference for ARL/AFFDL, Mach 6 fin interaction (see Ref. 15)															
57	?	Axi Cyl/Flare	X												X
Geometry too complex; wall-pressure data only															
58	3	3-D Sharp Fin	✓	X	✓	X	✓	✓	✓	X	X	X	X	X	X
Mainly surface-flow visualization data															
59	3	3-D Sharp Fin	✓	✓	✓	?	✓	✓	✓	X	X	X	X	✓	X
$\alpha = 10^\circ$ interaction too weak to be useful test of turbulence model															
60	3	3-D Sharp Fin	✓	✓	✓	?	✓	✓	✓	X	X	X	X	✓	X
$\alpha = 10^\circ$ interaction too weak to be useful test of turbulence model															
61	7.2	Axi IS	✓	✓	✓	?	?	✓	✓	✓	✓	X	X	✓	X
Consistency and frequency response of "webbed" hot-wires is highly questionable															
62	2.9	2-D IS	X	✓	✓	?	✓	✓	✓	✓	X	✓	✓	✓	X
Flowfield not actually 2-D, and not simple															
63	2.9	2-D CC	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure fluctuation data only															
64	2.9	2-D IS	✓	X	?	?	?	?	✓	X	X	X	X	X	X
Consistency and accuracy of buried-wire heat transfer data questionable															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
65	6, 8, 10	2-D IS 3-D Fin	✓	✓	X	X	✓	✓	✓	X	✓	X	X	X	X
Summary report only; no new data															
66	3	3-D sharp fin	✓	✓	✓	✓	✓	X	✓	X	X	X	X	✓	X
Data not available; $\alpha = 10^\circ$ interaction too weak to be useful test															
67	3	3-D sharp fin	✓	✓	✓	✓	✓	X	✓	X	X	X	X	✓	X
Data not available; $\alpha = 10^\circ$ interaction too weak to be useful test															
68	3	3-D sharp fin	✓	✓	✓	✓	✓	X	✓	X	X	X	X	✓	X
Data not available; $\alpha = 10^\circ$ interaction too weak to be useful test															
69	2-3	3-D IS	✓	X	?	?	?	X	✓	X	X	X	X	X	X
Data not available															
70	2-3	3-D IS	✓	X	?	?	?	X	✓	X	X	X	X	X	X
Data not available															
71	2-3	3-D IS	✓	X	?	?	?	X	✓	X	X	X	X	X	X
Data not available															
72	3	3-D Sharp Fin	✓	X	?	?	✓	✓	✓	X	X	X	X	X	X
Report is primarily concerned with facility effects															
73	2-4.3	2-D IS	✓	✓	✓	?	?	✓	✓	X	X	X	X	✓	X
Level of this experiment is not up to current code-validation standards															
74	5, 6, 7	Axi. Cyl/Flare	✓	X	X	X	?	?	✓	X	X	X	X	X	X
Level of this experiment is not up to current code-validation standards															
75	2.9	2-D IS	X	✓	✓	?	X	?	✓	X	X	X	X	✓	X
Actually a 3-D interaction; too complex for a useful test case															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
76	2.9	2-D IS	X	✓	✓	?	X	?	✓	X	X	X	X	✓	X
Actually a 3-D interaction; too complex for a useful test case															
77	2.9	2-D IS	✓	✓	✓	X	✓	X	✓	✓	X	✓	✓	X	X
The level of this experiment is not up to current code-validation standards															
78	4	Axi. IS	✓	✓	✓	?	X	?	✓	✓	X	X	X	✓	X
Serious questions have arisen about the accuracy of these hot-wire data															
79	2-4.5	Axi. Cyl/Flare	✓	X	✓	?	✓	?	✓	X	X	X	X	X	X
Measurements are primarily wall-pressure distributions															
80	2-4	2-D CC	✓	X	✓	?	✓	?	✓	X	X	X	X	X	X
Measurements are primarily wall-pressure distributions															
81	?	2-D IS	✓					X							X
Source document and data not available															
82	2-5	2-D IS	✓	X	✓	?	✓	X	✓	X	X	X	X	X	X
Results suitable for turbulence modeling extremely limited															
83	2.8, 3.8	Cone In Axi. Channel	✓	X	✓	?	?	?	✓	X	X	X	X	X	X
The level of this experiment is not up to present code-validation standards															
84	3	2-D CC	✓	✓	✓	✓	✓	✓	✓	✓	X	X	X	✓	✓
Secondary reference for Princeton 2-D CC hot-wire data (see Ref. 29)															
85	3	SCC	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	✓	✓
Primary Ref. for Princeton Mach 3 - 24 - 40 SCC															
86	3	SCC	✓	X	✓	X	✓	✓	✓	X	X	X	X	X	X
Surface-pressure and flow visualization data only															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
87	3	2-D CC	✓	✓	✓	✓	✓	✓	✓	X	X	X	X	✓	✓
Report contains mean flowfield data for combination with Refs. 29, 84, and 88															
88	2.9	2-D CC	✓	✓	✓	✓	✓	✓	✓	✓	X	X	X	✓	✓
Secondary ref. for Princeton 2-D CC experiments (see Ref. 29)															
89	2.9	2-D CC	✓	✓	✓	?	✓	?	✓	X	X	X	X	X	X
Mainly surface-pressure data															
90	4.8-6.2	2-D CC	✓	X	?	?	?	?	✓	X	X	X	X	X	X
Primarily surface-pressure data															
91	2, 3, 4	Axi. IS	✓	X	✓	X	?	?	✓	X	X	X	X	✓	X
The level of this experiment does not appear to be up to current code-validation standards															
92	3	SCC	✓	X	✓	✓	✓	✓	✓	X	X	X	X	X	X
Primarily surface flow-visualization data															
93	3.7	3-D Sharp Fin	✓	✓	X	✓	?	✓	✓	X	X	X	X	X	X
No incoming boundary-layer definition															
94	?	IS	X					X							X
Data not available															
95	3	3-D Sharp Fin	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure fluctuation data only															
96	3	3-D Sharp Fin	✓	X	✓	?	✓	✓	✓	X	X	X	X	X	X
Wall-pressure fluctuation data only															
97	?	Blunt LE		X				X							X
Superseded by more recent work of Dolling et al															

REF NO	MACH NO.	TEST GEOMETRY	NECESSARY CRITERIA								DESIRABLE CRITERIA				
			#2	#3	#4	#5	#6	#7	#8	#1	#2	#3	#4	#5	ACCEPT?
98	7.8	FF Step	✓	✓	X	X	?	X	✓	X	✓	✓	X	X	X
	Insufficient documentation														
99	7-10	2-D IS	✓	X	✓	?	?	?	✓	X	✓	X	X	✓	X
	No flowfield data inside interaction; data not available														
100	7.3, 10.4	2-D IS	✓	X	✓	?	?	?	✓	X	✓	X	X	✓	X
	No flowfield data inside interaction; data not available														
101	8.8, 13.5	Axi. Cone/flare	✓	X	X	X	?	?	✓	X	✓	X	X	X	X
	Incoming boundary-layer not defined														
102	3.7, 6.3	2-D CC	✓	✓	X	X	?	?	✓	X	✓	X	X	X	X
	Incoming boundary-layer not defined														
103	2-4	2-D CC	✓	✓	✓	✓	✓	✓	✓	✓	X	X	X	✓	✓
	Features extensive pitot and hot-wire surveys, heat transfer data														
104	3	3-D Sharp Fin	✓	X	?	?	✓	?	✓	X	X	X	X	✓	X
	Mainly surface and flow-visualization data; data not available														
105	3	3-D Blunt Cyl	✓	X	?	?	?	X	✓	X	X	X	X	✓	X
	Mainly wall-pressure distribution data														

## Appendix C: Data Tabulation

There follows a tabulation of pertinent data from the 5 Category IA and 7 Category IB studies which make up the database. For each study a brief discussion of the data is given for the benefit of users of the data. However, users are strongly encouraged to consult the original references for more detail on what was measured and how it was accomplished. Similarly, no attempt has been made to tabulate all available data from each of these studies, but rather only those data most pertinent to the issues of turbulence modeling and code validation. In several cases, additional data may be had from the original publications. A 5.25" double-sided high-density "floppy" disk is also provided in original copies of this report. This disk contains the data-tables of this Appendix in machine-readable ASCII files, formatted for MS-DOS computers. Individual ASCII files are given for each of the 12 Category I datasets with filenames as follows:

### Category IA:

LAW.DAT  
COLEMAN.DAT  
HOLDEN.DAT  
KUSSOY1.DAT  
KUSSOY2.DAT

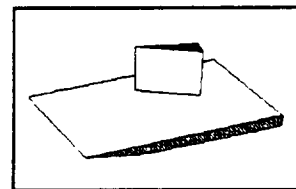
### Category IB:

SMITS.DAT, SETTLES.DAT  
ZHELT.DAT  
BOGDONOF.DAT  
KIM.DAT  
DUNAGAN.DAT  
BROWN.DAT  
MCKENZIE.DAT

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Ref.: 56, 15  
Author: Law, C. H.  
Geometry: 3-D Fin  
Mach number: 6  
Data:  $p_{\text{wall}}$ ,  $c_h$

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Law, C.H., "3-D Shock Wave-Turbulent Boundary Layer Interactions At Mach 6," *ARL TR 75-0191*, 1974.

Christophel, R.G., Rockwell, W.A. and Neumann, R.D., "Tabulated Mach 6 3-D Shock Wave-Turbulent Boundary Layer Interaction Heat Transfer Data," *AFFDL-TM-74-212-FXG and AFFDL-TM-74-212-FXG-Supplement*, 1975.

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The data consist of surface pressures and heat transfer coefficients beneath the swept interaction generated by an oblique shock impinging on a flat-plate turbulent boundary layer. The freestream Mach number was 5.85 for all tests. The coordinate system used is  $x \equiv$  streamwise coordinate and  $y \equiv$  spanwise coordinate, with origin at the top left corner of the flat plate as viewed from above. Coordinates are given in inches.

Surface data were measured along five spanwise rows in the interaction region. Fin angles-of-attack of 6, 8, 10, 12, 16, and 20 degrees were tested, spanning a broad range of interaction strength. These experiments were done at two distinct values of freestream Reynolds number, 10 and 30 million/foot. Natural transition occurred on the flat plate.

Only the higher-Reynolds number data are tabulated here. Of these, only one spanwise row of data is tabulated, generally corresponding to the last row taken (row 5). Since it is now widely accepted that such interactions are quasi-conical in nature, the other data rows are felt to be redundant. Redundant data in terms of repeated test conditions is also contained in the data reports listed above, but not tabulated here.

This experiment suffers two problems worthy of mention. First, although an acceptable boundary-layer profile was measured, it was obtained at the *end* of the flat plate in the absence of the fin, not near the fin leading-edge position. That boundary-layer profile is tabulated here even though it is not a proper incoming-flow boundary condition. This problem is not believed to be a major one, since a boundary-layer code may be used to match the given profile and then provide interpolated profiles at any location.

Secondly, the thin-skin thermocouple heat transfer technique used here is susceptible to errors due to lateral heat conduction in regions of strong skin-temperature gradient. Such strong gradients did occur for the stronger interactions represented here. No corrections for conduction were made, although the experimenter did estimate the worst-case magnitude of the errors at 15% to 25% of the peak heating value. This should be regarded as the accuracy band of the peak heat transfer coefficient.

FREESTREAM MACH NUMBER = 5.85

REFERENCE PRESSURE PREF = 1.432 PSIA

REFERENCE HEAT TRANSFER COEFFICIENT HREF = 0.0108 BTU/(FT\*2-SEC-DEG R)

DEL = FIN ANGLE, DEGREES

XLE = X-LOCATION OF FIN LEADING EDGE, INCHES

YLE = Y-LOCATION OF FIN LEADING-EDGE, INCHES

PO = WIND TUNNEL STAGNATION PRESSURE, PSIA

TO = WIND TUNNEL STAGNATION TEMPERATURE, DEGREES RANKINE

X = X-LOCATION OF SPANWISE INSTRUMENTATION ROW, INCHES

Y = Y-LOCATION OF MEASUREMENT POINT, INCHES

Z = HEIGHT ABOVE FLAT PLATE, INCHES

M = MACH NUMBER

P = WALL STATIC PRESSURE, PSIA

## \*\*\*\*\*BOUNDARY LAYER PROFILE\*\*\*\*\*

Z	M
0.018	2.116
0.043	2.89
0.066	3.24
0.091	3.58
0.120	4.33
0.145	4.78
0.170	5.25
0.195	5.65
0.220	5.85
0.246	5.91
0.276	5.92
0.301	5.95
0.329	5.92

## \*\*\*\*\*WALL PRESSURE DATA\*\*\*\*\*

DEL=6.0	XLE=8.48	DEL=8	XLE=8.52	DEL=10	XLE=8.53
YLE=3.13	PO=2116.0	YLE=3.13	PO=2105.0	YLE=3.06	PO=2115.0
TO=1128.0	X=15.0	TO=1126.0	X=15.0	TO=1091.0	X=15.0
Y	P/PREF	Y	P/PREF	Y	P/PREF
4.25	2.451	4.25	2.953	4.25	3.673
4.75	1.431	4.75	1.738	4.75	3.826
5.25	1.508	5.25	1.627	5.25	1.452
5.75	1.333	5.75	1.529	5.75	1.780
6.25	0.991	6.25	1.194	6.25	1.696
6.75	0.991	6.75	0.998	6.75	1.152
7.25	1.012	7.25	1.026	7.25	1.026
7.75	1.061	7.75	1.068	7.75	1.082
8.25	1.089	8.25	1.068	8.25	1.096
DEL=12	XLE=8.46	DEL=16	XLE=8.5	DEL=20	XLE=8.53
YLE=2.75	PO=2112.0	YLE=2.75	PO=2115.0	YLE=2.73	PO=2111.0
TO=1126.0	X=15.0	TO=1136.0	X=15.0	TO=1108.0	X=13.0
Y	P/PREF	Y	P/PREF	Y	P/PREF
4.25	4.825	4.75	6.927	4.75	8.840
4.75	3.079	5.25	4.406	5.25	1.731
5.25	1.515	5.75	1.850	5.75	2.241
5.75	1.857	6.25	2.039	6.25	2.458
6.25	1.766	6.75	2.241	6.75	2.269
6.75	1.256	7.25	2.143	7.25	1.096
7.25	0.984	7.75	1.270	7.75	1.187
7.75	1.033	8.25	1.277	8.25	1.277
8.25	1.040				



\*\*\*\*\*HEAT TRANSFER DATA\*\*\*\*\*

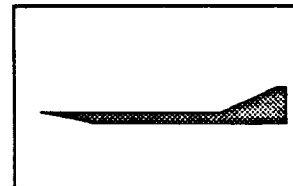
DEL=6.0	XLE=8.49	DEL=8.0	XLE=8.58	DEL=10.0	XLE=8.60
YLE=2.8	PO=2121.0	YLE=3.1	PO=2119.0	YLE=3.13	PO=2124.0
TO=1129.0	X=15.0	TO=1117.0	X=15.00	TO=1104.0	X=15.0
Y	H/HREF	Y	H/HREF	Y	H/HREF
3.25	0.117	3.75	0.037	4.00	0.102
3.50	0.529	4.00	0.852	4.25	1.192
3.75	2.717	4.25	3.150	4.50	4.031
4.00	2.636	4.50	3.307	4.75	3.743
4.25	2.071	4.75	2.331	5.00	2.199
4.50	1.482	5.00	1.662	5.25	1.766
4.75	0.965	5.25	1.371	5.50	1.634
5.00	1.048	5.50	1.157	5.75	1.414
5.25	1.099	5.75	1.076	6.25	1.050
5.50	1.141	6.25	0.789	6.75	1.197
		6.75	1.054	7.25	1.354
		7.25	1.295	7.75	1.199
		7.75	1.220	8.25	1.058
		8.25	1.043		
DEL=12.0	XLE=8.52	DEL=16.0	XLE=8.56	DEL=20.0	XLE=8.58
YLE=2.88	PO=2129.0	YLE=2.88	PO=2129.0	YLE=2.63	PO=2122.0
TO=1098.0	X=15.0	TO=1120.0	X=15.0	TO=1109.0	X=15.0
Y	H/HREF	Y	H/HREF	Y	H/HREF
3.75	0.007	4.00	0.025	4.25	0.005
4.00	0.164	4.25	0.075	4.50	0.079
4.25	1.458	4.50	0.297	4.75	0.626
4.50	4.713	4.75	3.010	5.00	3.752
4.75	4.352	5.00	5.838	5.25	6.827
5.00	2.361	5.25	4.670	5.50	5.625
5.25	1.805	5.50	2.735	5.75	3.195
5.50	1.855	5.75	2.183	6.25	2.363
5.75	1.756	6.25	1.844	6.75	3.176
6.25	1.065	6.75	2.085	7.25	2.629
6.75	1.203	7.25	1.491	7.75	1.467
7.25	1.398	7.75	1.216	8.25	0.883
7.75	1.252	8.25	1.058		
8.25	1.121				

\*\*\*\*\*END OF FILE\*\*\*\*\*

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Ref.: 16, 17, 28  
Author: Coleman, G. T.  
Geometry: 2-D Compression Corner  
Mach number: 9  
Data:  $p_{wall}$ ,  $C_h$

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Coleman, G.T., "Hypersonic Turbulent Boundary Layer Studies," *Ph.D. Thesis*, Univ. of London, 1973.

Coleman, G.T., and Stollery, J.L., "Heat Transfer From Hypersonic Turbulent Flow At a Wedge Compression Corner," *Journal of Fluid Mechanics*, Vol. 56, 1972, pp. 741-752.

Elfstrom, G.M., "Turbulent Hypersonic Flow at a Wedge-Compression Corner," *Journal of Fluid Mechanics*, Vol. 53, 1972, pp. 113-127.

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A flat plate and wedge compression corner were used to generate this dataset in a hypersonic gun tunnel with nitrogen as the test gas. Experiments are reported in Refs. 16 and 17 at Mach numbers of 8.96 and 9.22, though only the latter are included in this database. The slightly-lower Mach number data were also at a lower Reynolds number where boundary-layer tripping was thought necessary, hence these data are not included. Similarly, data obtained on a hollow-cylinder-flare model were not included because no incoming boundary-layer profile was measured.

The Mach 9.22 data include compression corner angles of 15, 30, 34, and 38 degrees for which surface-pressure and heat-transfer data are tabulated here. Users are encouraged to consult the cited References for a complete discussion of these data.

Note that the x-dimension cited in the data tables is always in the streamwise direction, even for points located on the sloped surface of the compression ramp. The origin of x is fixed at the compression corner, so that locations upstream of the corner have negative values.

The tabulated surface pressure data were actually obtained by Elfstrom in a companion study (Ref. 28). In none of these references are the stagnation and freestream static pressure conditions of the flow explicitly given. Rather, static pressure distributions are only cited in normalized form. This is thought to be typical of gun-tunnel operations, where a constant value of stagnation pressure is never actually achieved and thus normalization is a necessity. Users of these pressure data should bear in mind the possible adverse effect of such a procedure on the accuracy of the data.

The measured boundary-layer profile was actually taken "near the end of" the 76 cm flat plate. No boundary-layer integral parameters are stated in the cited references. The boundary-layer wake-strength parameter, however, is stated to be 0.2.

\*\*\*\*\*Coleman Mach 9 Compression Corner Data\*\*\*\*\*

\*\*\*\*\*Boundary-Layer Profile\*\*\*\*\*

Minf = 9.22  
 Re\_inf/cm = 4.7E+05  
 Tstagnation = 1070 deg K  
 Tinf = 64.5 deg K  
 Twall = 295 deg K  
 Twall/Trecovery = 0.28  
 Delta = 0.72 cm

y/Delta	M/Minf	U/Uinf
0.040	0.253	0.620
0.078	0.349	0.760
0.083	0.370	0.780
0.125	0.365	0.780
0.142	0.386	0.795
0.189	0.404	0.812
0.210	0.470	0.860
0.230	0.473	0.862
0.277	0.509	0.884
0.294	0.539	0.900
0.322	0.561	0.910
0.332	0.574	0.915
0.341	0.578	0.917
0.364	0.584	0.918
0.406	0.599	0.924
0.424	0.658	0.941
0.461	0.667	0.944
0.490	0.679	0.947
0.517	0.672	0.945
0.525	0.696	0.952
0.542	0.720	0.957
0.560	0.760	0.965
0.578	0.760	0.965
0.590	0.775	0.968
0.640	0.802	0.974
0.669	0.847	0.981
0.709	0.868	0.984
0.730	0.902	0.989
0.770	0.952	0.995
0.794	0.960	0.996
0.798	0.966	0.997
0.859	0.974	0.998
0.888	0.980	0.998
0.920	0.988	0.998
0.978	0.990	0.999
0.998	0.995	0.999
1.04	0.996	0.999
1.13	1.00	1.00
1.25	1.00	1.00

\*\*\*\*\*Coleman Mach 9 Compression Corner Data\*\*\*\*\*

Minf = 9.22  
 Re\_inf/cm = 4.7E+05  
 ReL = 26.2E+06  
 Tstagnation = 1070 deg K  
 Tinf = 64.5 deg K  
 Twall = 295 deg K  
 Twall/Trecovery = 0.28  
 Pinf/Pstagnation = 0.45  
 Delta = 0.72 cm

Alpha = 15 degrees  
 Qin = 6.07 W/cm\*\*2

Alpha = 30 degrees  
 Qin = 6.17 W/cm\*\*2

X, cm	Pw/Pinf	X, cm	Qw/Qinf	X, cm	Pw/Pinf	X, cm	Qw/Qinf
-1.9	0.99	-4.2	1.06	-1.65	1.05	-5.7	0.99
-1.4	1.0	-3.65	1.11	-1.14	1.05	-4.85	0.96
-0.9	1.02	-3.1	0.96	-1.14	1.06	-4.4	1.09
-0.4	1.04	-2.6	0.96	-0.89	1.08	-4.2	0.94
0.3	3.47	-2.05	1.03	-0.64	1.11	-3.65	1.0
0.84	4.7	-1.65	0.86	-0.38	1.28	-3.1	0.94
1.1	5.86	-1.35	0.94	0.25	7.45	-2.6	0.95

1.6	7.4	-1.1	1.0	0.56	15.8	-2.35	0.87
1.87	7.4	-0.85	0.92	0.84	24.2	-2.05	0.95
2.4	10.4	-0.6	0.8	1.15	31.8	-1.9	0.95
2.67	11.5	0.35	3.16	1.73	34.2	-1.35	0.93
3.2	11.8	0.6	2.91	2.0	36.0	-1.1	0.94
3.45	11.9	0.85	3.78	2.61	36.2	-0.85	0.96
4.24		1.35	4.27	2.89	36.4	-0.6	0.81
		1.6	4.68	3.5	36.4	-0.35	0.77
		1.85	5.42	3.8	36.0	0.25	5.8
		2.1	6.3	4.65	36.0	0.45	6.7
		2.35	6.67	5.6	36.4	0.7	8.6
		2.6	6.65			1.0	10.6
		3.15	7.69			1.2	13.1
		3.4	8.05			1.5	15.6
		3.9	8.5			1.7	15.7
		4.4	8.44			2.05	17.7
		4.9	8.61			2.24	18.1
		6.7	8.65			2.5	19.6
		7.2	8.04			2.7	18.6
						3.0	18.5
						3.3	17.7
						3.6	17.5
						3.7	17.0
						4.1	15.9
						4.25	16.6
						4.7	17.1
						5.3	16.3
						6.1	14.8
						6.6	15.4

Alpha = 34 degrees  
Qinf = 6.29 W/cm\*\*2

X, cm	Pw/Pinf	X, cm	Qw/Qinf
-2.67	1.02	-7.6	-0.95
-2.4	1.02	-6.65	1.03
-2.16	1.33	-5.7	0.98
-1.9	1.9	-5.15	1.03
-1.65	2.55	-4.65	1.03
-1.4	3.36	-4.2	1.0
-1.15	4.56	-3.9	0.89
-0.88	4.98	-3.65	0.93
-0.64	4.75	-3.35	0.77
-0.38	5.31	-3.1	0.9
0.25	6.97	-2.85	1.74
0.56	9.9	-2.6	1.35
0.87	12.6	-2.35	1.79
1.17	19.1	-2.05	2.24
1.47	26.9	-1.9	2.16
1.78	53.0	-1.65	2.08
2.08	62.3	-1.35	1.8
2.4	49.3	-1.1	1.67
3.0	51.0	-0.85	1.71
3.63	46.8	-0.6	1.52
3.94	46.2	-0.35	2.14
		0.25	3.22
		0.45	2.85
		0.7	5.85
		1.0	6.04
		1.2	9.91
		1.5	11.1
		1.7	14.3
		2.05	18.0
		2.24	21.1
		2.5	26.4
		2.7	27.6
		3.0	29.3
		3.3	24.3
		3.6	23.9
		3.7	21.8
		4.25	20.3
		4.7	21.4
		5.3	21.4
		7.1	19.2

Alpha = 38 degrees  
Qinf = 6.56 W/cm\*\*2

X, cm	Pw/Pinf	X, cm	Qw/Qinf
-4.7	1.03	-11.2	0.9
-4.45	1.11	-10.2	0.9
-4.2	1.24	-7.6	1.0
-3.93	1.46	-7.1	1.02
-3.68	1.44	-6.65	0.95
-3.43	2.54	-6.1	1.1
-3.18	2.67	-5.7	0.86
-2.92	4.45	-5.45	1.22
-2.66	5.25	-5.15	1.86
-2.41	5.7	-4.85	1.8
-2.16	5.7	-4.65	2.22
-1.9	5.7	-4.4	2.4
-1.4	6.2	-4.2	2.05
-1.14	6.2	-3.9	2.24
-0.89	6.2	-3.65	2.15
-0.64	6.6	-3.35	2.18
-0.38	6.7	-2.85	1.92
0.25	7.2	-2.6	1.82
0.5	8.5	-2.35	1.65
0.75	11.4	-2.05	1.93
1.0	15.1	-1.9	1.83
1.25	20.8	-1.65	1.7
1.5	29.0	-1.35	1.89
1.75	49.0	-1.1	1.88
2.0	59.0	-0.85	1.98
2.25	75.5	-0.6	1.94
2.54	86.0	0.25	2.86
2.8	93.0	0.45	3.25
3.05	77.5	0.70	4.16
3.3	69.5	1.0	4.82
3.81	59.0	1.2	6.47
4.06	61.5	1.6	10.0
4.32	59.6	1.7	10.6
4.57	57.5	2.24	16.4
5.33	59.0	2.5	18.7
		2.7	23.3
		3.0	29.0
		3.15	27.2
		3.3	31.3
		3.6	31.0
		3.7	28.5
		4.25	25.1

4.4	23.7
4.65	25.2
4.7	23.7
4.9	24.3
5.2	22.9
5.3	23.4
5.45	22.5
5.75	22.4
6.7	22.7

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Ref.: 40, 39, 38

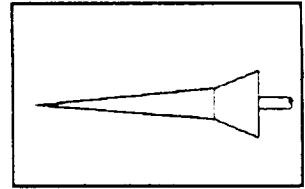
Author: Holden, M. S.

Geometry: Axisymmetric Cone-Flare

Mach number: 11, 13

Data:  $p_{wall}$ ,  $c_h$

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Holden, M.S., "Experimental Studies of Quasi-Two-Dimensional and Three- Dimensional Viscous Interaction Regions Induced by Skewed- Shock and Swept-Shock Boundary Layer Interaction," *AIAA Paper 84-1677*, 1984.

Holden, M.S., Havener, A.G. and Lee, C.H., "Shock Wave/Turbulent Boundary Layer Interaction in High- Reynolds-Number Hypersonic Flows," *CUBRC-86681*, 1986.

Holden, M.S., Bergman, R.C., Harvey, J., Duryea, G.R. and Moselle, J.R., "Studies of the Structure of Attached and Separated Regions of Viscous/Inviscid Interaction and the Effects of Combined Surface Roughness and Blowing in High Reynolds Number Hypersonic Flows," *AFOSR-89-0033TR*, 1988

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The data consist of surface pressures and heat transfer coefficients beneath the axisymmetric interaction generated by the shock due to a flare interacting with the turbulent boundary layer on a large, sharp-nosed 6 degree half-angle cone in the Calspan 96-inch Shock Tunnel. Flare angles of 30 and 36 degrees were tested at both Mach 11 and Mach 13, comprising four shock-tunnel runs as tabulated below. The data given below were not published previously, but were obtained directly from M. S. Holden

Boundary-layer profiles just ahead of the cone-flare intersection were measured via 11-tube pitot- and temperature-rakes and are plotted in CUBRC Report 86681 referenced above. The locations of the surveys were 1.2 inches upstream of the corner for Mach 11 and 2.0 inches upstream for Mach 13. These profiles were necessarily rather coarse due to the manner in which they were obtained. In their place, Holden has provided "interpolated" profiles in which points are given at increments of 0.02 in  $y/\delta$ . Users of these data should bear in mind that the original, measured boundary-layer profiles consisted of only 11 data points across the boundary-layer thickness. Also, no boundary-layer integral parameters were provided by the experimenter, so none are tabulated here.

The location of the flare compression corner was 104.6 inches along the cone surface from the cone tip. It appears that the given x-dimensions of the instrumentation sites are all given in terms of distance along the cone surface from its tip, even for  $x > 104.6$  inches where the instrumentation was actually mounted on the flare. Y is the corresponding dimension normal to the cone surface.

*Important note:* Although the flare angles cited by Holden and tabulated here are given as 30 and 36 degrees, their actual angles with respect to the symmetry axis of the cone-flare model are 36 and 42 degrees, respectively. The 30 and 36 degree angles are relative to the surface of the 6 degree forecone.

Run number = 8  
 Cone Angle of Attack (degrees) = 0  
 Angle of Cone (degrees) = 6  
 Length Along Cone (inches) = 104.6  
 Angle of Flare (degrees) = 30  
 Length of Flare (inches) = 9  
 Reservoir Total Pressure = 7234 PSIA  
 Reservoir Total Temperature = 2727 deg R  
 Freestream Mach number = 10.98  
 Freestream Velocity = 5931 ft/sec  
 Freestream Static Pressure = 0.09078 PSIA  
 Freestream Static Temperature = 121.3 deg R  
 Freestream Reynolds number = 3651000/ft<sup>2</sup>  
 Wall Enthalpy (Cp\*Tw) = 3183000 (Ft/sec)<sup>2</sup>

\*\*\*\*\*WALL PRESSURE & HEAT TRANSFER DATA\*\*\*\*\*BOUNDARY-LAYER DATA\*\*\*\*\*

Gage #	X inches	Press. PSIA	Y inches	U/UE	MACH#	TT/TTE
P1	98.9105	2.366E-01	0.014	0.493	2.186	0.469
P3	100.1195	2.672E-01	0.028	0.572	2.613	0.535
P7	101.9185	2.700E-01	0.042	0.618	2.890	0.576
P9	102.5125	2.686E-01	0.056	0.651	3.105	0.607
P11	103.1135	2.803E-01	0.070	0.676	3.285	0.631
P13	103.7165	2.677E-01	0.084	0.697	3.444	0.652
P14	104.0165	2.669E-01	0.098	0.716	3.589	0.670
P16	104.6765	2.250E+00	0.112	0.731	3.725	0.687
P17	105.1835	5.297E+00	0.127	0.746	3.855	0.702
P18	105.6825	6.390E+00	0.141	0.760	3.981	0.716
P19	106.1825	1.043E+01	0.155	0.772	4.104	0.729
P20	106.6825	9.707E+00	0.169	0.784	4.226	0.742
P21	107.1835	"Null"	0.183	0.795	4.347	0.754
P22	107.6835	9.788E+00	0.197	0.806	4.469	0.766
P24	108.6805	1.063E+01	0.211	0.816	4.591	0.778
P26	109.6825	1.135E+01	0.225	0.826	4.715	0.789
P27	110.6825	9.166E+00	0.239	0.836	4.840	0.800
P28	111.6825	9.016E+00	0.253	0.845	4.967	0.810
P29	112.6825	8.899E+00	0.267	0.854	5.097	0.821
			0.281	0.863	5.229	0.831
Gage #	X inches	BTU/FT2/SEC	0.295	0.871	5.363	0.841
HT1	98.9105	5.806E+00	0.309	0.880	5.500	0.851
HT5	101.3175	5.603E+00	0.323	0.887	5.639	0.860
HT6	101.6065	6.656E+00	0.337	0.895	5.781	0.869
HT7	101.9185	6.411E+00	0.351	0.902	5.925	0.877
HT8	102.2195	5.153E+00	0.366	0.910	6.072	0.886
HT9	102.5125	"Null"	0.380	0.916	6.221	0.895
HT10	102.8135	5.930E+00	0.394	0.923	6.371	0.902
HT11	103.1135	5.495E+00	0.408	0.929	6.523	0.910
HT12	103.4165	4.726E+00	0.422	0.935	6.677	0.917
HT13	103.7165	6.010E+00	0.436	0.941	6.831	0.924
HT14	104.0165	5.796E+00	0.450	0.946	6.985	0.931
HT16	104.6765	1.124E+02	0.464	0.952	7.140	0.938
HT17	105.1835	1.181E+02	0.478	0.956	7.294	0.944
HT18	105.6825	"Null"	0.492	0.961	7.447	0.950
HT19	106.1825	1.443E+02	0.506	0.965	7.597	0.955
HT20	106.6825	1.323E+02	0.520	0.969	7.746	0.960
HT21	107.1835	1.316E+02	0.534	0.973	7.891	0.965
HT22	107.6835	1.522E+02	0.548	0.976	8.033	0.970
HT23	108.1765	1.392E+02	0.562	0.980	8.170	0.974
HT24	108.6805	1.208E+02	0.576	0.983	8.301	0.978
HT25	109.1835	1.048E+02	0.590	0.986	8.427	0.981
HT26	109.6825	1.056E+02	0.605	0.988	8.547	0.985
HT27	110.6825	1.005E+02	0.619	0.991	8.659	0.988
HT28	111.6825	9.250E+01	0.633	0.993	8.763	0.990
HT29	112.6825	9.706E+01	0.647	0.995	8.859	0.993
			0.661	0.996	8.946	0.995
			0.675	0.998	9.024	0.997
			0.689	0.999	9.092	0.998
			0.703	1.000	9.150	1.000

Run number = 4  
 Cone Angle of Attack (degrees) = 0  
 Angle of Cone (degrees) = 6  
 Length Along Cone (inches) = 104.6  
 Angle of Flare (degrees) = 36  
 Length of Flare (inches) = 9  
 Reservoir Total Pressure = 7001 PSIA  
 Reservoir Total Temperature = 2649 deg R  
 Freestream Mach number = 10.97  
 Freestream Velocity = 5836 ft/sec  
 Freestream Static Pressure = 0.08943 PSIA  
 Freestream Static temperature = 117.8 deg R  
 Freestream Reynolds number = 3755000/ft  
 Wall Enthalpy (Cp\*Tw) = 3183000 (Ft/sec)<sup>2</sup>

\*\*\*\*\*WALL PRESSURE & HEAT TRANSFER DATA\*\*\*\*\*BOUNDARY-LAYER DATA\*\*\*\*\*

Gage #	X inches	Press. PSIA	Y inches	U/UE	MACH#	TT/TTE
P1	98.9105	2.6728E-01	0.014	0.493	2.186	0.469
P3	100.1195	2.8851E-01	0.028	0.572	2.613	0.535
P7	101.9185	3.4467E-01	0.042	0.618	2.890	0.576
P9	102.5125	8.0524E-01	0.056	0.651	3.105	0.607
P11	103.1135	1.29314E+00	0.070	0.676	3.285	0.631
P13	103.7165	1.38824E+00	0.084	0.697	3.444	0.652
P14	104.0165	1.53297E+00	0.098	0.716	3.589	0.670
P16	104.6765	1.30152E+00	0.112	0.731	3.725	0.687
P17	105.1835	"Null"	0.127	0.746	3.855	0.702
P18	105.6825	"Null"	0.141	0.760	3.981	0.716
P19	106.1825	1.49127E+01	0.155	0.772	4.104	0.729
P20	106.6825	1.51135E+01	0.169	0.784	4.226	0.742
P21	107.1835	1.26621E+01	0.183	0.795	4.347	0.754
P22	107.6835	1.39913E+01	0.197	0.806	4.469	0.766
P24	108.6805	1.45225E+01	0.211	0.816	4.591	0.778
P26	109.6825	1.54217E+01	0.225	0.826	4.715	0.789
P27	110.6825	1.29523E+01	0.239	0.836	4.840	0.800
P28	111.6825	1.19923E+01	0.253	0.845	4.967	0.810
P29	112.6825	1.22768E+01	0.267	0.854	5.097	0.821
			0.281	0.863	5.229	0.831
Gage #	X inches	BTU/FT2/SEC	0.295	0.871	5.363	0.841
HT1	98.9105	5.85107E+00	0.309	0.880	5.500	0.851
HT5	101.3175	5.71897E+00	0.323	0.887	5.639	0.860
HT6	101.6065	5.80157E+00	0.337	0.895	5.781	0.869
HT7	101.9185	6.16079E+00	0.351	0.902	5.925	0.877
HT8	102.2195	9.01738E+00	0.366	0.910	6.072	0.886
HT9	102.5125	1.47118E+01	0.380	0.916	6.221	0.895
HT10	102.8135	1.46640E+01	0.394	0.923	6.371	0.902
HT11	103.1135	1.32007E+01	0.408	0.929	6.523	0.910
HT12	103.4165	1.33030E+01	0.422	0.935	6.677	0.917
HT13	103.7165	1.19369E+01	0.436	0.941	6.831	0.924
HT14	104.0165	1.74385E+01	0.450	0.946	6.985	0.931
HT16	104.6765	"Null"	0.464	0.952	7.140	0.938
HT17	105.1835	9.00276E+01	0.478	0.956	7.294	0.944
HT18	105.6825	1.72171E+02	0.492	0.961	7.447	0.950
HT19	106.1825	2.03513E+02	0.506	0.965	7.597	0.955
HT20	106.6825	1.97140E+02	0.520	0.969	7.746	0.960
HT21	107.1835	1.77999E+02	0.534	0.973	7.891	0.965
HT22	107.6835	1.75618E+02	0.548	0.976	8.033	0.970
HT23	108.1765	1.65347E+02	0.562	0.980	8.170	0.974
HT24	108.6805	1.46599E+02	0.576	0.983	8.301	0.978
HT25	109.1835	1.44159E+02	0.590	0.986	8.427	0.981
HT26	109.6825	1.36357E+02	0.605	0.988	8.547	0.985
HT27	110.6825	1.29164E+02	0.619	0.991	8.659	0.988
HT28	111.6825	1.04948E+02	0.633	0.993	8.763	0.990
HT29	112.6825	1.02294E+02	0.647	0.995	8.859	0.993
			0.661	0.996	8.946	0.995
			0.675	0.998	9.024	0.997
			0.689	0.999	9.092	0.998
			0.703	1.000	9.150	1.000



Run number = 7

Cone Angle of Attack (degrees) = 0

Angle of Cone (degrees) = 6

Length Along Cone (inches) = 104.6

Angle of Flare (degrees) = 30

Length of Flare (inches) = 9

Reservoir Total Pressure = 17230 PSIA

Reservoir Total Temperature = 3246 deg R

Freestream Mach number = 12.92

Freestream Velocity = 6634 ft/sec

Freestream Static Temperature = 109.7 deg R

Freestream Static Pressure = 0.07272 PSIA

Freestream Reynolds number = 4000000/ft

Wall Enthalpy ( $C_p T_w$ ) = 3183000 (Ft/sec)<sup>2</sup>

## \*\*\*\*\*WALL PRESSURE &amp; HEAT TRANSFER DATA\*\*\*\*\*BOUNDARY-LAYER DATA\*\*\*\*\*

Gage #	X inches	Press. PSIA	Y inches	U/UE	MACH#	TT/TTE
P1	98.9105	2.577E-01	0.015	0.486	2.266	0.445
P3	100.1195	3.032E-01	0.029	0.567	2.718	0.515
P7	101.9185	3.241E-01	0.044	0.614	3.014	0.559
P9	102.5125	3.040E-01	0.059	0.648	3.243	0.591
P11	103.1135	3.605E-01	0.074	0.674	3.437	0.617
P13	103.7165	3.650E-01	0.088	0.696	3.609	0.639
P14	104.0165	4.115E-01	0.103	0.714	3.766	0.659
P16	104.6765	"Null"	0.118	0.731	3.914	0.676
P17	105.1835	6.149E+00	0.133	0.746	4.056	0.692
P18	105.6825	7.350E+00	0.147	0.759	4.194	0.707
P19	106.1825	1.358E+01	0.162	0.772	4.329	0.722
P20	106.6825	1.243E+01	0.177	0.784	4.463	0.735
P21	107.1835	"Null"	0.191	0.795	4.598	0.748
P22	107.6835	1.218E+01	0.206	0.806	4.732	0.760
P24	108.6805	1.394E+01	0.221	0.817	4.869	0.772
P26	109.6825	1.431E+01	0.236	0.827	5.007	0.784
P27	110.6825	1.188E+01	0.250	0.837	5.148	0.795
P28	111.6825	1.175E+01	0.265	0.847	5.291	0.807
P29	112.6825	1.219E+01	0.280	0.856	5.437	0.817
			0.295	0.864	5.587	0.828
Gage #	X inches	BTU/FT <sup>2</sup> /SEC	0.309	0.873	5.740	0.838
HT1	98.9105	"Null"	0.324	0.881	5.896	0.848
HT5	101.3175	8.217E+00	0.339	0.889	6.056	0.858
HT6	101.6065	9.402E+00	0.353	0.897	6.220	0.867
HT7	101.9185	9.852E+00	0.368	0.904	6.387	0.877
HT8	102.2195	8.042E+00	0.383	0.911	6.557	0.885
HT9	102.5125	9.473E+00	0.398	0.918	6.731	0.894
HT10	102.8135	8.531E+00	0.412	0.925	6.907	0.902
HT11	103.1135	8.600E+00	0.427	0.931	7.086	0.910
HT12	103.4165	7.930E+00	0.442	0.937	7.268	0.918
HT13	103.7165	1.013E+01	0.457	0.943	7.451	0.925
HT14	104.0165	8.931E+00	0.471	0.948	7.636	0.932
HT16	104.6765	4.236E+02	0.486	0.953	7.821	0.938
HT17	105.1835	3.791E+02	0.501	0.958	8.006	0.944
HT18	105.6825	"Null"	0.515	0.962	8.191	0.950
HT19	106.1825	3.199E+02	0.530	0.966	8.374	0.955
HT20	106.6825	2.818E+02	0.545	0.970	8.555	0.960
HT21	107.1835	4.178E+02	0.560	0.974	8.733	0.965
HT22	107.6835	3.879E+02	0.574	0.978	8.907	0.970
HT23	108.1765	3.152E+02	0.589	0.981	9.076	0.974
HT24	108.6805	2.296E+02	0.604	0.984	9.239	0.978
HT25	109.1835	2.287E+02	0.619	0.987	9.395	0.981
HT26	109.6825	1.925E+02	0.633	0.989	9.543	0.985
HT27	110.6825	1.819E+02	0.648	0.991	9.683	0.988
HT28	111.6825	1.624E+02	0.663	0.993	9.814	0.990
HT29	112.6825	1.455E+02	0.677	0.995	9.934	0.993
			0.692	0.996	10.043	0.995
			0.707	0.998	10.141	0.997
			0.722	0.999	10.227	0.998
			0.736	1.000	10.300	1.000

Run number = 6  
 Cone Angle of Attack (degrees) = 0  
 Angle of Cone (degrees) = 6  
 Length Along Cone (inches) = 104.6  
 Angle of Flare (degrees) = 36  
 Length of Flare (inches) = 9  
 Reservoir Total Pressure = 17600 PSIA  
 Reservoir Total Temperature = 3104 deg R  
 Freestream Mach number = 13.10  
 Freestream Velocity = 6458 ft/sec  
 Freestream Static Pressure = 0.07345 PSIA  
 Freestream Static Temperature = 102.6 deg R  
 Freestream Reynolds number = 5090000/ft  
 Wall Enthalpy ( $C_p T_w$ ) = 3183000 (Ft/sec)<sup>2</sup>

\*\*\*\*\*WALL PRESSURE & HEAT TRANSFER DATA\*\*\*\*\*BOUNDARY-LAYER DATA\*\*\*\*\*

Gage #	X inches	Press. PSIA	Y inches	U/UE	MACH#	TT/TTE
P1	98.9105	2.8474E-01	0.015	0.486	2.266	0.445
P3	100.1195	2.9218E-01	0.029	0.567	2.718	0.515
P7	101.9185	4.9656E-01	0.044	0.614	3.014	0.559
P9	102.5125	9.9969E-01	0.059	0.648	3.243	0.591
P11	103.1135	1.3961E+00	0.074	0.674	3.437	0.617
P13	103.7165	1.5696E+00	0.088	0.696	3.609	0.639
P14	104.0165	1.7067E+00	0.103	0.714	3.766	0.659
P16	104.6765	1.5295E+00	0.118	0.731	3.914	0.676
P17	105.1835	5.0433E+00	0.133	0.746	4.056	0.692
P18	105.6825	"Null"	0.147	0.759	4.194	0.707
P19	106.1825	1.8144E+01	0.162	0.772	4.329	0.722
P20	106.6825	2.1674E+01	0.177	0.784	4.463	0.735
P21	107.1835	"Null"	0.191	0.795	4.598	0.748
P22	107.6835	1.7335E+01	0.206	0.806	4.732	0.760
P24	108.6805	1.8623E+01	0.221	0.817	4.869	0.772
P26	109.6825	2.0844E+01	0.236	0.827	5.007	0.784
P27	110.6825	1.4811E+01	0.250	0.837	5.148	0.795
P28	111.6825	1.6215E+01	0.265	0.847	5.291	0.807
P29	112.6825	1.6687E+01	0.280	0.856	5.437	0.817
			0.295	0.864	5.587	0.828
			0.309	0.873	5.740	0.838
Gage #	X inches	BTU/FT2/SEC				
HT1	98.9105	8.8680E+00	0.324	0.881	5.896	0.848
HT5	101.3175	7.2142E+00	0.339	0.889	6.056	0.858
HT6	101.6065	7.6842E+00	0.353	0.897	6.220	0.867
HT7	101.9185	1.0260E+01	0.368	0.904	6.387	0.877
HT8	102.2195	1.3595E+01	0.383	0.911	6.557	0.885
HT9	102.5125	1.7688E+01	0.398	0.918	6.731	0.894
HT10	102.8135	1.9199E+01	0.412	0.925	6.907	0.902
HT11	103.1135	2.0174E+01	0.427	0.931	7.086	0.910
HT12	103.4165	1.7752E+01	0.442	0.937	7.268	0.918
HT13	103.7165	2.1149E+01	0.457	0.943	7.451	0.925
HT14	104.0165	2.3570E+01	0.471	0.948	7.636	0.932
HT16	104.6765	1.9627E+02	0.486	0.953	7.821	0.938
HT17	105.1835	2.0182E+02	0.501	0.958	8.006	0.944
HT18	105.6825	"Null"	0.515	0.962	8.191	0.950
HT19	106.1825	4.3151E+02	0.530	0.966	8.374	0.955
HT20	106.6825	3.8457E+02	0.545	0.970	8.555	0.960
HT21	107.1835	4.3251E+02	0.560	0.974	8.733	0.965
HT22	107.6835	4.5576E+02	0.574	0.978	8.907	0.970
HT23	108.1765	3.1065E+02	0.589	0.981	9.076	0.974
HT24	108.6805	3.5936E+02	0.604	0.984	9.239	0.978
HT25	109.1835	3.2652E+02	0.619	0.987	9.395	0.981
HT26	109.6825	2.4472E+02	0.633	0.989	9.543	0.985
HT27	110.6825	2.1926E+02	0.648	0.991	9.683	0.988
HT28	111.6825	1.6218E+02	0.663	0.993	9.814	0.990
HT29	112.6825	1.7518E+02	0.677	0.995	9.934	0.993
			0.692	0.996	10.043	0.995
			0.707	0.998	10.141	0.997
			0.722	0.999	10.227	0.998
			0.736	1.000	10.300	1.000

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Ref.: 51

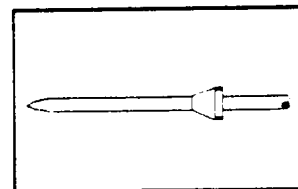
Author: Kussoy, M. I., *et al*

Geometry: Axisymmetric Ogive-Cylinder-Flare

Mach number: 7

Data:  $p_{wall}$ ,  $c_h$ , flowfield surveys

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Kussoy, M.I. and Horstman, C.C., "Documentation of Two- and Three-Dimensional Hypersonic Shock Wave/Turbulent Boundary Layer Interaction Flows," *NASA TM 101075*, 1989.

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This experiment provides data on an axisymmetric ogive-cylinder body with flares of various angles at Mach 7. The range of flare angles spans conditions from fully unseparated to well-separated flow. The data include wall pressures, heat transfer, and limited flowfield surveys obtained by pressure probes. The following tabulation includes freestream conditions, the incoming boundary-layer profile, three additional profiles through the 20 degree flare interaction, and wall pressure and heat transfer data for the 20, 30, 32.5, and 35 degree flare interactions. Note that the given dimension  $X$  is actually measured *along the surface of the model* with origin at the cylinder-flare corner. Points upstream of this location have negative  $X$ -values, while those downstream lie along the flare and have positive values. The cylinder-flare corner itself lies at 139 cm downstream from the tip of the ogival nose.

Users of these data are encouraged to read Ref. 51, which includes additional pertinent information. The uncertainty estimates placed on the data by the experimenters are also discussed therein. Note that Ref. 51 also contains data on an ogive-cylinder model with a fin, but these data are not included in the database due to the complexity of the geometry.

\*\*\*\*\*Kussoy et al, Ogive-Cylinder-Flare, Mach 7\*\*\*\*\*

\*\*\*\*\*Freestream Flow Conditions\*\*\*\*\*

Minf = 7.05  
Tinf = 81.2 deg K  
Pinf = 576 Pa  
RHOinf = 0.0252 kg/m\*\*3  
Twall = 311 deg K  
Uinf = 1274 m/s  
Re/m = 5.8E+06

\*\*\*\*\*Incoming Boundary-Layer Conditions\*\*\*\*\*

DELTA = 2.5 cm  
DELTA\* = 0.74 cm  
THETA = 0.065 cm  
Cfinf = 1.22E-03  
Chinf = 0.59E-03

Y, cm	M	U/Uinf	TT/TTinf
0.000	0.000	0.000	0.350
0.065	1.547	0.470	0.638
0.093	2.177	0.601	0.690
0.120	2.745	0.699	0.752
0.180	3.111	0.759	0.805
0.250	3.356	0.791	0.830
0.320	3.610	0.822	0.858
0.390	3.835	0.836	0.858
0.460	4.070	0.858	0.877
0.620	4.626	0.896	0.905
0.770	5.248	0.929	0.930
0.940	5.739	0.954	0.954
1.090	6.070	0.967	0.966
1.260	6.340	0.982	0.986
1.450	6.599	0.986	0.986
1.640	6.820	0.992	0.991
1.900	6.962	0.999	1.000
2.150	7.022	1.000	1.000
2.400	7.048	1.000	1.000
2.700	7.050	1.000	1.000
3.000	7.050	1.000	1.000

\*\*\*\*\*Wall Pressure and Heat Transfer Distributions\*\*\*\*\*

Alpha = 20 degrees

Alpha = 30 degrees

X, cm	Pw/Pinf	Qw/Qinf	X, cm	Pw/Pinf	Qw/Qinf
-11.3	0.97	0.98	-11.3	1.00	0.99
-10.3	0.98	1.05	-10.3	0.98	0.99
-9.3	0.96	1.06	-9.3	0.97	1.0
-8.3	0.98	1.02	-8.3	0.98	1.0
-7.3	0.97	1.03	-7.3	1.0	1.01
-6.3	0.99	1.0	-6.3	1.0	1.01
-5.3	1.03	1.03	-5.3	0.98	1.0
-4.3	1.00	1.01	-4.3	1.02	0.99
-3.3	1.02	0.99	-3.3	1.09	0.99
-2.3	1.0	0.86	-2.3	1.39	1.02
-1.3	1.09	4.79	-1.3	1.73	8.2
1.1	2.02	4.17	1.1	7.75	8.97
1.6	3.68	4.99	1.6	9.58	10.09
2.1	5.31	5.27	2.1	12.85	12.05
2.6	5.96	5.94	2.6	15.06	13.42
3.6	7.42	5.74	3.6	19.51	14.39
4.1	8.27	6.04	4.1	21.33	15.25
4.6	9.10	7.99	4.6	21.94	14.86
5.1	9.95	8.59	6.1	22.82	14.9
6.1	10.8	8.74	7.1	22.88	14.6
7.1	11.3	8.64	8.1	23.71	14.75
10.1	12.14	8.77	10.1	24.4	14.45
12.1	12.26	9.18	12.1	23.62	14.41
14.1	12.5	9.68	14.1	22.6	13.98

13.1 9.59  
14.1 9.36  
15.1 8.55

13.1 13.02  
14.1 12.87  
15.1 12.21

Alpha = 32.5 degrees

Alpha = 35 degrees

X, cm	Pw/Pinf	X, cm	Qw/Qinf	X, cm	Pw/Pinf	X, cm	Qw/Qinf
-11.3	1.01	-12.06	1.0	-11.3	0.97	-12.06	0.99
-10.3	0.99	-10.8	0.99	-10.3	0.98	-10.8	1.07
-9.3	0.98	-9.52	1.0	-9.3	1.0	-9.52	1.0
-8.3	1.01	-8.26	1.01	-8.3	1.15	-8.26	1.04
-7.3	0.97	-6.98	1.03	-7.3	1.45	-6.98	1.02
-6.3	1.01	-5.73	1.01	-6.3	2.01	-5.73	1.26
-5.3	1.03	-4.44	1.05	-5.3	2.53	-4.44	1.83
-4.3	1.12	-3.18	1.09	-4.3	3.25	-3.18	2.39
-3.3	1.23	-1.9	1.33	-3.3	3.63	-1.9	2.63
-2.3	1.71	-0.64	2.16	-2.3	4.41	-0.64	2.55
-1.3	2.56	1.05	6.79	-1.3	4.56	1.07	6.4
0.55	5.85	1.55	7.44	1.07	6.95	1.57	7.9
1.05	7.50	2.05	9.23	1.57	8.82	2.07	9.65
1.55	8.42	2.55	10.64	2.07	11.07	2.57	11.75
2.05	12.02	3.05	12.82	2.57	14.05	3.07	13.82
2.55	14.4	3.55	14.55	3.07	18.45	3.57	16.1
3.55	21.19	4.55	16.28	3.57	21.79	5.07	20.03
4.05	23.57	5.05	17.05	4.07	25.71	6.07	21.96
4.55	25.83	6.05	17.05	4.57	28.93	7.07	21.37
6.05	27.14	7.05	16.67	7.07	33.68	9.07	19.72
7.05	27.74	8.05	16.41	8.07	33.69	10.07	19.97
8.05	27.62	9.05	16.54	10.07	30.04		
10.05	27.74	10.05	15.90				
		11.05	15.77				

\*\*\*\*\*Flowfield Surveys\*\*\*\*\*

Alpha = 20 degrees, X = 5.5 cm

Y, cm	M	P/Pinf	U/Uinf	TT/TTinf
0.000	0.000	10.539	0.000	0.350
0.053	2.768	10.539	0.763	0.881
0.096	2.800	10.539	0.778	0.907
0.145	2.878	10.539	0.797	0.931
0.195	2.983	10.539	0.812	0.939
0.290	3.266	10.539	0.851	0.968
0.395	3.701	10.120	0.895	0.992
0.495	3.980	9.760	0.914	0.998
0.590	4.043	9.461	0.918	1.000
0.790	7.050	1.000	1.004	1.006
0.990	7.050	1.000	1.004	1.006
1.175	7.050	1.000	1.003	1.004
1.360	7.050	1.000	1.003	1.004

Alpha = 20 degrees, X = 10.3 cm

Y, cm	M	P/Pinf	U/Uinf	TT/TTinf
0.000	0.000	11.976	0.000	0.350
0.055	2.250	11.976	0.718	0.938
0.100	2.859	11.976	0.806	0.955
0.140	3.107	11.976	0.835	0.963
0.185	3.157	11.976	0.840	0.965
0.270	3.253	11.976	0.856	0.980
0.360	3.336	11.976	0.865	0.986
0.450	3.417	11.976	0.875	0.994
0.650	3.605	11.976	0.892	1.000
0.850	3.763	11.976	0.902	1.000
1.050	3.835	11.976	0.904	0.995
1.200	4.089	9.581	0.916	0.989
1.420	7.052	1.000	1.001	1.001
1.620	7.052	1.000	1.001	1.001

Alpha = 20 degrees, X = 15.5 cm

Y, cm	M	P/Pinf	U/Uinf	TT/TTinf
0.000	0.000	12.335	0.000	0.350
0.065	2.533	12.335	0.735	0.881
0.083	2.747	12.335	0.772	0.906
0.100	2.997	12.335	0.810	0.933
0.138	3.121	12.335	0.832	0.954
0.169	3.168	12.335	0.844	0.972
0.198	3.216	12.335	0.852	0.980
0.250	3.286	12.335	0.862	0.988
0.300	3.354	12.335	0.873	0.999
0.400	3.431	12.335	0.880	1.002
0.520	3.486	12.335	0.884	1.001
0.660	3.540	12.335	0.887	1.000
0.710	3.566	12.335	0.889	1.000
0.800	3.613	12.335	0.892	1.000
0.900	3.676	12.335	0.897	1.000
1.000	3.737	12.335	0.901	1.000
1.100	3.777	12.335	0.903	1.000
1.200	3.777	12.335	0.903	1.000

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Ref.: 53

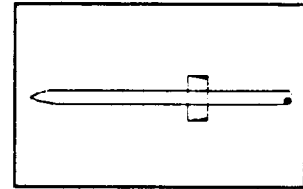
Author: Kussoy, M. I., *et al*

Geometry: Axisymmetric Impinging Shock

Mach number: 7

Data:  $p_{wall}$ ,  $c_h$ ,  $c_f$ , flowfield surveys

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Kussoy, M.I. and Horstman, C.C., "An Experimental Documentation of a Hypersonic Shock-Wave Turbulent Boundary Layer Interaction Flow - With and Without Separation," *NASA TM X-62412*, 1975.

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Experiments were conducted on a cone-ogive-cylinder model 3.3 m long and 0.203 m in diameter. Concentric with this model was mounted an annular shock-wave generator of 0.51 m outside diameter. The model was water-cooled to maintain a constant 300° surface temperature during tests. Surface instrumentation included static pressures, thermocouples, and a skin friction balance. Pitot, static pressure, and total temperature were measured at regular stations through the interaction using a probe drive built into the cylindrical model.

Two different shock wave strengths were produced by shock generator deflection angles of 7.5° and 15°. However, because of geometrical constraints the annular shock generator was rather short in streamwise extent, leading to a merging of the incident shock and its trailing expansion fan before the shock impinged on the test surface. This makes even the inviscid flowfield complicated, and renders a precise statement of the interaction strength problematic. Nonetheless the data are felt to have continuing value for some code validation and turbulence modeling purposes.

Unfortunately, these data are not available in machine-readable form and we have thus chosen to enter only a subset of the measurements into the database. Since the 7.5° case did not result in boundary-layer separation, it is omitted. All the surface data for the 15° case and a selection of the flowfield profiles for 15° are tabulated below. Users of these data should consult Ref. 53 for more complete information and data.

The experimenters claim an accuracy of  $\pm 10\%$  for surface data, which degrades, however, near separation. Other confidence limits quoted in Ref. 53 are  $\pm 1.5\%$  for total temperature (TT),  $\pm 10\%$  for static pressure (P),  $\pm 6\%$  for static temperature (T),  $\pm 12\%$  for density (RHO),  $\pm 3\%$  for x-velocity component (U), and  $\pm 0.02$  cm for vertical position (Y) above the cylindrical model surface. The U uncertainty is further placed at  $\pm 8\%$  near the surface inside the interaction, and  $\pm 35\%$  in the region of reverse flow.

The nomenclature of the data tables is largely standard and self-explanatory. Twm and Twc refer to as-measured and corrected wall shear stress, respectively. The correction was for bouyancy effects. Twc should be used for all-purposes unless an attempt at re-correcting the raw Twm data is contemplated. The term RHOU in the tables refer to mass flux per unit area. The streamwise coordinate X is measured along the surface of the cylindrical test body, with its origin at a position on the body corresponding to the leading-edge of the annular shock generator. Delta\*i and Thetai refer to kinematic displacement and momentum thicknesses, respectively.

*****Kussey and Morstman M-7 Impinging Shock Data*****									
*****15 Degree Shock Wave Generator*****									
*****Incoming freestream Conditions*****									
M <sub>∞</sub> = 6.86									
P <sub>∞</sub> = 607 N/m <sup>2</sup>									
T <sub>∞</sub> = 67.8 deg. K									
ρ <sub>∞</sub> = 0.0312 kg/m <sup>3</sup>									
U <sub>∞</sub> = 1132 m/s									
T <sub>0∞</sub> = 695 deg. K									
*****SURFACE DATA*****									
X (cm)	P <sub>w</sub> (N/m <sup>2</sup> )	T <sub>w</sub>	T <sub>w</sub> (K)	q <sub>w</sub> (W/m <sup>2</sup> )	h <sub>w</sub> (K)	h <sub>w</sub> (K)	h <sub>w</sub> (K)	h <sub>w</sub> (K)	h <sub>w</sub> (K)
20.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
21.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
22.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
23.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
24.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
25.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
26.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
27.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
28.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
29.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
30.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
31.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
32.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
33.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
34.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
35.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
36.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
37.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
38.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
39.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
40.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
41.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
42.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
43.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
44.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
45.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
46.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
47.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
48.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
49.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
50.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
51.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
52.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
53.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
54.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
55.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
56.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
57.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
58.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
59.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
60.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
61.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
62.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
63.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
64.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
65.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
66.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
67.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
68.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
69.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
70.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
71.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
72.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
73.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
74.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
75.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0
76.0	607	16.7	16.7	6240	42.0	42.0	42.0	42.0	42.0

Y (cm)	M	P / P <sub>∞</sub>	h <sub>w</sub> / h <sub>∞</sub>	T / T <sub>∞</sub>	U / U <sub>∞</sub>	h <sub>w</sub> / h <sub>∞</sub>	T / T <sub>∞</sub>
0.000	0.000	1.000	0.226	4.430	0.000	0.000	0.432
0.050	1.503	1.000	0.219	4.376	0.468	0.102	0.648
0.100	1.777	1.000	0.230	4.357	0.540	0.124	0.693
0.150	2.000	1.000	0.241	4.155	0.598	0.144	0.733
0.200	2.142	1.000	0.239	4.177	0.638	0.153	0.778
0.250	2.283	1.000	0.247	4.047	0.663	0.164	0.796
0.300	2.424	1.000	0.252	3.968	0.679	0.171	0.807
0.350	2.565	1.000	0.261	3.834	0.694	0.181	0.813
0.400	2.706	1.000	0.270	3.709	0.713	0.192	0.824
0.450	2.847	1.000	0.277	3.610	0.727	0.202	0.838
0.500	2.988	1.000	0.287	3.480	0.741	0.213	0.852
0.550	3.129	1.000	0.295	3.395	0.754	0.222	0.867
0.600	3.270	1.000	0.298	3.359	0.762	0.227	0.885
0.650	3.411	1.000	0.309	3.233	0.776	0.240	0.861
0.700	3.552	1.000	0.322	3.108	0.794	0.256	0.875
0.750	3.693	1.000	0.334	2.911	0.814	0.280	0.894
0.800	3.834	1.000	0.366	2.732	0.833	0.305	0.895
0.850	3.975	1.000	0.388	2.560	0.850	0.330	0.906
0.900	4.116	1.000	0.412	2.431	0.866	0.356	0.916
0.950	4.257	1.000	0.436	2.293	0.881	0.384	0.926
1.000	4.398	1.000	0.465	2.153	0.896	0.416	0.935
1.050	4.539	1.000	0.496	2.016	0.910	0.451	0.944
1.100	4.680	1.000	0.528	1.894	0.923	0.488	0.953
1.150	4.821	1.000	0.563	1.777	0.936	0.527	0.965
1.200	4.962	1.000	0.599	1.670	0.948	0.568	0.975
1.250	5.103	1.000	0.648	1.545	0.959	0.621	0.980
1.300	5.244	1.000	0.689	1.453	0.967	0.666	0.986
1.350	5.385	1.000	0.734	1.362	0.975	0.716	0.991
1.400	5.526	1.000	0.789	1.269	0.982	0.775	0.994
1.450	5.667	1.000	0.834	1.199	0.988	0.824	0.997
1.500	5.808	1.000	0.873	1.146	0.992	0.866	0.999
1.550	5.949	1.000	0.913	1.095	0.995	0.909	1.000
1.600	6.090	1.000	0.941	1.064	0.997	0.937	1.000
1.650	6.231	1.000	0.964	1.037	0.998	0.963	1.000
1.700	6.372	1.000	0.981	1.019	0.999	0.980	1.000
1.750	6.513	1.000	0.995	1.005	1.000	0.995	1.000
1.800	6.654	1.000	1.000	1.000	1.000	1.000	1.000
1.850	6.795	1.000	1.000	1.000	1.000	1.000	1.000
1.900	6.936	1.000	1.000	1.000	1.000	1.000	1.000
1.950	7.077	1.000	1.000	1.000	1.000	1.000	1.000
2.000	7.218	1.000	1.000	1.000	1.000	1.000	1.000
2.050	7.359	1.000	1.000	1.000	1.000	1.000	1.000
2.100	7.500	1.000	1.000	1.000	1.000	1.000	1.000
2.150	7.641	1.000	1.000	1.000	1.000	1.000	1.000
2.200	7.782	1.000	1.000	1.000	1.000	1.000	1.000
2.250	7.923	1.000	1.000	1.000	1.000	1.000	1.000
2.300	8.064	1.000	1.000	1.000	1.000	1.000	1.000
2.350	8.205	1.000	1.000	1.000	1.000	1.000	1.000
2.400	8.346	1.000	1.000	1.000	1.000	1.000	1.000
2.450	8.487	1.000	1.000	1.000	1.000	1.000	1.000
2.500	8.628	1.000	1.000	1.000	1.000	1.000	1.000
2.550	8.769	1.000	1.000	1.000	1.000	1.000	1.000
2.600	8.910	1.000	1.000	1.000	1.000	1.000	1.000
2.650	9.051	1.000	1.000	1.000	1.000	1.000	1.000
2.700	9.192	1.000	1.000	1.000	1.000	1.000	1.000
2.750	9.333	1.000	1.000	1.000	1.000	1.000	1.000
2.800	9.474	1.000	1.000	1.000	1.000	1.000	1.000
2.850	9.615	1.000	1.000	1.000	1.000	1.000	1.000
2.900	9.756	1.000	1.000	1.000	1.000	1.000	1.000
2.950	9.897	1.000	1.000	1.000	1.000	1.000	1.000
3.000	10.038	1.000	1.000	1.000	1.000	1.000	1.000

15 DEGREE SHOCK WAVE GENERATOR, α = 20.0 cm



15 DEGREE SHOCK WAVE GENERATOR,  $\alpha = 28.0^\circ$  cm

1.000	1.841	0.725	5.209	0.597	0.431	0.840
1.500	1.841	0.824	4.662	0.672	0.564	0.883
2.000	1.841	0.965	3.978	0.717	0.686	0.906
2.500	1.841	1.203	3.122	0.813	0.978	0.906
3.000	1.841	1.519	2.529	0.871	1.323	0.912
3.500	1.841	1.958	2.093	0.903	1.769	0.957
4.000	1.841	2.769	1.742	0.918	2.562	0.980
4.500	1.841	4.042	1.629	0.930	3.758	0.980
5.000	1.841	6.008	2.466	0.919	4.233	1.000
5.500	1.841	6.625	2.457	0.919	4.252	1.000
6.000	1.841	6.625	2.457	0.912	4.252	1.000
6.500	1.841	6.404	2.580	0.912	4.017	1.000
7.000	1.841	3.713	2.356	0.925	3.435	1.000
7.500	1.841	2.860	1.973	0.947	2.727	1.000
8.000	1.841	2.664	1.934	0.949	2.509	1.000
8.500	1.841	2.545	1.875	0.952	2.424	1.000
9.000	1.841	2.498	1.842	0.954	2.383	1.000
9.500	1.841	2.459	1.849	0.954	2.345	1.000
10.000	1.841	2.422	1.854	0.954	2.309	1.000
10.500	1.841	2.384	1.859	0.953	2.273	1.000
11.000	1.841	2.363	1.863	0.953	2.238	1.000
11.500	1.841	2.318	1.867	0.953	2.204	1.000

15 DEGREE SHOCK WAVE GENERATOR,  $x = 38.0$  cm

$\gamma(\text{CH})$	M	P / P <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO <sub>0</sub> / RHO <sub>inf</sub>	IT / IT <sub>inf</sub>
0.000	0.000	6.818	1.340	4.428	0.000	0.000	0.432
0.050	1.238	6.761	1.157	3.845	0.436	0.504	0.724
0.075	1.285	6.705	1.126	3.594	0.448	0.505	0.764
0.100	1.285	6.648	1.105	6.015	0.459	0.517	0.778
0.125	1.314	6.591	1.087	6.062	0.462	0.512	0.793
0.150	1.358	6.534	1.085	6.020	0.485	0.526	0.801
0.175	1.398	6.500	1.092	5.952	0.496	0.542	0.804
0.200	1.441	6.477	1.103	5.871	0.508	0.561	0.807
0.250	1.464	6.386	1.136	5.623	0.539	0.612	0.812
0.300	1.681	6.307	1.172	5.381	0.588	0.666	0.818
0.350	1.816	6.250	1.222	5.116	0.599	0.731	0.824
0.400	1.962	6.193	1.282	4.832	0.629	0.806	0.830
0.450	2.131	6.136	1.357	4.521	0.660	0.896	0.836
0.500	2.362	6.080	1.476	4.119	0.699	1.031	0.844
0.550	2.773	6.023	1.718	3.505	0.757	1.301	0.861
0.600	3.276	5.966	2.064	2.890	0.812	1.676	0.880
0.650	3.895	5.966	2.582	2.311	0.863	2.229	0.901
0.700	4.628	5.966	3.324	1.944	0.900	2.762	0.923
0.750	5.421	5.795	3.324	1.744	0.928	3.085	0.949
0.800	6.320	5.568	3.212	1.734	0.945	3.033	0.974
0.850	7.434	5.000	2.845	1.757	0.954	2.713	0.991
0.900	8.939	3.750	2.235	1.678	0.963	2.152	0.999
0.950	11.253	3.125	1.957	1.597	0.968	1.894	1.000
1.000	15.428	2.841	1.881	1.510	0.972	1.830	1.000
1.050	20.898	2.841	1.881	1.510	0.972	1.830	1.000
1.100	28.170	2.898	1.920	1.509	0.973	1.867	1.000
1.200	45.329	3.068	1.969	1.558	0.970	1.910	1.000
1.300	55.218	3.293	2.040	1.615	0.967	1.972	1.000
1.400	55.118	3.525	2.112	1.648	0.964	2.035	1.000
1.500	4.954	3.864	2.193	1.762	0.959	2.102	1.000
1.600	4.890	4.091	2.272	1.800	0.956	2.174	1.000
1.700	4.758	4.432	2.354	1.883	0.952	2.241	1.000
1.800	4.642	4.773	2.435	1.960	0.948	2.307	1.000
1.900	4.539	5.114	2.516	2.032	0.944	2.374	1.000
2.000	4.442	5.511	2.594	2.095	0.938	2.434	1.000
2.100	4.326	5.852	2.667	2.155	0.934	2.491	1.000
2.200	4.265	6.136	2.734	2.244	0.932	2.547	1.000
2.300	4.259	6.250	2.778	2.249	0.931	2.587	1.000
2.400	4.272	6.307	2.817	2.238	0.932	2.625	1.000
2.500	4.305	6.307	2.852	2.212	0.933	2.662	1.000
2.600	4.321	6.307	2.869	2.199	0.934	2.679	1.000
2.700	4.340	6.475	2.866	2.080	0.941	2.656	1.000
2.800	4.355	6.475	2.866	2.080	0.941	2.656	1.000
2.900	4.375	6.475	2.866	2.080	0.941	2.656	1.000
3.000	4.390	6.475	2.866	2.080	0.941	2.656	1.000
3.100	4.400	6.475	2.866	2.080	0.941	2.656	1.000
3.200	4.400	6.475	2.866	2.080	0.941	2.656	1.000
3.300	4.400	6.475	2.866	2.080	0.941	2.656	1.000
3.400	4.400	6.475	2.866	2.080	0.941	2.656	1.000
3.500	4.400	6.475	2.866	2.080	0.941	2.656	1.000

5 DEGREE SHOCK WAVE GENERATOR,  $x = 33.0$  cm

(cm)	M	P / Pinf	RNO / RNOinf	T / Tinf	U / Uinf	RNOU / RNOUinf	TT / Tinf
0.00	0.000	3.841	0.848	4.426	0.000	0.000	0.432
0.05	0.301	3.841	0.848	4.364	0.000	0.000	0.432
0.10	0.550	3.841	0.848	4.302	0.000	0.000	0.432
0.15	0.733	3.841	0.848	4.240	0.000	0.000	0.432
0.20	0.861	3.841	0.848	4.178	0.000	0.000	0.432
0.25	0.945	3.841	0.848	4.116	0.000	0.000	0.432
0.30	1.000	3.841	0.848	4.054	0.000	0.000	0.432
0.35	1.036	3.841	0.848	3.992	0.000	0.000	0.432
0.40	1.062	3.841	0.848	3.930	0.000	0.000	0.432
0.45	1.080	3.841	0.848	3.868	0.000	0.000	0.432
0.50	1.090	3.841	0.848	3.806	0.000	0.000	0.432
0.55	1.094	3.841	0.848	3.744	0.000	0.000	0.432
0.60	1.094	3.841	0.848	3.682	0.000	0.000	0.432
0.65	1.090	3.841	0.848	3.620	0.000	0.000	0.432
0.70	1.080	3.841	0.848	3.558	0.000	0.000	0.432
0.75	1.062	3.841	0.848	3.496	0.000	0.000	0.432
0.80	1.036	3.841	0.848	3.434	0.000	0.000	0.432
0.85	1.000	3.841	0.848	3.372	0.000	0.000	0.432
0.90	0.945	3.841	0.848	3.310	0.000	0.000	0.432
0.95	0.861	3.841	0.848	3.248	0.000	0.000	0.432
1.00	0.733	3.841	0.848	3.186	0.000	0.000	0.432
1.05	0.550	3.841	0.848	3.124	0.000	0.000	0.432
1.10	0.301	3.841	0.848	3.062	0.000	0.000	0.432
1.15	0.000	3.841	0.848	3.000	0.000	0.000	0.432
1.20		3.841	0.848	2.938	0.000	0.000	0.432
1.25		3.841	0.848	2.876	0.000	0.000	0.432
1.30		3.841	0.848	2.814	0.000	0.000	0.432
1.35		3.841	0.848	2.752	0.000	0.000	0.432
1.40		3.841	0.848	2.690	0.000	0.000	0.432
1.45		3.841	0.848	2.628	0.000	0.000	0.432
1.50		3.841	0.848	2.566	0.000	0.000	0.432
1.55		3.841	0.848	2.504	0.000	0.000	0.432
1.60		3.841	0.848	2.442	0.000	0.000	0.432
1.65		3.841	0.848	2.380	0.000	0.000	0.432
1.70		3.841	0.848	2.318	0.000	0.000	0.432
1.75		3.841	0.848	2.256	0.000	0.000	0.432
1.80		3.841	0.848	2.194	0.000	0.000	0.432
1.85		3.841	0.848	2.132	0.000	0.000	0.432
1.90		3.841	0.848	2.070	0.000	0.000	0.432
1.95		3.841	0.848	2.008	0.000	0.000	0.432
2.00		3.841	0.848	1.946	0.000	0.000	0.432
2.05		3.841	0.848	1.884	0.000	0.000	0.432
2.10		3.841	0.848	1.822	0.000	0.000	0.432
2.15		3.841	0.848	1.760	0.000	0.000	0.432
2.20		3.841	0.848	1.698	0.000	0.000	0.432
2.25		3.841	0.848	1.636	0.000	0.000	0.432
2.30		3.841	0.848	1.574	0.000	0.000	0.432
2.35		3.841	0.848	1.512	0.000	0.000	0.432
2.40		3.841	0.848	1.450	0.000	0.000	0.432
2.45		3.841	0.848	1.388	0.000	0.000	0.432
2.50		3.841	0.848	1.326	0.000	0.000	0.432
2.55		3.841	0.848	1.264	0.000	0.000	0.432
2.60		3.841	0.848	1.202	0.000	0.000	0.432
2.65		3.841	0.848	1.140	0.000	0.000	0.432
2.70		3.841	0.848	1.078	0.000	0.000	0.432
2.75		3.841	0.848	1.016	0.000	0.000	0.432
2.80		3.841	0.848	0.954	0.000	0.000	0.432
2.85		3.841	0.848	0.892	0.000	0.000	0.432
2.90		3.841	0.848	0.830	0.000	0.000	0.432
2.95		3.841	0.848	0.768	0.000	0.000	0.432
3.00		3.841	0.848	0.706	0.000	0.000	0.432

15 DEGREE SHOCK WAVE GENERATOR,  $x = 42.5\text{cm}$

Y(CM)	M	P / P <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>
1.400	4.223	6.500	2.843	2.286	0.931	2.647	1.003			
1.500	4.310	6.477	2.912	2.225	0.937	2.729	1.007			
1.600	4.381	6.443	2.969	2.170	0.941	2.794	1.009			
1.700	4.436	6.398	3.014	2.123	0.942	2.840	1.006			
1.800	4.483	6.341	3.049	2.080	0.943	2.874	1.003			
1.900	4.524	6.273	3.069	2.044	0.943	2.894	1.000			
2.000	4.557	6.182	3.061	2.020	0.944	2.890	1.000			
2.100	4.583	6.102	3.028	2.015	0.944	2.860	1.000			
2.200	4.559	6.011	2.978	2.019	0.944	2.812	1.000			
2.300	4.570	5.908	2.934	2.010	0.945	2.772	1.000			
2.400	4.568	5.790	2.878	2.008	0.945	2.721	1.000			
2.500	4.645	5.511	2.814	1.998	0.948	2.667	1.000			
2.600	4.727	5.227	2.747	1.993	0.951	2.612	1.000			
2.700	4.789	5.000	2.684	1.963	0.953	2.558	1.000			
2.800	5.043	4.432	2.591	1.710	0.961	2.491	1.000			
2.900	5.201	3.864	2.379	1.624	0.966	2.299	1.000			
3.000	5.193	3.068	1.885	1.628	0.966	1.821	1.000			
3.100	5.111	2.500	1.495	1.672	0.964	1.440	1.000			
3.200	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.300	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.400	4.981	2.455	1.406	1.746	0.960	1.369	1.000			
3.500	4.958	2.477	1.408	1.760	0.959	1.350	1.000			

15 DEGREE SHOCK WAVE GENERATOR,  $x = 60.0\text{cm}$

Y(CM)	M	P / P <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>
1.400	4.223	6.500	2.843	2.286	0.931	2.647	1.003			
1.500	4.310	6.477	2.912	2.225	0.937	2.729	1.007			
1.600	4.381	6.443	2.969	2.170	0.941	2.794	1.009			
1.700	4.436	6.398	3.014	2.123	0.942	2.840	1.006			
1.800	4.483	6.341	3.049	2.080	0.943	2.874	1.003			
1.900	4.524	6.273	3.069	2.044	0.943	2.894	1.000			
2.000	4.557	6.182	3.061	2.020	0.944	2.890	1.000			
2.100	4.583	6.102	3.028	2.015	0.944	2.860	1.000			
2.200	4.559	6.011	2.978	2.019	0.944	2.812	1.000			
2.300	4.570	5.908	2.934	2.010	0.945	2.772	1.000			
2.400	4.568	5.790	2.878	2.008	0.945	2.721	1.000			
2.500	4.645	5.511	2.814	1.998	0.948	2.667	1.000			
2.600	4.727	5.227	2.747	1.993	0.951	2.612	1.000			
2.700	4.789	5.000	2.684	1.963	0.953	2.558	1.000			
2.800	5.043	4.432	2.591	1.710	0.961	2.491	1.000			
2.900	5.201	3.864	2.379	1.624	0.966	2.299	1.000			
3.000	5.193	3.068	1.885	1.628	0.966	1.821	1.000			
3.100	5.111	2.500	1.495	1.672	0.964	1.440	1.000			
3.200	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.300	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.400	4.981	2.455	1.406	1.746	0.960	1.369	1.000			
3.500	4.958	2.477	1.408	1.760	0.959	1.350	1.000			

15 DEGREE SHOCK WAVE GENERATOR,  $x = 50.0\text{cm}$

Y(CM)	M	P / P <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>	U / U <sub>inf</sub>	RHO / RHO <sub>inf</sub>	T / T <sub>inf</sub>
1.400	4.223	6.500	2.843	2.286	0.931	2.647	1.003			
1.500	4.310	6.477	2.912	2.225	0.937	2.729	1.007			
1.600	4.381	6.443	2.969	2.170	0.941	2.794	1.009			
1.700	4.436	6.398	3.014	2.123	0.942	2.840	1.006			
1.800	4.483	6.341	3.049	2.080	0.943	2.874	1.003			
1.900	4.524	6.273	3.069	2.044	0.943	2.894	1.000			
2.000	4.557	6.182	3.061	2.020	0.944	2.890	1.000			
2.100	4.583	6.102	3.028	2.015	0.944	2.860	1.000			
2.200	4.559	6.011	2.978	2.019	0.944	2.812	1.000			
2.300	4.570	5.908	2.934	2.010	0.945	2.772	1.000			
2.400	4.568	5.790	2.878	2.008	0.945	2.721	1.000			
2.500	4.645	5.511	2.814	1.998	0.948	2.667	1.000			
2.600	4.727	5.227	2.747	1.993	0.951	2.612	1.000			
2.700	4.789	5.000	2.684	1.963	0.953	2.558	1.000			
2.800	5.043	4.432	2.591	1.710	0.961	2.491	1.000			
2.900	5.201	3.864	2.379	1.624	0.966	2.299	1.000			
3.000	5.193	3.068	1.885	1.628	0.966	1.821	1.000			
3.100	5.111	2.500	1.495	1.672	0.964	1.440	1.000			
3.200	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.300	5.005	2.432	1.404	1.732	0.960	1.348	1.000			
3.400	4.981	2.455	1.406	1.746	0.960	1.369	1.000			
3.500	4.958	2.477	1.408	1.760	0.959	1.350	1.000			

15 DEGREE SHOCK WAVE GENERATOR, x = 70.0cm

Y(CM)	M	P / Pinf	RHO / RHOinf	T / Tinf	U / Uinf	RHO / RHOinf	TT / TTinf
0.000	0.000	2.455	0.555	4.426	0.000	0.000	0.432
0.050	2.240	2.455	0.614	3.995	0.653	0.401	0.778
0.075	2.680	2.455	0.694	3.536	0.735	0.510	0.835
0.100	2.874	2.455	0.730	3.363	0.769	0.561	0.864
0.125	2.936	2.455	0.729	3.365	0.785	0.573	0.886
0.150	2.997	2.455	0.739	3.322	0.796	0.588	0.898
0.175	3.039	2.455	0.745	3.296	0.804	0.599	0.906
0.200	3.115	2.455	0.762	3.223	0.815	0.621	0.915
0.250	3.212	2.455	0.782	3.139	0.829	0.649	0.927
0.300	3.305	2.455	0.804	3.053	0.842	0.677	0.938
0.350	3.402	2.455	0.829	2.962	0.854	0.707	0.946
0.400	3.485	2.455	0.850	2.886	0.863	0.734	0.953
0.450	3.565	2.443	0.868	2.814	0.872	0.757	0.960
0.500	3.625	2.432	0.881	2.759	0.878	0.774	0.964
0.600	3.732	2.409	0.903	2.668	0.889	0.802	0.972
0.700	3.813	2.386	0.918	2.600	0.896	0.823	0.977
0.800	3.885	2.364	0.929	2.543	0.903	0.840	0.983
0.900	3.938	2.352	0.940	2.501	0.908	0.854	0.986
1.000	3.995	2.341	0.953	2.456	0.913	0.870	0.990
1.100	4.052	2.341	0.971	2.411	0.917	0.890	0.993
1.200	4.112	2.341	0.991	2.363	0.921	0.913	0.995
1.300	4.166	2.341	1.009	2.319	0.925	0.934	0.997
1.400	4.235	2.330	1.029	2.265	0.929	0.956	0.998
1.500	4.289	2.330	1.048	2.222	0.932	0.977	0.999
1.600	4.341	2.330	1.068	2.181	0.935	0.998	0.999
1.700	4.408	2.318	1.088	2.130	0.938	1.021	1.000
1.800	4.464	2.318	1.110	2.088	0.940	1.044	1.000
1.900	4.524	2.318	1.134	2.044	0.943	1.069	1.000
2.000	4.586	2.307	1.154	2.000	0.945	1.091	1.000
2.100	4.640	2.307	1.176	1.962	0.947	1.114	1.000
2.200	4.693	2.307	1.198	1.926	0.949	1.137	1.000
2.300	4.757	2.295	1.219	1.883	0.952	1.160	1.000
2.400	4.813	2.295	1.242	1.848	0.954	1.185	1.000
2.500	4.865	2.295	1.265	1.815	0.956	1.208	1.000
2.600	4.929	2.284	1.285	1.777	0.958	1.231	1.000
2.700	4.983	2.284	1.309	1.745	0.960	1.256	1.000
2.800	5.038	2.273	1.327	1.713	0.961	1.276	1.000
2.900	5.081	2.273	1.346	1.689	0.963	1.295	1.000
3.000	5.119	2.273	1.363	1.668	0.964	1.313	1.000
3.100	5.170	2.261	1.378	1.641	0.965	1.331	1.000
3.200	5.203	2.261	1.394	1.623	0.966	1.347	1.000
3.300	5.243	2.250	1.405	1.602	0.967	1.359	1.000
3.400	5.276	2.250	1.420	1.585	0.968	1.375	1.000
3.500	5.306	2.250	1.433	1.570	0.969	1.389	1.000

\*\*\*\*\*BOUNDARY-LAYER PARAMETERS THROUGH THE INTERACTION\*\*\*\*\*

15 DEGREE SHOCK WAVE GENERATOR

x,cm	Delta	Delta*	theta	Delta*i	thetai
20.0	2.70	1.393	0.096	0.354	0.254
25.5	2.70	1.374	0.108	0.369	0.266
28.0	2.80	1.332	0.123	0.444	0.271
30.5	2.85	2.250	0.098	0.913	0.412
33.0	2.20	2.018	0.026	1.399	0.158
35.5	1.70	0.952	0.104	0.544	0.245
38.0	1.30	0.304	0.126	0.312	0.182
40.0	1.32	0.438	0.088	0.264	0.157
42.5	1.45	0.251	0.101	0.247	0.151
45.0	1.60	0.485	0.080	0.203	0.138
50.0	2.00	0.577	0.083	0.175	0.130
55.0	2.40	0.660	0.095	0.181	0.137
60.0	2.70	0.781	0.106	0.207	0.160
65.0	2.95	0.925	0.101	0.207	0.160
70.0	3.10	0.897	0.098	0.191	0.148
75.0	3.25	0.861	0.099	0.190	0.146
85.0	3.40	1.020	0.106	0.224	0.171
95.0	3.50	0.919	0.117	0.235	0.178

\*\*\*\*\*END OF FILE\*\*\*\*\*

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Ref.: 88, 29, 87

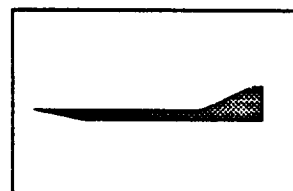
Author: Smits, A. J., *et al*

Geometry: 2-D Compression Corner

Mach number: 3

Data:  $p_{wall}$ ,  $c_f$ , mean & fluctuating flowfield surveys (pitot and hot-wire anemometry)

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Smits, A.J. and Muck, K.C., "Experimental Study of Three Shock Wave/Turbulent Boundary Layer Interactions," *Journal of Fluid Mechanics*, Vol. 182, Sept. 1987, pp. 291-314.

Fernholz, H.H., Finley, P.J., Dussauge, J.P. and Smits, A.J., "A Survey of Measurements and Measuring Techniques in Rapidly Distorted Compressible Turbulent Boundary Layers," *AGARDograph* 315, 1989.

Settles, G.S., Gilbert, R.B. and Bogdonoff, S.M., "Data Compilation For Shock Wave/Turbulent Boundary Layer Interaction Experiments On Two-Dimensional Compression Corners," *Princeton University Report 1489-MAE*, Princeton Univ. 1980.

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The data consist of both mean and fluctuation surveys of flow properties before and after two-dimensional compression corners at Mach numbers in the vicinity of 2.9. Compression corner angles of 8, 16, 20, and 24 degrees span the range from attached flow to large boundary-layer separation. Two-dimensionality of the experiments was demonstrated by studies of spanwise oil-flow patterns for all but the largest compression corner angle, where significant 3-D perturbations were observed.

The mean data include surface pressure and skin friction distributions, as well as pitot and static pressure distributions from which velocity and Mach number were deduced (the total temperature distribution through these interactions was nearly constant). The fluctuation data are derived from constant-temperature hot-wire anemometer surveys using both normal and inclined wires, and yielding both mass-flux fluctuation and Reynolds shear-stress profiles.

All units in the tables are SI. The x-coordinate is defined in the streamwise direction along the wind tunnel floor and compression corner surfaces. Thus locations upstream of the corner have negative x-values and those downstream have positive values. The compression corners were all located at 1.205 m downstream of the wind tunnel nozzle exit with the exception of the 24 degree corner for hot-wire measurements only, which was located 1.17 m downstream of the nozzle exit. The y-coordinate is measured upward from the test surface with its origin at that surface. The origin of the z-coordinate is on the wind tunnel centerline. It is taken positive to the left when looking downstream.

Some difficulties arose purely insofar as the data were taken by several different experimenters over a period of several years. For example, the mean survey locations did not always match those of the hot-wire, requiring some interpolation in order to reduce the latter data.

These data have already been presented to the research community in tabulated form, first in the data tapes of the 1980-81 AFOSR-HTTM-Stanford Conference on Complex Turbulent Flows (mean data only), and more recently in the data tape accompanying AGARDograph 315 (Ref. 88). Nonetheless, on account of their high value for compressible turbulence modeling, we are including the full set of profiles here as well.

Nothing has been added to this dataset since the publication of Ref. 88, so that the current listing is a verbatim recount of these data. Users are highly encouraged to consult Refs. 88 and 29 for detailed discussions of the data and their significance, which are beyond the present scope, as well as estimates of the various errors and discrepancies which serve to set confidence limits upon the data.



1.070E-01	6.894E+05	2.780E+02	9.300E+04	8.130E-04	1.130E-03	2380	1	1.980E+02	1.487E-03
1.170E-01	6.894E+05	2.780E+02	9.424E+04	8.130E-04	1.154E-03	2380	1	2.022E+02	1.518E-03
1.270E-01	6.894E+05	2.780E+02	9.424E+04	8.130E-04	1.154E-03	2380	1	2.022E+02	1.518E-03
1.370E-01	6.894E+05	2.780E+02	9.645E+04	8.130E-04	1.184E-03	2380	1	2.075E+02	1.558E-03

ALPHA=16 DEG RAMP, Z = - 0.0127

XLOC	PT1	TT1	PRESTON	D	CF	RUN	TEST	TAUWAL	CFINF
-3.810E-02	6.894E+05	2.650E+02	5.749E+04	8.380E-04	1.016E-03	2572	1	1.338E+02	9.835E-04
-2.540E-02	6.894E+05	2.650E+02	5.749E+04	8.380E-04	1.016E-03	2572	1	1.338E+02	9.835E-04
-1.270E-02	6.894E+05	2.650E+02	5.904E+04	8.380E-04	1.052E-03	2572	1	1.385E+02	1.018E-03
-9.530E-03	6.894E+05	2.650E+02	5.475E+04	8.380E-04	9.607E-04	2572	1	1.265E+02	9.299E-04
-6.350E-03	6.894E+05	2.650E+02	4.437E+04	8.380E-04	5.799E-04	2572	1	8.351E+01	6.139E-04
-3.180E-03	6.894E+05	2.650E+02	3.494E+04	8.380E-04	2.140E-05	2572	1	3.498E+00	2.572E-05
0.000E+00	6.894E+05	2.650E+02	4.561E+04	8.380E-04	1.852E-04	2572	1	3.270E+01	2.404E-04
3.180E-03	6.894E+05	2.650E+02	5.141E+04	8.380E-04	1.864E-04	2572	1	3.493E+01	2.568E-04
6.350E-03	6.894E+05	2.650E+02	6.241E+04	8.380E-04	3.725E-04	2572	1	7.186E+01	5.282E-04
9.530E-03	6.894E+05	2.650E+02	7.046E+04	8.380E-04	4.826E-04	2572	1	9.512E+01	6.993E-04
1.270E-02	6.894E+05	2.650E+02	7.516E+04	8.380E-04	5.347E-04	2572	1	1.069E+02	7.858E-04
1.910E-02	6.894E+05	2.650E+02	8.686E+04	8.380E-04	6.801E-04	2572	1	1.388E+02	1.021E-03
2.540E-02	6.894E+05	2.650E+02	9.204E+04	8.380E-04	7.068E-04	2572	1	1.473E+02	1.083E-03
3.810E-02	6.894E+05	2.650E+02	1.027E+05	8.380E-04	8.128E-04	2572	1	1.729E+02	1.271E-03
5.080E-02	6.894E+05	2.650E+02	1.129E+05	8.380E-04	9.155E-04	2572	1	1.973E+02	1.451E-03
7.620E-02	6.894E+05	2.650E+02	1.285E+05	8.380E-04	1.074E-03	2572	1	2.346E+02	1.725E-03
1.020E-01	6.894E+05	2.650E+02	1.379E+05	8.380E-04	1.178E-03	2572	1	2.574E+02	1.892E-03
1.270E-01	6.894E+05	2.650E+02	1.441E+05	8.380E-04	1.252E-03	2572	1	2.735E+02	2.011E-03
1.400E-01	6.894E+05	2.650E+02	1.484E+05	8.380E-04	1.293E-03	2572	1	2.825E+02	2.077E-03

ALPHA=20 DEG RAMP, Z = -0.0127

XLOC	PT1	TT1	PRESTON	D	CF	RUN	TEST	TAUWAL	CFINF
-3.810E-02	6.894E+05	2.670E+02	6.010E+04	8.130E-04	1.067E-03	2305	1	1.417E+02	9.968E-04
-2.220E-02	6.894E+05	2.670E+02	5.840E+04	8.130E-04	8.531E-04	2305	1	1.258E+02	8.854E-04
-1.910E-02	6.894E+05	2.670E+02	5.120E+04	8.130E-04	4.935E-04	2305	1	8.131E+01	5.721E-04
-1.590E-02	6.894E+05	2.670E+02	4.620E+04	8.130E-04	2.620E-04	2305	1	4.521E+01	3.181E-04
-1.270E-02	6.894E+05	2.670E+02	4.400E+04	8.130E-04	1.156E-04	2305	1	2.056E+01	1.447E-04
-1.110E-02	6.894E+05	2.670E+02	4.450E+04	8.130E-04	1.101E-04	2305	1	1.973E+01	1.388E-04
3.970E-03	6.894E+05	2.670E+02	5.170E+04	8.130E-04	5.743E-05	2305	1	1.124E+01	7.907E-05
6.350E-03	6.894E+05	2.670E+02	5.670E+04	8.130E-04	8.338E-05	2305	1	1.690E+01	1.189E-04
9.530E-03	6.894E+05	2.670E+02	6.330E+04	8.130E-04	1.876E-04	2305	1	3.883E+01	2.732E-04
1.270E-02	6.894E+05	2.670E+02	6.840E+04	8.130E-04	2.153E-04	2305	1	4.580E+01	3.222E-04
1.590E-02	6.894E+05	2.670E+02	7.450E+04	8.130E-04	3.465E-04	2305	1	7.370E+01	5.185E-04
1.910E-02	6.894E+05	2.670E+02	7.997E+04	8.130E-04	3.686E-04	2305	1	8.051E+01	5.665E-04
2.220E-02	6.894E+05	2.670E+02	8.520E+04	8.130E-04	4.012E-04	2305	1	8.936E+01	6.287E-04
2.540E-02	6.894E+05	2.670E+02	8.955E+04	8.130E-04	4.354E-04	2305	1	9.822E+01	6.911E-04
3.180E-02	6.894E+05	2.670E+02	9.790E+04	8.130E-04	5.137E-04	2305	1	1.181E+02	8.308E-04
4.130E-02	6.894E+05	2.670E+02	1.031E+05	8.130E-04	5.746E-04	2305	1	1.329E+02	9.351E-04
4.450E-02	6.894E+05	2.670E+02	1.158E+05	8.130E-04	6.992E-04	2305	1	1.647E+02	1.159E-03
5.720E-02	6.894E+05	2.670E+02	1.241E+05	8.130E-04	7.489E-04	2305	1	1.796E+02	1.263E-03
7.620E-02	6.894E+05	2.670E+02	1.400E+05	8.130E-04	9.302E-04	2305	1	2.243E+02	1.578E-03
9.530E-02	6.894E+05	2.670E+02	1.510E+05	8.130E-04	1.057E-03	2305	1	2.550E+02	1.794E-03
1.140E-01	6.894E+05	2.670E+02	1.579E+05	8.130E-04	1.130E-03	2305	1	2.726E+02	1.918E-03

ALPHA=24 DEG RAMP

XLOC	PT1	TT1	PRESTON	D	CF	RUN	TEST	TAUWAL	CFINF
-6.350E-02	6.894E+05	2.620E+02	6.949E+04	1.570E-03	1.050E-03	972	1	1.406E+02	1.034E-03
-5.080E-02	6.894E+05	2.620E+02	6.970E+04	1.570E-03	1.050E-03	972	1	1.406E+02	1.034E-03
-4.570E-02	6.894E+05	2.620E+02	5.970E+04	1.570E-03	8.815E-04	972	1	1.180E+02	8.676E-04
-4.320E-02	6.894E+05	2.620E+02	5.060E+04	1.570E-03	7.213E-04	972	1	9.654E+01	7.100E-04
-4.060E-02	6.894E+05	2.620E+02	4.880E+04	1.570E-03	6.854E-04	972	1	9.174E+01	6.746E-04
-3.810E-02	6.894E+05	2.620E+02	4.680E+04	1.570E-03	6.464E-04	972	1	8.652E+01	6.363E-04
-3.450E-02	6.894E+05	2.620E+02	4.640E+04	1.570E-03	6.388E-04	972	1	8.549E+01	6.287E-04
-3.300E-02	6.894E+05	2.620E+02	4.640E+04	1.570E-03	6.388E-04	972	1	8.549E+01	6.287E-04
1.020E-02	6.894E+05	2.620E+02	6.360E+04	1.570E-03	5.811E-05	972	1	1.244E+01	9.152E-05
1.520E-02	6.894E+05	2.620E+02	7.080E+04	1.570E-03	1.472E-04	972	1	3.216E+01	2.365E-04
2.030E-02	6.894E+05	2.620E+02	7.840E+04	1.570E-03	1.861E-04	972	1	4.197E+01	3.087E-04
3.050E-02	6.894E+05	2.620E+02	9.240E+04	1.570E-03	2.758E-04	972	1	6.496E+01	4.777E-04
5.080E-02	6.894E+05	2.620E+02	1.188E+05	1.570E-03	4.784E-04	972	1	1.180E+02	8.679E-04

6.100E-02	6.894E+05	2.620E+02	1.325E+05	1.570E-03	6.203E-04	972	1	1.539E+02	1.132E-03
1.090E-01	6.894E+05	2.620E+02	1.588E+05	1.570E-03	8.505E-04	972	1	2.133E+02	1.569E-03
1.020E-01	6.894E+05	2.620E+02	1.802E+05	1.570E-03	9.920E-04	972	1	2.528E+02	1.859E-03
1.220E-01	6.894E+05	2.620E+02	1.984E+05	1.570E-03	1.142E-03	972	1	2.910E+02	2.140E-03
1.420E-01	6.894E+05	2.620E+02	2.127E+05	1.570E-03	1.250E-03	972	1	3.185E+02	2.342E-03

\*\*\*\*\*MEAN PROFILE DATA\*\*\*\*\*

MEAN PROFILE TABULATION ALPHA = 0 DEG

SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL

X = -.2540E-01

Z = -.1270E-01

Stagnation Pressure pitot = 0.6804E+06

Stagnation Temperature pitot = 282.9

M ref = 2.870

U ref pitot = 594.9

P wall = 0.2300E+05

TAU wall preston = 133.5

Y	PT/PWALL	PS/PWALL	U/UREF	M
0.7026E-03	2.060	0.9848	0.5630	1.083
0.8319E-03	2.519	0.9848	0.6292	1.250
0.9934E-03	2.780	0.9844	0.6607	1.335
0.1026E-02	3.010	0.9843	0.6847	1.402
0.1316E-02	3.204	0.9836	0.7037	1.458
0.1478E-02	3.313	0.9831	0.7130	1.487
0.1672E-02	3.487	0.9826	0.7289	1.536
0.1995E-02	3.639	0.9818	0.7413	1.575
0.2383E-02	3.834	0.9824	0.7565	1.625
0.2835E-02	3.996	0.9847	0.7665	1.660
0.3190E-02	4.176	0.9866	0.7784	1.703
0.3546E-02	4.387	0.9884	0.7909	1.749
0.3932E-02	4.552	0.9873	0.8016	1.789
0.4613E-02	4.713	0.9821	0.8110	1.825
0.5192E-02	4.909	0.9782	0.8250	1.874
0.5872E-02	5.096	0.9757	0.8353	1.917
0.6518E-02	5.348	0.9731	0.8483	1.974
0.7163E-02	5.470	0.9700	0.8545	2.002
0.7681E-02	5.709	0.9680	0.8651	2.051
0.8456E-02	5.883	0.9707	0.8698	2.080
0.9134E-02	6.057	0.9712	0.8782	2.115
0.9749E-02	6.243	0.9686	0.8863	2.151
0.1039E-01	6.400	0.9659	0.8940	2.186
0.1117E-01	6.578	0.9664	0.8997	2.214
0.1172E-01	6.809	0.9670	0.9082	2.257
0.1230E-01	7.004	0.9671	0.9149	2.292
0.1295E-01	7.226	0.9657	0.9227	2.335
0.1362E-01	7.409	0.9644	0.9277	2.363
0.1427E-01	7.600	0.9651	0.9340	2.398
0.1492E-01	7.835	0.9668	0.9402	2.434
0.1556E-01	7.939	0.9686	0.9425	2.448
0.1647E-01	8.096	0.9721	0.9458	2.469
0.1686E-01	8.404	0.9736	0.9540	2.519
0.1773E-01	8.683	0.9764	0.9591	2.554
0.1870E-01	8.943	0.9789	0.9642	2.597
0.1954E-01	9.117	0.9808	0.9668	2.618
0.2041E-01	9.417	0.9824	0.9733	2.660
0.2141E-01	9.709	0.9832	0.9797	2.703
0.2235E-01	10.03	0.9825	0.9868	2.752
0.2344E-01	10.15	0.9822	0.9886	2.766
0.2451E-01	10.16	0.9822	0.9880	2.766
0.2558E-01	10.49	0.9806	0.9939	2.816
0.2654E-01	10.62	0.9790	0.9956	2.837
0.2807E-01	10.66	0.9764	0.9965	2.851
0.2924E-01	10.71	0.9744	0.9973	2.859
0.3045E-01	10.74	0.9753	0.9973	2.859
0.3203E-01	10.76	0.9740	0.9972	2.866
0.3399E-01	10.75	0.9689	0.9987	2.873
0.3602E-01	10.78	0.9702	0.9982	2.873
0.3815E-01	10.84	0.9741	0.9987	2.880
0.4056E-01	10.76	0.9730	0.9973	2.866



0.4205E-01	10.73	0.0689	0.0981	2.873	
MEAN PROFILE TABULATION ALPHA = 0 DEG					
SURVEY NORMAL TO TUNNEL FLOOR VERTICAL					
X	= 0.0000E+00				
Z	= -1270E-01				
Stagnation Pressure	pitot = 0.6700E+06				
Stagnation Temperature	pitot = 238.6				
M ref	= 2.870				
U ref	pitot = 546.4				
P wall	= 0.3151E+05				
TAU wall	preston = 103.9				
Y	PT/PMALL	PS/PMALL	U/UREF	M	
0.2485E-02	3.011	0.8296	0.7343	1.558	
0.3099E-02	3.162	0.7685	0.7598	1.643	
0.3846E-02	3.332	0.7535	0.7807	1.745	
0.4816E-02	3.573	0.7311	0.8167	1.866	
0.5766E-02	3.780	0.7204	0.8337	1.924	
0.7361E-02	4.059	0.7168	0.8513	2.002	
0.8054E-02	4.307	0.7152	0.8670	2.075	
0.8961E-02	4.487	0.7130	0.8779	2.122	
0.9898E-02	4.614	0.7119	0.8855	2.158	
0.1084E-01	4.802	0.7113	0.8955	2.207	
0.1181E-01	5.040	0.7119	0.9065	2.264	
0.1255E-01	5.151	0.7122	0.9124	2.292	
0.1336E-01	5.322	0.7122	0.9196	2.328	
0.1420E-01	5.605	0.7116	0.9326	2.398	
0.1521E-01	5.763	0.7100	0.9366	2.434	
0.1637E-01	6.036	0.7120	0.9478	2.490	
0.1744E-01	6.223	0.7150	0.9530	2.526	
0.1874E-01	6.534	0.7165	0.9637	2.589	
0.1990E-01	6.693	0.7175	0.9698	2.625	
0.2081E-01	6.947	0.7172	0.9774	2.674	
0.2178E-01	7.242	0.7169	0.9858	2.731	
0.2291E-01	7.277	0.7178	0.9840	2.738	
0.2405E-01	7.502	0.7192	0.9899	2.781	
0.2521E-01	7.648	0.7197	0.9920	2.809	
0.2657E-01	7.728	0.7197	0.9925	2.823	
0.2784E-01	7.826	0.7185	0.9951	2.844	
0.2941E-01	7.858	0.7171	0.9967	2.859	
0.3078E-01	7.864	0.7160	0.9966	2.866	
0.3241E-01	7.871	0.7157	0.9958	2.866	
0.3396E-01	7.902	0.7158	0.9943	2.866	
0.3548E-01	7.909	0.7159	0.9972	2.866	
0.3701E-01	7.902	0.7157	0.9976	2.866	
0.3846E-01	7.883	0.7147	0.9971	2.866	
0.3990E-01	7.880	0.7133	0.9948	2.859	
0.4145E-01	7.848	0.7124	0.9958	2.866	
0.4285E-01	7.832	0.7096	0.9945	2.873	
0.4409E-01	7.848	0.7067	0.9972	2.873	
0.4478E-01	7.826	0.7057	0.9972	2.873	
0.4678E-01	7.845	0.7057	0.9972	2.873	

MEAN PROFILE TABULATION ALPHA = 0 DEG

SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL  
X  
Z  
Stagnation Pressure  
Stagnation Temperature  
M ref  
U ref  
P wall  
TAU wall

Y PT/PMALL PS/PMALL U/UREF M

0.2380E-02 2.990 0.9393 0.6946 1.437  
0.3155E-02 3.141 0.9110 0.7171 1.508  
0.3962E-02 3.292 0.8218 0.7601 1.508  
0.4641E-02 3.390 0.7676 0.7871 1.745

MEAN PROFILE TABULATION ALPHA = 0 DEG

SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL  
X  
Z  
Stagnation Pressure  
Stagnation Temperature  
M ref  
U ref  
P wall  
TAU wall

Y PT/PMALL PS/PMALL U/UREF M

0.2348E-02 3.046 0.9583 0.6938 1.437  
0.2756E-02 3.150 0.9498 0.7052 1.473  
0.3122E-02 3.327 0.9454 0.7237 1.529  
0.3607E-02 3.511 0.9385 0.7415 1.586  
0.4026E-02 3.608 0.9279 0.7524 1.621  
0.4415E-02 3.694 0.9078 0.7653 1.664  
0.4963E-02 3.694 0.8694 0.7780 1.706  
0.5575E-02 3.730 0.8121 0.7995 1.782  
0.6190E-02 3.747 0.7529 0.8219 1.867  
0.6725E-02 3.833 0.7234 0.8379 1.931  
0.7188E-02 3.943 0.6996 0.8548 2.002  
0.7663E-02 4.022 0.6830 0.8653 2.051  
0.8611E-02 4.114 0.6735 0.8722 2.087  
0.9352E-02 4.309 0.6675 0.8849 2.151  
0.1006E-01 4.466 0.6668 0.8954 2.200  
0.1074E-01 4.626 0.6669 0.9028 2.235  
0.1138E-01 4.771 0.6656 0.9114 2.278  
0.1200E-01 4.892 0.6641 0.9182 2.313  
0.1271E-01 4.966 0.6626 0.9220 2.335  
0.1355E-01 5.126 0.6617 0.9296 2.377  
0.1452E-01 5.288 0.6618 0.9356 2.412  
0.1526E-01 5.288 0.6631 0.9450 2.469  
0.1590E-01 5.661 0.6648 0.9494 2.497  
0.1674E-01 5.768 0.6658 0.9524 2.519  
0.1729E-01 5.883 0.6663 0.9572 2.547  
0.1816E-01 6.099 0.6669 0.9660 2.597  
0.1881E-01 6.241 0.6671 0.9710 2.625  
0.1939E-01 6.294 0.6674 0.9737 2.639  
0.1981E-01 6.365 0.6678 0.9760 2.653  
0.2052E-01 6.528 0.6698 0.9794 2.682  
0.2102E-01 6.644 0.6717 0.9828 2.710  
0.2178E-01 6.755 0.6733 0.9842 2.724

0.2259E-01	6.977	0.6743	0.9893	2.766	0.2503E-01	7.202	0.6666	0.9938	2.830
0.2316E-01	7.030	0.6752	0.9904	2.781	0.2602E-01	7.234	0.6667	0.9938	2.837
0.2430E-01	7.093	0.6757	0.9903	2.780	0.2701E-01	7.265	0.6660	0.9954	2.844
0.2520E-01	7.128	0.6761	0.9915	2.802	0.2802E-01	7.275	0.6652	0.9955	2.851
0.2604E-01	7.246	0.6761	0.9938	2.837	0.3185E-01	7.307	0.6665	0.9951	2.851
0.2700E-01	7.320	0.6750	0.9950	2.837	0.3393E-01	7.286	0.6659	0.9947	2.851
0.2804E-01	7.374	0.6757	0.9959	2.851	0.3630E-01	7.266	0.6650	0.9951	2.851
0.2916E-01	7.397	0.6758	0.9971	2.859	0.3868E-01	7.304	0.6654	0.9973	2.859
0.3012E-01	7.439	0.6758	0.9985	2.866	0.4102E-01	7.330	0.6661	0.9985	2.859
0.3147E-01	7.439	0.6745	0.9985	2.866	MEAN PROFILE TABULATION ALPHA = 0 DEG				
0.3269E-01	7.436	0.6754	0.9968	2.859	SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL				
0.3416E-01	7.465	0.6776	0.9956	2.859	X	= 0.2540E-01			
0.3612E-01	7.465	0.6799	0.9957	2.859	Z	= -0.1270E-01			
0.3787E-01	7.477	0.6797	0.9962	2.859	Stagnation Pressure	= 0.6822E+06			
0.3975E-01	7.498	0.6793	0.9980	2.866	Stagnation Temperature	= 272.1			
MEAN PROFILE TABULATION ALPHA = 0 DEG					M ref	= 2.870			
SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL					U ref	= 583.5			
X	= 0.1016E-01				P wall	= 0.3565E+05			
Z	= -0.1270E-01				TAU wall	= 154.3			
Stagnation Pressure	= 0.6822E+06				Y	PT/PWALL	PS/PWALL	U/UREF	M
Stagnation Temperature	= 272.1				0.2348E-02	3.173	0.9698	0.7013	1.462
M ref	= 2.870				0.2799E-02	3.310	0.9713	0.7126	1.497
U ref	= 584.1				0.3188E-02	3.405	0.9692	0.7213	1.526
P wall	= 0.3427E+05				0.3543E-02	3.534	0.9669	0.7321	1.561
Preston	= 125.8				0.3995E-02	3.624	0.9643	0.7404	1.589
Y	PT/PWALL	PS/PWALL	U/UREF	M	0.4348E-02	3.798	0.9637	0.7528	1.632
0.2510E-02	3.192	0.9718	0.7028	1.466	0.4834E-02	3.969	0.9599	0.7661	1.678
0.2832E-02	3.344	0.9715	0.7167	1.508	0.5286E-02	4.076	0.9557	0.7752	1.710
0.3221E-02	3.379	0.9670	0.7210	1.522	0.5705E-02	4.261	0.9509	0.7890	1.759
0.3574E-02	3.560	0.9618	0.7375	1.575	0.6208E-02	4.387	0.9460	0.7989	1.797
0.3894E-02	3.729	0.9595	0.7583	1.621	0.6543E-02	4.544	0.9433	0.8080	1.832
0.4219E-02	3.826	0.9570	0.7583	1.646	0.7092E-02	4.634	0.9427	0.8129	1.853
0.4801E-02	3.974	0.9507	0.7701	1.689	0.7513E-02	4.825	0.9420	0.8232	1.896
0.5189E-02	4.111	0.9441	0.7818	1.731	0.7838E-02	4.912	0.9412	0.8285	1.917
0.5512E-02	4.190	0.9379	0.7888	1.756	0.8158E-02	5.097	0.9390	0.8349	1.959
0.5834E-02	4.275	0.9324	0.7962	1.782	0.8481E-02	5.189	0.9369	0.8449	1.981
0.6157E-02	4.421	0.9285	0.8056	1.818	0.9030E-02	5.307	0.9337	0.8522	2.009
0.6609E-02	4.497	0.9181	0.8134	1.846	0.9482E-02	5.445	0.9314	0.8592	2.037
0.7125E-02	4.578	0.8996	0.8244	1.889	0.9771E-02	5.579	0.9307	0.8653	2.066
0.7447E-02	4.517	0.8803	0.8260	1.896	0.1009E-01	5.734	0.9302	0.8728	2.101
0.7770E-02	4.500	0.8612	0.8399	1.952	0.1042E-01	5.820	0.9296	0.8755	2.115
0.8415E-02	4.566	0.8412	0.8514	2.002	0.1087E-01	5.978	0.9263	0.8827	2.151
0.9286E-02	4.517	0.7585	0.8642	2.058	0.1125E-01	6.115	0.9233	0.8901	2.186
0.9835E-02	4.485	0.7303	0.8720	2.094	0.1138E-01	6.132	0.9223	0.8901	2.186
0.1035E-01	4.584	0.7057	0.8858	2.158	0.1171E-01	6.261	0.9184	0.8960	2.214
0.1081E-01	4.637	0.6866	0.8960	2.207	0.1203E-01	6.325	0.9144	0.9003	2.235
0.1125E-01	4.756	0.6763	0.9060	2.257	0.1232E-01	6.450	0.9108	0.9101	2.285
0.1164E-01	4.856	0.6709	0.9128	2.292	0.1255E-01	6.527	0.9076	0.9182	2.285
0.1209E-01	4.990	0.6661	0.9208	2.335	0.1290E-01	6.597	0.9028	0.9150	2.299
0.1253E-01	5.171	0.6575	0.9308	2.335	0.1323E-01	6.583	0.8947	0.9158	2.315
0.1317E-01	5.363	0.6565	0.9393	2.391	0.1355E-01	6.555	0.8851	0.9173	2.320
0.1435E-01	5.404	0.6559	0.9414	2.441	0.1387E-01	6.468	0.8676	0.9187	2.328
0.1500E-01	5.605	0.6552	0.9495	2.455	0.1429E-01	6.452	0.8606	0.9254	2.363
0.1555E-01	5.693	0.6549	0.9528	2.505	0.1465E-01	6.295	0.7994	0.9318	2.398
0.1610E-01	5.754	0.6553	0.9548	2.526	0.1506E-01	6.199	0.7555	0.9406	2.448
0.1668E-01	5.897	0.6557	0.9593	2.568	0.1555E-01	6.143	0.6976	0.9491	2.497
0.1722E-01	6.067	0.6562	0.9659	2.611	0.1603E-01	6.126	0.6976	0.9561	2.540
0.1800E-01	6.186	0.6571	0.9691	2.631	0.1648E-01	6.121	0.6772	0.9617	2.575
0.1842E-01	6.294	0.6581	0.9723	2.653	0.1687E-01	6.182	0.6642	0.9683	2.618
0.1926E-01	6.428	0.6602	0.9766	2.682	0.1722E-01	6.258	0.6474	0.9759	2.670
0.2008E-01	6.566	0.6620	0.9809	2.710	0.1816E-01	6.328	0.6370	0.9800	2.714
0.2065E-01	6.697	0.6633	0.9842	2.738	0.1878E-01	6.429	0.6328	0.9845	2.745
0.2119E-01	6.735	0.6647	0.9853	2.758	0.1942E-01	6.474	0.6313	0.9868	2.759
0.2213E-01	6.819	0.6661	0.9861	2.782	0.2013E-01	6.606	0.6304	0.9912	2.788
0.2275E-01	6.942	0.6683	0.9891	2.781	0.2078E-01	6.628	0.6302	0.9917	2.795
0.2352E-01	7.029	0.6685	0.9897	2.795	0.2139E-01	6.743	0.6299	0.9937	2.816
0.2426E-01	7.053	0.6685	0.9897	2.802	0.2203E-01	6.760	0.6293	0.9936	2.823
0.2488E-01	7.132	0.6665	0.9917	2.816	0.2291E-01	6.791	0.6286	0.9938	2.830
					0.2358E-01	6.872	0.6286	0.9960	2.851

0.2430E-01	6.909	0.6286	0.9963	2.859	0.2280E-01	6.498	0.5740	1.004	2.901
0.2526E-01	6.934	0.6289	0.9966	2.866	MEAN PROFILE TABULATION				
0.2624E-01	6.937	0.6290	0.9968	2.866	SURVEY NORMAL TO RAMP SURFACE -8 DEG OFF VERTICAL				
0.2720E-01	6.948	0.6284	0.9976	2.866	X				
0.2817E-01	6.959	0.6272	0.9987	2.873	Z				
0.2979E-01	6.942	0.6243	0.9983	2.873	Stagnation Pressure pitot = 0.6604E-01				
					Stagnation Temperature pitot = 281.9				
					M ref = 2.870				
					U ref = 593.9				
					P wall = 0.3910E+05				
					TAU wall = 181.0				
					Y				
0.4400E-03	2.157	0.9579	0.5895	1.143	0.4400E-03	2.157	0.9579	0.5895	1.143
0.4575E-03	2.365	0.9608	0.6193	1.216	0.4575E-03	2.365	0.9608	0.6193	1.216
0.4791E-03	2.518	0.9627	0.6400	1.271	0.4791E-03	2.518	0.9627	0.6400	1.271
0.5009E-02	2.683	0.9641	0.6583	1.320	0.5009E-02	2.683	0.9641	0.6583	1.320
0.5228E-02	2.801	0.9638	0.6687	1.359	0.5228E-02	2.801	0.9638	0.6687	1.359
0.5444E-02	2.944	0.9636	0.6775	1.402	0.5444E-02	2.944	0.9636	0.6775	1.402
0.5675E-02	3.018	0.9635	0.6855	1.423	0.5675E-02	3.018	0.9635	0.6855	1.423
0.5900E-02	3.087	0.9639	0.6901	1.444	0.5900E-02	3.087	0.9639	0.6901	1.444
0.6125E-02	3.238	0.9646	0.6964	1.483	0.6125E-02	3.238	0.9646	0.6964	1.483
0.6350E-02	3.309	0.9653	0.7017	1.504	0.6350E-02	3.309	0.9653	0.7017	1.504
0.6575E-02	3.337	0.9662	0.7174	1.565	0.6575E-02	3.337	0.9662	0.7174	1.565
0.6800E-02	3.716	0.9667	0.7494	1.611	0.6800E-02	3.716	0.9667	0.7494	1.611
0.7025E-02	3.867	0.9664	0.7613	1.646	0.7025E-02	3.867	0.9664	0.7613	1.646
0.7250E-02	4.043	0.9648	0.7759	1.692	0.7250E-02	4.043	0.9648	0.7759	1.692
0.7475E-02	4.197	0.9635	0.7867	1.731	0.7475E-02	4.197	0.9635	0.7867	1.731
0.7700E-02	4.343	0.9640	0.8096	1.768	0.7700E-02	4.343	0.9640	0.8096	1.768
0.7925E-02	4.486	0.9645	0.8279	1.809	0.7925E-02	4.486	0.9645	0.8279	1.809
0.8150E-02	4.629	0.9659	0.8519	1.931	0.8150E-02	4.629	0.9659	0.8519	1.931
0.8375E-02	4.771	0.9670	0.8601	2.023	0.8375E-02	4.771	0.9670	0.8601	2.023
0.8600E-02	4.914	0.9677	0.8822	2.122	0.8600E-02	4.914	0.9677	0.8822	2.122
0.8825E-02	5.056	0.9684	0.8939	2.179	0.8825E-02	5.056	0.9684	0.8939	2.179
0.9050E-02	5.199	0.9688	0.9019	2.221	0.9050E-02	5.199	0.9688	0.9019	2.221
0.9275E-02	5.341	0.9690	0.9167	2.271	0.9275E-02	5.341	0.9690	0.9167	2.271
0.9500E-02	5.484	0.9693	0.9290	2.299	0.9500E-02	5.484	0.9693	0.9290	2.299
0.9725E-02	5.626	0.9695	0.9417	2.363	0.9725E-02	5.626	0.9695	0.9417	2.363
0.9950E-02	5.769	0.9698	0.9472	2.448	0.9950E-02	5.769	0.9698	0.9472	2.448
1.0175E-02	5.911	0.9700	0.9595	2.483	1.0175E-02	5.911	0.9700	0.9595	2.483
1.0400E-02	6.054	0.9703	0.9675	2.490	1.0400E-02	6.054	0.9703	0.9675	2.490
1.0625E-02	6.196	0.9705	0.9753	2.505	1.0625E-02	6.196	0.9705	0.9753	2.505
1.0850E-02	6.339	0.9708	0.9823	2.519	1.0850E-02	6.339	0.9708	0.9823	2.519
1.1075E-02	6.481	0.9710	0.9892	2.526	1.1075E-02	6.481	0.9710	0.9892	2.526
1.1300E-02	6.624	0.9713	0.9941	2.540	1.1300E-02	6.624	0.9713	0.9941	2.540
1.1525E-02	6.766	0.9715	0.9971	2.582	1.1525E-02	6.766	0.9715	0.9971	2.582
1.1750E-02	6.909	0.9718	0.9984	2.660	1.1750E-02	6.909	0.9718	0.9984	2.660
1.1975E-02	7.051	0.9720	0.9975	2.653	1.1975E-02	7.051	0.9720	0.9975	2.653
1.2200E-02	7.194	0.9723	0.9961	2.696	1.2200E-02	7.194	0.9723	0.9961	2.696
1.2425E-02	7.336	0.9725	0.9943	2.894	1.2425E-02	7.336	0.9725	0.9943	2.894
1.2650E-02	7.479	0.9728	0.9928		1.2650E-02	7.479	0.9728	0.9928	
1.2875E-02	7.621	0.9730	0.9909		1.2875E-02	7.621	0.9730	0.9909	
1.3100E-02	7.764	0.9733	0.9884		1.3100E-02	7.764	0.9733	0.9884	
1.3325E-02	7.906	0.9735	0.9859		1.3325E-02	7.906	0.9735	0.9859	
1.3550E-02	8.049	0.9738	0.9828		1.3550E-02	8.049	0.9738	0.9828	
1.3775E-02	8.191	0.9740	0.9796		1.3775E-02	8.191	0.9740	0.9796	
1.4000E-02	8.334	0.9743	0.9759		1.4000E-02	8.334	0.9743	0.9759	
1.4225E-02	8.476	0.9745	0.9718		1.4225E-02	8.476	0.9745	0.9718	
1.4450E-02	8.619	0.9748	0.9677		1.4450E-02	8.619	0.9748	0.9677	
1.4675E-02	8.761	0.9750	0.9637		1.4675E-02	8.761	0.9750	0.9637	
1.4900E-02	8.904	0.9753	0.9596		1.4900E-02	8.904	0.9753	0.9596	
1.5125E-02	9.046	0.9755	0.9555		1.5125E-02	9.046	0.9755	0.9555	
1.5350E-02	9.189	0.9758	0.9514		1.5350E-02	9.189	0.9758	0.9514	
1.5575E-02	9.331	0.9760	0.9473		1.5575E-02	9.331	0.9760	0.9473	
1.5800E-02	9.474	0.9763	0.9432		1.5800E-02	9.474	0.9763	0.9432	
1.6025E-02	9.616	0.9765	0.9391		1.6025E-02	9.616	0.9765	0.9391	
1.6250E-02	9.759	0.9768	0.9350		1.6250E-02	9.759	0.9768	0.9350	
1.6475E-02	9.901	0.9770	0.9309		1.6475E-02	9.901	0.9770	0.9309	
1.6700E-02	10.044	0.9773	0.9268		1.6700E-02	10.044	0.9773	0.9268	
1.6925E-02	10.186	0.9775	0.9227		1.6925E-02	10.186	0.9775	0.9227	
1.7150E-02	10.329	0.9778	0.9186		1.7150E-02	10.329	0.9778	0.9186	
1.7375E-02	10.471	0.9780	0.9145		1.7375E-02	10.471	0.9780	0.9145	
1.7600E-02	10.614	0.9783	0.9104		1.7600E-02	10.614	0.9783	0.9104	
1.7825E-02	10.756	0.9785	0.9063		1.7825E-02	10.756	0.9785	0.9063	
1.8050E-02	10.899	0.9788	0.9022		1.8050E-02	10.899	0.9788	0.9022	
1.8275E-02	11.041	0.9790	0.8981		1.8275E-02	11.041	0.9790	0.8981	
1.8500E-02	11.184	0.9793	0.8940		1.8500E-02	11.184	0.9793	0.8940	
1.8725E-02	11.326	0.9795	0.8900		1.8725E-02	11.326	0.9795	0.8900	
1.8950E-02	11.469	0.9798	0.8859		1.8950E-02	11.469	0.9798	0.8859	
1.9175E-02	11.611	0.9800	0.8818		1.9175E-02	11.611	0.9800	0.8818	
1.9400E-02	11.754	0.9803	0.8777		1.9400E-02	11.754	0.9803	0.8777	
1.9625E-02	11.896	0.9805	0.8736		1.9625E-02	11.896	0.9805	0.8736	
1.9850E-02	12.039	0.9808	0.8695		1.9850E-02	12.039	0.9808	0.8695	
2.0075E-02	12.181	0.9810	0.8654		2.0075E-02	12.181	0.9810	0.8654	
2.0300E-02	12.324	0.9813	0.8613		2.0300E-02	12.324	0.9813	0.8613	
2.0525E-02	12.466	0.9815	0.8572		2.0525E-02	12.466	0.9815	0.8572	
2.0750E-02	12.609	0.9818	0.8531		2.0750E-02	12.609	0.9818	0.8531	
2.0975E-02	12.751	0.9820	0.8490		2.0975E-02	12.751	0.9820	0.8490	
2.1200E-02	12.894	0.9823	0.8449		2.1200E-02	12.894	0.9823	0.8449	
2.1425E-02	13.036	0.9825	0.8408		2.1425E-02	13.036	0.9825	0.8408	
2.1650E-02	13.179	0.9828	0.8367		2.1650E-02	13.179	0.9828	0.8367	
2.1875E-02	13.321	0.9830	0.8326		2.1875E-02	13.321	0.9830	0.8326	
2.2100E-02	13.464	0.9833	0.8285		2.2100E-02	13.464	0.9833	0.8285	
2.2325E-02	13.606	0.9835	0.8244		2.2325E-02	13.606	0.9835	0.8244	
2.2550E-02	13.749	0.9838	0.8203		2.2550E-02	13.749	0.9838	0.8203	
2.2775E-02	13.891	0.9840	0.8162		2.2775E-02	13.891	0.9840	0.8162	
2.3000E-02	14.034	0.9843	0.8121		2.3000E-02	14.034	0.9843	0.8121	
2.3225E-02	14.176	0.9845	0.8080		2.3225E-02	14.176	0.9845	0.8080	
2.3450E-02	14.319	0.9848	0.8039		2.3450E-02	14.319	0.9848	0.8039	
2.3675E-02	14.461	0.9850	0.8000		2.3675E-02	14.461	0.9850	0.8000	
2.3900E-02	14.604	0.9853	0.7959		2.3900E-02	14.604	0.9853	0.7959	
2.4125E-02	14.746	0.9855	0.7918		2.4125E-02	14.746	0.9855	0.7918	
2.4350E-02	14.889	0.9858	0.7877		2.4350E-02	14.889	0.9858	0.7877	
2.4575E-02	15.031	0.9860	0.7836		2.4575E-02	15.031	0.9860	0.7836	
2.4800E-02	15.174	0.9863	0.7795		2.4800E-02	15.174	0.9863	0.7795	
2.5025E-02	15.316	0.9865	0.7754		2.5025E-02	15.316	0.9865	0.7754	
2.5250E-02	15.459	0.9868	0.7713		2.5250E-02	15.459	0.9868	0.7713	
2.5475E-02	15.601	0.9870	0.7672		2.5475E-02	15.601	0.9870	0.7672	
2.5700E-02	15.744	0.9873	0.7631		2.5700E-02	15.744	0.9873	0.7631	
2.5925E-02	15.886	0.9875	0.7590		2.5925E-02	15.886			

0.1268E-01	8.093	1.068	0.9382	2.369	0.5408E-03	2.636	1.005	0.6387	1.271
0.1345E-01	8.433	1.066	0.9491	2.405	0.6053E-03	2.775	1.005	0.6537	1.313
0.1415E-01	8.484	1.068	0.9644	2.405	0.8311E-03	2.894	1.003	0.6651	1.352
0.1487E-01	8.870	1.043	0.9592	2.497	0.1154E-02	3.076	1.001	0.6839	1.405
0.1563E-01	9.298	1.067	0.9643	2.497	0.1347E-02	3.183	0.9990	0.6949	1.437
0.1634E-01	9.298	1.067	0.9721	2.561	0.1702E-02	3.359	0.9960	0.7128	1.490
0.1712E-01	9.516	1.066	0.9752	2.561	0.2154E-02	3.411	0.9941	0.7173	1.504
0.1799E-01	9.800	1.055	0.9833	2.618	0.2540E-02	3.610	0.9947	0.7335	1.558
0.1884E-01	10.04	1.063	0.9854	2.639	0.3188E-02	3.675	0.9954	0.7514	1.572
0.1983E-01	10.27	1.058	0.9922	2.682	0.3607E-02	3.764	0.9945	0.7643	1.596
0.2077E-01	10.42	1.057	0.9939	2.703	0.4089E-02	3.851	0.9934	0.7514	1.618
0.2161E-01	10.56	1.064	0.9934	2.710	0.4575E-02	3.953	0.9920	0.7597	1.643
0.2236E-01	10.92	1.051	1.003	2.774	0.5123E-02	4.063	0.9900	0.7674	1.671
0.2364E-01	11.09	1.060	1.005	2.788	0.5901E-02	4.229	0.9860	0.7782	1.713
0.2457E-01	11.20	1.058	1.004	2.802	0.6736E-02	4.385	0.9819	0.7877	1.749
0.2550E-01	11.23	1.058	1.004	2.802	0.7581E-02	4.554	0.9784	0.8006	1.797
0.2652E-01	11.27	1.061	1.005	2.809	0.8316E-02	4.703	0.9779	0.8099	1.832
0.2779E-01	11.49	1.064	1.007	2.830	0.9091E-02	4.924	0.9779	0.8230	1.881
0.2898E-01	11.54	1.069	1.006	2.830	0.9898E-02	5.193	0.9766	0.8376	1.938
0.3068E-01	11.61	1.076	1.006	2.830	0.1083E-01	5.436	0.9759	0.8493	1.988
0.3317E-01	11.61	1.066	1.008	2.844	0.1157E-01	5.684	0.9761	0.8604	2.037
					0.1248E-01	5.814	0.9765	0.8646	2.058
					0.1342E-01	6.042	0.9770	0.8715	2.101
					0.1419E-01	6.437	0.9769	0.8915	2.179
					0.1490E-01	6.651	0.9767	0.8991	2.214
					0.1577E-01	6.825	0.9761	0.9063	2.250
					0.1655E-01	7.130	0.9761	0.9160	2.299
					0.1729E-01	7.158	0.9753	0.9261	2.306
					0.1803E-01	7.434	0.9757	0.9261	2.356
					0.1895E-01	7.414	0.9778	0.9242	2.349
					0.1907E-01	7.804	0.9816	0.9342	2.405
					0.2084E-01	7.853	0.9846	0.9351	2.412
					0.2158E-01	8.042	0.9867	0.9398	2.441
					0.2235E-01	8.067	0.9892	0.9392	2.441
					0.2342E-01	8.216	0.9894	0.9420	2.462
					0.2445E-01	8.276	0.9908	0.9439	2.476
					0.2570E-01	8.343	0.9867	0.9459	2.490
					0.2713E-01	8.283	0.9815	0.9459	2.483
					0.2878E-01	8.320	0.9741	0.9467	2.505
					0.3051E-01	8.281	0.9707	0.9457	2.497

MEAN PROFILE TABULATION ALPHA = 16 DEGS									
SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL									
X									
Z									
Stagnation Pressure									
Stagnation Temperature									
M ref									
U ref									
P wall									
TAU wall									
Y									
0.5192E-03	3.007	1.073	0.6698	1.327	0.6084E-03	3.510	1.092	0.7128	1.455
0.9164E-03	3.590	1.067	0.7206	1.487	0.1033E-02	3.999	1.077	0.7502	1.579
0.1438E-02	3.934	1.068	0.7471	1.568	0.1775E-02	4.390	1.082	0.7759	1.660
0.1683E-02	4.165	1.058	0.7674	1.632	0.2250E-02	4.744	1.086	0.7855	1.692
0.2445E-02	4.486	1.058	0.7883	1.703	0.2584E-02	5.019	1.094	0.7951	1.727
0.2891E-02	4.550	1.058	0.7917	1.717	0.3368E-02	5.293	1.103	0.8079	1.775
0.3571E-02	4.935	1.058	0.8141	1.797	0.3960E-02	5.493	1.085	0.8273	1.846
0.4173E-02	5.143	1.057	0.8253	1.839	0.4564E-02	5.670	1.084	0.8364	1.881
0.4796E-02	5.323	1.057	0.8362	1.881	0.5235E-02	5.953	1.072	0.8605	1.981
0.5418E-02	5.562	1.056	0.8467	1.924	0.5956E-02	6.270	1.079	0.8718	2.030
0.6050E-02	5.783	1.046	0.8604	1.981	0.6541E-02	6.330	1.075	0.8771	2.051
0.6673E-02	5.991	1.046	0.8685	2.016	0.7330E-02	6.563	1.073	0.8864	2.094
0.7226E-02	6.074	1.045	0.8722	2.030	0.8024E-02	6.730	1.074	0.8907	2.115
0.7917E-02	6.470	1.044	0.8900	2.108	0.8755E-02	7.149	1.067	0.9091	2.200
0.8608E-02	6.502	1.048	0.8892	2.151	0.9606E-02	7.274	1.072	0.9121	2.214
0.9299E-02	6.724	1.045	0.8989	2.193	0.1039E-01	7.470	1.070	0.9173	2.243
0.9931E-02	6.977	1.043	0.9076	2.243	0.1187E-01	7.702	1.068	0.9257	2.285
0.1064E-01	7.249	1.043	0.9177	2.285					
0.1161E-01	7.479	1.037	0.9256	2.328					
0.1197E-01	7.765	1.039	0.9341	2.328					
0.1287E-01	7.825	1.045	0.9353	2.335					
0.1372E-01	8.203	1.039	0.9469	2.398					
0.1446E-01	8.401	1.040	0.9511	2.427					
0.1530E-01	8.783	1.043	0.9615	2.483					
0.1613E-01	8.917	1.045	0.9651	2.497					
0.1693E-01	9.175	1.048	0.9707	2.533					
0.1769E-01	9.410	1.049	0.9755	2.568					
0.1846E-01	9.816	1.048	0.9826	2.625					
0.1916E-01	10.06	1.046	0.9898	2.667					
0.1987E-01	10.07	1.048	0.9891	2.660					
0.2072E-01	10.25	1.053	0.9907	2.682					
0.2155E-01	10.67	1.056	0.9985	2.738					
0.2223E-01	10.79	1.057	1.000	2.752					
0.2300E-01	10.88	1.057	1.001	2.759					
0.2373E-01	10.99	1.057	1.003	2.774					
0.2451E-01	11.01	1.057	1.002	2.781					
0.2530E-01	11.29	1.052	1.007	2.823					

MEAN PROFILE TABULATION  
SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL  
X  
Z  
Stagnation Pressure pitot = 0.0000E+00  
Stagnation Temperature pitot = -1.270E-01  
M ref = 0.6853E+06  
U ref = 276.2  
P wall = 2.850  
TAU wall = 586.6  
preston = 0.3918E+05  
preston = 32.70

Y	PT/PWALL	PS/PWALL	U/UREF	M
0.6600E-03	1.208	1.059	0.2519	0.4380
0.1097E-02	1.315	1.056	0.3217	0.5674
0.1518E-02	1.648	1.054	0.4524	0.8252
0.2064E-02	1.989	1.053	0.5317	0.9990
0.2453E-02	2.343	1.044	0.5920	1.140
0.3109E-02	2.715	1.025	0.6459	1.278
0.3498E-02	2.814	1.004	0.6642	1.327
0.4013E-02	3.093	0.9743	0.7019	1.434
0.4559E-02	3.276	0.9383	0.7319	1.522
0.5230E-02	3.347	0.8560	0.7636	1.625
0.5573E-02	3.390	0.8401	0.7736	1.657
0.6353E-02	3.415	0.8040	0.7869	1.704
0.7000E-02	3.415	0.7243	0.8151	1.804
0.7521E-02	3.446	0.7046	0.8255	1.846
0.8176E-02	3.530	0.6783	0.8418	1.910
0.8677E-02	3.593	0.6540	0.8554	1.966
0.9255E-02	3.753	0.6477	0.8703	2.030
0.1025E-01	3.847	0.6405	0.8796	2.073
0.1087E-01	3.944	0.6356	0.8879	2.108
0.1165E-01	4.002	0.6294	0.8943	2.136
0.1232E-01	4.167	0.6238	0.9059	2.193
0.1337E-01	4.294	0.6168	0.9161	2.243
0.1415E-01	4.429	0.6119	0.9246	2.285
0.1517E-01	4.548	0.6119	0.9302	2.320
0.1642E-01	4.837	0.6082	0.9470	2.412
0.1776E-01	4.995	0.6039	0.9554	2.462
0.1936E-01	5.272	0.5990	0.9691	2.540
0.2106E-01	5.559	0.5930	0.9810	2.625
0.2386E-01	5.653	0.5928	0.9806	2.653
0.2616E-01	6.026	0.5849	0.9954	2.739
0.2806E-01	6.196	0.5848	1.001	2.802
0.3106E-01	6.242	0.5848	1.003	2.816
0.3249E-01	6.242	0.5848	1.002	2.816
0.3434E-01	6.295	0.5848	1.004	2.830

MEAN PROFILE TABULATION  
SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL  
X  
Z  
Stagnation Pressure pitot = 0.6350E-02  
Stagnation Temperature pitot = -1.270E-01  
M ref = 0.6873E+06  
U ref = 276.2  
P wall = 2.850  
TAU wall = 586.3  
preston = 0.4802E+05  
preston = 71.86

Y	PT/PWALL	PS/PWALL	U/UREF	M
0.5121E-03	1.347	1.037	0.3550	0.6240
0.8667E-03	1.452	1.024	0.4047	0.7236
0.1183E-02	1.574	1.017	0.4489	0.8154
0.1581E-02	1.696	1.014	0.4836	0.8896
0.1956E-02	1.869	1.009	0.5253	0.9824
0.2275E-02	2.022	1.003	0.5567	1.055
0.2517E-02	2.128	0.9958	0.5761	1.101
0.2743E-02	2.218	0.9953	0.5907	1.136
0.3109E-02	2.376	0.9945	0.6142	1.193
0.3485E-02	2.530	0.9800	0.6373	1.253
0.3907E-02	2.691	0.9788	0.6593	1.313

0.2639E-01 11.18 1.045 1.006 2.816  
0.2743E-01 11.41 1.041 1.011 2.851  
0.2858E-01 11.61 1.044 1.010 2.851  
0.2974E-01 11.45 1.049 1.009 2.851  
0.3098E-01 11.49 1.049 1.010 2.851  
0.3195E-01 11.49 1.049 1.009 2.851  
0.3330E-01 11.49 1.046 1.010 2.859

MEAN PROFILE TABULATION ALPHA = 16 DEG  
SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL  
X  
Z  
Stagnation Pressure pitot = 0.6350E-02  
Stagnation Temperature pitot = -1.270E-01  
M ref = 0.6885E+06  
U ref = 264.7  
P wall = 2.850  
TAU wall = 574.0  
preston = 0.2626E+05  
preston = 83.51

Y	PT/PWALL	PS/PWALL	U/UREF	M
0.4204E-03	2.462	1.108	0.5907	1.133
0.8946E-03	3.069	1.095	0.6654	1.327
0.1359E-02	3.462	1.084	0.7057	1.441
0.1882E-02	3.822	1.070	0.7390	1.540
0.2297E-02	3.996	1.057	0.7553	1.593
0.2781E-02	4.233	1.041	0.7771	1.664
0.3078E-02	4.345	1.036	0.7854	1.692
0.3551E-02	4.448	1.019	0.7965	1.731
0.3927E-02	4.650	1.002	0.8141	1.797
0.4341E-02	4.670	0.9653	0.8196	1.818
0.4757E-02	4.856	0.9495	0.8351	1.874
0.5319E-02	4.930	0.9432	0.8404	1.896
0.5763E-02	5.103	0.9602	0.8510	1.938
0.6298E-02	5.251	0.9495	0.8611	1.981
0.6772E-02	5.379	0.9504	0.8656	2.002
0.7295E-02	5.552	0.9507	0.8744	2.037
0.7818E-02	5.775	0.9445	0.8864	2.094
0.8392E-02	5.853	0.9467	0.8880	2.101
0.9004E-02	6.035	0.9444	0.8974	2.143
0.9685E-02	6.175	0.9436	0.9038	2.172
0.1039E-01	6.301	0.9412	0.9129	2.214
0.1106E-01	6.529	0.9408	0.9183	2.243
0.1171E-01	6.826	0.9449	0.9265	2.285
0.1237E-01	6.966	0.9369	0.9358	2.320
0.1303E-01	7.222	0.9367	0.9422	2.370
0.1364E-01	7.383	0.9387	0.9462	2.391
0.1431E-01	7.519	0.9405	0.9509	2.412
0.1493E-01	7.609	0.9355	0.9544	2.441
0.1569E-01	7.857	0.9415	0.9599	2.469
0.1630E-01	8.141	0.9424	0.9683	2.519
0.1691E-01	8.269	0.9371	0.9727	2.547
0.1766E-01	8.520	0.9433	0.9766	2.575
0.1833E-01	8.739	0.9455	0.9812	2.611
0.1898E-01	8.755	0.9437	0.9819	2.618
0.1974E-01	9.097	0.9441	0.9905	2.667
0.2036E-01	9.060	0.9453	0.9881	2.660
0.2112E-01	9.406	0.9412	0.9973	2.717
0.2179E-01	9.514	0.9389	0.9984	2.738
0.2255E-01	9.662	0.9435	1.001	2.752
0.2332E-01	9.847	0.9419	1.004	2.781
0.2410E-01	9.869	0.9434	1.004	2.788
0.2488E-01	9.930	0.9422	1.003	2.795
0.2563E-01	10.04	0.9393	1.006	2.816
0.2637E-01	9.930	0.9375	1.004	2.802
0.2727E-01	10.14	0.9424	1.007	2.823
0.2797E-01	10.09	0.9461	1.005	2.816
0.2896E-01	10.23	0.9444	1.008	2.837
0.3015E-01	10.24	0.9401	1.009	2.844
0.3132E-01	10.26	0.9410	1.008	2.844
0.3261E-01	10.26	0.9450	1.008	2.837

0.4389E-02	2.822	0.9675	0.6772	1.363	0.1423E-01	3.966	0.4803	0.9555	2.455
0.4794E-02	3.040	0.9571	0.7034	1.437	0.1488E-01	3.960	0.4519	0.9681	2.540
0.5161E-02	3.138	0.9445	0.7164	1.476	0.1585E-01	4.028	0.4312	0.9802	2.625
0.5700E-02	3.249	0.9158	0.7363	1.536	0.1663E-01	4.085	0.4296	0.9843	2.666
0.6276E-02	3.382	0.8798	0.7592	1.611	0.1735E-01	4.153	0.4266	0.9889	2.682
0.6764E-02	3.496	0.8323	0.7850	1.696	0.1813E-01	4.223	0.4238	0.9935	2.717
0.7369E-02	3.521	0.7738	0.8079	1.775	0.1902E-01	4.278	0.4236	0.9956	2.731
0.7932E-02	3.538	0.7102	0.8323	1.867	0.1983E-01	4.300	0.4236	1.001	2.774
0.8539E-02	3.451	0.6492	0.8507	1.938	0.2093E-01	4.464	0.4236	1.003	2.795
0.9017E-02	3.434	0.6120	0.8639	1.995	0.2193E-01	4.466	0.4236	1.002	2.795
0.9741E-02	3.397	0.5702	0.8783	2.058	0.2281E-01	4.510	0.4236	1.003	2.809
0.1026E-01	3.399	0.5500	0.8872	2.101	0.2379E-01	4.525	0.4236	1.004	2.816
0.1099E-01	3.467	0.5308	0.9005	2.165	0.2538E-01	4.562	0.4236	1.005	2.830
0.1175E-01	3.521	0.5090	0.9090	2.207	0.2683E-01	4.571	0.4236	1.006	2.837
0.1253E-01	3.609	0.5137	0.9184	2.257	0.2804E-01	4.582	0.4236	1.005	2.830
0.1338E-01	3.742	0.5096	0.9272	2.306	0.2964E-01	4.567	0.4236	1.005	2.837
0.1416E-01	3.840	0.5071	0.9342	2.349	0.3139E-01	4.582	0.4236	1.005	2.837
0.1494E-01	3.948	0.5049	0.9423	2.384	0.3317E-01	4.582	0.4236	1.005	2.837
0.1572E-01	4.067	0.5029	0.9493	2.434					
0.1656E-01	4.130	0.5027	0.9509	2.448					
0.1759E-01	4.229	0.5026	0.9587	2.483					
0.1859E-01	4.357	0.5014	0.9655	2.526					
0.1946E-01	4.467	0.4993	0.9698	2.561					
0.2024E-01	4.559	0.4975	0.9745	2.597					
0.2108E-01	4.629	0.4955	0.9797	2.625					
0.2204E-01	4.750	0.4930	0.9844	2.667					
0.2297E-01	4.835	0.4904	0.9891	2.703					
0.2474E-01	4.908	0.4796	0.9952	2.752					
0.2779E-01	5.094	0.4796	1.002	2.809					
0.3045E-01	5.117	0.4796	1.003	2.816					
0.3303E-01	5.142	0.4796	1.003	2.823					

MEAN PROFILE TABULATION  
SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL

X = 0.3810E-01  
Z = -1.1270E-01  
Stationation Pressure pitot = 0.4884E+06  
Stationation Temperature pitot = 274.3  
M ref = 2.850  
P ref = 584.3  
P wall = 0.6005E+05  
TAU wall = 172.9

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.5667E-03	1.852	1.006	0.5270	0.9756
0.1050E-02	2.002	0.9972	0.5575	1.051
0.1581E-02	2.089	0.9886	0.5749	1.094
0.2095E-02	2.168	0.9793	0.5900	1.129
0.2678E-02	2.240	0.9731	0.6022	1.161
0.3233E-02	2.360	0.9690	0.6211	1.211
0.3843E-02	2.516	0.9692	0.6418	1.284
0.4435E-02	2.572	0.9637	0.6468	1.285
0.4950E-02	2.764	0.9627	0.6724	1.349
0.5482E-02	2.925	0.9592	0.6913	1.402
0.6284E-02	3.024	0.9572	0.7011	1.430
0.6876E-02	3.195	0.9550	0.7186	1.483
0.7541E-02	3.428	0.9564	0.7381	1.543
0.8179E-02	3.548	0.9553	0.7499	1.579
0.8872E-02	3.796	0.9538	0.7695	1.643
0.9474E-02	3.982	0.9545	0.7825	1.689
0.1009E-01	4.174	0.9556	0.7963	1.735
0.1072E-01	4.358	0.9548	0.8076	1.775
0.1141E-01	4.544	0.9543	0.8192	1.818
0.1210E-01	4.846	0.9529	0.8364	1.889
0.1273E-01	5.011	0.9540	0.8461	1.924
0.1343E-01	5.206	0.9459	0.8576	1.974
0.1410E-01	5.377	0.9500	0.8645	2.002
0.1468E-01	5.607	0.9479	0.8757	2.051
0.1537E-01	5.719	0.9404	0.8826	2.087
0.1587E-01	5.936	0.9296	0.8928	2.136
0.1648E-01	6.035	0.8957	0.9038	2.186
0.1696E-01	6.035	0.8316	0.9251	2.292
0.1743E-01	5.852	0.7608	0.9383	2.370
0.1805E-01	5.428	0.6488	0.9574	2.476
0.1851E-01	4.991	0.5687	0.9679	2.540
0.1905E-01	4.373	0.5258	0.9537	2.469
0.1959E-01	4.128	0.4828	0.9591	2.505
0.2013E-01	4.084	0.4517	0.9684	2.575
0.2077E-01	4.064	0.4261	0.9794	2.653
0.2133E-01	4.049	0.4036	0.9928	2.731
0.2173E-01	4.043	0.3932	0.9950	2.759
0.2289E-01	4.067	0.3866	0.9999	2.795
0.2387E-01	4.079	0.3815	1.004	2.816
0.2474E-01	4.085	0.3788	1.004	2.830

MEAN PROFILE TABULATION  
SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL

X = 0.1905E-01  
Z = -1.1270E-01  
Stationation Pressure pitot = 0.6929E+06  
Stationation Temperature pitot = 269.9  
M ref = 2.850  
P ref = 579.6  
P wall = 0.5437E+05  
TAU wall = 138.8

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.5354E-03	1.713	1.033	0.4819	0.8818
0.9723E-03	1.826	1.024	0.5116	0.9482
0.1440E-02	1.929	1.016	0.5348	1.002
0.1885E-02	2.010	1.008	0.5539	1.044
0.2204E-02	2.117	1.003	0.5725	1.090
0.2564E-02	2.226	0.9977	0.5913	1.136
0.3000E-02	2.334	0.9920	0.6097	1.182
0.3472E-02	2.466	0.9888	0.6247	1.221
0.3772E-02	2.553	0.9865	0.6396	1.260
0.4247E-02	2.676	0.9849	0.6551	1.303
0.4830E-02	2.867	0.9832	0.6769	1.363
0.5090E-02	2.996	0.9798	0.6913	1.402
0.5418E-02	3.046	0.9760	0.6973	1.419
0.5916E-02	3.188	0.9766	0.7100	1.458
0.6353E-02	3.395	0.9742	0.7308	1.519
0.6899E-02	3.519	0.9706	0.7423	1.554
0.7468E-02	3.723	0.9722	0.7587	1.607
0.7842E-02	3.872	0.9670	0.7717	1.650
0.8319E-02	3.940	0.9719	0.7743	1.660
0.8959E-02	4.124	0.9686	0.7862	1.706
0.9528E-02	4.326	0.9584	0.8060	1.766
1.0022E-01	4.547	0.9443	0.8228	1.832
1.0968E-01	4.672	0.8669	0.8330	1.952
0.1173E-01	4.611	0.7649	0.8817	2.073
0.1228E-01	4.363	0.6748	0.9007	2.158
0.1288E-01	4.124	0.5843	0.9204	2.257
0.1353E-01	4.013	0.5181	0.9428	2.377

0.2540E-01 4.089 0.3786 1.005 2.830  
 0.2672E-01 4.071 0.3785 1.004 2.823  
 0.2822E-01 4.069 0.3785 1.003 2.823

MEAN PROFILE TABULATION  
 SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL  
 X  
 Z

Stagnation Pressure pitot = 0.7620E-01  
 Stagnation Temperature pitot = 0.6800E+06  
 M ref = 269.9  
 U ref = 2.850  
 P well = 579.6  
 TAU well = 0.6280E+05  
 preston = 234.6

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.5888E-03	2.293	1.033	0.5895	1.133
0.1231E-02	2.494	1.018	0.6218	1.214
0.1835E-02	2.540	1.002	0.6327	1.242
0.2418E-02	2.639	0.9866	0.6464	1.278
0.3020E-02	2.691	0.9674	0.6553	1.303
0.3752E-02	2.876	0.9874	0.6783	1.366
0.4366E-02	2.873	0.9763	0.6791	1.370
0.4999E-02	3.070	0.9705	0.7006	1.434
0.5789E-02	3.194	0.9681	0.7134	1.469
0.6284E-02	3.261	0.9653	0.7197	1.490
0.6874E-02	3.432	0.9466	0.7342	1.536
0.7430E-02	3.548	0.9445	0.7443	1.568
0.7958E-02	3.731	0.9627	0.7603	1.618
0.8574E-02	3.841	0.9606	0.7666	1.646
0.9152E-02	4.100	0.9602	0.7873	1.710
0.9931E-02	4.221	0.9625	0.7948	1.738
0.1061E-01	4.503	0.9634	0.8136	1.804
0.1127E-01	4.769	0.9611	0.8266	1.860
0.1200E-01	5.033	0.9641	0.8446	1.917
0.1253E-01	4.982	0.9659	0.8403	1.903
0.1320E-01	5.395	0.9675	0.8603	1.980
0.1387E-01	5.529	0.9686	0.8671	2.016
0.1459E-01	5.616	0.9701	0.8705	2.030
0.1522E-01	5.904	0.9740	0.8818	2.080
0.1584E-01	6.104	0.9768	0.8891	2.115
0.1659E-01	6.185	0.9766	0.8966	2.129
0.1738E-01	6.231	0.9771	0.8918	2.136
0.1805E-01	6.349	0.9777	0.8966	2.165
0.1874E-01	6.392	0.9812	0.8951	2.179
0.1943E-01	6.433	0.9776	0.8970	2.172
0.2070E-01	6.416	0.9773	0.8999	2.186
0.2149E-01	6.497	0.9804	0.8995	2.193
0.2228E-01	6.473	0.9791	0.8967	2.179
0.2298E-01	6.497	0.9770	0.8981	2.186
0.2375E-01	6.522	0.9789	0.9001	2.193
0.2456E-01	6.498	0.9768	0.8998	2.184
0.2527E-01	6.535	0.9762	0.9007	2.193
0.2576E-01	6.535	0.9753	0.9023	2.200
0.2644E-01	6.510	0.9747	0.9016	2.207
0.2733E-01	6.503	0.9650	0.9033	2.207
0.2804E-01	6.510	0.9358	0.9093	2.243
0.2873E-01	6.328	0.9623	0.9233	2.306
0.2931E-01	5.419	0.9743	0.9196	2.299
0.3007E-01	4.113	0.9612	0.9210	2.306
0.3091E-01	4.080	0.9455	0.9824	2.660
0.3203E-01	4.064	0.9238	1.001	2.802
0.3348E-01	4.054	0.9722	1.006	2.844
0.3703E-01	4.037	0.9667	1.007	2.859

MEAN PROFILE TABULATION  
 SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL  
 X  
 Z  
 Stagnation Pressure pitot = 0.1397  
 Stagnation Temperature pitot = 0.1270E-01  
 M ref = 269.9  
 U ref = 2.850  
 P well = 579.6  
 TAU well = 0.6280E+05  
 preston = 234.6

Stagnation Temperature pitot = 273.6  
 M ref = 2.850  
 U ref = 583.5  
 P well = 0.5889E+05  
 TAU well = 282.5  
 preston = 282.5

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.4999E-03	2.656	1.070	0.4215	1.225
0.9545E-03	2.926	1.055	0.6552	1.317
0.1540E-02	3.060	1.051	0.6713	1.359
0.2141E-02	3.167	1.046	0.6853	1.395
0.2695E-02	3.279	1.040	0.6975	1.430
0.3653E-02	3.524	1.040	0.7184	1.494
0.4384E-02	3.537	1.037	0.7199	1.497
0.5235E-02	3.734	1.031	0.7371	1.554
0.6027E-02	3.671	1.024	0.7333	1.543
0.6650E-02	3.851	1.022	0.7480	1.589
0.7607E-02	4.043	1.019	0.7626	1.639
0.8331E-02	4.130	1.016	0.7697	1.664
0.9209E-02	4.288	1.014	0.7818	1.703
0.1015E-01	4.459	1.012	0.7921	1.742
0.1114E-01	4.702	1.012	0.8082	1.797
0.1223E-01	5.076	1.012	0.8284	1.874
0.1344E-01	5.242	1.013	0.8378	1.910
0.1443E-01	5.644	1.017	0.8543	1.981
0.1543E-01	5.880	1.020	0.8639	2.023
0.1654E-01	6.147	1.021	0.8749	2.073
0.1772E-01	6.441	1.021	0.8869	2.129
0.1888E-01	6.646	1.022	0.8946	2.165
0.2007E-01	6.736	1.024	0.8949	2.172
0.2115E-01	6.843	1.026	0.8991	2.193
0.2225E-01	6.845	1.027	0.8972	2.193
0.2334E-01	6.847	1.028	0.8966	2.193
0.2443E-01	6.889	1.029	0.8971	2.193
0.2543E-01	6.845	1.030	0.8937	2.186
0.2642E-01	6.889	1.031	0.8964	2.193
0.2741E-01	6.876	1.033	0.8935	2.186
0.2833E-01	6.848	1.035	0.8928	2.179
0.2927E-01	6.901	1.037	0.8938	2.186
0.3020E-01	6.833	1.037	0.8931	2.179
0.3115E-01	6.874	1.038	0.8940	2.186
0.3216E-01	6.908	1.040	0.8940	2.186
0.3305E-01	6.859	1.040	0.8939	2.186
0.3393E-01	6.913	1.038	0.8944	2.193
0.3480E-01	6.862	1.037	0.8915	2.179
0.3569E-01	6.882	1.037	0.8915	2.179
0.3650E-01	6.903	1.037	0.8930	2.186
0.3759E-01	6.826	1.035	0.8944	2.193
0.3829E-01	6.882	1.033	0.8915	2.179
0.3917E-01	6.847	1.031	0.8915	2.179
0.4034E-01	6.847	1.029	0.8915	2.179
0.4125E-01	6.843	1.027	0.8915	2.179
0.4211E-01	6.803	1.031	0.8915	2.179
0.4305E-01	6.816	1.032	0.8944	2.193
0.4399E-01	6.789	1.027	0.8915	2.179
0.4494E-01	6.847	1.025	0.8915	2.179
0.4590E-01	6.826	1.029	0.8944	2.193
0.4687E-01	6.826	1.032	0.8915	2.179
0.4789E-01	6.836	1.031	0.8915	2.179
0.4905E-01	6.860	0.9689	0.9088	2.186
0.5001E-01	6.789	0.8711	0.9317	2.264
0.5077E-01	4.564	0.7066	0.8855	2.384
0.5113E-01	4.327	0.6287	0.9017	2.228
0.5207E-01	4.349	0.4673	0.9713	2.618
0.5286E-01	4.335	0.4292	0.9884	2.731
0.5431E-01	4.334	0.4097	0.9984	2.802

MEAN PROFILE TABULATION  
 SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL  
 X  
 Z  
 Stagnation Pressure pitot = 0.1397  
 Stagnation Temperature pitot = 0.1270E-01  
 M ref = 269.9  
 U ref = 2.850  
 P well = 579.6  
 TAU well = 0.6280E+05  
 preston = 234.6





0.2565E-01 6.053 0.6023 0.9956 2.724  
 0.2621E-01 6.104 0.6024 0.9969 2.738  
 0.2682E-01 6.109 0.6025 0.9984 2.758  
 0.2690E-01 6.116 0.6026 0.9991 2.778  
 0.2713E-01 6.162 0.6028 0.9983 2.752  
 0.2781E-01 6.126 0.6030 0.9974 2.745  
 0.2799E-01 6.155 0.6031 0.9985 2.752  
 0.2797E-01 6.155 0.6031 0.9985 2.752  
 0.2855E-01 6.162 0.6034 0.9984 2.752  
 0.2906E-01 6.172 0.6036 0.9977 2.752  
 0.2936E-01 6.177 0.6039 0.9973 2.752  
 0.2962E-01 6.189 0.6041 0.9969 2.752  
 0.3010E-01 6.194 0.6045 0.9966 2.752  
 0.3068E-01 6.206 0.6050 0.9977 2.759

MEAN PROFILE TABULATION ALPHA = 2.0 DEG

SURVEY IS 5.5 DEGREES OFF VERTICAL  
 X = 0.0000E+00  
 Z = -1270E-01  
 Stagnation Pressure pitot = 254.4  
 Stagnation Temperature pitot = 254.4  
 M ref = 2.790  
 U ref = 558.1  
 P wall = 0.4615E+05  
 TAU wall = 13.47

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.1579E-02	1.006	0.9804	0.9354E-01	0.1594
0.1624E-02	1.033	0.9864	0.1509	0.2583
0.1624E-02	1.048	0.9925	0.1632	0.2798
0.2380E-02	1.076	0.9938	0.1966	0.3364
0.2380E-02	1.115	0.9935	0.2367	0.4097
0.2380E-02	1.165	0.9896	0.2806	0.4893
0.2913E-02	1.215	0.9927	0.3101	0.5439
0.3492E-02	1.294	0.9906	0.3566	0.6318
0.3492E-02	1.400	0.9851	0.4055	0.7275
0.4459E-02	1.648	0.9839	0.4423	0.8018
0.4459E-02	1.770	0.9868	0.4835	0.8877
0.4930E-02	1.869	0.9874	0.5133	0.9521
0.5116E-02	1.993	0.9861	0.5340	0.9980
0.5405E-02	2.101	0.9845	0.5590	1.055
0.5649E-02	2.173	0.9842	0.5785	1.101
0.5850E-02	2.306	0.9837	0.6102	1.179
0.6027E-02	2.368	0.9837	0.6214	1.207
0.6294E-02	2.472	0.9837	0.6322	1.235
0.6916E-02	2.687	0.9837	0.6592	1.306
0.7160E-02	2.860	0.9837	0.6670	1.327
0.7650E-02	2.927	0.9818	0.6787	1.359
0.7717E-02	3.086	0.9796	0.6863	1.381
0.7940E-02	3.144	0.9773	0.7039	1.430
0.8250E-02	3.294	0.9739	0.7246	1.448
0.8539E-02	3.317	0.9698	0.7282	1.490
0.8629E-02	3.430	0.9657	0.7400	1.501
0.9208E-02	3.556	0.9603	0.7526	1.536
0.9586E-02	3.703	0.9548	0.7657	1.575
0.9741E-02	3.779	0.9525	0.7770	1.618
1.0007E-01	3.935	0.9487	0.7843	1.681
1.0079E-01	3.957	0.9452	0.7870	1.692
1.0744E-01	3.983	0.9412	0.8067	1.703
1.1010E-01	4.217	0.9372	0.8067	1.763
0.1165E-01	4.236	0.9330	0.8080	1.768
0.1194E-01	4.377	0.9276	0.8196	1.811
0.1205E-01	4.438	0.9233	0.8252	1.832
0.1232E-01	4.446	0.9218	0.8326	1.832
0.1261E-01	4.546	0.9183	0.8362	1.860
0.1294E-01	4.596	0.9147	0.8379	1.874
0.1328E-01	4.583	0.9094	0.8379	1.881
0.1379E-01	4.652	0.9036	0.8432	1.903
		0.8833	0.8483	1.924

0.1417E-01 4.583 0.8660 0.8499 1.931  
 0.1440E-01 4.568 0.8513 0.8533 1.945  
 0.1483E-01 4.535 0.8289 0.8584 1.966  
 0.1535E-01 4.531 0.7764 0.8752 2.037  
 0.1579E-01 4.559 0.7217 0.8895 2.101  
 0.1617E-01 4.459 0.6608 0.9093 2.193  
 0.1663E-01 4.496 0.6264 0.9264 2.278  
 0.1699E-01 4.483 0.5984 0.9372 2.335  
 0.1746E-01 4.529 0.5628 0.9540 2.427  
 0.1786E-01 4.563 0.5382 0.9651 2.490  
 0.1830E-01 4.622 0.5150 0.9779 2.568

MEAN PROFILE TABULATION ALPHA = 2.0 DEG

SURVEY IS 5.5 DEGREES OFF VERTICAL  
 X = 0.3909E-02  
 Z = -1270E-01  
 Stagnation Pressure pitot = 254.4  
 Stagnation Temperature pitot = 254.4  
 M ref = 2.790  
 U ref = 560.3  
 P wall = 0.5073E+05  
 TAU wall = 11.24

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.1624E-02	1.006	0.9473	0.1715	0.2035
0.2044E-02	1.051	0.9571	0.2136	0.3677
0.2581E-02	1.134	0.9648	0.2796	0.4868
0.3114E-02	1.247	0.9602	0.3518	0.6221
0.3713E-02	1.385	0.9536	0.4179	0.7510
0.4204E-02	1.498	0.9586	0.4544	0.8252
0.4493E-02	1.619	0.9537	0.4905	0.9016
0.4803E-02	1.782	0.9536	0.5309	0.9902
0.5070E-02	1.881	0.9517	0.5514	1.037
0.5316E-02	2.024	0.9503	0.5770	1.097
0.5603E-02	2.137	0.9494	0.5959	1.143
0.6050E-02	2.273	0.9450	0.6188	1.200
0.6406E-02	2.431	0.9455	0.6397	1.253
0.6716E-02	2.594	0.9442	0.6612	1.310
0.7071E-02	2.691	0.9421	0.6763	1.345
0.7364E-02	2.803	0.9380	0.6883	1.384
0.8273E-02	3.126	0.9312	0.6969	1.409
0.8608E-02	3.215	0.9191	0.7249	1.490
0.9030E-02	3.320	0.9121	0.7353	1.522
0.9520E-02	3.416	0.9041	0.7467	1.558
0.9964E-02	3.578	0.8962	0.7577	1.593
1.050E-01	3.735	0.8856	0.7740	1.646
1.1051E-01	3.804	0.8799	0.7894	1.699
0.1132E-01	3.948	0.8749	0.7964	1.724
0.1174E-01	4.256	0.8700	0.8082	1.766
0.1228E-01	4.264	0.8620	0.8297	1.846
0.1288E-01	4.479	0.8505	0.8330	1.860
0.1326E-01	4.457	0.8432	0.8473	1.917
0.1366E-01	4.524	0.8364	0.8529	1.938
0.1406E-01	4.498	0.8250	0.8567	1.952
0.1433E-01	4.502	0.8166	0.8604	1.966
0.1472E-01	4.457	0.8079	0.8623	1.974
0.1503E-01	4.418	0.7935	0.8653	1.998
0.1535E-01	4.354	0.7764	0.8667	1.995
0.1572E-01	4.331	0.7459	0.8749	2.030
0.1619E-01	4.256	0.6952	0.8891	2.094
0.1661E-01	4.264	0.6470	0.9070	2.179
0.1704E-01	4.274	0.5734	0.9210	2.250
0.1744E-01	4.274	0.5425	0.9358	2.328
0.1790E-01	4.311	0.5162	0.9488	2.398
0.1839E-01	4.376	0.4755	0.9624	2.476
0.1895E-01	4.410	0.4589	0.9814	2.604
0.1939E-01	4.524	0.4452	0.9918	2.660
0.1999E-01	4.526	0.4370	1.003	2.745
0.2044E-01	4.526	0.4370	1.007	2.774

# MEAN PROFILE TABULATION ALPHA = 2.0 DEG

SURVEY IS 5.5 DEGREES OFF VERTICAL

X = 0.1270E-01  
 Y = -0.1270E-01  
 Stagnation Pressure pitot = 0.6934E+06  
 Stagnation Temperature pitot = 247.7  
 M ref = 2.790  
 U ref = 550.7  
 P wall = 0.5830E+05  
 TAU wall = 45.80

Y	PI/PMALL	PS/PMALL	U/UREF	M
0.1333E-02	1.310	0.9780	0.3198	0.5596
0.1421E-02	1.232	0.9752	0.3354	0.5809
0.1535E-02	1.249	0.9664	0.3498	0.6162
0.1644E-02	1.260	0.9672	0.3549	0.6260
0.1733E-02	1.267	0.9671	0.3589	0.6338
0.1799E-02	1.268	0.9643	0.3711	0.6572
0.1932E-02	1.297	0.9615	0.3770	0.6689
0.2021E-02	1.317	0.9595	0.3870	0.6885
0.2132E-02	1.331	0.9573	0.3939	0.7021
0.2212E-02	1.342	0.9537	0.3998	0.7139
0.2354E-02	1.331	0.9537	0.3957	0.7041
0.2443E-02	1.366	0.9524	0.4115	0.7373
0.2574E-02	1.390	0.9507	0.4212	0.7568
0.2664E-02	1.434	0.9497	0.4375	0.7900
0.2842E-02	1.454	0.9476	0.4450	0.8057
0.2977E-02	1.466	0.9435	0.4506	0.8174
0.3132E-02	1.520	0.9430	0.4684	0.8545
0.3287E-02	1.559	0.9402	0.4812	0.8818
0.3421E-02	1.559	0.9376	0.4820	0.8838
0.3576E-02	1.590	0.9374	0.4910	0.9033
0.3754E-02	1.584	0.9395	0.4881	0.8975
0.3952E-02	1.653	0.9411	0.5052	0.9346
0.4130E-02	1.719	0.9400	0.5211	0.9697
0.4288E-02	1.798	0.9391	0.5384	1.009
0.4465E-02	1.849	0.9375	0.5492	1.034
0.4643E-02	1.919	0.9354	0.5645	1.069
0.4841E-02	1.968	0.9285	0.5798	1.104
0.5085E-02	2.005	0.9261	0.5826	1.112
0.5243E-02	2.081	0.9129	0.5929	1.136
0.5470E-02	2.154	0.9138	0.6046	1.165
0.5707E-02	2.238	0.9316	0.6189	1.218
0.5931E-02	2.293	0.9308	0.6259	1.250
0.6129E-02	2.374	0.9301	0.6381	1.274
0.6333E-02	2.459	0.9279	0.6474	1.296
0.6619E-02	2.532	0.9250	0.6591	1.306
0.6866E-02	2.614	0.9272	0.6680	1.331
0.7196E-02	2.741	0.9262	0.6831	1.373
0.7485E-02	2.804	0.9239	0.6904	1.395
0.7795E-02	2.957	0.9226	0.7075	1.444
0.8019E-02	3.081	0.9221	0.7194	1.480
0.8374E-02	3.194	0.9204	0.7309	1.515
0.8641E-02	3.250	0.9200	0.7344	1.533
0.8928E-02	3.328	0.9170	0.7444	1.558
0.9373E-02	3.521	0.9100	0.7622	1.614
0.9594E-02	3.552	0.9070	0.7665	1.618
0.9840E-02	3.736	0.9060	0.7815	1.678
1.022E-01	3.724	0.9037	0.7903	1.674
1.057E-01	3.928	0.9000	0.7968	1.731
1.090E-01	4.015	0.8950	0.8048	1.759
1.1131E-01	4.137	0.8901	0.8150	1.797
1.120E-01	4.175	0.8864	0.8188	1.811
1.120E-01	4.376	0.8816	0.8320	1.860
1.1253E-01	4.506	0.8748	0.8430	1.903
1.1288E-01	4.611	0.8642	0.8501	1.931
1.1333E-01	4.758	0.8580	0.8625	1.981
1.1370E-01	4.930	0.8476	0.8762	2.037
1.1402E-01	5.026	0.8397	0.8829	2.066
1.1453E-01	5.098	0.8317	0.8908	2.101
1.1490E-01	5.194	0.8258	0.8958	2.165

Y	PI/PMALL	PS/PMALL	U/UREF	M
0.4869E-03	1.338	1.032	0.3534	0.6201
0.6861E-03	1.338	1.023	0.3584	0.6299
0.9294E-03	1.348	1.012	0.3704	0.6533
0.1062E-02	1.390	1.007	0.3915	0.6943
0.1173E-02	1.396	1.002	0.3973	0.7061
0.1394E-02	1.419	0.9963	0.4092	0.7295
0.1681E-02	1.425	0.9927	0.4130	0.7373
0.1969E-02	1.454	0.9843	0.4275	0.7646
0.2190E-02	1.459	0.9794	0.4321	0.7764
0.2478E-02	1.508	0.9754	0.4500	0.8135
0.2654E-02	1.537	0.9743	0.4603	0.8350
0.2898E-02	1.570	0.9662	0.4732	0.8623
0.3076E-02	1.573	0.9630	0.4758	0.8682
0.3297E-02	1.587	0.9594	0.4810	0.8799
0.3495E-02	1.620	0.9580	0.4899	0.8994
0.3673E-02	1.648	0.9567	0.4979	0.9170
0.3894E-02	1.733	0.9515	0.5200	0.9658
0.4049E-02	1.728	0.9464	0.5216	0.9697
0.4224E-02	1.745	0.9433	0.5254	0.9785
0.4425E-02	1.761	0.9451	0.5287	0.9843
0.4623E-02	1.899	0.9439	0.5573	1.051
0.4735E-02	1.907	0.9439	0.5588	1.055
0.4999E-02	1.922	0.9405	0.5633	1.065
0.5154E-02	1.998	0.9412	0.5769	1.097
0.5352E-02	2.103	0.9416	0.5901	1.129
0.5731E-02	2.098	0.9404	0.5923	1.136
0.5951E-02	2.219	0.9392	0.6123	1.186
0.6068E-02	2.322	0.9380	0.6279	1.225
0.6261E-02	2.362	0.9357	0.6335	1.239
0.6436E-02	2.368	0.9338	0.6377	1.250
0.6703E-02	2.483	0.9350	0.6501	1.281
0.6901E-02	2.583	0.9360	0.6620	1.313
0.7122E-02	2.671	0.9366	0.6722	1.342
0.7343E-02	2.723	0.9347	0.6785	1.359
0.7498E-02	2.829	0.9334	0.6911	1.395
0.7655E-02	2.845	0.9312	0.6934	1.402
0.8009E-02	2.967	0.9292	0.7070	1.441
0.8230E-02	3.042	0.9280	0.7106	1.451
0.8715E-02	3.186	0.9286	0.7284	1.462
0.9047E-02	3.360	0.9285	0.7457	1.504
0.9268E-02	3.397	0.9284	0.7478	1.558
0.9512E-02	3.485	0.9282	0.7546	1.586
0.9822E-02	3.644	0.9280	0.7679	1.628
0.9997E-02	3.728	0.9278	0.7754	1.653
1.0311E-01	3.785	0.9257	0.7797	1.667
1.053E-01	3.861	0.9242	0.7860	1.689
1.082E-01	3.976	0.9222	0.7952	1.720
1.102E-01	4.091	0.9209	0.8033	1.749

MEAN PROFILE TABULATION  
 SURVEY NORMAL TO BAMP SURFACE-20 DEG OFF VERTICAL  
 X = 0.2540E-01  
 Y = -0.1270E-01  
 Stagnation Pressure pitot = 0.6865E+06  
 Stagnation Temperature pitot = 259.4  
 M ref = 2.790  
 U ref = 563.7  
 P wall = 0.6397E+05  
 TAU wall = 98.22

0.1537E-01 5.208 0.7564 0.9317 2.250  
 0.1579E-01 5.201 0.7142 0.9311 2.299  
 0.1617E-01 5.192 0.6761 0.9451 2.363  
 0.1666E-01 5.038 0.6716 0.9372 2.335  
 0.1710E-01 4.883 0.6221 0.9475 2.391  
 0.1755E-01 4.628 0.4973 0.9859 2.618  
 0.1781E-01 4.477 0.4585 0.9963 2.609  
 0.1799E-01 4.422 0.4321 1.005 2.752

0.1135E-01	4.231	0.9186	0.8146	1.789	0.5596E-02	2.168	0.9806	0.5918	1.136
0.1135E-01	4.257	0.9170	0.8166	1.797	0.5839E-02	2.200	0.9783	0.5945	1.143
0.1192E-01	4.452	0.9129	0.8300	1.846	0.6106E-02	2.247	0.9759	0.6029	1.165
0.1212E-01	4.533	0.9129	0.8338	1.860	0.6482E-02	2.303	0.9730	0.6111	1.186
0.1223E-01	4.475	0.9121	0.8319	1.853	0.6746E-02	2.376	0.9716	0.6223	1.214
0.1252E-01	4.633	0.9095	0.8408	1.889	0.7033E-02	2.475	0.9642	0.6361	1.250
0.1267E-01	4.667	0.9080	0.8443	1.903	0.7346E-02	2.632	0.9640	0.6576	1.304
0.1292E-01	4.707	0.9055	0.8458	1.910	0.7531E-02	2.705	0.9635	0.6654	1.327
0.1325E-01	4.772	0.9006	0.8509	1.931	0.7742E-02	2.798	0.9628	0.6770	1.359
0.1347E-01	4.981	0.8966	0.8631	1.981	0.8029E-02	2.785	0.9613	0.6730	1.349
0.1362E-01	4.955	0.8936	0.8630	1.981	0.8230E-02	2.765	0.9594	0.6896	1.395
0.1385E-01	5.034	0.8878	0.8698	2.009	0.8471E-02	3.002	0.9574	0.6995	1.423
0.1422E-01	5.107	0.8753	0.8744	2.037	0.8781E-02	3.019	0.9564	0.7018	1.430
0.1444E-01	5.175	0.8703	0.8813	2.058	0.9114E-02	3.150	0.9556	0.7151	1.469
0.1467E-01	5.222	0.8644	0.8846	2.073	0.9423E-02	3.214	0.9558	0.7209	1.487
0.1495E-01	5.259	0.8603	0.8878	2.087	0.9733E-02	3.325	0.9541	0.7314	1.519
0.1531E-01	5.333	0.8507	0.8958	2.122	0.1011E-01	3.544	0.9519	0.7507	1.579
0.1555E-01	5.374	0.8436	0.8989	2.136	0.1018E-01	3.596	0.9518	0.7551	1.593
0.1593E-01	5.360	0.8262	0.9035	2.158	0.1031E-01	3.740	0.9517	0.7672	1.632
0.1615E-01	5.413	0.8145	0.9094	2.186	0.1071E-01	3.783	0.9515	0.7702	1.643
0.1641E-01	5.301	0.7989	0.9085	2.166	0.1097E-01	3.868	0.9515	0.7765	1.664
0.1657E-01	5.287	0.7889	0.9110	2.200	0.1132E-01	3.988	0.9515	0.7849	1.692
0.1686E-01	5.207	0.7724	0.9115	2.207	0.1137E-01	3.876	0.9515	0.7764	1.664
0.1694E-01	5.178	0.7679	0.9113	2.207	0.1172E-01	4.245	0.9515	0.8023	1.752
0.1717E-01	5.066	0.7525	0.9095	2.200	0.1201E-01	4.391	0.9497	0.7994	1.742
0.1732E-01	4.846	0.7414	0.9019	2.165	0.1221E-01	4.381	0.9483	0.8129	1.789
0.1732E-01	4.922	0.7248	0.9123	2.214	0.1250E-01	4.540	0.9483	0.8226	1.825
0.1761E-01	4.764	0.7169	0.9063	2.184	0.1283E-01	4.545	0.9491	0.8227	1.825
0.1761E-01	4.644	0.6998	0.9047	2.186	0.1298E-01	4.674	0.9493	0.8303	1.853
0.1794E-01	4.395	0.6771	0.9041	2.179	0.1321E-01	4.748	0.9493	0.8359	1.874
0.1808E-01	4.114	0.6536	0.9089	2.200	0.1338E-01	4.801	0.9493	0.8378	1.881
0.1854E-01	3.982	0.6153	0.8995	2.158	0.1367E-01	4.927	0.9488	0.8451	1.910
0.1867E-01	3.890	0.6038	0.8975	2.151	0.1385E-01	4.976	0.9483	0.8487	1.924
0.1885E-01	3.913	0.5850	0.9062	2.193	0.1407E-01	4.905	0.9477	0.8451	1.910
0.1907E-01	3.832	0.5631	0.9103	2.214	0.1433E-01	4.994	0.9464	0.8505	1.931
0.1938E-01	3.709	0.5342	0.9158	2.243	0.1467E-01	5.177	0.9442	0.8591	1.966
0.1973E-01	3.670	0.4670	0.9443	2.391	0.1500E-01	5.201	0.9416	0.8605	1.974
					0.1524E-01	5.391	0.9401	0.8706	2.016
					0.1557E-01	5.417	0.9385	0.8720	2.023
					0.1584E-01	5.532	0.9368	0.8780	2.058
					0.1615E-01	5.633	0.9347	0.8844	2.080
					0.1650E-01	5.678	0.9315	0.8854	2.087
					0.1680E-01	5.635	0.9297	0.8834	2.080
					0.1703E-01	5.724	0.9254	0.8876	2.101
					0.1741E-01	5.682	0.9198	0.8868	2.101
					0.1783E-01	5.728	0.9122	0.8911	2.122
					0.1814E-01	5.754	0.9053	0.8938	2.136
					0.1871E-01	5.774	0.8998	0.8978	2.158
					0.1933E-01	5.731	0.8643	0.9034	2.186
					0.1984E-01	5.688	0.8240	0.9135	2.235
					0.2013E-01	5.618	0.7808	0.9234	2.285
					0.2041E-01	5.499	0.7257	0.9356	2.349
					0.2073E-01	5.634	0.7044	0.9395	2.370
					0.2090E-01	5.262	0.6756	0.9420	2.384
					0.2104E-01	5.181	0.6525	0.9458	2.405
					0.2115E-01	5.177	0.6395	0.9508	2.434
					0.2123E-01	5.148	0.6280	0.9533	2.448
					0.2152E-01	4.825	0.5953	0.9506	2.434
					0.2152E-01	4.839	0.5677	0.9616	2.497
					0.2160E-01	4.497	0.5334	0.9591	2.483
					0.2179E-01	4.373	0.5079	0.9638	2.512
					0.2185E-01	4.357	0.4940	0.9697	2.547
					0.2212E-01	4.148	0.4488	0.9798	2.611
					0.2227E-01	3.847	0.4286	0.9728	2.568
					0.2247E-01	3.637	0.3963	0.9773	2.597
					0.2278E-01	3.528	0.3391	1.004	2.774

MEAN PROFILE TABULATION  
SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL  
X = 0.7620E-01  
Z = -0.1270E-01  
Stagnation Pressure pitot = 0.6657E+06



0.291E-02	2.437	0.9743	0.6117	1.232	0.2611E-01	5.418	0.9741	0.8500	1.980
0.3251E-02	2.470	0.9684	0.6364	1.250	0.2680E-01	5.403	0.9738	0.8556	1.981
0.3627E-02	2.509	0.9660	0.6436	1.264	0.2715E-01	5.392	0.9735	0.8555	1.981
0.3871E-02	2.516	0.9654	0.6447	1.267	0.2786E-01	5.371	0.9701	0.8553	1.981
0.4313E-02	2.561	0.9626	0.6511	1.285	0.2809E-01	5.371	0.9691	0.8553	1.981
0.4646E-02	2.623	0.9608	0.6588	1.306	0.2868E-01	5.342	0.9676	0.8536	1.974
0.4933E-02	2.699	0.9568	0.6691	1.335	0.2911E-01	5.324	0.9680	0.8537	1.974
0.5220E-02	2.671	0.9563	0.6649	1.324	0.2979E-01	5.349	0.9686	0.8539	1.974
0.5509E-02	2.679	0.9547	0.6660	1.327	0.3018E-01	5.364	0.9682	0.8539	1.974
0.5839E-02	2.765	0.9528	0.6762	1.356	0.3111E-01	5.367	0.9629	0.8559	1.981
0.6191E-02	2.766	0.9475	0.6785	1.363	0.3200E-01	5.368	0.9639	0.8583	1.988
0.6502E-02	2.808	0.9456	0.6834	1.377	0.3284E-01	5.392	0.9605	0.8590	1.995
0.6835E-02	2.885	0.9468	0.6908	1.398	0.3371E-01	5.375	0.9570	0.8574	1.995
0.7211E-02	2.882	0.9449	0.6918	1.402	0.3465E-01	5.349	0.9560	0.8563	1.988
0.7610E-02	2.964	0.9427	0.6991	1.423	0.3548E-01	5.354	0.9529	0.8586	1.995
0.7977E-02	3.017	0.9411	0.7063	1.444	0.3632E-01	5.336	0.9491	0.8580	1.995
0.8273E-02	3.019	0.9415	0.7061	1.444	0.3724E-01	5.306	0.9471	0.8574	1.995
0.8672E-02	3.129	0.9414	0.7164	1.476	0.3777E-01	5.332	0.9471	0.8578	1.995
0.8915E-02	3.206	0.9411	0.7230	1.497	0.3805E-01	5.320	0.9462	0.8580	1.995
0.9002E-02	3.257	0.9410	0.7275	1.512	0.3840E-01	5.304	0.9444	0.8582	1.995
0.9202E-02	3.297	0.9400	0.7317	1.526	0.3884E-01	5.291	0.9422	0.8585	1.995
0.9423E-02	3.423	0.9385	0.7427	1.561	0.3965E-01	5.306	0.9383	0.8597	2.002
0.9578E-02	3.440	0.9375	0.7440	1.565	0.4044E-01	5.294	0.9298	0.8606	2.009
0.9667E-02	3.457	0.9368	0.7464	1.572	0.4128E-01	5.274	0.8905	0.8709	2.051
0.9888E-02	3.503	0.9372	0.7500	1.582	0.4219E-01	5.046	0.8713	0.9060	2.207
0.1009E-01	3.410	0.9380	0.7423	1.558	0.4254E-01	4.526	0.6713	0.9039	2.207
0.1040E-01	3.485	0.9391	0.7483	1.575	0.4254E-01	4.348	0.6713	0.8954	2.158
0.1075E-01	3.481	0.9396	0.7482	1.575	0.4260E-01	4.265	0.6583	0.8954	2.158
0.1099E-01	3.809	0.9399	0.7744	1.660	0.4265E-01	4.122	0.6476	0.8907	2.136
0.1110E-01	3.718	0.9397	0.7669	1.635					
0.1121E-01	3.823	0.9393	0.7754	1.664					
0.1144E-01	3.853	0.9385	0.7775	1.671					
0.1179E-01	4.004	0.9374	0.7868	1.710					
0.1201E-01	4.066	0.9366	0.7929	1.724					
0.1212E-01	4.023	0.9391	0.7898	1.713					
0.1245E-01	4.160	0.9409	0.7978	1.742					
0.1261E-01	4.070	0.9417	0.7927	1.726					
0.1276E-01	4.248	0.9417	0.8048	1.766					
0.1298E-01	4.309	0.9417	0.8092	1.782					
0.1321E-01	4.404	0.9417	0.8149	1.804					
0.1336E-01	4.404	0.9417	0.8045	1.766					
0.1354E-01	4.375	0.9420	0.8127	1.797					
0.1380E-01	4.425	0.9428	0.8144	1.804					
0.1405E-01	4.536	0.9431	0.8219	1.832					
0.1424E-01	4.640	0.9442	0.8273	1.853					
0.1451E-01	4.603	0.9457	0.8252	1.846					
0.1482E-01	4.695	0.9474	0.8291	1.860					
0.1513E-01	4.726	0.9481	0.8310	1.867					
0.1544E-01	4.883	0.9486	0.8403	1.903					
0.1582E-01	4.907	0.9486	0.8422	1.910					
0.1610E-01	5.087	0.9487	0.8492	1.938					
0.1641E-01	4.951	0.9503	0.8437	1.917					
0.1659E-01	5.039	0.9513	0.8471	1.931					
0.1688E-01	5.173	0.9522	0.8540	1.959					
0.1719E-01	5.254	0.9528	0.8573	1.974					
0.1748E-01	5.320	0.9522	0.8609	1.968					
0.1779E-01	5.280	0.9516	0.8594	1.981					
0.1803E-01	5.309	0.9512	0.8612	1.988					
0.1836E-01	5.294	0.9521	0.8614	1.988					
0.1876E-01	5.390	0.9536	0.8647	2.002					
0.1915E-01	5.354	0.9558	0.8624	1.995					
0.1949E-01	5.396	0.9588	0.8616	2.002					
0.2026E-01	5.425	0.9603	0.8631	2.002					
0.2079E-01	5.461	0.9615	0.8652	2.009					
0.2139E-01	5.453	0.9640	0.8640	2.002					
0.2196E-01	5.467	0.9642	0.8633	2.002					
0.2256E-01	5.467	0.9725	0.8604	1.995					
0.2336E-01	5.476	0.9715	0.8610	2.002					
0.2398E-01	5.449	0.9746	0.8587	1.988					
0.2455E-01	5.438	0.9751	0.8598	1.988					
0.2504E-01	5.442	0.9748	0.8599	1.988					
0.2578E-01	5.403	0.9737	0.8569	1.981					

MEAN PROFILE TABULATION FOR 24 DEGREE CORNER

SURVEY IS 5.5 DEGREES OFF VERTICAL

X = -.6350E-01

Z = -.1270E-01

Stagnation Pressure pitot = 0.6954E+06

Stagnation Temperature pitot = 265.0

M ref = 2.840

U ref pitot = 551.4

P wall preston = 0.2115E+05

TAU wall

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.4343E-03	2.785	1.082	0.6407	1.257
0.7198E-03	3.252	1.082	0.6874	1.388
0.7948E-03	3.564	1.082	0.7155	1.469
0.1080E-02	3.801	1.082	0.7331	1.526
0.1336E-02	3.942	1.082	0.7422	1.558
0.1366E-02	4.046	1.082	0.7501	1.582
0.1583E-02	4.190	1.082	0.7612	1.618
0.1756E-02	4.288	1.082	0.7679	1.639
0.1981E-02	4.336	1.082	0.7713	1.650
0.2167E-02	4.512	1.082	0.7827	1.689
0.2364E-02	4.574	1.082	0.7855	1.699
0.2462E-02	4.679	1.082	0.7925	1.724
0.2698E-02	4.790	1.082	0.7983	1.745
0.3006E-02	4.899	1.082	0.8047	1.768
0.3183E-02	5.055	1.082	0.8145	1.804
0.3374E-02	5.098	1.082	0.8164	1.811
0.3800E-02	5.254	1.082	0.8241	1.839
0.4100E-02	5.372	1.082	0.8295	1.860
0.4265E-02	5.443	1.082	0.8330	1.874
0.4641E-02	5.599	1.082	0.8416	1.910
0.5197E-02	5.859	1.082	0.8517	1.952
0.6068E-02	6.105	1.082	0.8632	2.002
0.6721E-02	6.384	1.082	0.8748	2.051
0.7389E-02	6.587	1.082	0.8823	2.087
0.7816E-02	6.800	1.082	0.8901	2.122
0.8375E-02	6.965	1.082	0.8963	2.151
0.9017E-02	7.301	1.082	0.9081	2.207

U. 0.911E-02	7.495	1.082	0.9134	2.235
0.1018E-01	7.646	1.082	0.9180	2.264
0.1078E-01	7.902	1.082	0.9249	2.299
0.1146E-01	8.134	1.082	0.9317	2.335
0.1191E-01	8.369	1.082	0.9390	2.377
0.1270E-01	8.696	1.082	0.9461	2.420
0.1331E-01	8.923	1.082	0.9529	2.455
0.1409E-01	9.160	1.082	0.9587	2.490
0.1484E-01	9.403	1.082	0.9629	2.519
0.1588E-01	9.688	1.082	0.9745	2.597
0.1698E-01	10.20	1.082	0.9803	2.639
0.1748E-01	10.46	1.082	0.9849	2.674
0.1825E-01	10.80	1.082	0.9906	2.717
0.1898E-01	10.82	1.082	0.9896	2.717
0.1952E-01	11.08	1.082	0.9942	2.752
0.2025E-01	11.31	1.082	0.9976	2.781
0.2090E-01	11.37	1.082	0.9980	2.788
0.2164E-01	11.47	1.082	0.9998	2.802
0.2261E-01	11.62	1.002	1.002	2.823
0.2359E-01	11.73	1.002	1.003	2.837
0.2432E-01	11.77	1.002	1.003	2.844
0.2504E-01	11.78	1.002	1.002	2.846

MEAN PROFILE TABULATION FOR 24 DEGREE CORNER				
SURVEY IS 5.5 DEGREES OFF VERTICAL				
X	= -3048E-01			
Z	= -1270E-01			
Stagnation Pressure	pitot = 0.7019E+06			
Stagnation Temperature	pitot = 250.1			
M ref	= 2.840			
U ref	pitot = 557.1			
P well	preston = 0.3669E+05			
TAU well				

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.1905E-03	1.197	1.064	0.2385	0.4136
0.7008E-02	1.231	1.064	0.2635	0.4614
0.1338E-02	1.353	1.063	0.3348	0.5967
0.1962E-02	1.601	1.060	0.4328	0.7900
0.2598E-02	1.990	1.056	0.5290	0.9980
0.3208E-02	2.398	1.045	0.5978	1.161
0.3830E-02	2.716	1.035	0.6410	1.271
0.4481E-02	3.065	1.011	0.6660	1.395
0.5095E-02	3.365	0.9879	0.7204	1.497
0.5763E-02	3.577	0.963	0.7542	1.604
0.6353E-02	3.791	0.8907	0.7851	1.706
0.6957E-02	3.975	0.8404	0.8122	1.804
0.7592E-02	4.107	0.8039	0.8361	1.896
0.8222E-02	4.244	0.7571	0.8607	1.995
0.8872E-02	4.298	0.7087	0.8804	2.090
0.9474E-02	4.368	0.6746	0.8973	2.158
0.1012E-01	4.412	0.6383	0.9124	2.235
0.1079E-01	4.456	0.6210	0.9207	2.278
0.1141E-01	4.538	0.6063	0.9318	2.335
0.1205E-01	4.577	0.5989	0.9348	2.356
0.1267E-01	4.608	0.5919	0.9435	2.405
0.1330E-01	4.748	0.5916	0.9479	2.427
0.1394E-01	4.876	0.5916	0.9527	2.435
0.1458E-01	5.011	0.5916	0.9583	2.490
0.1529E-01	5.134	0.5916	0.9636	2.526
0.1651E-01	5.306	0.5916	0.9676	2.554
0.1720E-01	5.396	0.5916	0.9727	2.589
0.1783E-01	5.598	0.5916	0.9754	2.611
0.1914E-01	5.836	0.5916	0.9793	2.639
0.2039E-01	6.030	0.5916	0.9873	2.703
0.2168E-01	6.143	0.5916	0.9924	2.745
0.2288E-01	6.247	0.5916	0.9958	2.774
0.2429E-01	6.322	0.5916	0.9983	2.802
0.2568E-01	6.360	0.5916	0.9991	2.816
		0.5916	0.9985	2.823

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.1905E-03	1.097	1.038	0.1905E+00	0.2817
0.1712E-02	1.073	1.038	= -1270E-01	0.1254
0.2832E-02	1.071	1.038	= 0.7033E+06	0.2107
0.3620E-02	1.081	1.038	= 245.9	0.1369
0.4794E-02	1.081	1.038	= 2.840	0.2405
0.5601E-02	1.180	1.038	= 552.5	0.1911
0.6584E-02	1.260	1.038	= 0.5162E+05	0.3325
			preston =	0.2472
				0.5342

0.2692E-01	6.366	0.5916	0.9900	2.830
0.2832E-01	6.420	0.5916	1.000	2.837
0.2962E-01	6.474	0.5916	1.001	2.851

MEAN PROFILE TABULATION FOR 24 DEGREE CORNER				
SURVEY IS 5.5 DEGREES OFF VERTICAL				
X	= -1016E-01			
Z	= -1270E-01			
Stagnation Pressure	pitot = 0.6950E+06			
Stagnation Temperature	pitot = 253.3			
M ref	= 2.840			
U ref	pitot = 562.9			
P well	preston = 0.4694E+05			
TAU well				

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.2705E-02	1.132	1.081	0.1492	0.2573
0.3571E-02	1.196	1.081	0.2194	0.3613
0.4359E-02	1.277	1.081	0.2808	0.4932
0.4966E-02	1.360	1.081	0.3287	0.5839
0.5558E-02	1.497	1.081	0.3681	0.6902
0.6199E-02	1.653	1.081	0.4393	0.8018
0.6716E-02	1.760	1.081	0.4691	0.8643
0.7361E-02	1.980	1.081	0.5186	0.9717
0.7859E-02	2.160	1.081	0.5507	1.044
0.8512E-02	2.373	1.081	0.5838	1.122
0.9174E-02	2.629	1.081	0.6180	1.207
0.9718E-02	2.814	1.081	0.6399	1.264
0.1033E-01	3.051	1.081	0.6662	1.335
0.1091E-01	3.289	1.081	0.6888	1.398
0.1159E-01	3.573	1.081	0.7142	1.473
0.1209E-01	3.815	1.081	0.7331	1.529
0.1278E-01	4.028	1.081	0.7490	1.579
0.1346E-01	4.320	1.081	0.7690	1.646
0.1413E-01	4.535	1.081	0.7828	1.692
0.1468E-01	4.759	1.081	0.7976	1.742
0.1534E-01	5.077	1.081	0.8140	1.804
0.1611E-01	5.498	1.081	0.8367	1.889
0.1704E-01	5.801	1.074	0.8528	1.952
0.1804E-01	6.231	1.054	0.8747	2.051
0.1909E-01	6.449	1.008	0.8941	2.143
0.2002E-01	6.510	0.8901	0.9272	2.306
0.2072E-01	6.214	0.7903	0.9424	2.391
0.2107E-01	6.093	0.7392	0.9534	2.455
0.2165E-01	5.809	0.6529	0.9691	2.554
0.2215E-01	5.645	0.5846	0.9858	2.647
0.2260E-01	5.432	0.5231	1.000	2.774
0.2309E-01	5.337	0.4876	1.010	2.851

0.2692E-01 4.664 0.6425 0.9129 2.292  
 0.2743E-01 4.291 0.5697 0.9217 2.342  
 0.2804E-01 4.092 0.4900 0.9431 2.476  
 0.2865E-01 3.977 0.4279 0.9581 2.554  
 0.2954E-01 3.933 0.4037 0.9780 2.682  
 0.3051E-01 3.914 0.3882 0.9949 2.731

# MEAN PROFILE TABULATION FOR 24 DEGREE CORNER

SURVEY IS 24.0 DEGREES OFF VERTICAL  
 X = 0.3048E-01  
 Z = -1.1270E-01  
 Stagnation Pressure pitot = 0.6687E+06  
 Stagnation Temperature pitot = 254.4  
 M ref = 2.840  
 U ref = 561.9  
 P wall = 0.7288E+05  
 TAU wall = preston =

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.1275E-02	1.163	1.050	0.2226	0.3843
0.1990E-02	1.172	1.039	0.2414	0.4185
0.2908E-02	1.198	1.027	0.2715	0.4741
0.3553E-02	1.218	1.020	0.2899	0.5088
0.4442E-02	1.270	1.010	0.3284	0.5811
0.5329E-02	1.384	0.9990	0.3882	0.6982
0.6368E-02	1.548	0.9837	0.4533	0.8311
0.7468E-02	1.733	0.9713	0.5075	0.9482
0.8552E-02	1.979	0.9627	0.5596	1.0690
0.9472E-02	2.263	0.9537	0.6090	1.189
0.1041E-01	2.563	0.9423	0.6521	1.303
0.1141E-01	2.872	0.9274	0.6920	1.412
0.1235E-01	3.255	0.9168	0.7345	1.536
0.1321E-01	3.603	0.9069	0.7680	1.643
0.1418E-01	3.978	0.8940	0.7999	1.752
0.1512E-01	4.338	0.8730	0.8417	1.867
0.1571E-01	4.488	0.8630	0.8628	1.995
0.1650E-01	4.791	0.8515	0.8843	2.087
0.1738E-01	5.056	0.8319	0.8937	2.129
0.1825E-01	5.129	0.8125	0.9041	2.179
0.1892E-01	5.254	0.7976	0.9095	2.207
0.1952E-01	5.299	0.7842	0.9167	2.243
0.2031E-01	5.339	0.7665	0.9258	2.278
0.2118E-01	5.340	0.7471	0.9258	2.292
0.2201E-01	5.311	0.7316	0.9248	2.306
0.2296E-01	5.256	0.7175	0.9248	2.320
0.2366E-01	5.211	0.7040	0.9276	2.320
0.2475E-01	5.153	0.6954	0.9234	2.320
0.2558E-01	5.104	0.6892	0.9241	2.299
0.2647E-01	4.975	0.6826	0.9183	2.221
0.2741E-01	4.588	0.6701	0.9028	2.166
0.2816E-01	4.419	0.6671	0.8955	2.151
0.2894E-01	4.330	0.6634	0.8910	2.165
0.2978E-01	4.049	0.6302	0.8840	2.151
0.2807E-01	3.681	0.5491	0.8984	2.200
0.2870E-01	3.551	0.4754	0.9236	2.328
0.2913E-01	3.278	0.3697	0.9635	2.554
0.2959E-01	3.222	0.3393	0.9781	2.646
0.3012E-01	3.142	0.3243	0.9781	2.674
0.3094E-01	3.116	0.3161	0.9827	2.710

# MEAN PROFILE TABULATION FOR 24 DEGREE CORNER

SURVEY IS 24.0 DEGREES OFF VERTICAL  
 X = 0.6094E-01  
 Z = -1.1270E-01  
 Stagnation Pressure pitot = 0.6943E+06  
 Stagnation Temperature pitot = 248.2  
 M ref = 2.840  
 U ref = 555.0  
 P wall = 0.8638E+05  
 TAU wall = preston =

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.7889E-03	1.529	1.026	0.4321	0.7783
0.1548E-02	1.595	1.014	0.4578	0.8311
0.2097E-02	1.614	1.005	0.4663	0.8506
0.2817E-02	1.665	0.9946	0.4839	0.8896
0.3528E-02	1.690	0.9868	0.4930	0.9111
0.4214E-02	1.755	0.9793	0.5121	0.9521
0.4915E-02	1.824	0.9714	0.5296	0.9941
0.5685E-02	1.902	0.9650	0.5463	1.034
0.6337E-02	1.981	0.9599	0.5628	1.073
0.7068E-02	2.109	0.9542	0.5862	1.129
0.7777E-02	2.210	0.9494	0.6032	1.172
0.8458E-02	2.341	0.9472	0.6220	1.221
0.9190E-02	2.527	0.9449	0.6477	1.289
0.9832E-02	2.739	0.9429	0.6732	1.359
0.1093E-01	2.993	0.9382	0.7017	1.441
0.1217E-01	3.314	0.9323	0.7343	1.536
0.1295E-01	3.574	0.9294	0.7583	1.611
0.1393E-01	3.942	0.9279	0.7868	1.706
0.1490E-01	4.169	0.9264	0.8026	1.763
0.1590E-01	4.473	0.9228	0.8233	1.839
0.1686E-01	4.737	0.9182	0.8399	1.903
0.1795E-01	5.161	0.9115	0.8559	1.966
0.1953E-01	5.260	0.8946	0.8666	2.023
0.2049E-01	5.308	0.8833	0.8768	2.058
0.2181E-01	5.296	0.8625	0.8853	2.101
0.2294E-01	5.296	0.8441	0.8867	2.122
0.2431E-01	5.271	0.8119	0.8923	2.158
0.2588E-01	5.212	0.7703	0.9012	2.207
0.2715E-01	5.153	0.7284	0.9114	2.264
0.2850E-01	5.077	0.6707	0.9278	2.349
0.2995E-01	4.976	0.5887	0.9530	2.490
0.3117E-01	4.866	0.4789	0.9920	2.745
0.3228E-01	4.702	0.3506	1.046	3.170
0.3297E-01	4.590	0.2708	1.085	3.584
0.3368E-01	4.283	0.2650	1.076	3.489
0.3386E-01	4.083	0.2650	1.069	3.404
0.3416E-01	3.598	0.2650	1.048	3.191
0.3437E-01	3.184	0.2650	1.025	2.993
0.3465E-01	2.883	0.2650	1.006	2.844
0.3490E-01	2.746	0.2650	0.9962	2.774
0.3523E-01	2.649	0.2650	0.9879	2.717

# MEAN PROFILE TABULATION FOR 24 DEGREE CORNER

SURVEY IS 24.0 DEGREES OFF VERTICAL  
 X = 0.1016  
 Z = -1.1270E-01  
 Stagnation Pressure pitot = 0.4920E+06  
 Stagnation Temperature pitot = 241.4  
 M ref = 2.840  
 U ref = 547.4  
 P wall = 0.9264E+05  
 TAU wall = preston =

Y	PT/PMALL	PS/PMALL	U/UREF	M
0.9898E-03	1.943	1.037	0.5230	0.9902
0.1655E-02	2.052	1.032	0.5454	1.041
0.2274E-02	2.098	1.023	0.5588	1.069
0.2860E-02	2.165	1.018	0.5713	1.097
0.3543E-02	2.238	1.011	0.5839	1.129
0.3947E-02	2.271	1.011	0.5887	1.143
0.4669E-02	2.361	1.007	0.6034	1.179
0.5420E-02	2.438	1.003	0.6137	1.207
0.5989E-02	2.537	0.9994	0.6271	1.242
0.6464E-02	2.653	0.9966	0.6433	1.285
0.7102E-02	2.783	0.9929	0.6588	1.327
0.7981E-02	2.927	0.9888	0.6761	1.371
0.8349E-02	3.064	0.9877	0.6908	1.416





\*\*\*\*\*ALPHA=8 DEGREES, INCLINED WIRE SURVEYS\*\*\*\*\*

AGADOGGRAPH 315 DATA TAPE FILE: MEUCOB0101.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS VERTICAL

( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL, X = .2540E-01 )

X

Z

M ref

U ref hu

RHO ref hu

TAU wall prestion

P wall mean flow

Y

T

(rho)u<sup>2</sup>

RHO U<sup>2</sup>

U

Uref

M

Pa

Pwall

(u<sup>2</sup>)

(u<sup>2</sup>)

RHOref Uref<sup>2</sup>

RHOref Uref<sup>2</sup>

0.190E-02

0.3261E-02

0.4699E-02

0.6127E-02

0.7575E-02

0.9025E-02

0.1047E-01

0.1191E-01

0.1337E-01

0.1482E-01

0.1626E-01

0.1772E-01

0.1917E-01

0.2062E-01

0.2209E-01

0.2354E-01

0.2498E-01

0.2642E-01

0.2786E-01

0.2933E-01

0.3079E-01

0.3223E-01

0.3365E-01

0.3502E-01

0.3642E-01

0.3782E-01

0.3922E-01

0.4062E-01

0.4202E-01

0.4342E-01

0.4482E-01

0.4622E-01

0.4762E-01

0.4902E-01

0.5042E-01

0.5182E-01

0.5322E-01

0.5462E-01

0.5602E-01

0.5742E-01

0.5882E-01

0.6022E-01

0.6162E-01

0.6302E-01

0.6442E-01

0.6582E-01

0.6722E-01

0.6862E-01

0.7002E-01

0.7142E-01

0.7282E-01

0.7422E-01

0.7562E-01

0.7702E-01

0.7842E-01

0.7982E-01

0.8122E-01

0.8262E-01

0.8402E-01

0.190E-02

0.3261E-02

0.4699E-02

0.6127E-02

0.7575E-02

0.9025E-02

0.1047E-01

0.1191E-01

0.1337E-01

0.1482E-01

0.1626E-01

0.1772E-01

0.1917E-01

0.2062E-01

0.2209E-01

0.2354E-01

0.2498E-01

0.2642E-01

0.2786E-01

0.2933E-01

0.3079E-01

0.3223E-01

0.3365E-01

0.3502E-01

0.3642E-01

0.3782E-01

0.3922E-01

0.4062E-01

0.4202E-01

0.4342E-01

0.4482E-01

0.4622E-01

0.4762E-01

0.4902E-01

0.5042E-01

0.5182E-01

0.5322E-01

0.5462E-01

0.5602E-01

0.5742E-01

0.5882E-01

0.6022E-01

0.6162E-01

0.6302E-01

0.6442E-01

0.6582E-01

0.6722E-01

0.6862E-01

0.7002E-01

0.7142E-01

0.7282E-01

0.7422E-01

0.7562E-01

0.7702E-01

0.7842E-01

0.7982E-01

0.8122E-01

0.8262E-01

0.8402E-01

0.8542E-01

0.8682E-01

0.8822E-01

0.8962E-01

0.9102E-01

0.9242E-01

0.9382E-01

0.9522E-01

0.9662E-01

0.9802E-01

0.9942E-01

1.0082E-01

1.0222E-01

1.0362E-01

1.0502E-01

1.0642E-01

1.0782E-01

1.0922E-01

1.1062E-01

1.1202E-01

1.1342E-01

1.1482E-01

1.1622E-01

0.2060E-01

0.2205E-01

0.2351E-01

0.2497E-01

0.2642E-01

0.2787E-01

0.2932E-01

0.3077E-01

0.3222E-01

0.3367E-01

0.3512E-01

0.3657E-01

0.3802E-01

0.3947E-01

0.4092E-01

0.4237E-01

0.4382E-01

0.4527E-01

0.4672E-01

0.4817E-01

0.4962E-01

0.5107E-01

0.5252E-01

0.5397E-01

0.5542E-01

0.5687E-01

0.5832E-01

0.5977E-01

0.6122E-01

0.6267E-01

0.6412E-01

0.6557E-01

0.6702E-01

0.6847E-01

0.6992E-01

0.7137E-01

0.7282E-01

0.7427E-01

0.7572E-01

0.7717E-01

0.7862E-01

0.8007E-01

0.8152E-01

0.8297E-01

0.8442E-01

0.8587E-01

0.8732E-01

0.8877E-01

0.9022E-01

0.9167E-01

0.9312E-01

0.9457E-01

0.9602E-01

0.9747E-01

0.9892E-01

1.0037E-01

1.0182E-01

1.0327E-01

1.0472E-01

1.0617E-01

1.0762E-01

1.0907E-01

1.1052E-01

1.1197E-01

1.1342E-01

1.1487E-01

1.1632E-01

1.1777E-01

1.1922E-01

1.2067E-01

1.2212E-01

1.2357E-01

1.2502E-01

1.2647E-01

1.2792E-01

1.2937E-01

1.3082E-01

1.3227E-01

1.3372E-01

1.3517E-01

0.190E-03

0.3261E-03

0.4699E-03

0.6127E-03

0.7575E-03

0.9025E-03

0.1047E-02

0.1191E-02

0.1337E-02

0.1482E-02

0.1626E-02

0.1772E-02

0.1917E-02

0.2062E-02

0.2209E-02

0.2354E-02

0.2498E-02

0.2642E-02

0.2786E-02

0.2933E-02

0.3079E-02

0.3223E-02

0.3365E-02

0.3502E-02

0.3642E-02

0.3782E-02

0.3922E-02

0.4062E-02

0.4202E-02

0.4342E-02

0.4482E-02

0.4622E-02

0.4762E-02

0.4902E-02

0.5042E-02

0.5182E-02

0.5322E-02

0.5462E-02

0.5602E-02

0.5742E-02

0.5882E-02

0.6022E-02

0.6162E-02

0.6302E-02

0.6442E-02

0.6582E-02

0.6722E-02

0.6862E-02



AGARDGRAPH 315 DATA TAPE FILE: MEMCDB02.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS VERTICAL  
( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL. X = -.2540E-01 )

X  
Z  
M ref  
U ref hu  
RHO ref hu  
TAU wall prestion  
P wall mean flow

Y  
<(rho)u>  
RHO

U  
Uref

M  
Mref

Pa  
Pwall

Uref  
Uref

Uref  
Uref

Uref  
Uref

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Uref

AGARDGRAPH 315 DATA TAPE FILE: MEMCDB03.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS VERTICAL  
( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL. X = -.2540E-01 )

X  
Z  
M ref  
U ref hu  
RHO ref hu  
TAU wall prestion  
P wall mean flow

Y  
<(rho)u>  
RHO

U  
Uref

M  
Mref

Pa  
Pwall

Uref  
Uref

Uref  
Uref

Uref  
Uref

Uref  
Uref

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AGARDGRAPH 315 DATA TAPE FILE: MEMCDB04.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL. X = 0.0000E+00 )

X  
Z  
M ref  
U ref hu  
RHO ref hu  
TAU wall prestion  
P wall mean flow

Y  
<(rho)u>  
RHO

U  
Uref

M  
Mref

Pa  
Pwall

Uref  
Uref

Uref  
Uref

Uref  
Uref

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Uref





AGROGRAPH 315 DATA TAPE FILE: NEWCOMB12.DAT									
PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE ( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR-VERTICAL. X = 0.2540E-01 )									
Z	Y	<(rho)u> RHO	U	Uref	M	Pa	Puall	Uref	RHOref Uref**2
0.1102E-01	0.1585	0.8856	2.165	0.9251	0.3345E-02	0.2745E-02	0.9251	0.8856	0.3024E-02
0.1247E-01	0.1544	0.9102	2.285	0.9088	0.2749E-02	0.2461E-02	0.9088	0.3024E-02	0.3024E-02
0.1394E-01	0.1512	0.9346	2.316	0.8933	0.1951E-02	0.1737E-02	0.8933	0.3024E-02	0.3024E-02
0.1540E-01	0.1482	0.9644	2.482	0.7333	0.1218E-02	0.1042E-02	0.7333	0.3024E-02	0.3024E-02
0.1685E-01	0.1450	0.9879	2.615	0.6450	0.7976E-03	0.6871E-03	0.6450	0.3024E-02	0.3024E-02
0.1832E-01	0.1418	0.9811	2.719	0.6159	0.5208E-03	0.4546E-03	0.6159	0.3024E-02	0.3024E-02
0.1978E-01	0.1386	0.9890	2.774	0.6308	0.3372E-03	0.3283E-03	0.6308	0.3024E-02	0.3024E-02
0.2124E-01	0.1354	0.9932	2.811	0.6100	0.2015E-03	0.1899E-03	0.6100	0.3024E-02	0.3024E-02
0.2269E-01	0.1322	0.9917	2.828	0.6288	0.8289E-04	0.7894E-04	0.6288	0.3024E-02	0.3024E-02
0.2415E-01	0.1290	0.9942	2.857	0.6284	0.4843E-04	0.4707E-04	0.6284	0.3024E-02	0.3024E-02
0.2558E-01	0.1258	0.9967	2.864	0.6289	0.1975E-04	0.1932E-04	0.6289	0.3024E-02	0.3024E-02
0.2703E-01	0.1226	0.9975	2.864	0.6285	0.1342E-04	0.1313E-04	0.6285	0.3024E-02	0.3024E-02
0.2849E-01	0.1194	0.9966	2.873	0.6266	0.9535E-05	0.9339E-05	0.6266	0.3024E-02	0.3024E-02
0.2994E-01	0.1162	0.9946							
0.3139E-01	0.1130	0.9926							
0.3285E-01	0.1098	0.9906							
0.3432E-01	0.1066	0.9886							
0.3578E-01	0.1034	0.9866							

```

0.3382E-01 0.1046E-01
0.3527E-01 0.1075E-01
0.3674E-01 0.1160E-01

ACARDOGRAPH 315 DATA TAPE FILE: NEWCCOBM15.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-B DEG OFF VERTICAL. X = 0.4572E-01 )
X Y <(rhoU)> U Uref M Pa Pwall <u>+2 <u>+2 RHO-u+2 RHO-u+2 Uref+2
Z = -.1270E-01
M ref = 2.870
U ref hw = 573.9
RHO ref hw = 0.8002
TAU wall prestion = 179.6
P wall mean flow = 0.3864E+05

0.1400E-01 0.1356 0.8902 2.191 0.9665 0.2369E-02 0.2281E-02
0.1553E-01 0.1210 0.9123 2.276 0.9660 0.1700E-02 0.1767E-02
0.1698E-01 0.1037 0.9268 2.351 0.9652 0.1140E-02 0.1264E-02
0.1843E-01 0.7229E-01 0.9385 2.415 0.9627 0.5136E-03 0.5989E-03
0.1989E-01 0.4578E-01 0.9440 2.459 0.9604 0.1956E-03 0.2350E-03
0.2136E-01 0.2931E-01 0.9477 2.491 0.9583 0.7717E-04 0.9531E-04
0.2282E-01 0.2025E-01 0.9515 2.514 0.9567 0.3565E-04 0.4503E-04
0.2424E-01 0.1827E-01 0.9535 2.530 0.9513 0.2663E-04 0.3423E-04
0.2567E-01 0.2034E-01 0.9585 2.570 0.9191 0.3390E-04 0.4274E-04
0.2715E-01 0.1025 0.9705 2.655 0.7831 0.8946E-03 0.1354E-03
0.2860E-01 0.4897E-01 1.003 2.893 0.5704 0.1365E-03
0.3006E-01 0.1144E-01

```

ACARDOGRAPH 315 DATA TAPE FILE: NEWCCOBM17.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-B DEG OFF VERTICAL. X = 0.6604E-01 )

```

X Y <(rhoU)> U Uref M Pa Pwall <u>+2 <u>+2 RHO-u+2 RHO-u+2 Uref+2
Z = -.1270E-01
M ref = 2.870
U ref hw = 571.5
RHO ref hw = 0.8068
TAU wall prestion = 189.0
P wall mean flow = 0.3910E+05

0.1000E-03 0.1000 0.6450 1.284 0.9631 0.9754E-02 0.3219E-02
0.1678E-02 0.1559 0.6994 1.442 0.9639 0.8484E-02 0.3307E-02
0.2341E-02 0.1674 0.7230 1.523 0.9656 0.8373E-02 0.3697E-02
0.3010E-02 0.1805 0.7588 1.639 0.9665 0.8413E-02 0.4335E-02
0.3697E-02 0.1814 0.7845 1.723 0.9658 0.7634E-02 0.4537E-02
0.4401E-02 0.1871 0.8087 1.814 0.9645 0.7232E-02 0.4769E-02
0.5124E-02 0.1729 0.8333 1.919 0.9657 0.5411E-02 0.3996E-02
0.5867E-02 0.1713 0.8581 2.006 0.9668 0.4760E-02 0.3845E-02
0.6630E-02 0.1599 0.8787 2.106 0.9671 0.3662E-02 0.3261E-02
0.7413E-02 0.1492 0.8954 2.187 0.9665 0.2884E-02 0.2760E-02
0.8216E-02 0.1343 0.9119 2.274 0.9660 0.2100E-02 0.2179E-02
0.9039E-02 0.1137 0.9258 2.346 0.9653 0.1379E-02 0.1522E-02
0.9882E-02 0.9240E-01 0.9378 2.411 0.9628 0.8430E-03 0.9799E-03
0.1074E-01 0.6666E-01 0.9434 2.454 0.9606 0.4194E-03 0.5040E-03
0.1172E-01 0.3842E-01 0.9475 2.489 0.9585 0.1329E-03 0.1639E-03
0.1272E-01 0.2950E-01 0.9511 2.512 0.9569 0.7628E-04 0.9565E-04
0.1375E-01 0.1866E-01 0.9534 2.529 0.9517 0.2993E-04 0.3783E-04
0.1481E-01 0.1866E-01 0.9582 2.568 0.9212 0.2861E-04 0.3608E-04
0.1590E-01 0.1804E-01 0.9708 2.657 0.7939 0.2413E-04 0.2808E-04
0.1702E-01 0.2026E-01 1.001 2.876 0.5785 0.2381E-04 0.2366E-04
0.1816E-01 0.2594E-01
0.1932E-01 0.1106
0.2050E-01 0.1744E-01
0.2181E-01 0.1281E-01
0.2316E-01 0.1346E-01
0.2456E-01 0.1214E-01

```

ACARDOGRAPH 315 DATA TAPE FILE: NEWCCOBM16.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-B DEG OFF VERTICAL. X = 0.6604E-01 )

```

X Y <(rhoU)> U Uref M Pa Pwall <u>+2 <u>+2 RHO-u+2 RHO-u+2 Uref+2
Z = -.1270E-01
M ref = 2.870
U ref hw = 575.4
RHO ref hw = 0.7962
TAU wall prestion = 181.0
P wall mean flow = 0.3910E+05

0.1000E-03 0.1377 0.6532 1.322 0.9930 0.9202E-02 0.3110E-02
0.1610E-02 0.1349 0.6930 1.447 0.9640 0.8434E-02 0.3537E-02
0.2241E-02 0.1816 0.7315 1.545 0.9875 0.9143E-02 0.4289E-02
0.2900E-02 0.1922 0.7661 1.643 0.9821 0.7745E-02 0.4189E-02
0.3594E-02 0.1856 0.8017 1.727 0.9836 0.7220E-02 0.4312E-02
0.4312E-02 0.1887 0.8319 1.809 0.9755 0.6229E-02 0.4131E-02
0.5050E-02 0.1804 0.8585 1.894 0.9707 0.5107E-02 0.3787E-02
0.5812E-02 0.1689 0.8842 2.000 0.9685 0.4264E-02 0.3474E-02
0.6594E-02 0.1567 0.9093 2.083 0.9653 0.3083E-02 0.2750E-02
0.7397E-02 0.1453 0.9343 2.157 0.9635 0.2302E-02 0.2100E-02
0.8220E-02 0.1307 0.9596 2.243 0.9628 0.1615E-02 0.1522E-02
0.9063E-02 0.1107 0.9846 2.336 0.9620 0.1057E-02 0.1057E-02
0.9926E-02 0.8192E-01 0.9910 2.402 0.9577 0.7596E-03 0.9105E-03
0.1081E-01 0.6232E-01 0.9942 2.496 0.9535 0.4105E-03 0.4105E-03
0.1172E-01 0.4521E-01 0.9637 2.617 0.9008 0.1507E-03 0.1763E-03
0.1268E-01 0.2846E-01 0.9814 2.739 0.6502 0.3468E-04 0.3468E-04
0.1368E-01 0.8667E-01
0.1468E-01 0.7790E-01
0.1568E-01 0.1725E-01
0.1668E-01 0.1111E-01
0.1768E-01 0.1168E-01
0.1868E-01 0.1182E-01
0.1968E-01 0.1148E-01
0.2068E-01 0.1121E-01

```

ACARDOGRAPH 315 DATA TAPE FILE: NEWCCOBM18.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-B DEG OFF VERTICAL. X = 0.6604E-01 )

```

X Y <(rhoU)> U Uref M Pa Pwall <u>+2 <u>+2 RHO-u+2 RHO-u+2 Uref+2
Z = -.1270E-01
M ref = 2.870
U ref hw = 576.6
RHO ref hw = 0.7927
TAU wall prestion = 191.4
P wall mean flow = 0.3910E+05

```









AGROGRAPH 315 DATA TAPE FILE: NEWCC16105.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL. X = 0.7620E-01 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			

AGROGRAPH 315 DATA TAPE FILE: NEWCC16107.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL. X = 0.1397 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			

AGROGRAPH 315 DATA TAPE FILE: NEWCC16106.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL. X = 0.7620E-01 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			

AGROGRAPH 315 DATA TAPE FILE: NEWCC16108.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -16 DEG OFF VERTICAL. X = 0.7620E-01 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			

AGROGRAPH 315 DATA TAPE FILE: NEWCC16108.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS VERTICAL  
( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR - VERTICAL. X = -0.3810E-01 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			

AGROGRAPH 315 DATA TAPE FILE: NEWCC16108.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS VERTICAL  
( MEAN FLOW SURVEY NORMAL TO TUNNEL FLOOR - VERTICAL. X = -0.3810E-01 )

X	Y	M	U	Uref	(rho)U <sup>2</sup> RHO U <sup>2</sup>	Pa	Puall	(U <sup>2</sup> ) U <sup>2</sup>	RHO(U <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2973E-01	0.1683E-04	0.9204	2.303	0.6427	0.5301E-05	0.5408E-05			
0.3110E-01	0.1672E-04	0.9069	2.694	0.4154	0.3771E-05	0.3521E-05			



97

P well mean flow = 0.5437E+05

ACARDOGRAPH 315 DATA TAPE FILE: MEUCCLM08.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL. X = 0.1900E-01 )

Y	<(rho u)> RHO U	U Uref	M	Ps Pwell	<u> Uref	RHO<u> RHOref Uref
0.1500E-02	0.2392	0.5374	1.008	1.015	0.3702E-01	0.1084E-01
0.2681E-02	0.3021	0.5955	1.147	0.9904	0.4309E-01	0.1604E-01
0.4131E-02	0.3006	0.6513	1.292	0.9853	0.5592E-01	0.2680E-01
0.5554E-02	0.2800	0.7008	1.430	0.9762	0.7034E-01	0.4131E-01
0.7013E-02	0.2438	0.7456	1.565	0.9709	0.8566E-01	0.5656E-01
0.8474E-02	0.2165	0.7777	1.671	0.9711	0.1162E-01	0.8970E-02
0.9922E-02	0.1993	0.8194	1.819	0.9471	0.1575E-02	0.1263E-02
0.1156E-01	0.1808	0.8680	2.015	0.8136	0.5421E-02	0.5421E-02
0.1277E-01	0.2065	0.9169	2.239	0.6006	0.1676E-02	0.1676E-02
0.1570E-01	0.1258	0.9557	2.457	0.4344	0.1040E-02	0.1040E-02
0.1717E-01	0.1151	0.9878	2.612	0.4273	0.8343E-03	0.8343E-03
0.1863E-01	0.1019	0.9947	2.775	0.4236	0.7125E-03	0.7125E-03
0.2010E-01	0.8786E-01	1.001	2.952	0.4236	0.5828E-03	0.5828E-03
0.2158E-01	0.8686E-01	1.003	3.140	0.4236	0.4908E-03	0.4908E-03
0.2304E-01	0.8586E-01	1.003	3.330	0.4236	0.3988E-03	0.3988E-03
0.2448E-01	0.8486E-01	1.004	3.520	0.4236	0.3068E-03	0.3068E-03
0.2592E-01	0.8386E-01	1.005	3.710	0.4236	0.2148E-03	0.2148E-03
0.2736E-01	0.8286E-01	1.006	3.900	0.4236	0.1228E-03	0.1228E-03
0.2880E-01	0.8186E-01	1.007	4.090	0.4236	0.3068E-04	0.3068E-04
0.3024E-01	0.8086E-01	1.008	4.280	0.4236	0.2148E-04	0.2148E-04
0.3168E-01	0.7986E-01	1.009	4.470	0.4236	0.1228E-04	0.1228E-04
0.3312E-01	0.7886E-01	1.010	4.660	0.4236	0.3068E-05	0.3068E-05
0.3456E-01	0.7786E-01	1.011	4.850	0.4236	0.2148E-05	0.2148E-05

ACARDOGRAPH 315 DATA TAPE FILE: MEUCCLM08.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL. X = 0.1900E-01 )

X = 0.2540E-01  
Z = 0.1270E-01  
M ref = 2.850  
U ref hu = 564.9  
RHO ref hu = 0.8396  
TAU wall prestion = 147.3  
P well mean flow = 0.5669E+05

Y	<(rho u)> RHO U	U Uref	M	Ps Pwell	<u> Uref	RHO<u> RHOref Uref
0.1300E-02	0.2353	0.5278	0.9859	1.018	0.3122E-01	0.9156E-02
0.1810E-02	0.2440	0.5507	1.037	1.009	0.3178E-01	0.1022E-01
0.2320E-02	0.2963	0.6174	1.202	0.9904	0.3687E-01	0.1648E-01
0.2830E-02	0.3109	0.6759	1.360	0.9833	0.4190E-01	0.2207E-01
0.3340E-02	0.3025	0.7159	1.475	0.9759	0.4692E-01	0.2715E-01
0.3850E-02	0.2645	0.7559	1.604	0.9721	0.5194E-01	0.3223E-01
0.4360E-02	0.2190	0.7856	1.697	0.9692	0.5696E-01	0.3731E-01
0.4870E-02	0.1815	0.8292	1.858	0.9278	0.6198E-01	0.4239E-01
0.5380E-02	0.1443	0.8787	2.040	0.7755	0.6699E-01	0.4747E-01
0.5890E-02	0.1173	0.9271	2.294	0.5640	0.7199E-01	0.5255E-01
0.6400E-02	0.1044	0.9611	2.601	0.4677	0.7699E-01	0.5763E-01
0.6910E-02	0.1064	0.9907	2.927	0.4310	0.8199E-01	0.6271E-01
0.7420E-02	0.1316	0.9800	3.283	0.4236	0.8699E-01	0.6779E-01
0.7930E-02	0.1614E-01	0.9751	3.782	0.4236	0.9199E-01	0.7287E-01
0.8440E-02	0.1800E-01	1.002	4.340	0.4236	0.9699E-01	0.7795E-01
0.8950E-02	0.1800E-01	1.002	4.900	0.4236	0.1019E-01	0.8295E-01
0.9460E-02	0.1800E-01	1.003	5.460	0.4236	0.1519E-01	0.8795E-01
0.9970E-02	0.1800E-01	1.004	6.020	0.4236	0.2019E-01	0.9295E-01
0.1048E-01	0.1800E-01	1.005	6.580	0.4236	0.2519E-01	0.9795E-01
0.1099E-01	0.1800E-01	1.006	7.140	0.4236	0.3019E-01	0.1021E-01
0.1150E-01	0.1800E-01	1.007	7.700	0.4236	0.3519E-01	0.1071E-01
0.1201E-01	0.1800E-01	1.008	8.260	0.4236	0.4019E-01	0.1121E-01
0.1252E-01	0.1800E-01	1.009	8.820	0.4236	0.4519E-01	0.1171E-01
0.1303E-01	0.1800E-01	1.010	9.380	0.4236	0.5019E-01	0.1221E-01
0.1354E-01	0.1800E-01	1.011	9.940	0.4236	0.5519E-01	0.1271E-01
0.1405E-01	0.1800E-01	1.012	10.500	0.4236	0.6019E-01	0.1321E-01
0.1456E-01	0.1800E-01	1.013	11.060	0.4236	0.6519E-01	0.1371E-01
0.1507E-01	0.1800E-01	1.014	11.620	0.4236	0.7019E-01	0.1421E-01
0.1558E-01	0.1800E-01	1.015	12.180	0.4236	0.7519E-01	0.1471E-01
0.1609E-01	0.1800E-01	1.016	12.740	0.4236	0.8019E-01	0.1521E-01
0.1660E-01	0.1800E-01	1.017	13.300	0.4236	0.8519E-01	0.1571E-01
0.1711E-01	0.1800E-01	1.018	13.860	0.4236	0.9019E-01	0.1621E-01
0.1762E-01	0.1800E-01	1.019	14.420	0.4236	0.9519E-01	0.1671E-01
0.1813E-01	0.1800E-01	1.020	14.980	0.4236	0.1000E-01	0.1721E-01
0.1864E-01	0.1800E-01	1.021	15.540	0.4236	0.1050E-01	0.1771E-01
0.1915E-01	0.1800E-01	1.022	16.100	0.4236	0.1100E-01	0.1821E-01
0.1966E-01	0.1800E-01	1.023	16.660	0.4236	0.1150E-01	0.1871E-01
0.2017E-01	0.1800E-01	1.024	17.220	0.4236	0.1200E-01	0.1921E-01
0.2068E-01	0.1800E-01	1.025	17.780	0.4236	0.1250E-01	0.1971E-01
0.2119E-01	0.1800E-01	1.026	18.340	0.4236	0.1300E-01	0.2021E-01
0.2170E-01	0.1800E-01	1.027	18.900	0.4236	0.1350E-01	0.2071E-01
0.2221E-01	0.1800E-01	1.028	19.460	0.4236	0.1400E-01	0.2121E-01
0.2272E-01	0.1800E-01	1.029	20.020	0.4236	0.1450E-01	0.2171E-01
0.2323E-01	0.1800E-01	1.030	20.580	0.4236	0.1500E-01	0.2221E-01
0.2374E-01	0.1800E-01	1.031	21.140	0.4236	0.1550E-01	0.2271E-01
0.2425E-01	0.1800E-01	1.032	21.700	0.4236	0.1600E-01	0.2321E-01
0.2476E-01	0.1800E-01	1.033	22.260	0.4236	0.1650E-01	0.2371E-01
0.2527E-01	0.1800E-01	1.034	22.820	0.4236	0.1700E-01	0.2421E-01
0.2578E-01	0.1800E-01	1.035	23.380	0.4236	0.1750E-01	0.2471E-01
0.2629E-01	0.1800E-01	1.036	23.940	0.4236	0.1800E-01	0.2521E-01
0.2680E-01	0.1800E-01	1.037	24.500	0.4236	0.1850E-01	0.2571E-01
0.2731E-01	0.1800E-01	1.038	25.060	0.4236	0.1900E-01	0.2621E-01
0.2782E-01	0.1800E-01	1.039	25.620	0.4236	0.1950E-01	0.2671E-01
0.2833E-01	0.1800E-01	1.040	26.180	0.4236	0.2000E-01	0.2721E-01
0.2884E-01	0.1800E-01	1.041	26.740	0.4236	0.2050E-01	0.2771E-01
0.2935E-01	0.1800E-01	1.042	27.300	0.4236	0.2100E-01	0.2821E-01
0.2986E-01	0.1800E-01	1.043	27.860	0.4236	0.2150E-01	0.2871E-01
0.3037E-01	0.1800E-01	1.044	28.420	0.4236	0.2200E-01	0.2921E-01
0.3088E-01	0.1800E-01	1.045	28.980	0.4236	0.2250E-01	0.2971E-01
0.3139E-01	0.1800E-01	1.046	29.540	0.4236	0.2300E-01	0.3021E-01
0.3190E-01	0.1800E-01	1.047	30.100	0.4236	0.2350E-01	0.3071E-01
0.3241E-01	0.1800E-01	1.048	30.660	0.4236	0.2400E-01	0.3121E-01
0.3292E-01	0.1800E-01	1.049	31.220	0.4236	0.2450E-01	0.3171E-01
0.3343E-01	0.1800E-01	1.050	31.780	0.4236	0.2500E-01	0.3221E-01
0.3394E-01	0.1800E-01	1.051	32.340	0.4236	0.2550E-01	0.3271E-01
0.3445E-01	0.1800E-01	1.052	32.900	0.4236	0.2600E-01	0.3321E-01
0.3496E-01	0.1800E-01	1.053	33.460	0.4236	0.2650E-01	0.3371E-01
0.3547E-01	0.1800E-01	1.054	34.020	0.4236	0.2700E-01	0.3421E-01
0.3598E-01	0.1800E-01	1.055	34.580	0.4236	0.2750E-01	0.3471E-01
0.3649E-01	0.1800E-01	1.056	35.140	0.4236	0.2800E-01	0.3521E-01
0.3700E-01	0.1800E-01	1.057	35.700	0.4236	0.2850E-01	0.3571E-01
0.3751E-01	0.1800E-01	1.058	36.260	0.4236	0.2900E-01	0.3621E-01
0.3802E-01	0.1800E-01	1.059	36.820	0.4236	0.2950E-01	0.3671E-01
0.3853E-01	0.1800E-01	1.060	37.380	0.4236	0.3000E-01	0.3721E-01
0.3904E-01	0.1800E-01	1.061	37.940	0.4236	0.3050E-01	0.3771E-01
0.3955E-01	0.1800E-01	1.062	38.500	0.4236	0.3100E-01	0.3821E-01
0.4006E-01	0.1800E-01	1.063	39.060	0.4236	0.3150E-01	0.3871E-01
0.4057E-01	0.1800E-01	1.064	39.620	0.4236	0.3200E-01	0.3921E-01
0.4108E-01	0.1800E-01	1.065	40.180	0.4236	0.3250E-01	0.3971E-01
0.4159E-01	0.1800E-01	1.066	40.740	0.4236	0.3300E-01	0.4021E-01
0.4210E-01	0.1800E-01	1.067	41.300	0.4236	0.3350E-01	0.4071E-01
0.4261E-01	0.1800E-01	1.068	41.860	0.4236	0.3400E-01	0.4121E-01
0.4312E-01	0.1800E-01	1.069	42.420	0.4236	0.3450E-01	0.4171E-01
0.4363E-01	0.1800E-01	1.070	42.980	0.4236	0.3500E-01	0.4221E-01
0.4414E-01	0.1800E-01	1.071	43.540	0.4236	0.3550E-01	0.4271E-01
0.4465E-01	0.1800E-01	1.072	44.100	0.4236	0.3600E-01	0.4321E-01
0.4516E-01	0.1800E-01	1.073	44.660	0.4236	0.3650E-01	0.4371E-01
0.4567E-01	0.1800E-01	1.074	45.220	0.4236	0.3700E-01	0.4421E-01
0.4618E-01	0.1800E-01	1.075	45.780	0.4236	0.3750E-01	0.4471E-01
0.4669E-01	0.1800E-01	1.076	46.340	0.4236	0.3800E-01	0.4521E-01
0.4720E-01	0.1800E-01	1.077	46.900	0.4236	0.3850E-01	0.4571E-01
0.4771E-01	0.1800E-01	1.078	47.460	0.4236	0.3900E-01	0.4621E-01
0.4822E-01	0.1800E-01	1.079	48.020	0.4236	0.3950E-01	0.4671E-01
0.4873E-01	0.1800E-01	1.080	48.580	0.4236	0.4000E-01	0.4721E-01
0.4924E-01	0.1800E-01	1.081	49.140	0.4236	0.4050E-01	0.4771E-01
0.4975E-01	0.1800E-01	1.082	49.700	0.4236	0.4100E-01	0.4821E-01
0.5026E-01	0.1800E-01	1.083	50.260	0.4236	0.4150E-01	0.4871E-01
0.5077E-01	0.1800E-01	1.084	50.820	0.4236	0.4200E-01	0.4921E-01
0.5128E-01	0.1800E-01	1.085	51.380	0.4236	0.4250E-01	0.4971E-01
0.5179E-01	0.1800E-01	1.086	51.940	0.4236	0.4300E-01	0.5021E-01
0.5230E-01	0.1800E-01	1.087	52.500	0.4236	0.4350E-01	0.5071E-01
0.5281E-01	0.1800E-01	1.088	53.060	0.4236	0.4400E-01	0.5121E-01
0.5332E-01	0.1800E-01	1.089	53.620	0.4236	0.4450E-01	0.5171E-01
0.5383E-01	0.1800E-01	1.090	54.180	0.4236	0.4500E-01	0.5221E-01
0.5434E-01	0.1800E-01	1.091	54.740	0.4236	0.4550E-01	0.5271E-01
0.5485E-01	0.1800E-01	1.092	55.300	0.4236	0.4600E-01	0.5321E-01
0.5536E-01	0.1800E-01	1.093	55.860	0.4236	0.4650E-01	0.5371E-01
0.5587E-01	0.1800E-01	1.094	56.420	0.4236	0.4700E-01	0.5421E-01
0.5638E-01	0.1800E-01	1.095	56.980	0.4236	0.4750E-01	0.5471E-01
0.5689E-01	0.1800E-01	1.096	57.540	0.4236	0.4800E-01	0.5521E-01
0.5740E-01	0.1800E-01	1.097	58.100	0.4236	0.4850E-01	0.5571E-01
0.5791E-01	0.1800E-01	1.098	58.660	0.4236	0.4900E-01	0.5621E-01
0.5842E-01	0.1800E-01	1.099	59.220	0.4236	0.4950E-01	0.5671E-01
0.5893E-01	0.1800E-01	1.100	59.780	0.4236	0.5000E-01	0.5721E-01
0.5944E-01	0.1800E-01	1.101	60.340	0.4236	0.5050E-01	0.5771E-01
0.5995E-01	0.1800E-01	1.102	60.900	0.4236	0.5100E-01	0.5821E-01
0.6046E-01	0.1800E-01	1.103	61.460	0.4236	0.5150E-01	0.5871E-01
0.6097E-01	0.1800E-01	1.104	62.020	0.4236	0.5200E-01	0.5921E-01
0.6148E-01	0.1800E-01	1.105	62.580	0.4236	0.5250E-01	0.5971E-01
0.6199E-01	0.1800E-01	1.106	63.140	0.4236	0.5300E-01	0.6021E-01
0.6250E-01	0.1800E-01	1.107	63.700	0.4236	0.5350E-01	0.6071E-01
0.6301E-01	0.1800E-01	1.108	64.260	0.4236	0.5400E-01	0.6121E-01
0.6352E-01	0.1800E-01	1.109	64.820	0.4236	0.5450E-01	0.6171E-01
0.6403E-01	0.1800E-01	1.110	65.380	0.4236	0.5500E-01	0.6221E-01
0.6454E-01	0.1800E-01	1.111	65.940	0.4236	0.5550E-01	0.6271E-01
0.6505E-01	0.1800E-01	1.112	66.500	0.4236	0.5600E-01	0.6321E-01
0.6556E-01	0.1800E-01	1.113	67.060	0.4236	0.5650E-01	0.6371E-01
0.6607E-01	0.1800E-01	1.114	67.620	0.4236	0.5700E-01	0.6421E-01
0.6658E-01	0.1800E-01	1.115	68.180	0.4236	0.5750E-	

0.2549E-01 0.1313E-01 1.005 2.829 0.3766 0.1037E-04 0.1037E-04 0.2065E-01 0.5814E-01 0.8992 2.184 0.9800 0.4392E-03 0.6358E-03  
 0.2713E-01 0.1351E-01 1.004 2.823 0.3785 0.1101E-04 0.1101E-04 0.2210E-01 0.2468E-01 0.8987 2.184 0.9798 0.7901E-04 0.1144E-03  
 AGADGRAPH 315 DATA TAPE FILE: MEUC16M11.DAT  
 PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL. X = 0.7620E-01 )  
 X = 0.7620E-01  
 Z = -1.270E-01  
 M ref = 2.850  
 U ref hu = 562.8  
 RHO ref hu = 0.8457  
 TAU well prestn = 234.6  
 P well mean flow = 0.6280E+05

Y	<(rho)u> RHO	U	Uref	M	Ps	Pwell	Uref	Uref**2	RHOref	Uref**2	RHOref	Uref**2
0.1300E-02	0.2098	0.6230	0.6230	1.217	1.016	0.1914E-01	0.9457E-02	0.3558E-01	0.6646	1.341	1.053	0.2764E-03
0.1670E-02	0.2283	0.6297	0.6297	1.234	1.006	0.2221E-01	0.1118E-01	0.1064	0.6708	1.358	1.051	0.2395E-02
0.2470E-02	0.2488	0.6472	0.6472	1.280	0.9958	0.2497E-01	0.1337E-01	0.1771	0.6794	1.380	1.048	0.6802E-02
0.3890E-02	0.2700	0.6785	0.6785	1.367	0.9779	0.2647E-01	0.1587E-01	0.2115	0.6881	1.403	1.045	0.9829E-02
0.5380E-02	0.2866	0.7068	0.7068	1.451	0.9693	0.2668E-01	0.1800E-01	0.2137	0.6963	1.427	1.041	0.1004E-01
0.6830E-02	0.2889	0.7331	0.7331	1.532	0.9647	0.2665E-01	0.1833E-01	0.2175	0.7042	1.450	1.040	0.1043E-01
0.8300E-02	0.2819	0.7649	0.7649	1.634	0.9615	0.2213E-01	0.1772E-01	0.2294	0.7191	1.495	1.039	0.1649E-01
0.9780E-02	0.2856	0.7933	0.7933	1.733	0.9621	0.1870E-01	0.1604E-01	0.2356	0.7360	1.550	1.031	0.1201E-01
0.1126E-01	0.2569	0.8284	0.8284	1.859	0.9611	0.1288E-01	0.1404E-01	0.2487	0.7485	1.721	1.022	0.1574E-01
0.1274E-01	0.2308	0.8461	0.8461	1.928	0.9693	0.9531E-02	0.1123E-01	0.2548	0.7686	1.878	1.017	0.1320E-01
0.1422E-01	0.1940	0.8688	0.8688	2.023	0.9693	0.5976E-02	0.7779E-02	0.2648	0.7860	2.068	1.012	0.1090E-01
0.1567E-01	0.1651	0.8869	0.8869	2.105	0.9760	0.3909E-02	0.5545E-02	0.2748	0.8091	2.281	1.012	0.8757E-02
0.1717E-01	0.1167	0.8916	0.8916	2.135	0.9770	0.1881E-02	0.2749E-02	0.2848	0.8324	2.537	1.016	0.7022E-02
0.1866E-01	0.8128E-01	0.8953	0.8953	2.165	0.9810	0.8791E-03	0.1326E-02	0.2948	0.8511	2.817	1.021	0.5806E-02
0.2017E-01	0.4979E-01	0.8979	0.8979	2.177	0.9784	0.3250E-03	0.9468E-03	0.3048	0.8683	3.133	1.024	0.4777E-02
0.2165E-01	0.3268E-01	0.8972	0.8972	2.180	0.9815	0.1394E-03	0.2135E-03	0.3148	0.8859	3.488	1.027	0.3195E-02
0.2315E-01	0.2072E-01	0.8985	0.8985	2.187	0.9776	0.5556E-04	0.8528E-04	0.3248	0.9021	3.869	1.031	0.2138E-02
0.2465E-01	0.1795E-01	0.8990	0.8990	2.190	0.9776	0.4173E-04	0.6403E-04	0.3348	0.9194	4.281	1.034	0.1641E-02
0.2614E-01	0.1903E-01	0.9020	0.9020	2.200	0.9750	0.3993E-04	0.6184E-04	0.3448	0.9367	4.733	1.034	0.1201E-02
0.2764E-01	0.1814E-01	0.9061	0.9061	2.224	0.9514	0.3993E-04	0.6290E-04	0.3548	0.9532	5.217	1.034	0.8757E-02
0.2916E-01	0.2247E-01	0.9206	0.9206	2.301	0.7741	0.5691E-04	0.7653E-04	0.3648	0.9693	5.733	1.034	0.6413E-04

0.2549E-01 0.1313E-01 1.005 2.829 0.3766 0.1037E-04 0.1037E-04 0.2065E-01 0.5814E-01 0.8992 2.184 0.9800 0.4392E-03 0.6358E-03  
 0.2713E-01 0.1351E-01 1.004 2.823 0.3785 0.1101E-04 0.1101E-04 0.2210E-01 0.2468E-01 0.8987 2.184 0.9798 0.7901E-04 0.1144E-03  
 AGADGRAPH 315 DATA TAPE FILE: MEUC16M11.DAT  
 PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-16 DEG OFF VERTICAL. X = 0.7620E-01 )  
 X = 0.7620E-01  
 Z = -1.270E-01  
 M ref = 2.850  
 U ref hu = 562.8  
 RHO ref hu = 0.8457  
 TAU well prestn = 234.6  
 P well mean flow = 0.6280E+05

Y	<(rho)u> RHO	U	Uref	M	Ps	Pwell	Uref	Uref**2	RHOref	Uref**2	RHOref	Uref**2
0.1300E-02	0.2098	0.6230	0.6230	1.217	1.016	0.1914E-01	0.9457E-02	0.3558E-01	0.6646	1.341	1.053	0.2764E-03
0.1670E-02	0.2283	0.6297	0.6297	1.234	1.006	0.2221E-01	0.1118E-01	0.1064	0.6708	1.358	1.051	0.2395E-02
0.2470E-02	0.2488	0.6472	0.6472	1.280	0.9958	0.2497E-01	0.1337E-01	0.1771	0.6794	1.380	1.048	0.6802E-02
0.3890E-02	0.2700	0.6785	0.6785	1.367	0.9779	0.2647E-01	0.1587E-01	0.2115	0.6881	1.403	1.045	0.9829E-02
0.5380E-02	0.2866	0.7068	0.7068	1.451	0.9693	0.2668E-01	0.1800E-01	0.2137	0.6963	1.427	1.041	0.1004E-01
0.6830E-02	0.2889	0.7331	0.7331	1.532	0.9647	0.2665E-01	0.1833E-01	0.2175	0.7042	1.450	1.040	0.1043E-01
0.8300E-02	0.2819	0.7649	0.7649	1.634	0.9615	0.2213E-01	0.1772E-01	0.2294	0.7191	1.495	1.039	0.1649E-01
0.9780E-02	0.2856	0.7933	0.7933	1.733	0.9621	0.1870E-01	0.1604E-01	0.2356	0.7360	1.550	1.031	0.1201E-01
0.1126E-01	0.2569	0.8284	0.8284	1.859	0.9611	0.1288E-01	0.1404E-01	0.2487	0.7485	1.721	1.022	0.1574E-01
0.1274E-01	0.2308	0.8461	0.8461	1.928	0.9693	0.9531E-02	0.1123E-01	0.2548	0.7686	1.878	1.017	0.1320E-01
0.1422E-01	0.1940	0.8688	0.8688	2.023	0.9693	0.5976E-02	0.7779E-02	0.2648	0.7860	2.068	1.012	0.1090E-01
0.1567E-01	0.1651	0.8869	0.8869	2.105	0.9760	0.3909E-02	0.5545E-02	0.2748	0.8091	2.281	1.012	0.8757E-02
0.1717E-01	0.1167	0.8916	0.8916	2.135	0.9770	0.1881E-02	0.2749E-02	0.2848	0.8324	2.537	1.016	0.7022E-02
0.1866E-01	0.8128E-01	0.8953	0.8953	2.165	0.9810	0.8791E-03	0.1326E-02	0.2948	0.8511	2.817	1.021	0.5806E-02
0.2017E-01	0.4979E-01	0.8979	0.8979	2.177	0.9784	0.3250E-03	0.9468E-03	0.3048	0.8683	3.133	1.024	0.4777E-02
0.2165E-01	0.3268E-01	0.8972	0.8972	2.180	0.9815	0.1394E-03	0.2135E-03	0.3148	0.8859	3.488	1.027	0.3195E-02
0.2315E-01	0.2072E-01	0.8985	0.8985	2.187	0.9776	0.5556E-04	0.8528E-04	0.3248	0.9021	3.869	1.031	0.2138E-02
0.2465E-01	0.1795E-01	0.8990	0.8990	2.190	0.9776	0.4173E-04	0.6403E-04	0.3348	0.9194	4.281	1.034	0.1641E-02
0.2614E-01	0.1903E-01	0.9020	0.9020	2.200	0.9750	0.3993E-04	0.6290E-04	0.3448	0.9367	4.733	1.034	0.1201E-02
0.2764E-01	0.1814E-01	0.9061	0.9061	2.224	0.9514	0.3993E-04	0.7653E-04	0.3548	0.9532	5.217	1.034	0.8757E-02
0.2916E-01	0.2247E-01	0.9206	0.9206	2.301	0.7741	0.5691E-04	0.7653E-04	0.3648	0.9693	5.733	1.034	0.6413E-04

U ref hu = 566.2  
 RNO ref hu = 0.8775  
 TAU well prestion = 98.22  
 P well mean flow = 0.6597E+05

Y	(rho)u <sup>2</sup> RNO U <sup>2</sup>	U	Uref	M	Pa	Pwell	(u <sup>2</sup> ) U <sup>2</sup>	RNO(u <sup>2</sup> ) RNOref Uref <sup>2</sup>
0.2000E-02	-9.635E-02	0.4281	0.7680	0.7680	0.9854	0.9854	-7.796E-02	-1.487E-02
0.3440E-02	-1.321E-01	0.4874	0.8940	0.8940	0.9584	0.9584	-1.001E-01	-2.517E-02
0.4891E-02	-1.573E-01	0.5615	1.061	1.061	0.9419	0.9419	-9.666E-02	-3.294E-02
0.6322E-02	-1.83E-01	0.6352	1.243	1.243	0.9349	0.9349	-7.309E-02	-3.468E-02
0.7760E-02	-1.700E-01	0.6934	1.402	1.402	0.9312	0.9312	-9.519E-02	-5.718E-02
0.9123E-02	-1.245E-01	0.7469	1.542	1.542	0.9284	0.9284	-6.807E-02	-6.687E-02
0.1041E-01	-9.729E-02	0.7887	1.698	1.698	0.9236	0.9236	-4.519E-02	-3.950E-02
0.1208E-01	-8.783E-02	0.8330	1.857	1.857	0.9132	0.9132	-3.691E-02	-3.164E-02
0.1354E-01	-7.041E-02	0.8631	1.991	1.991	0.8952	0.8952	-2.740E-02	-2.167E-02
0.1490E-01	-5.389E-02	0.8884	2.090	2.090	0.8595	0.8595	-1.617E-02	-1.174E-02
0.1644E-01	-3.893E-02	0.9090	2.189	2.189	0.7971	0.7971	0.4766E-03	0.4608E-03
0.1792E-01	-2.1584E-01	0.9058	2.185	2.185	0.6818	0.6818	0.5464E-02	0.5315E-02
0.1937E-01	-6.439E-02	0.9156	2.242	2.242	0.5352	0.5352	0.2139E-02	0.1889E-02
0.2079E-01	-6.710E-03							
0.2226E-01	-9.979E-04							
0.2373E-01	-1.815E-04							
0.2520E-01	-9.281E-05							
0.2667E-01	-1.864E-05							
0.2814E-01	-1.821E-05							
0.2959E-01	0.1698E-05							
0.3108E-01	0.3778E-05							

AGARDGRAPH 315 DATA TAPE FILE: MEUC20102.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY IS 5.5 DEGREES OFF VERTICAL. X = 0.1270E-01 )

X  
 Z  
 M ref  
 U ref hu  
 RNO ref hu  
 TAU well prestion  
 P well mean flow

Y	(rho)u <sup>2</sup> RNO U <sup>2</sup>	U	Uref	M	Pa	Pwell	(u <sup>2</sup> ) U <sup>2</sup>	RNO(u <sup>2</sup> ) RNOref Uref <sup>2</sup>
0.2000E-02	-7.400E-02	0.3846	0.6839	0.6839	0.9400	0.9400	-6.361E-02	-8.185E-03
0.3380E-02	-1.019E-01	0.4518	0.8012	0.8012	0.9384	0.9384	-7.700E-02	-1.450E-02
0.4812E-02	-1.430E-01	0.5771	1.076	1.076	0.9296	0.9296	-9.700E-02	-3.156E-02
0.6252E-02	-1.379E-01	0.6432	1.283	1.283	0.9269	0.9269	-8.410E-02	-3.419E-02
0.7685E-02	-1.102E-01	0.7014	1.427	1.427	0.9231	0.9231	-6.510E-02	-3.353E-02
0.9131E-02	-1.244E-01	0.7525	1.584	1.584	0.9138	0.9138	-6.210E-02	-4.128E-02
0.1058E-01	-1.007E-01	0.7971	1.732	1.732	0.8998	0.8998	-4.943E-02	-3.871E-02
0.1201E-01	-6.271E-02	0.8295	1.851	1.851	0.8825	0.8825	-3.490E-02	-3.060E-02
0.1343E-01	-6.633E-02	0.8664	1.997	1.997	0.8550	0.8550	-2.550E-02	-2.528E-02
0.1485E-01	-2.032E-02	0.9028	2.157	2.157	0.7993	0.7993	-7.102E-03	-7.662E-03
0.1632E-01	-6.229E-02	0.9413	2.354	2.354	0.6747	0.6747	0.2101E-02	0.2101E-02
0.1777E-01	-6.770E-03	0.9947	2.678	2.678	0.4643	0.4643	0.1691E-03	0.1691E-03
0.1923E-01	-1.225E-02							
0.2067E-01	-7.252E-03							
0.2216E-01	-5.429E-03							
0.2361E-01	-1.725E-03							
0.2506E-01	-1.698E-03							
0.2651E-01	-1.015E-03							
0.2798E-01	-3.608E-04							
0.2944E-01	-1.623E-04							
0.3091E-01	-5.402E-05							
0.3238E-01	0.5139E-05							
0.3383E-01	0.5909E-05							
0.3532E-01	-3.679E-05							

AGARDGRAPH 315 DATA TAPE FILE: MEUC20103.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -20 DEG OFF VERTICAL. X = 0.2540E-01 )

X  
 Z  
 M ref

Y	(rho)u <sup>2</sup> RNO U <sup>2</sup>	U	Uref	M	Pa	Pwell	(u <sup>2</sup> ) U <sup>2</sup>	RNO(u <sup>2</sup> ) RNOref Uref <sup>2</sup>
0.2000E-02	-8.370E-02	0.4633	0.8422	0.8422	1.048	1.048	-6.520E-02	-1.738E-02
0.3412E-02	-9.564E-02	0.5113	0.9481	0.9481	1.012	1.012	-7.034E-02	-2.294E-02
0.4844E-02	-1.187E-01	0.5562	1.050	1.050	0.9887	0.9887	-8.235E-02	-3.222E-02
0.6277E-02	-1.430E-01	0.6066	1.175	1.175	0.9746	0.9746	-9.212E-02	-4.443E-02
0.7732E-02	-1.396E-01	0.6765	1.358	1.358	0.9628	0.9628	-8.038E-02	-5.117E-02
0.9157E-02	-1.140E-01	0.7159	1.472	1.472	0.9516	0.9516	-6.535E-02	-4.535E-02
0.1060E-01	-1.053E-01	0.7694	1.640	1.640	0.9495	0.9495	-5.073E-02	-4.657E-02
0.1203E-01	-9.023E-02	0.8009	1.747	1.747	0.9491	0.9491	-4.062E-02	-4.224E-02
0.1348E-01	-4.762E-02	0.8404	1.891	1.891	0.9420	0.9420	-2.899E-02	-3.421E-02
0.1495E-01	-6.820E-02	0.8603	1.973	1.973	0.9420	0.9420	-1.863E-02	-2.659E-02
0.1639E-01	-1.702E-02	0.8851	2.085	2.085	0.9326	0.9326	-4.244E-03	-9.037E-03
0.1781E-01	-2.862E-03	0.9025	2.121	2.121	0.9125	0.9125	-1.024E-03	-1.505E-03
0.1922E-01	0.3158E-02	0.9184	2.181	2.181	0.8646	0.8646	0.1088E-02	0.1612E-02
0.2070E-01	0.2644E-02	0.9364	2.344	2.344	0.7108	0.7108	0.4236E-02	0.1173E-01
0.2215E-01	0.3891E-02	0.9782	2.601	2.601	0.4441	0.4441	0.1050E-02	0.1131E-01
0.2363E-01	-1.136E-03							
0.2506E-01	-3.117E-05							
0.2654E-01	0.5068E-06							

AGARDGRAPH 315 DATA TAPE FILE: MEUC20105.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -20 DEG OFF VERTICAL. X = 0.4128E-01 )

X  
 Z  
 M ref

AGARDGRAPH 315 DATA TAPE FILE: MEUC20104.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
 ( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE -20 DEG OFF VERTICAL. X = 0.4128E-01 )

X  
 Z  
 M ref  
 U ref hu  
 RNO ref hu  
 TAU well prestion  
 P well mean flow



AGARDGRAPH 315 DATA TAPE FILE: MEUC20107.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL. X = 0.1143 )

X	Y	(rho)u <sup>2</sup> RHO Uref <sup>2</sup>	U	Uref	M	Ps	Puall	(u <sup>2</sup> ) Uref <sup>2</sup>	RHO(u <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2000E-02	0.8422	0.4633	0.4633	0.4633	1.187	0.9874	0.9874	1.000E-02	-5450E-03
0.3427E-02	0.9493	0.5118	0.5118	0.5118	1.257	0.9672	0.9672	-1815E-02	-1087E-02
0.4868E-02	1.051	0.5663	0.5663	0.5663	1.324	0.9572	0.9572	-2153E-02	-1415E-02
0.6303E-02	1.176	0.6272	0.6272	0.6272	1.363	0.9474	0.9474	-3528E-02	-2511E-02
0.7751E-02	1.359	0.6928	0.6928	0.6928	1.427	0.9424	0.9424	-4963E-02	-3711E-02
0.9211E-02	1.475	0.7697	0.7697	0.7697	1.521	0.9404	0.9404	-5345E-02	-4535E-02
0.1064E-01	1.641	0.8592	0.8592	0.8592	1.575	0.9393	0.9393	-5698E-02	-5176E-02
0.1208E-01	1.759	0.9492	0.9492	0.9492	1.724	0.9386	0.9386	-5508E-02	-4074E-02
0.1349E-01	1.892	0.9491	0.9491	0.9491	1.780	0.9418	0.9418	-5698E-02	-4432E-02
0.1497E-01	1.973	0.8406	0.8406	0.8406	1.842	0.9476	0.9476	-4735E-02	-4484E-02
0.1642E-01	1.973	0.8404	0.8404	0.8404	1.903	0.9515	0.9515	-3975E-02	-4484E-02
0.1789E-01	2.085	0.8852	0.8852	0.8852	1.995	0.9548	0.9548	-3274E-02	-4484E-02
0.1934E-01	2.125	0.8916	0.8916	0.8916	2.009	0.9616	0.9616	-3408E-02	-4484E-02
0.1934E-01	2.187	0.8634	0.8634	0.8634	1.993	0.9735	0.9735	-3528E-02	-4484E-02
0.2082E-01	2.377	0.9036	0.9036	0.9036	1.902	0.9738	0.9738	-3528E-02	-4484E-02
0.2229E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2374E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2520E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2666E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2814E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02

AGARDGRAPH 315 DATA TAPE FILE: MEUC20108.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL. X = 0.7402E-01 )

X	Y	(rho)u <sup>2</sup> RHO Uref <sup>2</sup>	U	Uref	M	Ps	Puall	(u <sup>2</sup> ) Uref <sup>2</sup>	RHO(u <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2000E-02	0.8422	0.4633	0.4633	0.4633	1.187	0.9874	0.9874	1.000E-02	-5450E-03
0.3427E-02	0.9493	0.5118	0.5118	0.5118	1.257	0.9672	0.9672	-1815E-02	-1087E-02
0.4868E-02	1.051	0.5663	0.5663	0.5663	1.324	0.9572	0.9572	-2153E-02	-1415E-02
0.6303E-02	1.176	0.6272	0.6272	0.6272	1.363	0.9474	0.9474	-3528E-02	-2511E-02
0.7751E-02	1.359	0.6928	0.6928	0.6928	1.427	0.9424	0.9424	-4963E-02	-3711E-02
0.9211E-02	1.475	0.7697	0.7697	0.7697	1.521	0.9404	0.9404	-5345E-02	-4535E-02
0.1064E-01	1.641	0.8592	0.8592	0.8592	1.575	0.9393	0.9393	-5698E-02	-5176E-02
0.1208E-01	1.759	0.9492	0.9492	0.9492	1.724	0.9386	0.9386	-5508E-02	-4074E-02
0.1349E-01	1.892	0.9491	0.9491	0.9491	1.780	0.9418	0.9418	-5698E-02	-4432E-02
0.1497E-01	1.973	0.8406	0.8406	0.8406	1.842	0.9476	0.9476	-4735E-02	-4484E-02
0.1642E-01	1.973	0.8404	0.8404	0.8404	1.903	0.9515	0.9515	-3975E-02	-4484E-02
0.1789E-01	2.085	0.8852	0.8852	0.8852	1.995	0.9548	0.9548	-3274E-02	-4484E-02
0.1934E-01	2.125	0.8916	0.8916	0.8916	2.009	0.9616	0.9616	-3408E-02	-4484E-02
0.1934E-01	2.187	0.8634	0.8634	0.8634	1.993	0.9735	0.9735	-3528E-02	-4484E-02
0.2082E-01	2.377	0.9036	0.9036	0.9036	1.902	0.9738	0.9738	-3528E-02	-4484E-02
0.2229E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2374E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2520E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2666E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2814E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02

AGARDGRAPH 315 DATA TAPE FILE: MEUC20109.DAT

PROFILE TABULATION - INCLINED WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL. X = 0.1143 )

X	Y	(rho)u <sup>2</sup> RHO Uref <sup>2</sup>	U	Uref	M	Ps	Puall	(u <sup>2</sup> ) Uref <sup>2</sup>	RHO(u <sup>2</sup> ) RHOref Uref <sup>2</sup>
0.2000E-02	0.8422	0.4633	0.4633	0.4633	1.187	0.9874	0.9874	1.000E-02	-5450E-03
0.3427E-02	0.9493	0.5118	0.5118	0.5118	1.257	0.9672	0.9672	-1815E-02	-1087E-02
0.4868E-02	1.051	0.5663	0.5663	0.5663	1.324	0.9572	0.9572	-2153E-02	-1415E-02
0.6303E-02	1.176	0.6272	0.6272	0.6272	1.363	0.9474	0.9474	-3528E-02	-2511E-02
0.7751E-02	1.359	0.6928	0.6928	0.6928	1.427	0.9424	0.9424	-4963E-02	-3711E-02
0.9211E-02	1.475	0.7697	0.7697	0.7697	1.521	0.9404	0.9404	-5345E-02	-4535E-02
0.1064E-01	1.641	0.8592	0.8592	0.8592	1.575	0.9393	0.9393	-5698E-02	-5176E-02
0.1208E-01	1.759	0.9492	0.9492	0.9492	1.724	0.9386	0.9386	-5508E-02	-4074E-02
0.1349E-01	1.892	0.9491	0.9491	0.9491	1.780	0.9418	0.9418	-5698E-02	-4432E-02
0.1497E-01	1.973	0.8406	0.8406	0.8406	1.842	0.9476	0.9476	-4735E-02	-4484E-02
0.1642E-01	1.973	0.8404	0.8404	0.8404	1.903	0.9515	0.9515	-3975E-02	-4484E-02
0.1789E-01	2.085	0.8852	0.8852	0.8852	1.995	0.9548	0.9548	-3274E-02	-4484E-02
0.1934E-01	2.125	0.8916	0.8916	0.8916	2.009	0.9616	0.9616	-3408E-02	-4484E-02
0.1934E-01	2.187	0.8634	0.8634	0.8634	1.993	0.9735	0.9735	-3528E-02	-4484E-02
0.2082E-01	2.377	0.9036	0.9036	0.9036	1.902	0.9738	0.9738	-3528E-02	-4484E-02
0.2229E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2374E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2520E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2666E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02
0.2814E-01	2.571	0.9733	0.9733	0.9733	1.902	0.9689	0.9689	-3528E-02	-4484E-02





Y	<(rho)u> RHO	U	Uref	M	Pa	Pwall	<u> <sup>2</sup> U <sup>2</sup>	RHO<u> <sup>2</sup> RHOref Uref <sup>2</sup>
0.1192E-01	0.2015	0.8265	1.840	0.8836	0.8122E-02	0.7047E-02	0.4633	0.1253E-01
0.1338E-01	0.1736	0.6637	1.966	0.8571	0.5012E-02	0.4915E-02	0.5065	0.1774E-01
0.1478E-01	0.1663	0.9002	2.145	0.8046	0.3804E-02	0.4150E-02	0.5557	0.2172E-01
0.1625E-01	0.1917	0.9421	2.358	0.6753	0.3863E-02	0.4211E-02	0.6045	0.2699E-01
0.1770E-01	0.1380	0.9920	2.660	0.4745	0.1407E-02	0.1370E-02	0.6716	0.3817E-01
0.1917E-01	0.9773E-01						0.7145	0.5322E-01
0.2063E-01	0.6413E-01						0.7691	0.7810E-01
0.2210E-01	0.6623E-01						0.8014	0.970E-01
0.2359E-01	0.4603E-01						0.8411	0.1354E-01
0.2505E-01	0.3893E-01						0.8916	0.2382E-01
0.2649E-01	0.2999E-01						0.9495	0.370E-01
0.2797E-01	0.2309E-01						0.9841	0.5123E-01
0.2945E-01	0.1771E-01						0.8404	0.7103E-01
0.3093E-01	0.1506E-01						0.8917	0.8622E-01
0.3240E-01	0.1399E-01						0.9044	0.1042E-01
0.3388E-01	0.1366E-01						0.9414	0.300E-01
0.3536E-01							0.9739	0.5390E-01

AGARDGRAPH 315 DATA TAPE FILE: MEKCC2D007.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL. X = 0.2540E-01 )

X = 0.2540E-01  
Z = -0.1270E-01  
U ref hu = 2.790  
RHO ref hu = 561.6  
TAU wall prestion = 0.8919  
P wall mean flow = 0.6597E-05

Y	<(rho)u> RHO	U	Uref	M	Pa	Pwall	<u> <sup>2</sup> U <sup>2</sup>	RHO<u> <sup>2</sup> RHOref Uref <sup>2</sup>
0.2000E-02	0.3189	0.4281	0.7680	0.9854	0.7094E-01	0.1354E-01	0.4633	0.1015E-01
0.3242E-02	0.3242	0.4797	0.8770	0.9403	0.6423E-01	0.1406E-01	0.5072	0.1408E-01
0.4692E-02	0.3413	0.5582	1.053	0.8444	0.4844E-01	0.2355E-01	0.5559	0.2003E-01
0.6152E-02	0.3718	0.6305	1.231	0.9349	0.5912E-01	0.2757E-01	0.6051	0.4921E-01
0.7592E-02	0.3573	0.6919	1.397	0.9126	0.4444E-01	0.2349E-01	0.6725	0.2487E-01
0.9022E-02	0.3360	0.7444	1.554	0.9285	0.3344E-01	0.2309E-01	0.7144	0.4306E-01
0.1045E-01	0.2949	0.7838	1.681	0.9217	0.2128E-01	0.1824E-01	0.7688	0.2671E-01
0.1193E-01	0.2307	0.8302	1.847	0.9143	0.1053E-01	0.1000E-01	0.7998	0.2408E-01
0.1338E-01	0.1955	0.8582	1.961	0.8982	0.6359E-02	0.1437E-02	0.8395	0.1743E-01
0.1486E-01	0.1565	0.8668	2.083	0.8623	0.3609E-02	0.737E-02	0.8910	0.952E-02
0.1631E-01	0.1314	0.8608	2.186	0.8048	0.2239E-02	0.4431E-02	0.8910	0.740E-02
0.1779E-01	0.1194	0.9049	2.180	0.6962	0.4895E-02	0.317E-02	0.9032	0.3305E-02
0.1924E-01	0.1170	0.9134	2.230	0.5471	0.3849E-02	0.3437E-02	0.9402	0.4090E-03
0.2072E-01	0.4950E-01						0.9749	0.1358E-03
0.2218E-01	0.2916E-01							0.3541E-04
0.2366E-01	0.2397E-01							
0.2512E-01	0.1579E-01							
0.2641E-01	0.1625E-01							
0.2807E-01	0.1347E-01							
0.2953E-01	0.1354E-01							
0.3102E-01	0.1623E-01							

AGARDGRAPH 315 DATA TAPE FILE: MEKCC2D008.DAT

PROFILE TABULATION - NORMAL WIRE SURVEY IS NORMAL TO THE RAMP SURFACE  
( MEAN FLOW SURVEY NORMAL TO RAMP SURFACE-20 DEG OFF VERTICAL. X = 0.4128E-01 )

X = 0.4128E-01  
Z = -0.1270E-01  
U ref hu = 2.790  
RHO ref hu = 561.6  
TAU wall prestion = 0.8919  
P wall mean flow = 0.6597E-05



0.1964E-01	0.1682	0.8695	2.027	0.8933	0.4469E-02	0.7361E-02
0.2124E-01	0.1151	0.8816	2.082	0.8715	0.1953E-02	0.3312E-02
0.2281E-01	0.5938E-01	0.8865	2.120	0.8462	0.4963E-03	0.8467E-03
0.2436E-01	0.5124E-01	0.8926	2.160	0.8106	0.3517E-03	0.5967E-03
0.2589E-01	0.3805E-01	0.9013	2.207	0.7700	0.1829E-03	0.3079E-03
0.2739E-01	0.4258E-01	0.9143	2.279	0.7181	0.2098E-03	0.3512E-03
0.2890E-01	0.5146E-01	0.9348	2.388	0.6481	0.2688E-03	0.4458E-03
0.3056E-01	0.5687E-01	0.9725	2.618	0.5338	0.2508E-03	0.4116E-03
0.3222E-01	0.1165	1.043	3.147	0.3575	0.5893E-03	0.9364E-03
0.3381E-01	0.2456	1.070	3.420	0.2650	0.1985E-02	0.2762E-02
0.3541E-01	0.1085					
0.3702E-01	0.3739E-01					
0.3860E-01	0.1253E-01					
0.4017E-01	0.1229E-01					

AGARDOGRAPH 315 DATA TAPE FILE: NEWCC24N03.DAT

PROFILE TABULATION - NORMAL WIRE IS 24 DEGREES OFF VERTICAL, NORMAL TO RAMP SURFACE  
( MEAN SURVEY 24 DEGREES OFF VERTICAL-NORMAL TO RAMP SURFACE. X = 0.1016 )

X = 0.9144E-01  
Z = 0.0000E+00  
M ref = 2.840  
U ref hw = 575.8  
RHO ref hw = 0.8028  
TAU wall preston =  
P wall mean flow = 0.9264E+05

Y	<(rho u)"> RHO U	U Uref	M	Ps Pwall	<u"***2 U**2	RHO<u"***2 RHOref Uref**2
0.2286E-02	0.2087	0.5591	1.070	1.023	0.2243E-01	0.1279E-01
0.3693E-02	0.2339	0.5857	1.134	1.013	0.2620E-01	0.1663E-01
0.5226E-02	0.2640	0.6110	1.200	1.004	0.3094E-01	0.2179E-01
0.6828E-02	0.3080	0.6521	1.309	0.9945	0.3697E-01	0.3069E-01
0.8429E-02	0.3414	0.6919	1.419	0.9875	0.3967E-01	0.3844E-01
0.1004E-01	0.3504	0.7306	1.535	0.9827	0.3614E-01	0.4079E-01
0.1168E-01	0.3542	0.7690	1.659	0.9779	0.3159E-01	0.4141E-01
0.1326E-01	0.3522	0.7962	1.750	0.9749	0.2782E-01	0.4046E-01
0.1487E-01	0.3402	0.8162	1.827	0.9749	0.2354E-01	0.3731E-01
0.1646E-01	0.2826	0.8290	1.880	0.9749	0.1519E-01	0.2549E-01
0.1807E-01	0.2462	0.8354	1.910	0.9749	0.1109E-01	0.1922E-01
0.1969E-01	0.1806	0.8378	1.923	0.9749	0.5874E-02	0.1031E-01
0.2129E-01	0.1437	0.8343	1.917	0.9735	0.3746E-02	0.6529E-02
0.2284E-01	0.9043E-01	0.8342	1.917	0.9705	0.1483E-02	0.2578E-02
0.2448E-01	0.5891E-01	0.8332	1.917	0.9641	0.6295E-03	0.1087E-02
0.2603E-01	0.4230E-01	0.8333	1.917	0.9583	0.3246E-03	0.5569E-03
0.2757E-01	0.3627E-01	0.8311	1.910	0.9540	0.2408E-03	0.4083E-03
0.2911E-01	0.3023E-01	0.8303	1.907	0.9543	0.1678E-03	0.2838E-03
0.3077E-01	0.3050E-01	0.8266	1.894	0.9551	0.1737E-03	0.2900E-03
0.3240E-01	0.3411E-01	0.8241	1.882	0.9478	0.2206E-03	0.3609E-03
0.3397E-01	0.3309E-01	0.8205	1.868	0.9368	0.2114E-03	0.3366E-03
0.3556E-01	0.3444E-01	0.8258	1.889	0.9131	0.2229E-03	0.3539E-03
0.3717E-01	0.4533E-01	0.8406	1.946	0.8568	0.3592E-03	0.5680E-03
0.3874E-01	0.2814	0.8845	2.141	0.6990	0.1086E-01	0.1695E-01
0.4033E-01	0.1084	0.8890	2.163	0.4266	0.1567E-02	0.1524E-02
0.4192E-01	0.2783E-01	0.9304	2.382	0.3091	0.7917E-04	0.6766E-04
0.4351E-01	0.1008E-01	0.9797	2.682	0.2471	0.7319E-05	0.6339E-05
0.4512E-01	0.9567E-02					

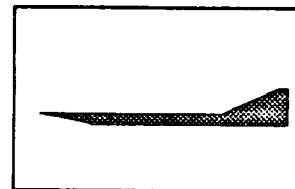
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Ref.: 103, private communication  
Author: Zheltovodov, A. A., *et al*  
Geometry: 2-D Compression Corner

Mach number: 3

Data:  $p_{wall}$ ,  $c_h$ , mean and fluctuating flowfield surveys (pitot and hot-wire anemometry)

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Zheltovodov, A.A., Zaylichny, E.G., Trofimov, V.M. and Yakovlev, V.N., "Investigation of Heat Transfer and Turbulence in Supersonic Separation," *ITPM Preprint* 22-87, 1987.

Zheltovodov, A.A., Trofimov, V.M., Shilein, E.H. and Yakovlev, V.N., "An Experimental Documentation of Supersonic Turbulent Flows in the Vicinity of Forward- and Backward-Facing Ramps," *Unpublished Report* of the Inst. of Theoretical and Applied Mechanics, Siberian Division, USSR Academy of Sciences, April 1990.

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The data consist of surface measurements and mean-flow and fluctuation profiles for a variety of wedge-compression and wedge-expansion corners. While the latter are interesting and useful data not previously available in the West, only the former match our present interest in shock/boundary-layer interactions. The tabulated data are clearly the result of an extensive, careful, detailed research program.

Compression corners of 8, 25, 45, and 90 degrees angle were tested at about Mach 2, 3, and 4. Of these test conditions, however, not all mean, turbulence, and heat transfer measurements were made for each condition represented. Only the 25 degree compression corner (designated FFS25) at Mach 3 includes all three types of data. For purposes of turbulence modeling this case is thus of most interest, since it provides far more detailed data than are available for the 24 degree ramp of Smits *et al*. The 25 degree case is the only one tabulated here.

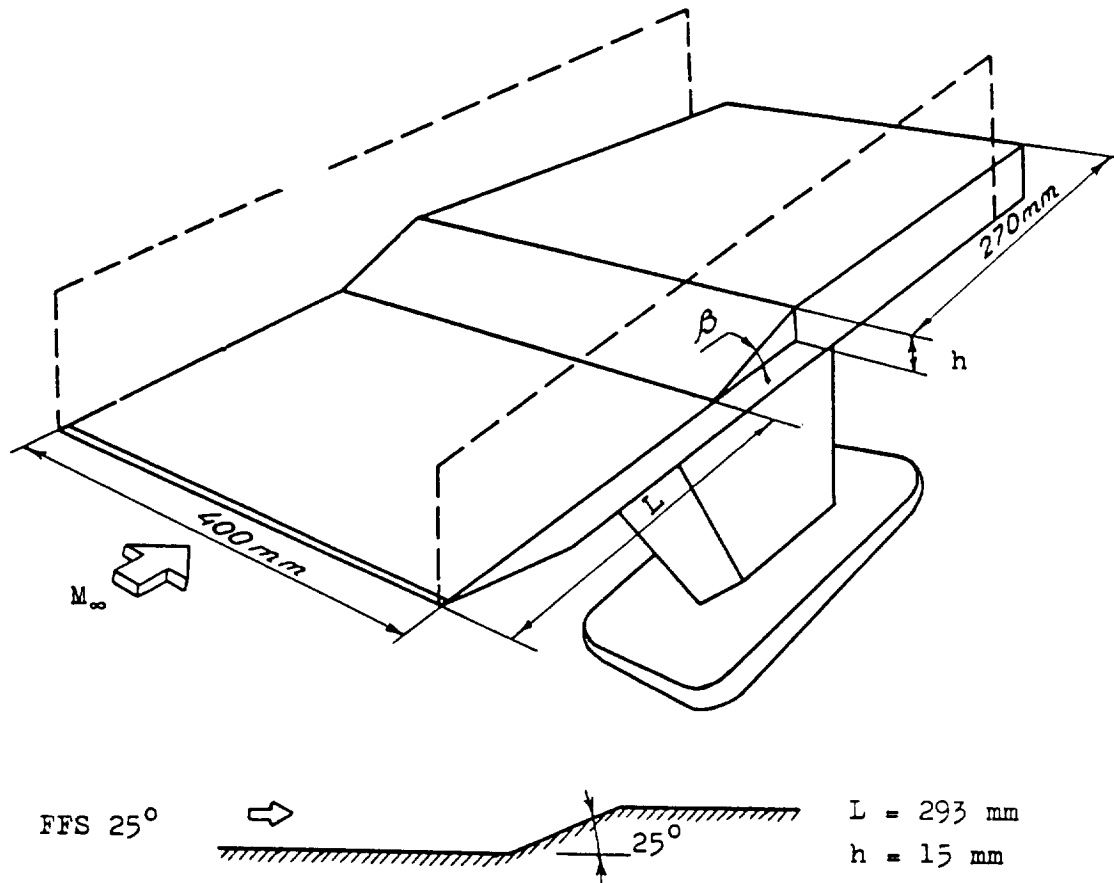
The data were actually taken in three different wind tunnel facilities. Thus, three sets of incoming conditions and boundary-layer profiles are given in the tabulated data. Note that, in the case of the hot-wire measurements, the boundary-layer on the flat plate was tripped near its leading edge. In all cases the cited x-coordinates are measured from an origin at the compression corner along the surface of the test model, whether horizontal or sloped. Similarly, y-coordinates have their origin at the model surface and are vertical in all cases (see model drawings reproduced below).

Confidence limits cited by the experimenters for these data are as follows. The mean-flow profiles have a maximum error of  $\pm 30\%$  in the vicinity of shock waves, but better than this elsewhere. The constant-current hot-wire-anemometer turbulence data are subject to several possible errors and are not meant to give the impression of high accuracy, but are given confidence limits of  $\pm 35\%$  to a maximum of  $\pm 50\%$  in regions of peak fluctuation intensity. The heat transfer data are repeatable to  $\pm 5\%$  and are believed accurate to  $\pm 15\%$  near flow reattachment and  $\pm 10\%$  elsewhere.

Since the background material on this experiment is unavailable to the research community in the West we have attempted to provide sufficiently thorough coverage of the 25 degree ramp case to make such material unnecessary. Nonetheless it is hoped that the

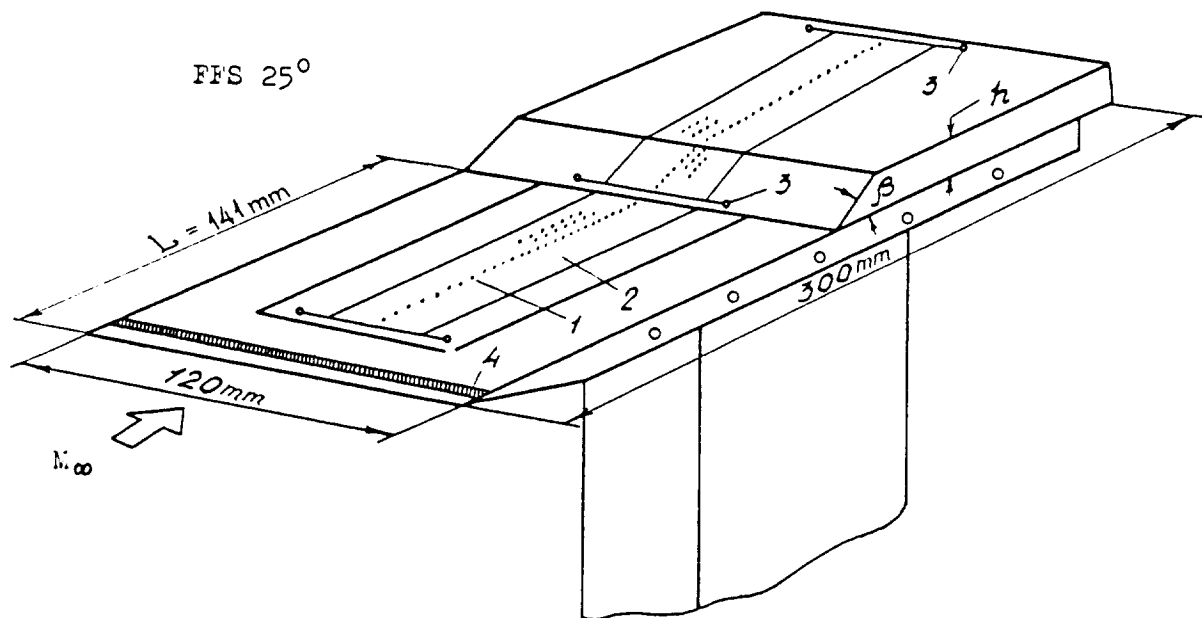
unpublished report listed above, which is written in English and is nearly 300 pages long, will eventually be made formally available. The content of this report is far broader than our present scope allows for inclusion in this Report. In the interim, informal copies of some or all of this material may be obtained through direct contact with personnel of the Penn State Gas Dynamics Laboratory.

The nomenclature of the tabulated data should be self-explanatory for the most part. The designation "inf" refers to wind tunnel freestream conditions, "1" refers to local incoming conditions ahead of the interaction, and "e" refers to conditions at the boundary-layer edge. The term "P0" is used in some tables to represent pitot pressure. Bracketed terms such as  $\langle u \rangle$  and  $\langle m \rangle$  represent rms values of velocity and mass-flux fluctuations. Heat transfer coefficients are denoted by  $\alpha$ , and  $T_w$  and  $T_{rw}$  denote actual wall temperature and adiabatic wall temperature, respectively.

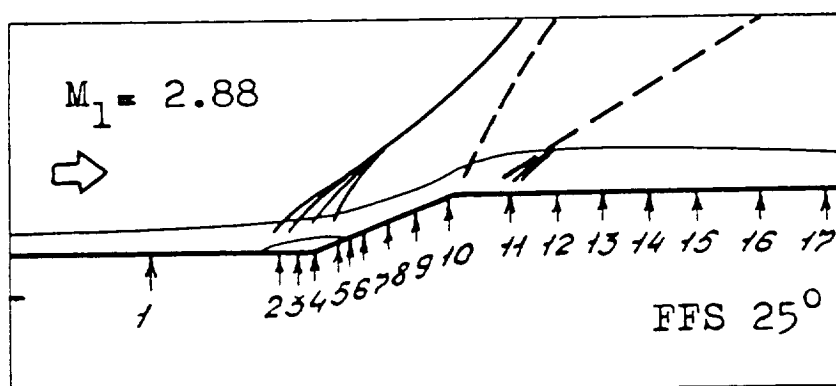
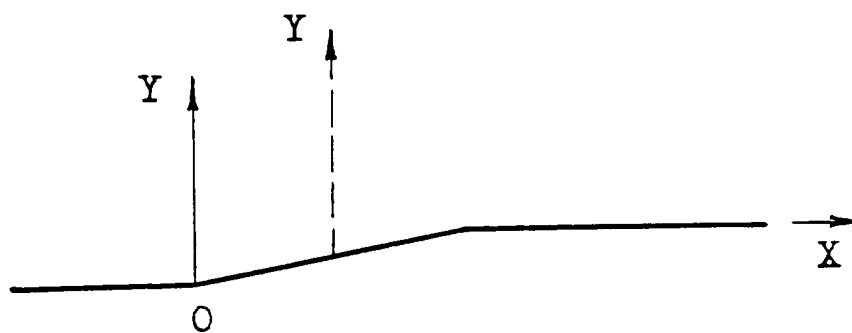


Test Geometry for Mean-Flow Measurements

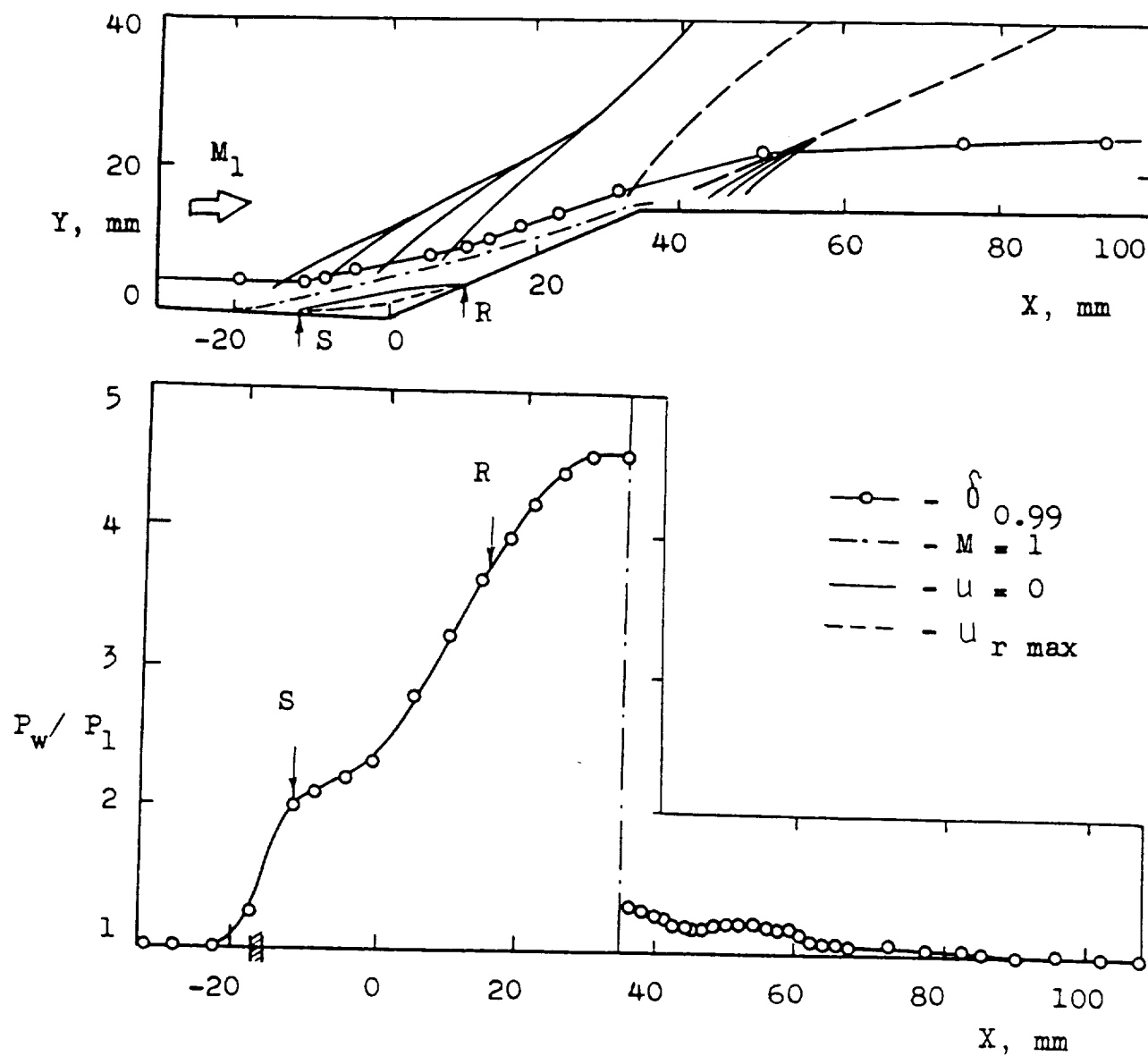




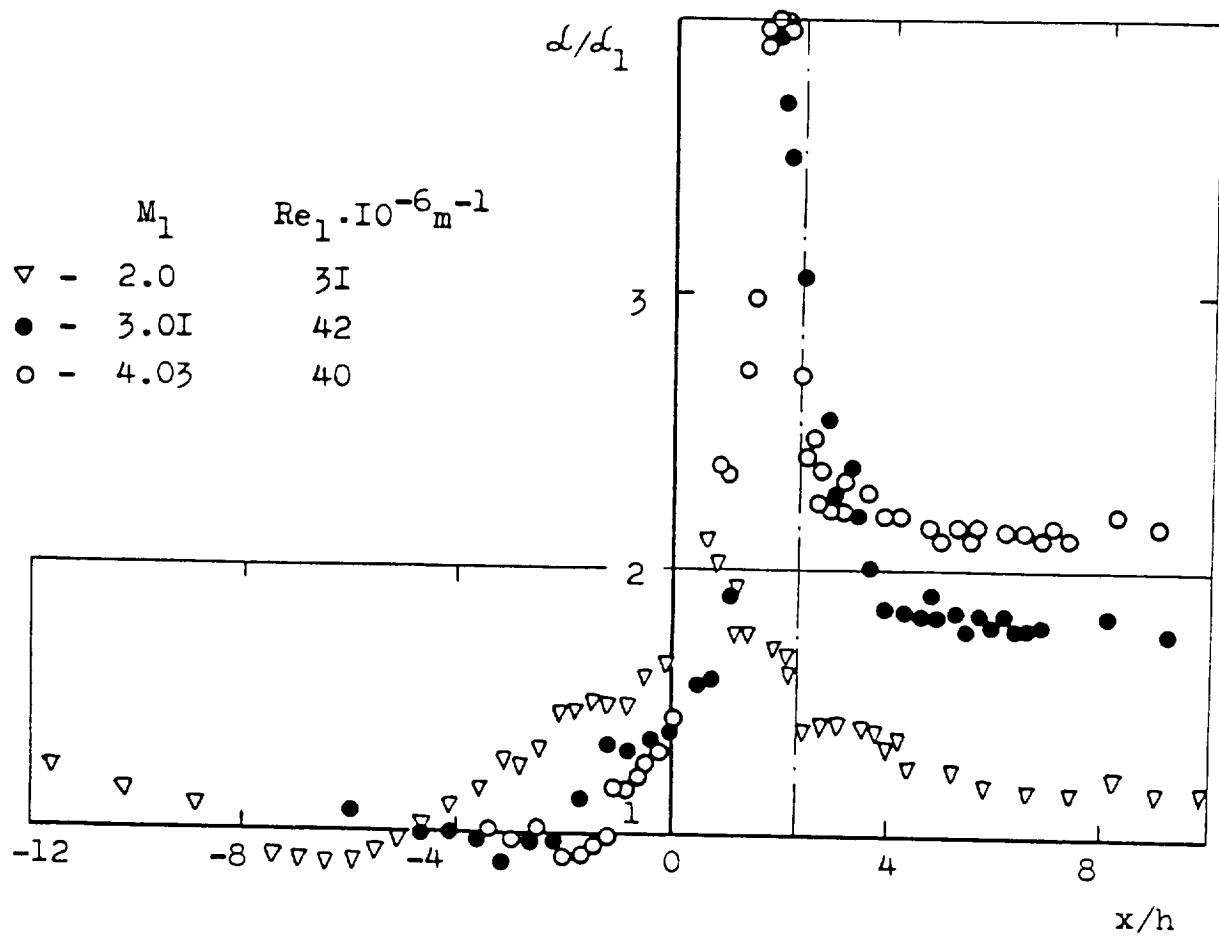
Test Geometry for Heat Transfer Measurements



Coordinate Scheme and Survey Locations



Flow field schematic, FFS 25°,  $M_1 = 2.88$



Heat Transfer Distributions for FFS25 Model at 3 Mach Numbers

\*\*\*\*\*1. Mean Flow Data\*\*\*\*\*  
 \*\*\*\*\*Nominal Freestream Conditions in Wind Tunnel T 315 for FFS25 Test\*\*\*\*\*  
 MInf = 3.01 +/- 0.003  
 PInf = 0.112 +/- 0.001 kg/cm<sup>2</sup>  
 Pstagnation = 4.223 +/- 0.15 kg/cm<sup>2</sup>  
 Re/m = (32.4 +/- 1.6)E+06  
 Istagnation = 294 +/- 12 K  
 \*\*\*\*\*Incoming Flow Parameters for FFS25 Test\*\*\*\*\*  
 M1 = 2.80  
 P1 = 0.122 kg/cm<sup>2</sup>  
 Delta = 4.10 mm  
 Delta\* = 1.43 mm  
 Theta = 0.30 mm  
 CFl = 1.47E-03  
 \*\*\*\*\*Locations of Survey Coordinates for FFS25 Test\*\*\*\*\*

Section #	x [cm]
1	-3.30
2	-1.20
3	-0.80
4	-0.50
5	0.50
6	1.10
7	1.25
8	1.70
9	2.35
10	3.10
11	5.00
12	6.25
13	7.85
14	9.85
15	11.70
16	13.60
17	15.05

\*\*\*\*\*SURFACE PRESSURES FFS25\*\*\*\*\*  
 M1 = 2.80, P1 = 0.1224 kg/cm<sup>2</sup>

X[cm]	P/P1	X[cm]	P/P1
-7.85	0.925	4.10	1.287
-7.05	0.959	4.20	1.265
-6.65	0.972	4.30	1.255
-6.25	0.982	4.40	1.266
-5.85	0.988	4.50	1.223
-5.45	0.989	4.60	1.225
-5.05	1.022	4.80	1.225
-4.65	0.997	5.00	1.204
-4.25	1.007	5.20	1.216
-3.85	0.998	5.40	1.181
-3.45	1.222	5.60	1.173
-3.05	2.077	5.80	1.150
-2.65	2.171	6.00	1.145
-2.25	2.304	6.20	1.119
-1.85	2.806	6.40	1.097
-1.45	3.205	6.60	1.094
-1.05	3.595	6.80	1.043
-0.65	3.895	7.00	1.105
-0.25	4.162	7.20	1.007
0.15	4.381	7.40	1.013
0.55	4.500	7.60	1.019
0.95	4.520	7.80	1.022
1.35	4.520	8.00	1.006
1.75	4.520	8.20	1.003
2.15	4.520	8.40	1.003
2.55	4.520	8.60	1.015
2.95	4.520	8.80	1.015
3.35	4.520	9.00	0.974

\*\*\*\*\*Static Pressure Profiles, FFS25\*\*\*\*\*  
 M1 = 2.80, P1 = 0.1224 kg/cm<sup>2</sup>

SECTION 2	SECTION 4	SECTION 6
Y [cm] P/P1	Y [cm] P/P1	Y [cm] P/P1
0.043 1.964	0.150 2.208	0.150 3.052
0.063 1.960	0.190 2.221	0.180 3.072
0.093 2.010	0.230 2.224	0.220 3.056
0.113 1.989	0.270 2.223	0.260 3.044
0.133 1.985	0.310 2.220	0.300 3.023
0.153 1.958	0.350 2.221	0.340 3.030
0.163 1.946	0.390 2.212	0.360 3.031
0.173 1.942	0.430 2.215	0.420 3.025
0.193 1.873	0.470 2.199	0.460 3.025
0.213 1.794	0.510 2.182	0.500 3.024
0.243 1.721	0.550 2.156	0.540 2.987
0.273 1.671	0.590 2.097	0.580 2.964
0.303 1.619	0.630 2.047	0.620 2.895
0.333 1.587	0.670 2.016	0.660 2.835
0.383 1.555	0.710 1.997	0.700 2.766
0.393 1.520	0.750 1.977	0.740 2.640
0.423 1.471	0.790 1.977	0.780 2.538
0.453 1.415	0.830 1.916	0.820 2.467
0.483 1.361	0.870 1.842	0.860 2.387
0.513 1.296	0.910 1.736	0.910 2.301
0.553 1.254	0.950 1.645	
0.593 1.190	0.990 1.463	
	1.030 1.310	
	1.070 1.149	
	1.110 0.938	

SECTION 8	SECTION 11
Y [cm] P/P1	Y [cm] P/P1
0.043 3.965	0.043 1.215
0.093 3.887	0.093 1.200
0.143 3.828	0.113 1.193
0.193 3.810	0.143 1.194
0.243 3.768	0.193 1.185
0.293 3.793	0.243 1.179
0.343 3.780	0.293 1.174
0.393 3.773	0.343 1.198
0.443 3.733	0.393 1.205
0.493 3.676	0.443 1.182
0.543 3.612	0.503 1.158
0.593 3.534	0.563 1.141
0.633 3.447	0.623 1.130
	0.683 1.154
	0.743 1.201
	0.803 1.266
	0.863 1.344
	0.943 1.470
	1.023 1.065
	1.103 1.744
	1.183 1.867
	1.263 1.977

\*\*\*\*\*Mean Flowfield Profiles FFS25\*\*\*\*\*  
 FFS25, M1 = 2.80, SECTION 1

X = -3.30 cm, Me = 2.89, Pu = 0.122 kg/cm<sup>2</sup>,  
 Ue = 618.1 m/s, Te = 116.8 K, RHOe = 0.0376 kg s<sup>3</sup>/m

Y	P0	P	M	1/Te	U/Ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]				
0.00	0.122	0.122	0.00	2.474	0.000	0.404
0.01	0.221	0.122	0.96	2.126	0.486	0.470

0.03	0.354	0.122	1.35	1.866	0.642	0.516
0.05	0.448	0.123	1.60	1.698	0.725	0.589
0.06	0.499	0.123	1.66	1.658	0.744	0.603
0.07	0.530	0.122	1.72	1.618	0.762	0.618
0.08	0.554	0.123	1.77	1.591	0.774	0.628
0.11	0.619	0.123	1.88	1.520	0.805	0.658
0.14	0.672	0.123	1.96	1.466	0.827	0.682
0.17	0.743	0.123	2.08	1.399	0.854	0.715
0.20	0.816	0.123	2.18	1.338	0.878	0.747
0.23	0.894	0.123	2.29	1.278	0.901	0.782
0.26	0.973	0.123	2.40	1.220	0.922	0.819
0.29	1.051	0.123	2.50	1.171	0.940	0.854
0.33	1.141	0.123	2.63	1.107	0.963	0.904
0.37	1.232	0.124	2.74	1.059	0.980	0.944
0.41	1.315	0.124	2.80	1.030	0.990	0.970
0.45	1.346	0.124	2.85	1.013	0.996	0.987
0.49	1.360	0.124	2.86	1.008	0.997	0.992
0.52	1.367	0.124	2.86	1.005	0.998	0.995
0.55	1.360	0.124	2.86	1.009	0.997	0.991
0.58	1.372	0.124	2.87	1.004	0.999	0.996
0.61	1.377	0.124	2.87	1.002	0.999	0.998
0.64	1.382	0.124	2.88	1.000	1.000	1.000

FF525\*, M1 = 2.88, SECTION 2

X = -1.20 cm, Me = 2.35, Pu = 0.247 kg/cm<sup>2</sup>,  
Ue = 555.2 m/s, Te = 139.1 K, RHOe = 0.0456 kg s<sup>2</sup>/m

Y	P0	P	M	1/Te	U/ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]			
0.00	0.247	0.247	0.00	1.615	0.000	0.589
0.01	0.257	0.247	0.24	1.681	0.158	0.595
0.02	0.267	0.247	0.00	-	0.000	-
0.03	0.252	0.247	0.16	1.972	0.097	0.507
0.05	0.257	0.247	0.24	1.961	0.145	0.510
0.07	0.283	0.247	0.45	1.913	0.264	0.523
0.09	0.316	0.247	0.60	1.863	0.347	0.537
0.11	0.306	0.247	0.85	1.756	0.479	0.569
0.13	0.428	0.247	0.92	1.722	0.515	0.581
0.14	0.538	0.218	1.22	1.566	0.450	0.638
0.19	0.721	0.212	1.50	1.417	0.759	0.708
0.22	0.828	0.220	1.59	1.367	0.791	0.732
0.25	0.954	0.211	1.77	1.273	0.849	0.785
0.28	1.039	0.204	1.89	1.212	0.866	0.825
0.31	1.163	0.199	2.04	1.139	0.927	0.878
0.34	1.220	0.195	2.12	1.102	0.946	0.907
0.37	1.270	0.192	2.18	1.073	0.962	0.932
0.40	1.360	0.188	2.31	1.017	0.991	0.983
0.43	1.380	0.182	2.35	1.000	1.000	1.000

FF525\*, M1 = 2.88, SECTION 3

X = -0.60 cm, Me = 2.24, Pu = 0.266 kg/cm<sup>2</sup>,  
Ue = 543.0 m/s, Te = 146.9 K, RHOe = 0.0631 kg s<sup>2</sup>/m

Y	P0	P	M	1/Te	U/ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]			
0.00	0.266	0.266	0.00	1.783	0.000	0.589
0.01	0.275	0.266	0.22	1.871	0.135	0.534
0.02	0.278	0.266	0.25	1.867	0.153	0.536
0.03	0.278	0.266	0.25	1.866	0.153	0.536
0.04	0.280	0.266	0.27	1.862	0.167	0.537
0.06	0.281	0.266	0.28	1.860	0.173	0.537
0.08	0.279	0.266	0.26	1.865	0.158	0.536
0.12	0.266	0.266	0.00	-	0.000	-
0.15	0.303	0.266	0.44	1.827	0.264	0.547
0.18	0.357	0.266	0.66	1.752	0.393	0.571
0.21	0.396	0.266	0.78	1.706	0.454	0.586
0.24	0.458	0.266	0.92	1.643	0.526	0.609
0.27	0.588	0.266	1.13	1.539	0.627	0.650

0.30	0.708	0.266	1.28	1.461	0.694	0.685
0.33	0.779	0.266	1.36	1.410	0.728	0.705
0.36	0.942	0.266	1.53	1.311	0.792	0.751
0.39	1.088	0.266	1.67	1.261	0.839	0.792
0.42	1.270	0.266	1.82	1.187	0.889	0.843
0.45	1.426	0.266	1.94	1.129	0.925	0.866
0.48	1.595	0.266	2.07	1.072	0.958	0.933
0.51	1.716	0.266	2.15	1.035	0.980	0.966
0.54	1.816	0.266	2.22	1.006	0.996	0.994
0.57	1.839	0.266	2.24	1.000	1.000	1.000

FF525\*, M1 = 2.88, SECTION 4

X = -0.50 cm, Me = 2.25, Pu = 0.277 kg/cm<sup>2</sup>,  
Ue = 545.9 m/s, Te = 146.1 K, RHOe = 0.0661 kg s<sup>2</sup>/m

Y	P0	P	M	1/Te	U/ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]			
0.00	0.277	0.277	0.00	1.796	0.000	0.526
0.01	0.291	0.277	0.27	1.877	0.164	0.533
0.03	0.293	0.277	0.29	1.874	0.175	0.534
0.05	0.300	0.277	0.34	1.864	0.204	0.536
0.07	0.298	0.277	0.32	1.867	0.196	0.536
0.09	0.295	0.277	0.30	1.872	0.181	0.534
0.11	0.277	0.277	0.00	-	0.000	-
0.23	0.339	0.277	0.54	1.808	0.325	0.553
0.27	0.399	0.277	0.74	1.734	0.434	0.577
0.31	0.437	0.277	0.84	1.694	0.482	0.590
0.36	0.606	0.277	1.12	1.556	0.621	0.643
0.41	0.750	0.277	1.30	1.464	0.697	0.683
0.45	0.999	0.277	1.55	1.334	0.794	0.750
0.49	1.190	0.277	1.72	1.249	0.851	0.801
0.52	1.324	0.277	1.82	1.196	0.885	0.836
0.55	1.458	0.277	1.92	1.147	0.915	0.872
0.58	1.647	0.277	2.06	1.085	0.952	0.922
0.61	1.773	0.277	2.14	1.047	0.974	0.955
0.64	1.878	0.277	2.21	1.018	0.990	0.982
0.67	1.924	0.277	2.24	1.003	0.997	0.995
0.70	1.945	0.277	2.25	1.000	1.000	1.000
0.73	1.945	0.277	2.25	1.000	1.000	1.000

FF525\*, M1 = 2.88, SECTION 5

X = 0.50 cm, Me = 2.00, Pu = 0.364 kg/cm<sup>2</sup>,  
Ue = 512.1 m/s, Te = 163.0 K, RHOe = 0.0737 kg s<sup>2</sup>/m

Y	P0	P	M	1/Te	U/ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]			
0.00	0.364	0.364	0.00	1.587	0.000	0.601
0.01	0.383	0.364	0.27	1.645	0.177	0.608
0.03	0.390	0.364	0.31	1.637	0.207	0.611
0.05	0.393	0.364	0.33	1.634	0.218	0.612
0.07	0.397	0.364	0.35	1.630	0.232	0.614
0.08	0.395	0.364	0.34	1.632	0.225	0.613
0.09	0.396	0.364	0.34	1.631	0.229	0.613
0.10	0.394	0.364	0.34	1.633	0.223	0.613
0.12	0.391	0.364	0.32	1.636	0.210	0.611
0.15	0.379	0.364	0.24	1.649	0.160	0.607
0.18	0.364	0.364	0.00	-	0.000	-
0.26	0.422	0.340	0.46	1.620	0.367	0.620
0.29	0.449	0.340	0.57	1.596	0.415	0.638
0.32	0.507	0.339	0.71	1.547	0.495	0.657
0.35	0.658	0.340	0.82	1.509	0.545	0.681
0.38	0.637	0.342	0.94	1.459	0.608	0.707
0.42	0.755	0.339	1.07	1.397	0.682	0.749
0.46	0.909	0.341	1.23	1.327	0.753	0.787
0.50	1.070	0.341	1.37	1.264	0.814	0.857
0.54	1.349	0.342	1.58	1.169	0.897	0.929
0.58	1.649	0.343	1.78	1.083	0.966	0.947
0.61	1.727	0.343	1.83	1.062	0.982	0.992

0.64 1.841 0.344 1.95 1.010 0.999 0.999  
0.67 1.941 0.345 2.00 1.000 1.000 1.000

# FFS25\*, M1 = 2.88, SECTION 6

X = 1.10 cm, Me = 1.93, Pu = 0.407 kg/cm²,  
Ue = 498.0 m/s, Te = 166.0 K, RHOe = 0.0757 kg s' /m

Y	PO	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]				
0.00	0.407	0.407	0.00	1.662	0.000	0.602
0.07	0.415	0.407	0.16	1.654	0.110	0.605
0.09	0.423	0.390	0.34	1.628	0.227	0.614
0.11	0.427	0.388	0.37	1.621	0.247	0.617
0.13	0.430	0.388	0.39	1.618	0.256	0.618
0.16	0.432	0.387	0.48	1.597	0.312	0.626
0.19	0.435	0.385	0.68	1.536	0.437	0.651
0.22	0.439	0.384	0.79	1.496	0.500	0.668
0.28	0.690	0.381	0.96	1.427	0.596	0.701
0.32	0.710	0.379	0.99	1.414	0.612	0.707
0.36	0.822	0.376	1.12	1.358	0.677	0.736
0.40	1.042	0.374	1.32	1.267	0.772	0.789
0.44	1.298	0.372	1.52	1.177	0.856	0.850
0.48	1.616	0.370	1.73	1.082	0.936	0.924
0.52	1.861	0.367	1.88	1.018	0.986	0.982
0.56	1.904	0.361	1.93	1.000	1.000	1.000

# FFS25\*, M1 = 2.88, SECTION 7

X = 1.25 cm, Me = 1.78, Pu = 0.448 kg/cm²,  
Ue = 477.0 m/s, Te = 178.0 K, RHOe = 0.0816 kg s' /m

Y	PO	P	M	T/Te	U/Ue	RHO/RHOe
(cm)	(g/cm <sup>2</sup> )	(g/cm <sup>2</sup> )				
0.00	0.448	0.448	0.00	1.566	0.000	0.638
0.01	0.466	0.448	0.24	1.550	0.169	0.645
0.03	0.502	0.446	0.42	1.519	0.288	0.658
0.04	0.494	0.446	0.38	1.526	0.268	0.655
0.06	0.524	0.444	0.49	1.501	0.339	0.666
0.08	0.532	0.442	0.52	1.494	0.356	0.669
0.10	0.569	0.441	0.61	1.468	0.417	0.681
0.12	0.580	0.441	0.64	1.460	0.432	0.685
0.16	0.577	0.439	0.64	1.461	0.431	0.684
0.20	0.626	0.437	0.74	1.428	0.494	0.700
0.24	0.760	0.434	0.93	1.357	0.608	0.737
0.27	0.874	0.432	1.06	1.307	0.677	0.765
0.30	0.952	0.430	1.13	1.276	0.716	0.784
0.33	1.069	0.428	1.23	1.234	0.766	0.811
0.36	1.236	0.426	1.36	1.179	0.826	0.848
0.39	1.399	0.423	1.47	1.131	0.877	0.884
0.42	1.527	0.423	1.55	1.097	0.911	0.912
0.45	1.743	0.419	1.68	1.040	0.964	0.961
0.48	1.914	0.417	1.78	1.000	1.000	1.000

# FFS25\*, M1 = 2.88, SECTION 8

X = 1.70 cm, Me = 1.63, Pu = 0.483 kg/cm²,  
Ue = 446.5 m/s, Te = 187.9 K, RHOe = 0.0893 kg s' /m

Y	PO	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]				
0.00	0.483	0.483	0.00	1.470	0.000	0.680
0.01	0.616	0.483	0.64	1.370	0.462	0.730
0.03	0.664	0.480	0.70	1.354	0.498	0.739
0.04	0.694	0.485	0.73	1.341	0.523	0.746
0.05	0.714	0.483	0.77	1.330	0.545	0.752
0.07	0.742	0.484	0.81	1.318	0.570	0.759
0.09	0.747	0.481	0.82	1.314	0.577	0.761

0.11	0.806	0.484	0.88	1.290	0.619	0.775
0.14	0.821	0.483	0.90	1.283	0.630	0.779
0.17	0.873	0.485	0.96	1.264	0.662	0.791
0.20	0.954	0.485	1.03	1.235	0.704	0.809
0.23	1.010	0.486	1.08	1.218	0.732	0.821
0.26	1.099	0.484	1.15	1.190	0.772	0.841
0.30	1.221	0.485	1.24	1.155	0.819	0.866
0.34	1.507	0.486	1.41	1.084	0.908	0.922
0.38	1.608	0.487	1.47	1.061	0.932	0.942
0.42	1.826	0.486	1.59	1.015	0.984	0.986
0.46	1.882	0.485	1.62	1.003	0.997	0.997
0.50	1.882	0.487	1.62	1.004	0.996	0.996
0.54	1.881	0.486	1.62	1.004	0.996	0.996
0.58	1.882	0.486	1.62	1.003	0.997	0.997
0.62	1.881	0.481	1.62	1.000	1.000	1.000

# FFS25\*, M1 = 2.88, SECTION 9

X = 2.35 cm, Me = 1.56, Pu = 0.518 kg/cm²,  
Ue = 436.0 m/s, Te = 194.5 K, RHOe = 0.0935 kg s' /m

Y	PO	P	U/Ue	M	T/Te	U/Ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]					
0.00	0.518	0.518	0.00	1.433	0.000	0.698	
0.01	0.716	0.518	0.70	1.319	0.512	0.758	
0.03	0.811	0.519	0.82	1.278	0.598	0.782	
0.05	0.876	0.519	0.90	1.253	0.645	0.798	
0.06	0.879	0.517	0.90	1.250	0.649	0.800	
0.07	0.869	0.517	0.89	1.254	0.642	0.797	
0.08	0.927	0.518	0.95	1.234	0.678	0.810	
0.09	0.950	0.519	0.97	1.227	0.690	0.815	
0.10	0.976	0.518	1.00	1.218	0.705	0.821	
0.12	0.981	0.517	1.00	1.215	0.709	0.823	
0.14	1.099	0.519	1.09	1.181	0.762	0.847	
0.16	1.093	0.520	1.09	1.183	0.759	0.845	
0.18	1.168	0.520	1.14	1.162	0.790	0.860	
0.20	1.215	0.520	1.18	1.150	0.808	0.870	
0.22	1.288	0.519	1.22	1.131	0.835	0.884	
0.25	1.365	0.520	1.27	1.113	0.860	0.899	
0.28	1.385	0.520	1.28	1.108	0.867	0.903	
0.32	1.770	0.520	1.50	1.024	0.972	0.976	
0.35	1.794	0.519	1.51	1.019	0.978	0.982	
0.38	1.881	0.520	1.55	1.002	0.997	0.998	
0.41	1.901	0.520	1.56	0.999	1.002	1.001	
0.44	1.902	0.520	1.56	0.999	1.002	1.001	
0.47	1.902	0.522	1.56	1.000	1.000	1.000	

# FFS25\*, M1 = 2.88, SECTION 10

X = 3.15 cm, Me = 1.49, Pu = 0.570 kg/cm²,  
Ue = 422.3 m/s, Te = 200.4 K, RHOe = 0.0982 kg s' /m

Y	PO	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	[kg/cm <sup>2</sup> ]	[kg/cm <sup>2</sup> ]				
0.00	0.570	0.570	0.00	1.394	0.000	0.717
0.01	0.684	0.570	0.82	1.246	0.613	0.802
0.04	0.968	0.569	0.90	1.217	0.671	0.822
0.06	1.046	0.569	0.97	1.193	0.715	0.838
0.07	1.081	0.569	1.00	1.182	0.733	0.846
0.08	1.106	0.568	1.02	1.175	0.746	0.851
0.09	1.126	0.570	1.04	1.171	0.753	0.854
0.11	1.172	0.569	1.07	1.158	0.775	0.864
0.13	1.184	0.567	1.08	1.154	0.781	0.867
0.15	1.202	0.566	1.10	1.149	0.789	0.870
0.18	1.248	0.567	1.13	1.137	0.807	0.879
0.21	1.328	0.566	1.18	1.118	0.837	0.894
0.24	1.394	0.566	1.22	1.103	0.860	0.907
0.25	1.521	0.566	1.29	1.075	0.900	0.930
0.31	1.599	0.565	1.34	1.058	0.923	0.945
0.35	1.733	0.566	1.41	1.028	0.964	0.972

0.39 1.858 0.566 1.46 1.109 0.989 0.991  
0.43 1.903 0.565 1.49 1.000 1.000 1.000  
0.48 1.903 0.565 1.49 1.000 1.000 1.000

FFS25\*, M1 = 2.80, SECTION 11

X = 5.00 cm, Me = 2.61, Pu = 0.150 kg/cm²,  
Ue = 519.7 m/s, Te = 122.5 K, RHOe = 0.0429 kg s¹/m

Y	PO	P	U/Ue	RHO/RHOe
[cm]	[kg/cm²]	[kg/cm²]	M	1/Te
0.00	0.150	0.150	0.00	2.215
0.01	0.318	0.150	1.10	1.824
0.03	0.696	0.150	1.79	1.410
0.06	0.817	0.150	1.96	1.316
0.07	0.829	0.151	1.97	1.309
0.08	0.838	0.150	1.99	1.301
0.09	0.854	0.150	2.01	1.289
0.10	0.869	0.150	2.03	1.280
0.11	0.891	0.150	2.06	1.264
0.13	0.912	0.150	2.08	1.251
0.15	0.936	0.151	2.11	1.236
0.18	0.968	0.151	2.15	1.217
0.21	0.969	0.150	2.16	1.211
0.23	1.004	0.150	2.19	1.193
0.29	1.011	0.151	2.20	1.191
0.33	1.053	0.150	2.25	1.164
0.39	1.088	0.151	2.29	1.148
0.45	1.137	0.150	2.35	1.119
0.51	1.160	0.151	2.37	1.109
0.57	1.202	0.150	2.42	1.086
0.63	1.228	0.150	2.45	1.072
0.69	1.246	0.150	2.46	1.064
0.75	1.296	0.151	2.51	1.043
0.77	1.320	0.151	2.54	1.033
0.80	1.338	0.150	2.56	1.024
0.84	1.398	0.151	2.61	1.000

FFS25\*, M1 = 2.80, SECTION 12

X = 6.25 cm, Me = 2.71, Pu = 0.143 kg/cm²,  
Ue = 590.2 m/s, Te = 118.2 K, RHOe = 0.0423 kg s¹/m

Y	PO	P	U/Ue	RHO/RHOe
[cm]	[kg/cm²]	[kg/cm²]	M	1/Te
0.00	0.143	0.143	0.00	2.306
0.01	0.278	0.143	1.02	1.945
0.03	0.319	0.144	1.13	1.877
0.04	0.531	0.144	1.57	1.600
0.05	0.630	0.144	1.74	1.499
0.07	0.721	0.143	1.88	1.417
0.09	0.766	0.144	1.91	1.398
0.11	0.769	0.144	1.94	1.379
0.13	0.789	0.144	1.97	1.364
0.15	0.819	0.144	2.01	1.339
0.19	0.853	0.144	2.06	1.315
0.23	0.875	0.143	2.09	1.298
0.25	0.868	0.143	2.11	1.288
0.30	0.911	0.143	2.16	1.273
0.35	0.941	0.143	2.17	1.252
0.40	0.971	0.143	2.21	1.233
0.46	1.009	0.143	2.26	1.210
0.52	1.051	0.143	2.31	1.184
0.58	1.093	0.143	2.36	1.160
0.64	1.144	0.144	2.41	1.133
0.70	1.198	0.144	2.47	1.105
0.76	1.248	0.143	2.53	1.079
0.82	1.295	0.144	2.58	1.058

0.88 1.350 0.144 2.63 1.032 0.980 0.969  
0.92 1.371 0.144 2.65 1.024 0.991 0.977  
0.96 1.398 0.144 2.68 1.011 0.996 0.989  
1.00 1.425 0.144 2.71 1.000 1.000 1.000

FFS25\*, M1 = 2.80, SECTION 13

X = 7.85 cm, Me = 2.78, Pu = 0.133 kg/cm²,  
Ue = 602.1 m/s, Te = 116.6 K, RHOe = 0.0402 kg s¹/m

Y	PO	P	U/Ue	RHO/RHOe
[cm]	[kg/cm²]	[kg/cm²]	M	1/Te
0.00	0.133	0.133	0.00	2.377
0.01	0.247	0.133	0.98	2.028
0.03	0.422	0.133	1.43	1.740
0.04	0.566	0.134	1.67	1.590
0.05	0.585	0.134	1.74	1.547
0.06	0.634	0.134	1.82	1.497
0.07	0.652	0.134	1.84	1.480
0.09	0.689	0.134	1.90	1.446
0.11	0.710	0.134	1.94	1.426
0.13	0.740	0.134	1.98	1.400
0.15	0.760	0.134	2.01	1.383
0.18	0.784	0.134	2.04	1.364
0.21	0.806	0.134	2.07	1.346
0.25	0.835	0.134	2.12	1.323
0.29	0.860	0.134	2.15	1.304
0.34	0.871	0.134	2.16	1.297
0.39	0.907	0.134	2.21	1.271
0.45	0.937	0.134	2.25	1.250
0.52	0.969	0.134	2.29	1.229
0.59	1.013	0.134	2.34	1.202
0.66	1.046	0.134	2.41	1.169
0.73	1.113	0.134	2.46	1.142
0.80	1.172	0.134	2.53	1.110
0.87	1.222	0.134	2.59	1.084
0.94	1.270	0.134	2.64	1.060
1.00	1.314	0.134	2.69	1.039
1.04	1.333	0.134	2.71	1.030
1.08	1.366	0.134	2.74	1.015
1.11	1.373	0.134	2.75	1.012
1.14	1.376	0.134	2.75	1.012
1.17	1.399	0.134	2.78	1.002
1.20	1.405	0.135	2.78	1.000

FFS25\*, M1 = 2.80, SECTION 14

X = 9.85 cm, Me = 2.80, Pu = 0.127 kg/cm²,  
Ue = 598.2 m/s, Te = 113.7 K, RHOe = 0.0392 kg s¹/m

Y	PO	P	U/Ue	RHO/RHOe
[cm]	[kg/cm²]	[kg/cm²]	M	1/Te
0.00	0.127	0.127	0.00	2.395
0.01	0.240	0.127	1.00	2.034
0.03	0.354	0.126	1.33	1.824
0.04	0.432	0.126	1.50	1.707
0.05	0.513	0.127	1.66	1.607
0.06	0.575	0.126	1.77	1.535
0.07	0.611	0.126	1.84	1.494
0.08	0.631	0.126	1.87	1.475
0.09	0.651	0.126	1.90	1.456
0.11	0.682	0.126	1.95	1.427
0.13	0.707	0.127	1.98	1.408
0.15	0.732	0.126	2.03	1.391
0.18	0.767	0.127	2.08	1.353
0.21	0.785	0.127	2.10	1.340
0.25	0.810	0.127	2.14	1.319
0.29	0.833	0.127	2.18	1.300

0.35	0.860	0.126	2.22	1.278	0.805	0.761
0.41	0.887	0.126	2.25	1.258	0.803	0.795
0.47	0.916	0.127	2.29	1.238	0.811	0.808
0.53	0.949	0.127	2.33	1.216	0.919	0.822
0.59	0.973	0.127	2.36	1.201	0.925	0.832
0.65	1.006	0.127	2.41	1.179	0.934	0.848
0.71	1.035	0.127	2.44	1.162	0.940	0.861
0.78	1.069	0.127	2.51	1.128	0.953	0.886
0.85	1.146	0.127	2.58	1.097	0.964	0.911
0.92	1.197	0.127	2.63	1.072	0.974	0.933
0.96	1.233	0.127	2.68	1.051	0.981	0.951
1.00	1.257	0.127	2.70	1.041	0.985	0.961
1.04	1.287	0.127	2.74	1.025	0.991	0.975
1.07	1.305	0.127	2.76	1.017	0.994	0.984
1.10	1.315	0.127	2.77	1.012	0.996	0.988
1.13	1.334	0.127	2.79	1.002	0.999	0.998
1.16	1.352	0.128	2.80	1.000	1.000	1.000

FFS25°, M1 = 2.88, SECTION 15

X = 11.70 cm, Me = 2.86, Pu = 0.121 kg/cm²,  
 Ue = 608.1 m/s, Te = 112.7 K, RHOe = 0.0379 kg s/m

Y	P0	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	(kg/cm²)	(kg/cm²)				
0.00	0.121	0.121	0.00	2.453	0.000	0.407
0.01	0.229	0.121	1.00	2.085	0.503	0.480
0.03	0.320	0.121	1.28	1.901	0.616	0.526
0.05	0.486	0.121	1.65	1.655	0.741	0.604
0.06	0.500	0.121	1.68	1.635	0.750	0.612
0.07	0.539	0.121	1.75	1.589	0.771	0.629
0.08	0.566	0.122	1.80	1.559	0.784	0.641
0.09	0.591	0.121	1.84	1.529	0.798	0.654
0.11	0.624	0.122	1.90	1.496	0.812	0.669
0.13	0.656	0.122	1.95	1.464	0.825	0.683
0.15	0.690	0.122	2.01	1.429	0.839	0.700
0.17	0.715	0.122	2.05	1.405	0.850	0.712
0.21	0.733	0.122	2.10	1.371	0.863	0.729
0.25	0.785	0.122	2.15	1.344	0.874	0.744
0.31	0.818	0.122	2.20	1.318	0.884	0.759
0.37	0.853	0.122	2.25	1.291	0.894	0.775
0.43	0.881	0.122	2.29	1.268	0.903	0.789
0.50	0.915	0.122	2.34	1.245	0.912	0.803
0.57	0.951	0.122	2.38	1.220	0.921	0.819
0.64	0.987	0.122	2.43	1.196	0.930	0.836
0.71	1.028	0.122	2.48	1.170	0.940	0.854
0.85	1.132	0.122	2.55	1.138	0.951	0.879
0.92	1.183	0.122	2.61	1.107	0.963	0.904
0.98	1.227	0.122	2.68	1.079	0.972	0.927
1.04	1.269	0.123	2.73	1.056	0.981	0.947
1.07	1.290	0.122	2.77	1.037	0.987	0.964
1.10	1.304	0.122	2.80	1.026	0.991	0.975
1.13	1.319	0.122	2.81	1.020	0.993	0.981
1.16	1.329	0.122	2.83	1.012	0.996	0.988
1.19	1.348	0.123	2.84	1.008	0.997	0.992
					1.000	1.000

FFS25°, M1 = 2.88, SECTION 16

X = 13.60 cm, Me = 2.86, Pu = 0.122 kg/cm²,  
 Ue = 609.2 m/s, Te = 113.1 K, RHOe = 0.0383 kg s/m

Y	P0	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	(kg/cm²)	(kg/cm²)				
0.00	0.122	0.122	0.00	2.453	0.000	0.408
0.01	0.232	0.122	1.00	2.082	0.505	0.480
0.03	0.343	0.123	1.32	1.870	0.633	0.535
0.05	0.429	0.123	1.52	1.739	0.701	0.575

0.06	0.492	0.123	1.65	1.654	0.742	0.605
0.08	0.560	0.123	1.77	1.572	0.778	0.636
0.10	0.598	0.124	1.84	1.532	0.796	0.653
0.12	0.637	0.123	1.90	1.490	0.814	0.671
0.14	0.670	0.123	1.96	1.457	0.828	0.686
0.17	0.715	0.124	2.03	1.417	0.844	0.706
0.20	0.753	0.124	2.09	1.382	0.859	0.724
0.24	0.795	0.124	2.15	1.347	0.872	0.742
0.28	0.827	0.124	2.19	1.322	0.882	0.756
0.32	0.847	0.124	2.22	1.306	0.889	0.766
0.36	0.879	0.124	2.26	1.282	0.898	0.780
0.41	0.900	0.124	2.29	1.266	0.904	0.790
0.47	0.932	0.124	2.34	1.242	0.913	0.805
0.53	0.955	0.124	2.37	1.227	0.918	0.815
0.59	0.984	0.124	2.41	1.207	0.926	0.829
0.65	1.017	0.124	2.45	1.187	0.933	0.842
0.71	1.054	0.124	2.49	1.164	0.942	0.859
0.77	1.089	0.124	2.54	1.142	0.950	0.876
0.83	1.132	0.124	2.59	1.118	0.959	0.895
0.89	1.173	0.124	2.64	1.094	0.967	0.914
0.95	1.210	0.124	2.68	1.075	0.974	0.930
0.98	1.229	0.124	2.70	1.066	0.977	0.938
1.01	1.259	0.124	2.74	1.050	0.982	0.952
1.04	1.277	0.124	2.76	1.041	0.986	0.960
1.07	1.299	0.124	2.78	1.031	0.989	0.969
1.10	1.316	0.124	2.80	1.022	0.992	0.978
1.13	1.335	0.124	2.83	1.012	0.996	0.988
1.16	1.342	0.124	2.83	1.011	0.996	0.989
1.19	1.365	0.124	2.86	1.000	1.000	1.000

FFS25°, M1 = 2.88, SECTION 17

X = 15.05 cm, Me = 2.91, Pu = 0.126 kg/cm²,  
 Ue = 621.3 m/s, Te = 113.3 K, RHOe = 0.0394 kg s/m

Y	P0	P	M	T/Te	U/Ue	RHO/RHOe
[cm]	(kg/cm²)	(kg/cm²)				
0.00	0.126	0.126	0.00	2.509	0.000	0.398
0.01	0.243	0.126	1.02	2.119	0.509	0.472
0.03	0.329	0.126	1.27	1.952	0.608	0.512
0.05	0.432	0.126	1.50	1.790	0.690	0.558
0.06	0.481	0.127	1.60	1.725	0.721	0.580
0.07	0.514	0.127	1.66	1.682	0.740	0.594
0.08	0.545	0.127	1.72	1.647	0.756	0.607
0.09	0.567	0.127	1.76	1.620	0.768	0.617
0.10	0.590	0.127	1.80	1.593	0.779	0.628
0.11	0.611	0.127	1.83	1.573	0.788	0.636
0.12	0.632	0.127	1.86	1.550	0.797	0.645
0.14	0.660	0.127	1.91	1.519	0.810	0.658
0.16	0.695	0.127	1.97	1.486	0.823	0.673
0.18	0.729	0.127	2.02	1.456	0.835	0.687
0.20	0.757	0.127	2.06	1.429	0.844	0.700
0.22	0.780	0.127	2.10	1.408	0.854	0.710
0.25	0.807	0.127	2.13	1.387	0.862	0.721
0.28	0.836	0.127	2.18	1.362	0.872	0.734
0.32	0.870	0.128	2.22	1.337	0.881	0.748
0.37	0.906	0.128	2.27	1.311	0.891	0.763
0.42	0.936	0.127	2.31	1.287	0.900	0.777
0.47	0.968	0.128	2.35	1.267	0.907	0.789
0.52	0.993	0.128	2.38	1.250	0.914	0.800
0.57	1.027	0.128	2.42	1.226	0.922	0.816
0.63	1.058	0.127	2.47	1.205	0.930	0.830
0.69	1.093	0.127	2.51	1.185	0.937	0.844
0.75	1.142	0.128	2.56	1.156	0.947	0.865
0.81	1.182	0.128	2.61	1.134	0.955	0.882
0.87	1.224	0.128	2.66	1.111	0.963	0.900
0.93	1.272	0.128	2.71	1.086	0.971	0.920
0.99	1.323	0.128	2.77	1.060	0.980	0.943
1.04	1.358	0.128	2.80	1.046	0.985	0.956
1.09	1.390	0.128	2.84	1.030	0.990	0.971



1.10 1.400 0.128 2.85 1.025 0.992 0.976  
 1.15 1.425 0.128 2.88 1.013 0.996 0.987  
 1.20 1.448 0.128 2.90 1.005 0.998 0.995  
 1.25 1.460 0.128 2.91 1.000 1.000 1.000

\*\*\*\*\*Boundary Layer Parameters Through FFS25 Interaction\*\*\*\*\*

FFS25°, M1 = 2.88

Delta Theta Cfl  
 SECT. (mm) (mm) a10\*\*3

1 4.10 1.43 0.30 1.47  
 2 3.97 - - -  
 3 5.26 - - -  
 4 6.40 - - -  
 5 6.34 - - -  
 6 5.20 - - -  
 7 4.27 1.80 0.57 0.54  
 8 4.91 1.51 0.54 1.41  
 9 3.26 1.05 0.41 1.88  
 10 2.53 0.92 0.40 2.26  
 11 10.59 1.59 0.41 2.01  
 12 9.09 2.31 0.56 1.72  
 13 10.43 2.71 0.64 1.68  
 14 10.33 2.72 0.64 1.61  
 15 10.59 2.89 0.66 1.54  
 16 10.78 2.89 0.66 1.56  
 17 10.89 3.07 0.67 1.58

\*\*\*\*\*Turbulence Data\*\*\*\*\*

\*\*\*\*\*Incoming Boundary-Layer Profiles in Wind Tunnel I-325\*\*\*\*\*

Minf Pstagnation P Inf To Inf Re1x10.6  
 (kg/cm²) (kg/cm²) (K) (1/m)

3.00 3.562±0.09 0.097±0.002 295±5 28.0±2

EXTERNAL FLOW AND UNDISTURBED BOUNDARY LAYER

PARAMETERS IN SECTION 1

M1 = 2.95 Tel = 108 K  
 P1 = 0.0994 kg/cm² Uel = 614.6 m/s  
 Delta = 2.27 mm RHOel=0.03205 kg s¹/m  
 Delta\* = 0.79 mm mel = 19.8 kg s/m  
 Theta = 0.15 mm Cfl = 0.00153

Y	P0	P	T/Tel	U/Uel	RHO/RHOel
0.00	0.099	0.099	0.00	2.549	0.000
0.13	0.252	0.099	1.24	1.998	0.596
0.18	0.298	0.099	1.38	1.901	0.647
0.23	0.332	0.099	1.48	1.832	0.680
0.28	0.376	0.099	1.59	1.755	0.716
0.33	0.414	0.099	1.69	1.692	0.744
0.38	0.442	0.099	1.75	1.649	0.762
0.43	0.470	0.099	1.81	1.608	0.779
0.48	0.494	0.099	1.86	1.574	0.793
0.53	0.518	0.099	1.91	1.542	0.806
0.63	0.552	0.099	1.98	1.499	0.823
0.73	0.584	0.099	2.05	1.460	0.838
0.83	0.616	0.099	2.11	1.424	0.852
0.93	0.643	0.099	2.16	1.395	0.863
1.05	0.666	0.099	2.23	1.350	0.880
1.15	0.723	0.099	2.30	1.314	0.893
1.30	0.774	0.099	2.38	1.268	0.909
1.45	0.818	0.099	2.48	1.214	0.928
1.60	0.894	0.099	2.57	1.171	0.943
1.80	0.965	0.099	2.68	1.120	0.961
2.00	1.030	0.099	2.77	1.077	0.975
2.20	1.095	0.099	2.86	1.037	0.988
2.40	1.127	0.099	2.90	1.019	0.994
2.60	1.142	0.099	2.92	1.010	0.997

2.80 1.152 0.099 2.94 1.004 0.999 0.994  
 3.00 1.160 0.099 2.95 1.000 1.000 1.000  
 3.20 1.160 0.099 2.95 1.000 1.000 1.000

Y/Delta cm/me1 uU/Uel RHO/RHOel

0.07 1.76 2.54 0.87  
 0.09 1.96 2.55 0.98  
 0.13 2.39 2.57 1.78  
 0.18 2.82 2.46 2.11  
 0.22 3.03 2.45 2.23  
 0.26 3.21 2.40 2.33  
 0.31 3.46 2.39 2.50  
 0.35 3.72 2.47 2.64  
 0.40 4.04 2.51 2.85  
 0.44 4.03 2.39 2.85  
 0.48 4.23 2.40 2.98  
 0.53 4.46 2.38 3.15  
 0.57 4.71 2.38 3.29  
 0.62 4.94 2.34 3.28  
 0.66 4.98 2.26 3.43  
 0.70 4.96 2.16 3.47  
 0.75 4.91 2.02 3.44  
 0.79 4.73 1.87 3.26  
 0.84 4.60 1.73 3.17  
 0.93 3.66 1.28 2.53  
 1.01 2.41 0.79 1.66  
 1.10 1.29 0.42 0.89  
 1.19 0.70 0.22 0.48  
 1.28 0.50 0.16 0.34  
 1.37 0.42 0.14 0.30  
 1.45 0.40 0.12 0.27  
 1.54 0.37 0.11 0.26

\*\*\*\*\*SURFACE PRESSURE, FFS25, M1 = 2.95, P1 = 0.0994 kg/cm²\*\*\*\*\*

X(mm) Pa/P1

-36 1.000  
 -31 1.000  
 -26 1.020  
 -21 0.980  
 -16 1.020  
 -11 1.000  
 -6 2.040  
 -1 2.300  
 5 3.240  
 9 3.810  
 13 3.950  
 16 1.380  
 21 1.210  
 26 1.090  
 31 1.090  
 36 1.020  
 41 0.990  
 46 1.000  
 51 1.000

\*\*\*\*\*HOT-WIRE ANEMOMETRY SURVEY LOCATIONS, FFS25\*\*\*\*\*

SECT. X (mm)

1 -35.0  
 2 10.0  
 3 18.0  
 4 34.0  
 5 59.0

\*\*\*\*\*NOT-WIRE DATA, FFS25\*\*\*\*\*

cm-e/m-e1=0.37%, <cm-max>/m-e1=5%									
SECT	1	2	3	4	5				
X(mm)	35	10	18	34	59				
cm-e/cm-e1	1.00	9.30	6.89	5.59	4.97				
cm-max/cm-max1	1.00			1.43	1.35				

#### FFS25, SECTION 1

Delta = Delta1 = 2.27 mm;  
 <cm-max/m1> = 5.0%, <cm-max> = <cm-max1>;  
 <cm-max/Ue1> = 2.4%, <U-max> = <U-max1>;  
 <RHO-max/RHOe1> = 3.5%, <RHO-max> = <RHO-max1>.

Y	<cm>	<U>	<RHO>	U	RHO	M
Delta1	<cm-max1>	<U-max1>	<RHO-max1>	Ue1	RHOe1	
0.1	0.41	1.06	0.37	0.680	0.550	1.48
0.2	0.57	1.02	0.61	0.785	0.595	1.85
0.3	0.69	1.02	0.70	0.830	0.680	2.02
0.4	0.80	1.02	0.80	0.860	0.715	2.16
0.5	0.90	1.00	0.88	0.890	0.755	2.29
0.6	0.97	0.98	0.95	0.915	0.805	2.43
0.7	1.00	0.90	1.00	0.940	0.850	2.56
0.8	0.96	0.76	0.95	0.960	0.900	2.68
0.9	0.78	0.58	0.77	0.975	0.945	2.79
1.0	0.68	0.33	0.49	0.990	0.975	2.88
1.1	0.25	0.17	0.25	0.990	0.990	2.92
1.2	0.14	0.10	0.13	1.000	0.995	2.94
1.3	0.10	0.07	0.09	1.000	1.000	2.95
1.4	0.08	0.06	0.07	1.000	1.000	2.95

#### FFS25, SECTION 2

Delta = 1.80 Delta1

Y	<cm>	<U>	<RHO>	U	RHO	M
Delta1	<cm-max1>	<U-max1>	<RHO-max1>	Ue1	RHOe1	
0.7	4.43	2.86	2.78	0.595	2.370	1.23
0.8	4.40	2.71	2.87	0.620	2.420	1.28
0.9	4.21	2.45	2.89	0.645	2.475	1.35
1.0	3.70	2.02	2.67	0.665	2.530	1.42
1.1	2.87	1.51	2.11	0.685	2.570	1.46
1.2	2.03	1.05	1.51	0.690	2.585	1.48
1.3	1.39	0.72	1.03	0.690	2.590	1.48
1.4	0.98	0.50	0.73	0.690	2.597	1.48
1.5	0.82	0.43	0.61	0.690	2.597	1.48
1.6	0.75	0.38	0.56	0.690	2.597	1.48
1.7	0.72	0.38	0.53	0.690	2.597	1.48

#### FFS25, SECTION 3

Y	<cm>	<U>	<RHO>	U	RHO	M
Delta1	<cm-max1>	<U-max1>	<RHO-max1>	Ue1	RHOe1	
0.1	0.94	1.37	0.83	0.810	0.715	1.93
0.2	1.03	1.37	0.93	0.847	0.735	2.06
0.3	1.15	1.42	1.05	0.867	0.753	2.16
0.4	1.25	1.47	1.15	0.879	0.768	2.22
0.5	1.35	1.53	1.24	0.890	0.781	2.25
0.6	1.45	1.58	1.35	0.892	0.800	2.28
0.7	1.56	1.64	1.46	0.898	0.815	2.32
0.8	1.66	1.70	1.57	0.905	0.825	2.36
0.9	1.78	1.76	1.67	0.915	0.833	2.40
1.0	1.90	1.82	1.80	0.925	0.840	2.45
1.1	2.01	1.87	1.92	0.933	0.845	2.50

1.2	2.12	1.93	2.03	0.942	0.851	2.54
1.3	2.25	2.00	2.15	0.945	0.865	2.55
1.4	2.26	2.03	2.16	0.942	0.885	2.55
1.5	2.26	1.98	2.16	0.940	0.915	2.54
1.6	2.12	1.71	2.03	0.940	0.955	2.54
1.7	1.87	1.37	1.70	0.940	1.000	2.54
1.8	1.60	1.18	1.53	0.940	1.050	2.54
1.9	1.36	0.96	1.30	0.937	1.100	2.54
2.0	1.18	0.80	1.13	0.933	1.150	2.53
2.1	1.06	0.68	1.02	0.930	1.210	2.52
2.2	0.94	0.58	0.91	0.925	1.270	2.52

#### FFS25, SECTION 4

Delta = 2.14 Delta1

Y	<cm>	<U>	<RHO>	U	RHO	M
Delta1	<cm-max1>	<U-max1>	<RHO-max1>	Ue1	RHOe1	
0.1	0.58	1.37	0.41	0.690	0.575	1.44
0.2	0.75	1.20	0.65	0.765	0.675	1.86
0.3	0.68	1.02	0.68	0.720	0.695	1.95
0.4	0.69	1.00	0.69	0.725	0.700	1.98
0.5	0.70	1.00	0.71	0.735	0.705	2.02
0.6	0.73	1.00	0.74	0.745	0.715	2.05
0.7	0.77	1.01	0.78	0.755	0.730	2.10
0.8	0.80	1.01	0.83	0.765	0.761	2.14
0.9	0.85	1.03	0.88	0.775	0.758	2.19
1.0	0.91	1.04	0.95	0.785	0.778	2.24
1.1	0.97	1.06	1.02	0.795	0.795	2.29
1.2	1.05	1.10	1.11	0.808	0.813	2.34
1.3	1.14	1.13	1.20	0.819	0.830	2.40
1.4	1.23	1.16	1.31	0.828	0.850	2.46
1.5	1.33	1.20	1.41	0.839	0.867	2.51
1.6	1.43	1.23	1.53	0.849	0.888	2.56
1.7	1.49	1.22	1.59	0.859	0.910	2.62
1.8	1.50	1.18	1.61	0.865	0.930	2.68
1.9	1.46	1.09	1.57	0.875	0.955	2.73
2.0	1.37	0.98	1.47	0.882	0.975	2.78
2.1	1.26	0.86	1.35	0.889	0.988	2.81
2.2	1.08	0.73	1.17	0.893	1.000	2.83
2.3	0.93	0.63	1.01	0.895	1.050	2.85
2.4	0.79	0.52	0.86	0.900	1.100	2.88
2.5	0.67	0.43	0.72	0.900	1.150	2.89
2.6	0.57	0.38	0.62	0.900	1.200	2.89
2.7	0.52	0.34	0.57	0.900	1.200	2.89
2.8	0.49	0.32	0.53	0.900	1.200	2.89

#### FFS25, SECTION 5

Delta = 2.12 Delta1

Y	<cm>	<U>	<RHO>	U	RHO	M
Delta1	<cm-max1>	<U-max1>	<RHO-max1>	Ue1	RHOe1	
0.1	0.63	1.17	0.58	0.690	0.625	1.74
0.2	0.71	1.02	0.74	0.700	0.700	2.00
0.3	0.76	1.00	0.80	0.730	0.730	2.08
0.4	0.76	0.96	0.80	0.740	0.750	2.14
0.5	0.77	0.93	0.81	0.750	0.765	2.18
0.6	0.78	0.91	0.83	0.760	0.775	2.22
0.7	0.79	0.90	0.84	0.770	0.785	2.25
0.8	0.81	0.88	0.86	0.780	0.795	2.29
0.9	0.83	0.88	0.89	0.790	0.810	2.33
1.0	0.86	0.88	0.92	0.800	0.820	2.37
1.1	0.92	0.91	0.99	0.810	0.830	2.41
1.2	0.98	0.94	1.05	0.820	0.845	2.45
1.3	1.05	0.97	1.13	0.830	0.855	2.50
1.4	1.13	1.00	1.21	0.840	0.870	2.54
1.5	1.21	1.03	1.30	0.850	0.885	2.58
1.6	1.30	1.06	1.39	0.860	0.905	2.64

```

1.7 1.19 1.09 1.49 0.870 0.925 2.68
1.8 1.42 1.07 1.53 0.875 0.940 2.74
1.9 1.33 0.96 1.45 0.880 0.960 2.79
2.0 1.18 0.82 1.28 0.890 0.975 2.83
2.1 1.00 0.68 1.09 0.895 0.990 2.87
2.2 0.82 0.54 0.90 0.900 1.000 2.90
2.3 0.71 0.44 0.77 0.900 1.005 2.92
2.4 0.62 0.41 0.68 0.900 1.010 2.93
2.5 0.57 0.32 0.60 0.900 1.015 2.94
2.6 0.54 0.29 0.59 0.900 1.020 2.95
2.7 0.50 0.27 0.54 0.900 1.020 2.95
2.8 0.46 0.27 0.54 0.900 1.020 2.95
2.9 0.45 0.25 0.50 0.900 1.020 2.95

*****Flow Conditions for FFS25 Heat Transfer Tests in Tunnel T-333*****
M1 = 3.01 +/- 0.02
Pstagnation = 4.520 kg/cm**2
Pinf = 0.123 kg/cm**2
Tstagnation = 263.0 K
Rel/m = 4.2E-06

*****Incoming Boundary-Layer Profiles in Wind Tunnel T-333*****
Me = 3.02, Pw = 0.125 kg/cm**2, Rel = 42x10E-06/m, Te = 93.18 K, Ue = 584.1 m/s,
RhoE = 0.0475 kg.s**2/m**4, Delta* = 2.66 mm, Delta* = 0.68 mm, theta = 0.12 mm.

```

Y Po P M 1/Te U/Ue RHO/RHOE

```

0.00 0.125 0.125 0.00 2.622 0.000 0.181
0.17 0.316 0.125 1.24 2.058 0.590 0.466
0.24 0.473 0.126 1.59 1.810 0.708 0.552
0.30 0.539 0.125 1.72 1.717 0.747 0.582
0.39 0.608 0.126 1.83 1.640 0.778 0.610
0.45 0.638 0.128 1.87 1.618 0.787 0.618
0.55 0.807 0.126 2.14 1.442 0.853 0.694
0.67 0.846 0.126 2.20 1.409 0.865 0.710
0.74 0.898 0.126 2.27 1.367 0.880 0.751
0.83 0.959 0.126 2.33 1.332 0.892 0.751
0.93 0.977 0.126 2.38 1.307 0.900 0.765
1.00 1.016 0.126 2.43 1.280 0.910 0.781
1.08 1.050 0.126 2.47 1.257 0.917 0.796
1.18 1.140 0.126 2.56 1.211 0.933 0.826
1.47 1.230 0.126 2.68 1.148 0.953 0.871
1.58 1.300 0.127 2.75 1.116 0.964 0.896
1.78 1.410 0.128 2.86 1.067 0.979 0.937
1.86 1.430 0.127 2.89 1.053 0.984 0.950
1.96 1.450 0.128 2.90 1.048 0.985 0.954
2.03 1.490 0.129 2.93 1.036 0.989 0.966
2.12 1.500 0.128 2.95 1.026 0.992 0.974
2.22 1.540 0.128 3.00 1.009 0.997 0.991
2.32 1.550 0.127 3.02 1.000 1.000 1.000

```

Me = 4.01, Pw = 0.052 kg/cm\*\*2, Rel = 40E-06/m, Te = 65.53 K, Ue = 650.3 m/s,  
RhoE = 0.0274 kg.s\*\*2/m\*\*4, Delta\* = 2.50 mm, Delta\* = 1.16 mm, theta = 0.14 mm.

Y PO P M 1/Te U/Ue RHO/RHOE

```

0.00 0.052 0.052 0.00 3.858 0.000 0.259
0.17 0.122 0.052 1.18 3.087 0.570 0.324
0.25 0.132 0.052 1.25 3.018 0.542 0.331
0.32 0.166 0.052 1.45 2.811 0.405 0.356
0.39 0.197 0.052 1.60 2.647 0.651 0.378
0.45 0.227 0.052 1.74 2.507 0.688 0.399
0.51 0.254 0.052 1.85 2.394 0.716 0.418
0.59 0.306 0.052 2.06 2.202 0.761 0.454
0.66 0.359 0.052 2.24 2.036 0.798 0.491
0.75 0.408 0.052 2.40 1.904 0.827 0.525
0.82 0.443 0.052 2.51 1.819 0.845 0.550

```

```

0.91 0.453 0.052 2.54
1.01 0.471 0.052 2.59
1.10 0.494 0.052 2.66
1.18 0.522 0.052 2.74
1.24 0.540 0.052 2.78
1.31 0.560 0.052 2.84
1.39 0.585 0.052 2.90
1.47 0.607 0.052 2.96
1.54 0.618 0.052 2.99
1.59 0.640 0.052 3.04
1.65 0.672 0.052 3.12
1.72 0.720 0.052 3.24
1.77 0.750 0.052 3.30
1.84 0.753 0.052 3.31
1.91 0.766 0.052 3.34
1.99 0.797 0.052 3.41
2.06 0.825 0.052 3.47
2.13 0.861 0.052 3.55
2.19 0.906 0.052 3.63
2.26 0.951 0.052 3.74
2.33 0.952 0.052 3.74
2.39 0.957 0.052 3.75
2.45 0.996 0.052 3.82
2.51 1.018 0.052 3.87
2.59 1.068 0.052 3.96
2.65 1.063 0.052 3.95
2.73 1.091 0.052 4.01

```

\*\*\*\*\*Heat Transfer Data on Model FFS25\*\*\*\*\*

```

M1 = 3.01, a1 = 280 W/m**2.K
X(mm) a/a1 Trw Tw X(mm) a/a1 Trw Tw
-36.5 1.16 248.2 256.8 22.5 2.00 244.3 260.9
-28.0 1.02 244.2 256.8 23.5 1.83 245.8 261.7
-21.0 1.00 244.4 258.1 25.5 1.84 250.6 265.8
-15.0 0.93 244.7 257.2 26.5 1.81 247.6 263.4
-19.0 0.91 250.1 259.4 28.0 1.89 247.7 264.6
-16.0 0.98 243.8 257.2 29.0 1.81 247.8 265.2
-13.0 0.89 247.0 259.4 30.5 1.80 247.3 265.2
-10.0 1.14 249.7 259.5 32.0 1.71 248.0 265.1
-7.0 1.31 250.1 261.6 33.0 1.63 247.8 265.8
-4.0 1.29 247.5 258.5 34.0 1.74 248.0 265.9
-2.0 1.30 251.0 261.0 35.5 1.82 248.2 265.9
0.0 1.34 248.1 259.1 37.0 1.76 248.4 266.2
3.0 1.65 253.7 264.2 39.0 1.75 248.3 266.5
6.5 1.93 253.8 263.7 41.0 1.82 248.1 266.8
10.5 2.33 253.6 262.1 49.0 1.82 248.4 263.7
12.5 3.66 253.9 261.7 61.0 1.54 247.9 264.1
13.5 3.68 250.8 262.0 69.0 1.30 247.6 263.4
14.5 3.17 242.8 256.8 74.5 1.18 248.0 272.3
15.0 3.11 242.7 256.7 78.5 1.11 247.8 272.3
17.0 2.55 243.0 258.7 83.5 1.10 247.8 271.1
18.0 2.33 242.4 258.6
19.5 2.37 242.8 258.6
20.5 2.19 244.3 259.3

```

```

M1 = 4.03, a1 = 189 W/m**2
X(mm) a/a1 Trw Tw X(mm) a/a1 Trw Tw
-24.0 0.98 249.4 274.3 14.5 2.42 239.4 251.7
-21.0 1.02 250.0 274.0 15.0 2.51 238.8 251.7
-18.0 0.95 259.6 287.0 16.0 2.33 238.5 252.8
-15.0 1.00 248.7 272.4 16.5 2.42 237.9 252.3
-12.0 0.90 248.5 271.7 17.0 2.50 238.8 254.2
-9.0 0.90 245.3 269.3 18.0 2.50 238.3 255.5
-8.0 0.94 246.6 272.3 19.0 2.60 239.4 255.0
-7.5 0.97 244.1 270.3 19.5 2.45 239.4 254.7
-6.0 1.17 234.0 262.1 20.0 2.55 239.4 254.7
-5.0 1.17 247.6 276.7 21.0 2.40 239.7 256.0
-3.5 1.23 248.1 277.0 22.5 2.50 241.7 258.2
-3.0 1.25 242.5 272.0 23.5 2.33 241.6 258.6
-2.0 1.29 247.5 277.9 24.5 2.45 242.0 258.9

```

0.0	1.40	242.5	272.9	26.0	2.45	243.3	260.1
5.5	2.35	246.4	277.0	28.5	2.40	243.9	261.4
8.0	2.78	246.0	263.5	30.0	2.34	243.8	261.3
9.0	3.04	245.1	252.2	31.0	2.30	244.2	262.0
10.0	3.90	245.8	253.2	32.0	2.45	244.7	262.0
11.0	4.01	245.8	253.2	33.5	2.30	245.1	262.8
12.0	4.00	243.6	251.4	34.5	2.40	245.4	262.9
13.0	3.90	242.9	251.4	38.0	2.40	246.3	263.6
13.5	3.00	240.8	251.6	39.5	2.40	246.1	263.6
14.0	2.69	244.4	243.0	41.0	2.30	246.3	263.9
				42.5	2.36	246.6	268.6
				44.0	2.30	249.6	272.3
				49.0	2.45	245.9	268.8
				54.0	2.36	246.9	271.3
				64.0	1.75	245.0	264.5
				69.0	1.65	247.3	265.4
				74.0	1.47	247.3	265.4
				79.0	1.34	247.6	265.7
				84.0	1.32	247.6	265.3

---

Ref.: 49, 32, 10

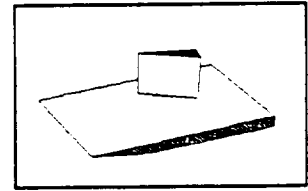
Author: Bogdonoff, S. M., *et al*

Geometry: 3-D Fin

Mach number: 3

Data:  $p_{wall}$ , mean flowfield surveys ("cobra" probe)

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Knight, D.D., Horstman, C.C., Bogdonoff, S.M. and Shapey, B.L., "The Flowfield Structure of the 3-D Shock Wave - Boundary Layer Interaction Generated by a 20 deg Sharp Fin at Mach 3," *AIAA Paper 86-0343*, 1986.

Goodwin, S. "An Exploratory Investigation of Sharp-Fin Induced Shock Wave/Turbulent Boundary Layer Interactions at High Shock Strengths," MS Thesis, Dept. of Mechanical and Aerospace Engineering, Princeton Univ., 1984.

Knight, D.D., Horstman, C.C., Shapey, B., and Bogdonoff, S. M., "Structure of Supersonic Turbulent Flow Past a Sharp Fin," *AIAA Journal*, Vol 25, Oct. 1987, pp. 1331-1337.

Shapey, B., MS Thesis, Dept. of Mechanical and Aerospace Engineering, Princeton Univ., 1986.

Bogdonoff, S.M. and Shapey, B.L., "Three-Dimensional Shock Wave Turbulent Boundary Layer Interaction for a 20 deg Sharp Fin at Mach 3," *AIAA Paper 87-0554*, 1987.

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The following is an edited version (edited by C. C. Horstman) of the data obtained at the Princeton Gas Dynamics Lab in 1985-1986 for a 20-degree sharp fin interaction at Mach 3. The data are presented in two parts. Part 1 includes profile data while part 2 contains surface pressure data. Each part is headed by a description of the data format.

Part 1 contains contains surveys carried out with a computer-controlled nulling yaw probe ("cobra probe"). This probe measured pitot pressure and yaw angle along survey lines in the y-direction, *ie* normal to the tunnel wall which supported the turbulent boundary layer. The yaw angles thus measured lie in the horizontal (x-z) plane. The survey locations tabulated in part 1 and shown in the diagram below were chosen to provide detailed information within the separated region produced by this relatively strong swept-shock/boundary layer interaction. Since the tunnel-floor boundary-layer thickness is significant compared to the spanwise locations of the survey stations from the fin leading edge, it appears likely that the data are at least partially within the non-conical, fully 3-D "inception zone" of the interaction. Users of the data should bear this in mind.

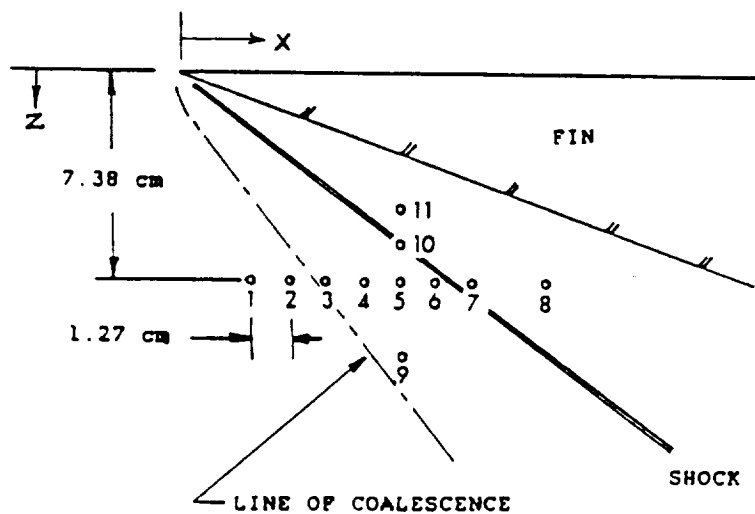
In part 1, where an undisturbed freestream exists, upstream of the inviscid shock and outside of the interaction, the freestream yaw angle is defined as zero. For stations 2, 3, 4, and 5 the yaw data show a non-zero freestream yaw. This is caused by a slight misalignment of the probe of around 1 to 2 degrees. A uniform shift of the yaw profile to bring the freestream yaw to zero should correct this.

Downstream of the inviscid shock it is difficult to tell exactly how well the probe is aligned since there is no undisturbed freestream to check against. For these stations the probe is aligned parallel to the tunnel wall as closely as possible, within about one degree.

*Note:* none of the corrections listed above have actually been applied to the yaw data tabulated below. Only the uncorrected yaw profiles are tabulated.

Users of these data should be careful of data very close to the floor. The point closest to the floor is taken when the probe touches the floor and the probe is not nulled.

Station coordinates  $X$  and  $Z$  are measured in inches from the fin apex, as illustrated in the diagram given below. The survey station coordinates are tabulated early in part 1. Surface pressures were measured along rows of orifices aligned with the freestream ( $x$ ) direction.



Location of Experimental Surveys

\*\*\*\*\*  
Data of Princeton Gas Dynamics Lab for 3-D Sharp fin at M = 3,  $\alpha = 20$  deg  
\*\*\*\*\*

TEST CONDITIONS

Freestream Mach number = 2.93  
Nominal freestream total pressure = 100psia or 6800Pa  
Wall temperature approx = 270K  
Reynolds number = 7.0E+08/meter  
Initial boundary layer parameters:  
Delta  $\theta$  at  $x=0$  1.3cm  
Delta star (compressible)  $\theta$  at  $x=0$  0.39cm  
Momentum thickness  $\theta$  at  $x=0$  0.072cm

PART ONE

STATION	X(IN)	Z(IN)	RUN
1	0.98	2.906	808
2	1.48	2.906	808
3	1.98	2.906	807
4	2.48	2.906	8091
5	2.98	2.906	809
6	3.48	2.906	730
7	3.98	2.906	807
8	4.98	2.906	8121
9	2.98	3.906	807
11	2.98	1.906	8081

Format of survey data files:

RUN TEST STATION NPIS DTIPE PONEAM TONEAM PIMEAN XLOC  
DATE COMMENT  
BLANK LINE  
YDATA PIDATA YADATA  
" " "  
" " "  
PONEAM, PIMEAN, PIDATA are in Pascals.  
TONEAM is in degrees Kelvin.  
XLOC is in inches.  
YDATA is in meters.  
YADATA is in degrees.

808 5 1 65 1 0.6898E+06 268.7 0.1993E+05 0.98  
8-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIN, BS

0.8900E+04 0.5127E+05 0.1012  
0.2900E+03 0.7669E+05 0.2047  
0.9348E+03 0.9048E+05 0.1194  
0.1223E+02 0.9910E+05 0.1803  
0.1682E+02 0.1026E+06 0.1012  
0.2138E+02 0.1121E+06 0.1743  
0.2594E+02 0.1169E+06 0.1073  
0.2795E+02 0.1206E+06 0.1864  
0.3498E+02 0.1259E+06 0.1134  
0.4008E+02 0.1366E+06 0.1864  
0.4449E+02 0.1489E+06 0.9508E-01  
0.4952E+02 0.1652E+06 0.1925  
0.5709E+02 0.1552E+06 0.9508E-01  
0.6150E+02 0.1541E+06 0.1438  
0.6521E+02 0.1610E+06 0.1560

0.7286E+02 0.1676E+06 0.8899E-01  
0.7974E+02 0.1765E+06 0.1682  
0.8523E+02 0.1814E+06 0.5265E-01  
0.9103E+02 0.1874E+06 0.1560  
0.9776E+02 0.1940E+06 0.9508E-01  
0.1032E+02 0.1995E+06 0.1621  
0.1105E+02 0.2082E+06 0.9508E-01  
0.1160E+02 0.2162E+06 0.1925  
0.1236E+02 0.2205E+06 0.9508E-01  
0.1286E+02 0.2285E+06 0.1925  
0.1346E+02 0.2274E+06 0.8290E-01  
0.1416E+02 0.2326E+06 0.1073  
0.1488E+02 0.2312E+06 0.1803  
0.1577E+02 0.2321E+06 0.8899E-01  
0.1594E+02 0.2334E+06 0.1925  
0.1666E+02 0.2320E+06 0.7681E-01  
0.1735E+02 0.2332E+06 0.1438  
0.1782E+02 0.2331E+06 0.2169  
0.1859E+02 0.2351E+06 0.1012  
0.1914E+02 0.2348E+06 0.2656  
0.1982E+02 0.2359E+06 0.1377  
0.2049E+02 0.2348E+06 0.1743  
0.2109E+02 0.2333E+06 0.1986  
0.2173E+02 0.2337E+06 0.2473  
0.2241E+02 0.2346E+06 0.1864  
0.2293E+02 0.2353E+06 0.2656  
0.2358E+02 0.2367E+06 0.2717  
0.2431E+02 0.2373E+06 0.1986  
0.2496E+02 0.2381E+06 0.2047  
0.2548E+02 0.2390E+06 0.2839  
0.2620E+02 0.2385E+06 0.1621  
0.2668E+02 0.2397E+06 0.2778  
0.2739E+02 0.2390E+06 0.1925  
0.2804E+02 0.2386E+06 0.2412  
0.2858E+02 0.2385E+06 0.2595  
0.2932E+02 0.2393E+06 0.1946  
0.2980E+02 0.2379E+06 0.3082  
0.3056E+02 0.2395E+06 0.1682  
0.3115E+02 0.2400E+06 0.2778  
0.3179E+02 0.2391E+06 0.2717  
0.3251E+02 0.2398E+06 0.2108  
0.3302E+02 0.2398E+06 0.2961  
0.3377E+02 0.2395E+06 0.2047  
0.3438E+02 0.2398E+06 0.2900  
0.3499E+02 0.2387E+06 0.2352  
0.3572E+02 0.2398E+06 0.1925  
0.3642E+02 0.2399E+06 0.2900  
0.3701E+02 0.2386E+06 0.1743  
0.3765E+02 0.2399E+06 0.2412  
0.3822E+02 0.2370E+06 0.1560

808 3 2 66 1 0.6898E+06 268.9 0.1993E+05 1.48  
8-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIN, BS

0.8900E+04 0.5353E+05 1.213  
0.2823E+03 0.7772E+05 1.216  
0.8002E+03 0.9365E+05 1.219  
0.1133E+02 0.9837E+05 1.222  
0.1519E+02 0.1033E+06 1.264  
0.1805E+02 0.1078E+06 1.301  
0.2076E+02 0.1117E+06 1.392  
0.2779E+02 0.1207E+06 1.307  
0.2996E+02 0.1245E+06 1.319  
0.3606E+02 0.1324E+06 1.325  
0.4287E+02 0.1391E+06 1.319  
0.4952E+02 0.1446E+06 1.161  
0.5508E+02 0.1522E+06 1.179  
0.6243E+02 0.1561E+06 1.143  
0.6745E+02 0.1682E+06 1.234  
0.7147E+02 0.1673E+06 1.258  
0.7681E+02 0.1765E+06 1.240

0.8003E-020 1796E+06 1.214  
0.8655E-020 1860E+06 1.277  
0.9358E-020 1911E+06 1.210  
0.9969E-020 1981E+06 1.277  
0.1056E-010 2068E+06 1.222  
0.1124E-010 2122E+06 1.197  
0.1179E-010 2179E+06 1.270  
0.1253E-010 2254E+06 1.197  
0.1307E-010 2279E+06 1.289  
0.1368E-010 2293E+06 1.216  
0.1437E-010 2317E+06 1.191  
0.1491E-010 2318E+06 1.289  
0.1564E-010 2339E+06 1.161  
0.1622E-010 2332E+06 1.246  
0.1679E-010 2344E+06 1.234  
0.1753E-010 2368E+06 1.203  
0.1866E-010 2369E+06 1.283  
0.1873E-010 2369E+06 1.197  
0.1939E-010 2378E+06 1.203  
0.1993E-010 2372E+06 1.252  
0.2071E-010 2381E+06 1.155  
0.2129E-010 2390E+06 1.246  
0.2194E-010 2395E+06 1.210  
0.2251E-010 2399E+06 1.295  
0.2323E-010 2394E+06 1.185  
0.2386E-010 2393E+06 1.270  
0.2441E-010 2370E+06 1.283  
0.2520E-010 2367E+06 1.216  
0.2562E-010 2384E+06 1.295  
0.2634E-010 2380E+06 1.216  
0.2700E-010 2395E+06 1.203  
0.2751E-010 2400E+06 1.283  
0.2828E-010 2416E+06 1.216  
0.2882E-010 2412E+06 1.307  
0.2941E-010 2416E+06 1.246  
0.3011E-010 2423E+06 1.210  
0.3074E-010 2419E+06 1.246  
0.3134E-010 2429E+06 1.252  
0.3205E-010 2419E+06 1.191  
0.3260E-010 2423E+06 1.283  
0.3332E-010 2425E+06 1.203  
0.3393E-010 2427E+06 1.277  
0.3453E-010 2426E+06 1.179  
0.3529E-010 2420E+06 1.179  
0.3587E-010 2419E+06 1.258  
0.3651E-010 2429E+06 1.216  
0.3721E-010 2421E+06 1.173  
0.3773E-010 2439E+06 1.246  
0.3849E-010 2440E+06 1.185

807 8 3 69 1 0.6890E+06 270.6 0.2031E+05 1.98  
7-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIM, BS

0.8900E-040 4400E+05 54.00  
0.2279E-030 5676E+05 40.39  
0.9147E-030 8041E+05 12.26  
0.9842E-030 8917E+05 11.01  
0.1139E-020 9246E+05 10.61  
0.1324E-020 1014E+06 8.632  
0.1825E-020 1128E+06 7.803  
0.1995E-020 1157E+06 7.608  
0.2219E-020 1238E+06 7.157  
0.2535E-020 1289E+06 7.029  
0.2929E-020 1359E+06 6.956  
0.3261E-020 1438E+06 6.895  
0.3485E-020 1457E+06 6.962  
0.4087E-020 1580E+06 6.554  
0.4364E-020 1612E+06 6.408  
0.5082E-020 1663E+06 6.261  
0.5622E-020 1679E+06 6.139  
0.6301E-020 1680E+06 5.115

0.6896E-020 1705E+06 4.043  
0.7613E-020 1744E+06 2.946  
0.8200E-020 1795E+06 2.306  
0.8856E-020 1858E+06 1.964  
0.9435E-020 1896E+06 1.934  
0.1018E-010 1980E+06 1.800  
0.1073E-010 2038E+06 1.861  
0.1133E-010 2124E+06 1.940  
0.1202E-010 2176E+06 1.855  
0.1262E-010 2243E+06 1.879  
0.1332E-010 2297E+06 1.891  
0.1386E-010 2291E+06 2.007  
0.1450E-010 2317E+06 1.934  
0.1519E-010 2342E+06 1.928  
0.1567E-010 2342E+06 1.964  
0.1646E-010 2359E+06 1.910  
0.1700E-010 2341E+06 1.977  
0.1761E-010 2342E+06 1.934  
0.1834E-010 2359E+06 1.934  
0.1864E-010 2350E+06 2.007  
0.1953E-010 2351E+06 1.928  
0.2015E-010 2355E+06 1.897  
0.2072E-010 2362E+06 1.944  
0.2164E-010 2358E+06 1.879  
0.2203E-010 2340E+06 1.964  
0.2264E-010 2346E+06 2.117  
0.2329E-010 2358E+06 1.970  
0.2400E-010 2350E+06 1.861  
0.2458E-010 2348E+06 1.903  
0.2519E-010 2351E+06 1.928  
0.2585E-010 2349E+06 1.861  
0.2637E-010 2351E+06 1.977  
0.2706E-010 2344E+06 1.873  
0.2769E-010 2346E+06 1.903  
0.2828E-010 2343E+06 1.934  
0.2901E-010 2351E+06 1.861  
0.2950E-010 2321E+06 1.964  
0.3019E-010 2316E+06 1.867  
0.3087E-010 2323E+06 1.916  
0.3142E-010 2331E+06 1.940  
0.3217E-010 2342E+06 1.867  
0.3270E-010 2342E+06 1.958  
0.3340E-010 2375E+06 1.861  
0.3409E-010 2394E+06 1.903  
0.3443E-010 2381E+06 1.928  
0.3540E-010 2382E+06 1.861  
0.3594E-010 2381E+06 1.958  
0.3661E-010 2379E+06 1.897  
0.3720E-010 2375E+06 1.940  
0.3792E-010 2369E+06 1.849  
0.3848E-010 2373E+06 1.891

8091 2 4 80 1 0.6892E+06 275.7 0.2014E+05 2.48  
9-AUG-85 TOTAL HEAD, 20 DEG FIM, BS

0.8900E-040 5538E+05 33.08  
0.2591E-030 6541E+05 37.07  
0.9881E-030 6946E+05 53.24  
0.1520E-020 7163E+05 50.84  
0.2162E-020 8453E+05 38.04  
0.2487E-020 9054E+05 27.41  
0.2734E-020 9663E+05 24.01  
0.3043E-020 1074E+06 17.86  
0.3190E-020 1112E+06 17.83  
0.3538E-020 1168E+06 15.50  
0.3871E-020 1272E+06 13.97  
0.4041E-020 1297E+06 12.22  
0.4374E-020 1381E+06 11.14  
0.4567E-020 1476E+06 9.434  
0.4768E-020 1542E+06 9.252  
0.4930E-020 1540E+06 9.240

0.5603E-020 1648E+06 8.624  
0.5864E-020 1699E+06 8.374  
0.6570E-020 1824E+06 7.479  
0.6872E-020 1881E+06 7.229  
0.7197E-020 1953E+06 7.101  
0.7514E-020 2016E+06 6.894  
0.7808E-020 2096E+06 6.632  
0.8112E-020 2104E+06 6.516  
0.8720E-020 2282E+06 6.395  
0.8898E-020 2304E+06 5.993  
0.9664E-020 2466E+06 5.030  
0.9934E-020 2510E+06 4.604  
0.1059E-010 2631E+06 4.384  
0.1095E-010 2694E+06 4.396  
0.1124E-010 2805E+06 4.311  
0.1162E-010 2723E+06 4.348  
0.1157E-010 2815E+06 4.348  
0.1160E-010 2770E+06 4.372  
0.1174E-010 2741E+06 4.305  
0.1180E-010 2772E+06 4.366  
0.1219E-010 2751E+06 4.403  
0.1280E-010 2721E+06 4.634  
0.1338E-010 2684E+06 5.030  
0.1402E-010 2502E+06 5.968  
0.1413E-010 2470E+06 6.401  
0.1459E-010 2407E+06 6.388  
0.1464E-010 2365E+06 6.194  
0.1558E-010 2356E+06 5.554  
0.1620E-010 2375E+06 3.397  
0.1667E-010 2352E+06 2.301  
0.1741E-010 2354E+06 1.947  
0.1792E-010 2357E+06 1.874  
0.1844E-010 2369E+06 1.850  
0.1918E-010 2362E+06 1.868  
0.1985E-010 2365E+06 1.838  
0.2048E-010 2370E+06 1.746  
0.2118E-010 2389E+06 1.734  
0.2178E-010 2360E+06 1.868  
0.2248E-010 2367E+06 1.673  
0.2307E-010 2357E+06 1.692  
0.2375E-010 2363E+06 1.655  
0.2423E-010 2355E+06 1.771  
0.2496E-010 2357E+06 1.649  
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0.2680E-010 2346E+06 1.655  
0.2755E-010 2344E+06 1.673  
0.2805E-010 2345E+06 1.752  
0.2875E-010 2373E+06 1.649  
0.2938E-010 2380E+06 1.584  
0.2992E-010 2367E+06 1.655  
0.3061E-010 2363E+06 1.618  
0.3125E-010 2358E+06 1.625  
0.3189E-010 2344E+06 1.612  
0.3262E-010 2337E+06 1.539  
0.3316E-010 2317E+06 1.637  
0.3362E-010 2304E+06 1.551  
0.3447E-010 2305E+06 1.618  
0.3507E-010 2301E+06 1.612  
0.3579E-010 2308E+06 1.545  
0.3643E-010 2328E+06 1.564  
0.3703E-010 2340E+06 1.631  
0.3764E-010 2343E+06 1.612  
0.3839E-010 2344E+06 1.564

809 5 5 85 1 0.6895E+06 274.1 0.2012E+05 2.98  
9-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIM, BS

0.8900E-040 7349E+05 53.45  
0.1277E-030 9212E+05 55.20  
0.4444E-030 1033E+06 60.58



0.8082E-030.1053E+06 60.05  
0.1303E-020.1053E+06 55.97  
0.1875E-020.1010E+06 50.59  
0.2440E-020.9641E+05 46.93  
0.2842E-020.9443E+05 44.55  
0.3523E-020.9335E+05 39.79  
0.4064E-020.9424E+05 32.44  
0.4520E-020.1023E+06 25.11  
0.4907E-020.1065E+06 21.87  
0.5580E-020.1181E+06 16.25  
0.5781E-020.1261E+06 13.66  
0.6152E-020.1315E+06 12.22  
0.6477E-020.1439E+06 9.160  
0.6655E-020.1455E+06 9.075  
0.7250E-020.1580E+06 7.340  
0.7552E-020.1653E+06 7.150  
0.7830E-020.1708E+06 6.748  
0.8225E-020.1763E+06 6.346  
0.8588E-020.1833E+06 6.102  
0.8751E-020.1897E+06 5.960  
0.8983E-020.1922E+06 5.682  
0.9710E-020.2043E+06 4.926  
0.1003E-010.2129E+06 4.744  
0.1031E-010.2138E+06 4.760  
0.1072E-010.2321E+06 4.555  
0.1093E-010.2328E+06 4.201  
0.1140E-010.2384E+06 3.336  
0.1214E-010.2541E+06 3.026  
0.1239E-010.2557E+06 3.038  
0.1300E-010.2700E+06 2.928  
0.1328E-010.2747E+06 2.993  
0.1365E-010.2833E+06 2.928  
0.1395E-010.2906E+06 2.928  
0.1422E-010.2900E+06 3.007  
0.1469E-010.2991E+06 3.001  
0.1499E-010.3054E+06 2.922  
0.1532E-010.3110E+06 3.014  
0.1611E-010.3211E+06 3.062  
0.1645E-010.3250E+06 3.062  
0.1703E-010.3251E+06 3.056  
0.1777E-010.3269E+06 2.977  
0.1828E-010.3291E+06 3.056  
0.1880E-010.3270E+06 3.007  
0.1940E-010.3253E+06 3.056  
0.2003E-010.3187E+06 3.836  
0.2053E-010.2885E+06 3.367  
0.2075E-010.2802E+06 3.664  
0.2089E-010.2625E+06 3.324  
0.2096E-010.2548E+06 3.324  
0.2101E-010.2489E+06 3.123  
0.2114E-010.2428E+06 2.270  
0.2115E-010.2424E+06 1.527  
0.2249E-010.2374E+06 4.223  
0.2249E-010.2363E+06 1.044  
0.2320E-010.2372E+06 1.080  
0.2367E-010.2363E+06 1.080  
0.2458E-010.2373E+06 1.117  
0.2491E-010.2372E+06 1.068  
0.2564E-010.2378E+06 1.190  
0.2621E-010.2371E+06 1.080  
0.2679E-010.2365E+06 1.184  
0.2708E-010.2383E+06 1.135  
0.2787E-010.2357E+06 1.031  
0.2856E-010.2362E+06 1.166  
0.2914E-010.2352E+06 1.208  
0.2972E-010.2352E+06 1.038  
0.3043E-010.2350E+06 1.135  
0.3098E-010.2351E+06 1.062  
0.3163E-010.2356E+06 1.147  
0.3233E-010.2368E+06 1.147  
0.3296E-010.2365E+06 1.251

0.3363E-010.2363E+06 1.172  
0.3410E-010.2358E+06 1.281  
0.3444E-010.2371E+06 1.348  
0.3514E-010.2374E+06 1.360  
0.3570E-010.2364E+06 1.348  
0.3631E-010.2359E+06 1.434  
0.3689E-010.2358E+06 1.330  
0.3754E-010.2357E+06 1.348  
0.3813E-010.2345E+06 1.354  
0.3863E-010.2340E+06 1.342

730 3 6 85 5 0.6085E+06 273.5  
30 JUL 85 COMRA TEST, 20 DEG F IM, BS

0.8711E-020.1280E+06 7.842  
0.9059E-020.1308E+06 7.805  
0.9141E-020.1337E+06 7.712  
0.9290E-020.1368E+06 6.989  
0.9609E-020.1391E+06 6.102  
0.1026E-010.1490E+06 5.221  
0.1090E-010.1573E+06 5.211  
0.1150E-010.1634E+06 5.190  
0.1214E-010.1652E+06 4.788  
0.1276E-010.1854E+06 4.797  
0.1338E-010.1941E+06 4.198  
0.1402E-010.2078E+06 4.202  
0.1463E-010.2147E+06 4.209  
0.1495E-010.2161E+06 4.529  
0.1558E-010.2298E+06 3.751  
0.1619E-010.2403E+06 3.751  
0.1683E-010.2470E+06 3.749  
0.1747E-010.2560E+06 3.744  
0.1808E-010.2615E+06 3.744  
0.1872E-010.2701E+06 3.720  
0.1934E-010.2802E+06 3.228  
0.1967E-010.2805E+06 3.219  
0.1998E-010.2837E+06 2.650  
0.2005E-010.2827E+06 2.731

807 2 7 90 1 0.6088E+06 271.1 0.2029E+05 3.98  
7 AUG 85 TOTAL HEAD SURVEY, 20 DEG F IM, BS

0.8900E-040.1064E+06 64.78  
0.2895E-030.1628E+06 65.10  
0.8987E-030.1673E+06 64.53  
0.1500E-020.1576E+06 61.08  
0.1867E-020.1502E+06 59.91  
0.2032E-020.1442E+06 59.60  
0.2456E-020.1380E+06 58.83  
0.2780E-020.1308E+06 58.42  
0.3073E-020.1251E+06 58.16  
0.3336E-020.1181E+06 57.44  
0.3713E-020.1129E+06 56.47  
0.3999E-020.1071E+06 54.77  
0.4284E-020.1036E+06 53.73  
0.4870E-020.9697E+05 49.18  
0.5533E-020.9206E+05 44.19  
0.6220E-020.9202E+05 40.92  
0.6775E-020.9387E+05 35.99  
0.7492E-020.9826E+05 30.30  
0.8094E-020.1067E+06 25.48  
0.8618E-020.1138E+06 20.04  
0.9266E-020.1261E+06 15.74  
0.9644E-020.1335E+06 15.11  
0.9890E-020.1377E+06 15.11  
0.1059E-010.1532E+06 12.16  
0.1085E-010.1592E+06 11.34  
0.1120E-010.1691E+06 10.78  
0.1121E-010.1738E+06 10.21  
0.1153E-010.1749E+06 10.05  
0.1206E-010.1936E+06 9.773  
0.1229E-010.1908E+06 9.553  
0.1263E-010.2083E+06 9.529  
0.1280E-010.2082E+06 9.425  
0.1335E-010.2175E+06 9.370  
0.1375E-010.2337E+06 9.267  
0.1385E-010.2305E+06 9.248  
0.1406E-010.2421E+06 9.224  
0.1427E-010.2459E+06 9.248  
0.1460E-010.2500E+06 9.279  
0.1509E-010.2620E+06 9.212  
0.1512E-010.2626E+06 9.206  
0.1569E-010.2862E+06 9.285  
0.1584E-010.2868E+06 9.297  
0.1647E-010.2961E+06 9.218

0.1684E-010.1028E+06 9.267  
0.1708E-010.1189E+06 9.230  
0.1736E-010.1308E+06 9.157  
0.1732E-010.1321E+06 9.261  
0.1744E-010.1321E+06 9.218  
0.1807E-010.1339E+06 9.297  
0.1837E-010.1364E+06 9.547  
0.1874E-010.1350E+06 9.913  
0.1888E-010.1355E+06 9.937  
0.1916E-010.1363E+06 10.32  
0.1954E-010.1369E+06 10.39  
0.1989E-010.1379E+06 10.44  
0.2001E-010.1375E+06 10.43  
0.2024E-010.1389E+06 10.61  
0.2049E-010.1390E+06 10.63  
0.2080E-010.1395E+06 10.94  
0.2144E-010.1406E+06 11.22  
0.2174E-010.1410E+06 11.28  
0.2211E-010.1413E+06 11.24  
0.2275E-010.1417E+06 11.58  
0.2330E-010.1427E+06 11.88  
0.2396E-010.1429E+06 12.19  
0.2455E-010.1432E+06 12.40  
0.2521E-010.1435E+06 12.83  
0.2587E-010.1439E+06 13.13  
0.2649E-010.1436E+06 13.96  
0.2709E-010.1428E+06 15.20  
0.2772E-010.1412E+06 16.21  
0.2813E-010.1405E+06 16.71  
0.2813E-010.1400E+06 16.92  
0.2844E-010.1397E+06 17.13  
0.2914E-010.1383E+06 17.67  
0.2976E-010.1369E+06 18.50  
0.3033E-010.1356E+06 18.75  
0.3094E-010.1350E+06 18.87  
0.3168E-010.1349E+06 18.93  
0.3234E-010.1343E+06 19.23  
0.3296E-010.1346E+06 19.45  
0.3359E-010.1400E+06 19.46  
0.3417E-010.1401E+06 20.90  
0.3468E-010.1402E+06 20.92  
0.3535E-010.1403E+06 20.97  
0.3616E-010.1403E+06 20.99  
0.3679E-010.1405E+06 21.01  
0.3746E-010.1405E+06 20.90  
0.3810E-010.1402E+06 21.00  
0.3871E-010.1407E+06 20.97

8121 3 8 72 1 0.6899E+06 273.2 0.2002E+05 4.98  
12-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIM, BS

0.8900E-040.1753E+06 48.58  
0.5850E-030.2428E+06 48.41  
0.1174E-020.2466E+06 47.34  
0.1848E-020.2409E+06 44.15  
0.2429E-020.2358E+06 43.16  
0.3094E-020.2322E+06 41.76  
0.3731E-020.2279E+06 40.29  
0.4367E-020.2248E+06 34.53  
0.4940E-020.2201E+06 36.81  
0.5599E-020.2212E+06 34.70  
0.6258E-020.2196E+06 33.13  
0.6793E-020.2191E+06 31.29  
0.7444E-020.2217E+06 29.24  
0.8157E-020.2299E+06 26.84  
0.8769E-020.2345E+06 25.25  
0.9404E-020.2399E+06 24.03  
0.9970E-020.2457E+06 22.31  
0.1079E-010.2512E+06 20.52  
0.1126E-010.2486E+06 20.04  
0.1164E-010.2497E+06 19.05

0.1226E-010.2808E+06 18.27  
0.1259E-010.2876E+06 17.94  
0.1292E-010.2910E+06 17.64  
0.1347E-010.3046E+06 16.43  
0.1379E-010.3068E+06 16.19  
0.1439E-010.3291E+06 15.22  
0.1453E-010.3283E+06 15.11  
0.1524E-010.3422E+06 15.02  
0.1561E-010.3555E+06 14.86  
0.1571E-010.3536E+06 14.66  
0.1639E-010.3646E+06 13.69  
0.1664E-010.3802E+06 13.57  
0.1687E-010.3754E+06 13.60  
0.1704E-010.3811E+06 13.58  
0.1775E-010.3827E+06 13.60  
0.1777E-010.3922E+06 13.12  
0.1812E-010.3982E+06 12.24  
0.1840E-010.4025E+06 12.15  
0.1901E-010.4073E+06 12.05  
0.1965E-010.4120E+06 11.97  
0.2028E-010.4134E+06 11.95  
0.2094E-010.4168E+06 12.05  
0.2163E-010.4174E+06 11.97  
0.2222E-010.4192E+06 12.01  
0.2294E-010.4191E+06 11.96  
0.2355E-010.4154E+06 11.96  
0.2421E-010.4042E+06 11.99  
0.2445E-010.3919E+06 12.07  
0.2462E-010.3948E+06 12.22  
0.2498E-010.3870E+06 12.58  
0.2504E-010.3878E+06 12.79  
0.2565E-010.3810E+06 13.49  
0.2627E-010.3796E+06 14.41  
0.2687E-010.3797E+06 14.98  
0.2751E-010.3813E+06 15.33  
0.2814E-010.3820E+06 15.61  
0.2872E-010.3823E+06 15.70  
0.2938E-010.3846E+06 15.78  
0.2999E-010.3864E+06 15.68  
0.3060E-010.3870E+06 15.74  
0.3121E-010.3862E+06 15.90  
0.3192E-010.3880E+06 15.97  
0.3255E-010.3905E+06 16.01  
0.3315E-010.3908E+06 16.06  
0.3387E-010.3894E+06 16.03  
0.3444E-010.3884E+06 16.48  
0.3510E-010.3890E+06 17.45  
0.3575E-010.3887E+06 17.66  
0.3643E-010.3901E+06 17.81  
0.3704E-010.3921E+06 17.85  
0.3776E-010.3949E+06 17.73  
0.3839E-010.3949E+06 17.70

807 6 9 73 1 0.6889E+06 270.6 0.2030E+05 2.98  
7-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIM, BS

0.8900E-040.4933E+05 58.11  
0.4363E-030.5485E+05 48.26  
0.9302E-030.4402E+05 33.23  
0.1254E-020.6928E+05 26.92  
0.1583E-020.7703E+05 17.90  
0.1987E-020.8410E+05 13.88  
0.2234E-020.9453E+05 10.32  
0.2412E-020.9772E+05 10.10  
0.2690E-020.1044E+06 8.827  
0.2968E-020.1118E+06 7.986  
0.3338E-020.1189E+06 7.401  
0.3724E-020.1252E+06 6.853  
0.3870E-020.1331E+06 6.566  
0.4303E-020.1397E+06 6.085  
0.4627E-020.1469E+06 5.944

0.4869E-020.1523E+06 5.646  
0.5159E-020.1567E+06 5.463  
0.5838E-020.1691E+06 5.164  
0.6078E-020.1759E+06 5.195  
0.6548E-020.1837E+06 5.109  
0.6803E-020.1880E+06 5.134  
0.7405E-020.2020E+06 5.201  
0.7629E-020.2110E+06 5.250  
0.7814E-020.2129E+06 5.231  
0.8539E-020.2192E+06 5.146  
0.9072E-020.2268E+06 5.189  
0.9667E-020.2234E+06 4.994  
0.1015E-010.2179E+06 4.689  
0.1111E-010.2158E+06 2.830  
0.1233E-010.2268E+06 2578  
0.1251E-010.2263E+06 3187  
0.1327E-010.2311E+06 2151  
0.1380E-010.2329E+06 2456  
0.1434E-010.2346E+06 1724  
0.1516E-010.2343E+06 1785  
0.1582E-010.2361E+06 1724  
0.1647E-010.2361E+06 2212  
0.1707E-010.2368E+06 2212  
0.1771E-010.2379E+06 2395  
0.1818E-010.2366E+06 3248  
0.1893E-010.2370E+06 2029  
0.1946E-010.2366E+06 2517  
0.2012E-010.2376E+06 2456  
0.2065E-010.2370E+06 3126  
0.2142E-010.2358E+06 2029  
0.2199E-010.2362E+06 2700  
0.2268E-010.2367E+06 2456  
0.2320E-010.2370E+06 3309  
0.2395E-010.2364E+06 2029  
0.2442E-010.2360E+06 2700  
0.2520E-010.2369E+06 2273  
0.2569E-010.2365E+06 3248  
0.2639E-010.2371E+06 2090  
0.2698E-010.2358E+06 3065  
0.2767E-010.2369E+06 2131  
0.2831E-010.2368E+06 2334  
0.2891E-010.2387E+06 3065  
0.2943E-010.2372E+06 3248  
0.3023E-010.2372E+06 2517  
0.3073E-010.2370E+06 3614  
0.3138E-010.2367E+06 3126  
0.3212E-010.2372E+06 2517  
0.3262E-010.2364E+06 3675  
0.3332E-010.2374E+06 3431  
0.3404E-010.2362E+06 2578  
0.3460E-010.2365E+06 3492  
0.3531E-010.2365E+06 3187  
0.3584E-010.2363E+06 3431  
0.3641E-010.2379E+06 2761  
0.3718E-010.2372E+06 3136  
0.3783E-010.2375E+06 3492  
0.3847E-010.2367E+06 3248

8081 311 88 1 0.6887E+06 273.2 0.1989E+05 2.98  
8-AUG-85 TOTAL HEAD SURVEY, 20 DEG FIM, BS

0.8900E-040.1616E+06 58.10  
0.1586E-030.1794E+06 57.43  
0.8157E-030.1750E+06 51.82  
0.1450E-020.1687E+06 49.17  
0.2099E-020.1644E+06 47.34  
0.2764E-020.1587E+06 45.73  
0.3359E-020.1574E+06 43.22  
0.3970E-020.1551E+06 40.54  
0.4604E-020.1583E+06 37.24

0.5277E-020.1842E+06 34.40  
0.5895E-020.1709E+06 31.37  
0.6506E-020.1834E+06 27.73  
0.6807E-020.1860E+06 25.64  
0.7564E-020.1940E+06 23.38  
0.8013E-020.2091E+06 20.92  
0.8106E-020.2150E+06 20.55  
0.8717E-020.2218E+06 19.03  
0.8949E-020.2263E+06 18.72  
0.9281E-020.2346E+06 17.98  
0.9683E-020.2413E+06 17.51  
0.9985E-020.2468E+06 17.00  
0.1032E-010.2562E+06 16.76  
0.1046E-010.2588E+06 16.34  
0.1065E-010.2646E+06 15.97  
0.1110E-010.2719E+06 15.51  
0.1131E-010.2806E+06 15.41  
0.1169E-010.2896E+06 15.37  
0.1177E-010.2924E+06 15.31  
0.1221E-010.3041E+06 14.93  
0.1222E-010.2999E+06 14.77  
0.1267E-010.3165E+06 14.24  
0.1282E-010.3197E+06 14.16  
0.1313E-010.3340E+06 14.07  
0.1319E-010.3295E+06 13.97  
0.1343E-010.3383E+06 13.64  
0.1356E-010.3429E+06 13.51  
0.1366E-010.3446E+06 13.23  
0.1428E-010.3651E+06 13.11  
0.1444E-010.3638E+06 13.06  
0.1513E-010.3834E+06 12.07  
0.1524E-010.3824E+06 11.74  
0.1590E-010.3897E+06 11.53  
0.1657E-010.4039E+06 11.49  
0.1685E-010.4044E+06 11.49  
0.1745E-010.4066E+06 11.59  
0.1804E-010.4047E+06 11.64  
0.1876E-010.3940E+06 11.62  
0.1909E-010.3927E+06 11.57  
0.1961E-010.3848E+06 12.47  
0.2031E-010.3832E+06 14.18  
0.2102E-010.3814E+06 15.48  
0.2165E-010.3845E+06 16.08  
0.2225E-010.3848E+06 16.36  
0.2289E-010.3882E+06 16.76  
0.2347E-010.3894E+06 16.82  
0.2407E-010.3887E+06 17.05  
0.2482E-010.3887E+06 17.31  
0.2534E-010.3878E+06 17.61  
0.2598E-010.3909E+06 17.88  
0.2661E-010.3954E+06 18.38  
0.2730E-010.3947E+06 18.58  
0.2797E-010.3975E+06 18.79  
0.2848E-010.3978E+06 18.94  
0.2910E-010.3975E+06 18.87  
0.2973E-010.3990E+06 18.89  
0.3044E-010.3991E+06 18.92  
0.3103E-010.3973E+06 19.14  
0.3165E-010.3994E+06 19.23  
0.3237E-010.3996E+06 19.30  
0.3291E-010.3990E+06 19.42  
0.3371E-010.3995E+06 19.31  
0.3429E-010.4000E+06 19.38  
0.3490E-010.4001E+06 19.42  
0.3550E-010.4013E+06 19.46  
0.3627E-010.4032E+06 19.41  
0.3688E-010.4073E+06 19.45  
0.3748E-010.4066E+06 19.65  
0.3823E-010.4116E+06 19.96  
0.3872E-010.4136E+06 20.39  
0.3942E-010.4126E+06 20.45  
0.4007E-010.4148E+06 20.55

0.4072E-010.4125E+06 20.53  
0.4140E-010.4128E+06 20.37  
0.4200E-010.4104E+06 20.40  
0.4276E-010.4101E+06 20.33  
0.4330E-010.4124E+06 20.42  
0.4395E-010.4129E+06 20.36  
0.4460E-010.4133E+06 20.29

PART 1400

20 DEGREE FIN DATA - SURFACE PRESSURES

PO=100 PSI

FREESTREAM MACH NUMBER=2.93

FORMAT OF DATA:

Z(CM) NUM. OF POINTS  
X PS/PINF

X IN CM, PS/PINF IS SURFACE PRESSURE DIVIDED BY FREESTREAM STATIC PRESSURE

\*\*\*\*\*SURFACE PRESSURE DATA\*\*\*\*\*

3.56 18

0.3 1.01  
1.4 1.02  
2.1 1.62  
2.4 1.73  
2.7 1.81  
3.0 1.85  
3.3 1.90  
3.6 1.89  
3.9 1.84  
4.2 1.81  
4.5 1.80  
5.7 2.35  
6.4 2.64  
7.1 3.07  
7.7 3.32  
8.3 3.54  
8.9 3.77

6.10 21

0.6 1.00  
1.2 0.99  
1.8 0.98  
2.4 0.99  
3.0 0.98  
3.3 1.03  
3.9 1.31  
5.0 1.78  
5.6 1.90  
6.2 1.93  
6.8 1.89  
7.4 1.83  
8.0 1.87  
8.9 2.08

9.5 2.34  
10.1 2.35  
10.7 2.95  
11.3 3.21  
12.2 3.58  
13.1 3.48  
13.7 3.50

8.64 18

2.0 1.00  
2.9 1.00  
3.8 1.00  
5.0 1.00  
6.3 1.63  
8.2 1.95  
9.5 1.96  
10.1 1.97  
10.8 1.94  
11.4 1.90  
12.1 1.99  
12.7 2.06  
13.3 2.30  
14.0 2.52  
14.7 2.82  
15.4 3.10  
16.2 3.22  
16.8 3.24

\*\*\*\*\*END OF FILE\*\*\*\*\*

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Ref.: 46

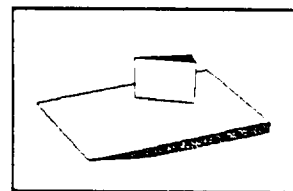
Author: Kim, K-S, *et al*

Geometry: 3-D Fin

Mach number: 3, 4

Data:  $p_{wall}$ ,  $c_f$ , surface-flow angles

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Kim, K-S, and Settles, G. S., "Skin Friction Measurements by Laser Interferometry in Swept Shock/ Turbulent Boundary-Layer Interactions," AIAA Paper 88-0497, Jan. 1988, and *AIAA Journal*, Vol. 28, Jan. 1990, pp. 133-139.

Kim, K-S., Lee, Y., Alvi, F. S., Settles, G. S., and Hortsman, C. C., "Laser Skin Friction Measurements and CFD Comparison of Weak-to-Strong Swept Shock/Boundary Layer Interactions," AIAA Paper 90-0378.

Kim, K.-S., and Settles, G. S., "Skin Friction Measurements by Laser Interferometry," Ch. 3 of AGARDograph No. 315, A Survey of Measurements and Measuring Techniques in Rapidly Distorted Compressible Turbulent Boundary Layers, eds. H. H. Fernholz, A. J. Smits, and J.-P. Dussauge, November 1988.

Kim, K.-S., "Skin Friction Measurements by Laser Interferometry in Supersonic Flows," Ph.D. Dissertation, M.E. Dept., Penn State University, May 1989.

(Incoming Boundary-Layers:)

Lu, F. K., "Fin-Generated Shock-Wave Boundary-Layer Interactions," Ph.D. Dissertation, M.E. Dept., Penn State University, May 1988.

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The data consist of measured surface pressure, skin friction, and surface flow angles from sharp-fin-induced shock wave/turbulent boundary-layer interactions tested at Mach 3 and Mach 4 in the Penn State Gas Dynamics Laboratory over the period 1985-1990. Tabulated flat-plate incoming boundary-layer profiles are also given. The geometry consisted of a sharp, unswept-leading-edge fin mounted at angle-of-attack  $\alpha$  to the freestream, with its leading-edge 216 mm downstream from the leading edge of a flat plate upon which the turbulent boundary layer was generated.

Assessments of the confidence limits placed on these data are as follows:

#### 1) Skin Friction Distributions

Each reported skin friction point is an ensemble of several experimental tests. The tolerances reported here reflect the repeatability of the measurement, and are stated in terms of one standard deviation from the mean. No absolute accuracy statement can be made, since there exists no prior knowledge of skin friction in such interacting flows and no other means of measuring it accurately for comparison. However, based on experience with a flat-plate boundary layer calibration (AGARDograph

315 mentioned above), the authors believe that the given tolerances also represent the accuracy of the data, since repeatability error appears to be the largest of the several possible error sources.

## 2) Surface Streamline Angles

These data were extracted by hand from kerosene-lampblack surface flow visualization patterns using a protractor. Due to inherent errors in this process, these data are believed to carry an overall accuracy of  $\pm 5\%$ .

## 3) Surface Pressure Distributions

These data were obtained in late 1989/early 1990 by S. Garg and Y. Lee and remain unpublished at this writing. An extensive error analysis has been performed. First, a weak streamwise pressure gradient which existed on the flat plate with no fin in place has been removed from the data prior to their tabulation here. Otherwise, errors due to electronic noise, repeatability, and calibration uncertainty are believed to be within the range of  $\pm 3\%$ .

The skin friction values tabulated below are given as  $cf$  vs.  $\beta$ , where  $cf$  is the skin friction coefficient defined as the absolute value (magnitude) of the wall shear stress normalized by the undisturbed freestream dynamic pressure.  $\beta$  is defined as the conical ray angle, measured from the fin leading edge, with respect to the freestream direction. For each of 4 combinations of fin angle and Mach number, Laser Interferometer Skin Friction (LISF) data were taken along a single circular arc at radius  $R$  from the fin leading edge.  $R$  was 114 mm (4.5 inches) in the case of Mach 3,  $\alpha = 10$  deg. For the other three cases,  $R$  was 89 mm (3.5 inches). The angle  $\beta$  is given in degrees below. The error bars are given without sign, and are one-sided rather than total error estimates (ie, for the first data point given below,  $cf = 0.001118 \pm 0.00002634$ ).

The surface streamline angle values given below represent the local angles of surface flow pattern streamlines,  $\phi$ , measured with respect to the freestream direction, vs. the conical ray angle  $\beta$ . All values are in degrees.

The surface pressure data tabulated below were read from pressure taps arranged along a circular-arc segment at radius  $R = 101.6$  mm (4 inches) from the fin leading edge. The pressure data are given in normalized form, ie  $P_w/P_{\infty}$ , where  $P_w$  is the measured wall pressure and  $P_{\infty}$  is the undisturbed flat-plate surface pressure measured at the upstream limit of the interaction. Angles  $\beta$  and  $\alpha$  are in degrees.

The incoming boundary-layer profiles given are largely self-explanatory. Tabulated values of UPLUS and YPLUS refer to the transformed incompressible wall-wake coordinates of the profile, which were obtained by way of a least-squares curvefit of the wall-wake similarity law to the boundary-layer profile after having transformed it to the incompressible plane by way of the Van Driest transformation.

Ongoing work on this interaction at the Penn State Gas Dynamics Lab is expected to produce additional data within the near future. These data will include heat transfer distributions and flowfield density profiles via conical holographic interferometry.

MACH 6, ALPHA = 20 DEGREES  
BETA PHI  
49.5 0.0  
48.8 45.5

27	2.896	0.918	2.862	0.9277	0.9471	0.1791	0.1665E-04	26.8163
28	2.972 <th>0.952</th> <th>2.875</th> <th>0.9090</th> <th>0.9891</th> <th>0.372</th> <th>0.1535E-04</th> <th>26.8643</th>	0.952	2.875	0.9090	0.9891	0.372	0.1535E-04	26.8643
29	3.023 <th>0.968</th> <th>2.881</th> <th>0.9095</th> <th>0.9971</th> <th>0.3759</th> <th>0.1529E-04</th> <th>26.8821</th>	0.968	2.881	0.9095	0.9971	0.3759	0.1529E-04	26.8821
30	3.122 <th>1.000</th> <th>2.897</th> <th>1.0000</th> <th>1.0000</th> <th>0.3747</th> <th>0.1579E-04</th> <th>26.9006</th>	1.000	2.897	1.0000	1.0000	0.3747	0.1579E-04	26.9006
31	3.150 <th>1.009</th> <th>2.890</th> <th>1.0005</th> <th>0.9999</th> <th>0.3744</th> <th>0.1559E-04</th> <th>26.9199</th>	1.009	2.890	1.0005	0.9999	0.3744	0.1559E-04	26.9199
32	3.251 <th>1.041</th> <th>2.898</th> <th>1.0011</th> <th>0.9999</th> <th>0.3732</th> <th>0.1645E-04</th> <th>26.9394</th>	1.041	2.898	1.0011	0.9999	0.3732	0.1645E-04	26.9394
33	3.327 <th>1.062</th> <th>2.882</th> <th>0.9968</th> <th>1.0010</th> <th>0.3758</th> <th>0.1681E-04</th> <th>26.8562</th>	1.062	2.882	0.9968	1.0010	0.3758	0.1681E-04	26.8562
34	3.378 <th>1.082</th> <th>2.903</th> <th>1.0013</th> <th>0.9963</th> <th>0.3724</th> <th>0.1709E-04</th> <th>26.9474</th>	1.082	2.903	1.0013	0.9963	0.3724	0.1709E-04	26.9474
35	3.429 <th>1.098</th> <th>2.893</th> <th>0.9998</th> <th>0.9995</th> <th>0.3740</th> <th>0.1735E-04</th> <th>26.8931</th>	1.098	2.893	0.9998	0.9995	0.3740	0.1735E-04	26.8931
36	3.505 <th>1.123</th> <th>2.885</th> <th>0.9983</th> <th>1.0008</th> <th>0.3753</th> <th>0.1773E-04</th> <th>26.8492</th>	1.123	2.885	0.9983	1.0008	0.3753	0.1773E-04	26.8492
37	3.556 <th>1.139</th> <th>2.884</th> <th>0.9978</th> <th>0.9947</th> <th>0.3754</th> <th>0.1799E-04</th> <th>26.8152</th>	1.139	2.884	0.9978	0.9947	0.3754	0.1799E-04	26.8152
38	3.611 <th>1.147</th> <th>2.884</th> <th>0.9967</th> <th>0.9949</th> <th>0.3754</th> <th>0.1812E-04</th> <th>26.8157</th>	1.147	2.884	0.9967	0.9949	0.3754	0.1812E-04	26.8157
39	3.632 <th>1.146</th> <th>2.874</th> <th>0.9970</th> <th>0.9942</th> <th>0.3771</th> <th>0.1837E-04</th> <th>26.7551</th>	1.146	2.874	0.9970	0.9942	0.3771	0.1837E-04	26.7551
40	3.714 <th>1.196</th> <th>2.885</th> <th>0.9972</th> <th>0.9891</th> <th>0.3753</th> <th>0.1880E-04</th> <th>26.7991</th>	1.196	2.885	0.9972	0.9891	0.3753	0.1880E-04	26.7991
41	3.810 <th>1.220</th> <th>2.909</th> <th>1.0000</th> <th>1.0000</th> <th>0.3715</th> <th>0.1927E-04</th> <th>26.9006</th>	1.220	2.909	1.0000	1.0000	0.3715	0.1927E-04	26.9006

```

MACH 4. (3.88)
RUN NO.: 5200
X, Z = 0.178, 0.000 =
M1 = 3.876
PU = 647.34 m/s
U1 = 1400.40
P11 = 110
P101 = 0.523 kg/m3
P1 = 0.999
RECOVERY FACTOR = 0.89
10.483 kPa
206.50 kPa
108.2 K
0.65
(Edge criterion is UE = 0.999 = Ut)

```

I	I <sub>1</sub>	I <sub>2</sub>	I <sub>3</sub>	I <sub>4</sub>	I <sub>5</sub>	I <sub>6</sub>	I <sub>7</sub>	I <sub>8</sub>	I <sub>9</sub>	I <sub>10</sub>	I <sub>11</sub>	I <sub>12</sub>	I <sub>13</sub>	I <sub>14</sub>	I <sub>15</sub>	I <sub>16</sub>	I <sub>17</sub>	I <sub>18</sub>	I <sub>19</sub>	I <sub>20</sub>	I <sub>21</sub>	I <sub>22</sub>	I <sub>23</sub>	I <sub>24</sub>	I <sub>25</sub>	I <sub>26</sub>	I <sub>27</sub>	I <sub>28</sub>	I <sub>29</sub>	I <sub>30</sub>	I <sub>31</sub>	I <sub>32</sub>	I <sub>33</sub>	I <sub>34</sub>	I <sub>35</sub>	I <sub>36</sub>	I <sub>37</sub>	I <sub>38</sub>	I <sub>39</sub>	I <sub>40</sub>	I <sub>41</sub>	I <sub>42</sub>	I <sub>43</sub>	I <sub>44</sub>	I <sub>45</sub>	I <sub>46</sub>	I <sub>47</sub>	I <sub>48</sub>	I <sub>49</sub>	I <sub>50</sub>	I <sub>51</sub>	I <sub>52</sub>	I <sub>53</sub>	I <sub>54</sub>	I <sub>55</sub>	I <sub>56</sub>	I <sub>57</sub>	I <sub>58</sub>	I <sub>59</sub>	I <sub>60</sub>	I <sub>61</sub>	I <sub>62</sub>	I <sub>63</sub>	I <sub>64</sub>	I <sub>65</sub>	I <sub>66</sub>	I <sub>67</sub>	I <sub>68</sub>	I <sub>69</sub>	I <sub>70</sub>	I <sub>71</sub>	I <sub>72</sub>	I <sub>73</sub>	I <sub>74</sub>	I <sub>75</sub>	I <sub>76</sub>	I <sub>77</sub>	I <sub>78</sub>	I <sub>79</sub>	I <sub>80</sub>	I <sub>81</sub>	I <sub>82</sub>	I <sub>83</sub>	I <sub>84</sub>	I <sub>85</sub>	I <sub>86</sub>	I <sub>87</sub>	I <sub>88</sub>	I <sub>89</sub>	I <sub>90</sub>	I <sub>91</sub>	I <sub>92</sub>	I <sub>93</sub>	I <sub>94</sub>	I <sub>95</sub>	I <sub>96</sub>	I <sub>97</sub>	I <sub>98</sub>	I <sub>99</sub>	I <sub>100</sub>	I <sub>101</sub>	I <sub>102</sub>	I <sub>103</sub>	I <sub>104</sub>	I <sub>105</sub>	I <sub>106</sub>	I <sub>107</sub>	I <sub>108</sub>	I <sub>109</sub>	I <sub>110</sub>	I <sub>111</sub>	I <sub>112</sub>	I <sub>113</sub>	I <sub>114</sub>	I <sub>115</sub>	I <sub>116</sub>	I <sub>117</sub>	I <sub>118</sub>	I <sub>119</sub>	I <sub>120</sub>	I <sub>121</sub>	I <sub>122</sub>	I <sub>123</sub>	I <sub>124</sub>	I <sub>125</sub>	I <sub>126</sub>	I <sub>127</sub>	I <sub>128</sub>	I <sub>129</sub>	I <sub>130</sub>	I <sub>131</sub>	I <sub>132</sub>	I <sub>133</sub>	I <sub>134</sub>	I <sub>135</sub>	I <sub>136</sub>	I <sub>137</sub>	I <sub>138</sub>	I <sub>139</sub>	I <sub>140</sub>	I <sub>141</sub>	I <sub>142</sub>	I <sub>143</sub>	I <sub>144</sub>	I <sub>145</sub>	I <sub>146</sub>	I <sub>147</sub>	I <sub>148</sub>	I <sub>149</sub>	I <sub>150</sub>	I <sub>151</sub>	I <sub>152</sub>	I <sub>153</sub>	I <sub>154</sub>	I <sub>155</sub>	I <sub>156</sub>	I <sub>157</sub>	I <sub>158</sub>	I <sub>159</sub>	I <sub>160</sub>	I <sub>161</sub>	I <sub>162</sub>	I <sub>163</sub>	I <sub>164</sub>	I <sub>165</sub>	I <sub>166</sub>	I <sub>167</sub>	I <sub>168</sub>	I <sub>169</sub>	I <sub>170</sub>	I <sub>171</sub>	I <sub>172</sub>	I <sub>173</sub>	I <sub>174</sub>	I <sub>175</sub>	I <sub>176</sub>	I <sub>177</sub>	I <sub>178</sub>	I <sub>179</sub>	I <sub>180</sub>	I <sub>181</sub>	I <sub>182</sub>	I <sub>183</sub>	I <sub>184</sub>	I <sub>185</sub>	I <sub>186</sub>	I <sub>187</sub>	I <sub>188</sub>	I <sub>189</sub>	I <sub>190</sub>	I <sub>191</sub>	I <sub>192</sub>	I <sub>193</sub>	I <sub>194</sub>	I <sub>195</sub>	I <sub>196</sub>	I <sub>197</sub>	I <sub>198</sub>	I <sub>199</sub>	I <sub>200</sub>	I <sub>201</sub>	I <sub>202</sub>	I <sub>203</sub>	I <sub>204</sub>	I <sub>205</sub>	I <sub>206</sub>	I <sub>207</sub>	I <sub>208</sub>	I <sub>209</sub>	I <sub>210</sub>	I <sub>211</sub>	I <sub>212</sub>	I <sub>213</sub>	I <sub>214</sub>	I <sub>215</sub>	I <sub>216</sub>	I <sub>217</sub>	I <sub>218</sub>	I <sub>219</sub>	I <sub>220</sub>	I <sub>221</sub>	I <sub>222</sub>	I <sub>223</sub>	I <sub>224</sub>	I <sub>225</sub>	I <sub>226</sub>	I <sub>227</sub>	I <sub>228</sub>	I <sub>229</sub>	I <sub>230</sub>	I <sub>231</sub>	I <sub>232</sub>	I <sub>233</sub>	I <sub>234</sub>	I <sub>235</sub>	I <sub>236</sub>	I <sub>237</sub>	I <sub>238</sub>	I <sub>239</sub>	I <sub>240</sub>	I <sub>241</sub>	I <sub>242</sub>	I <sub>243</sub>	I <sub>244</sub>	I <sub>245</sub>	I <sub>246</sub>	I <sub>247</sub>	I <sub>248</sub>	I <sub>249</sub>	I <sub>250</sub>	I <sub>251</sub>	I <sub>252</sub>	I <sub>253</sub>	I <sub>254</sub>	I <sub>255</sub>	I <sub>256</sub>	I <sub>257</sub>	I <sub>258</sub>	I <sub>259</sub>	I <sub>260</sub>	I <sub>261</sub>	I <sub>262</sub>	I <sub>263</sub>	I <sub>264</sub>	I <sub>265</sub>	I <sub>266</sub>	I <sub>267</sub>	I <sub>268</sub>	I <sub>269</sub>	I <sub>270</sub>	I <sub>271</sub>	I <sub>272</sub>	I <sub>273</sub>	I <sub>274</sub>	I <sub>275</sub>	I <sub>276</sub>	I <sub>277</sub>	I <sub>278</sub>	I <sub>279</sub>	I <sub>280</sub>	I <sub>281</sub>	I <sub>282</sub>	I <sub>283</sub>	I <sub>284</sub>	I <sub>285</sub>	I <sub>286</sub>	I <sub>287</sub>	I <sub>288</sub>	I <sub>289</sub>	I <sub>290</sub>	I <sub>291</sub>	I <sub>292</sub>	I <sub>293</sub>	I <sub>294</sub>	I <sub>295</sub>	I <sub>296</sub>	I <sub>297</sub>	I <sub>298</sub>	I <sub>299</sub>	I <sub>300</sub>	I <sub>301</sub>	I <sub>302</sub>	I <sub>303</sub>	I <sub>304</sub>	I <sub>305</sub>	I <sub>306</sub>	I <sub>307</sub>	I <sub>308</sub>	I <sub>309</sub>	I <sub>310</sub>	I <sub>311</sub>	I <sub>312</sub>	I <sub>313</sub>	I <sub>314</sub>	I <sub>315</sub>	I <sub>316</sub>	I <sub>317</sub>	I <sub>318</sub>	I <sub>319</sub>	I <sub>320</sub>	I <sub>321</sub>	I <sub>322</sub>
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	$P_w/P_{\text{infinity}}$
64.0	57.8
40.5	44.1
39.5	40.2
37.0	46.1
31.0	53.5
31.0	66.2
29.0	65.3
26.3	45.6
25.0	34.5
23.3	26.7
21.0	20.0

	BETA	ALPHA=10	ALPHA=16	ALPHA=20
1	1.931	14.0	1.905	16.0
2	1.864	18.0	1.844	18.0
3	1.797	22.0	1.782	20.0
4	1.730	26.0	1.720	22.0
5	1.663	30.0	1.658	24.0
6	1.596	34.0	1.597	26.0
7	1.529	38.0	1.530	28.0
8	1.462	42.0	1.462	30.0
9	1.395	46.0	1.395	32.0
10	1.328	50.0	1.328	34.0
11	1.261	54.0	1.261	36.0
12	1.194	58.0	1.194	38.0
13	1.127	62.0	1.127	40.0
14	1.060	66.0	1.060	42.0
15	0.993	70.0	0.993	44.0
16	0.926	74.0	0.926	46.0
17	0.859	78.0	0.859	48.0
18	0.792	82.0	0.792	50.0
19	0.725	86.0	0.725	52.0
20	0.658	90.0	0.658	54.0
21	0.591	94.0	0.591	56.0
22	0.524	98.0	0.524	58.0
23	0.457	102.0	0.457	60.0
24	0.390	106.0	0.390	62.0
25	0.323	110.0	0.323	64.0
26	0.256	114.0	0.256	66.0
27	0.189	118.0	0.189	68.0
28	0.122	122.0	0.122	70.0
29	0.055	126.0	0.055	72.0
30	0.000	130.0	0.000	74.0

INCOMING BOUNDARY-LAYER PROFILES									
MACH 3 (2.91)									
RUN NO.: 1177									
$x, z = 0.178, 0.040$ m									
$y_1$	=	2.909				RECOVERY FACTOR = 0.89			
$y_2$	=	601.07	m/s			RENI =	0.52912E+08	/m	
$y_3$	=	745.86	kPa			PU =	20.167	kPa	
$y_4$	=	110	K			P11 =	227.60	kPa	
$y_5$	=	108.2	K			T101 =	289.7	K	
$y_6$	=	0.656	kg/m <sup>3</sup>			P1 =	0.0882		
Edge criterion is UE = 0.999 * U1									

	Y <sub>1</sub>	m	V/Delta	MACH	NO. UJVE	RM/RADIAL	T/TITE	YPLUS	YMINUS	UPLUS
1	1.194	0.382	2.059		0.6647	0.4703	0.5451	0.46309E+03		22.4711
2	1.210	0.391	2.075		0.6698	0.4756	0.5376	0.46157E+03		22.5666
3	1.219	0.407	2.105		0.6763	0.4851	0.5302	0.46042E+03		22.7667
4	1.346	0.431	2.158		0.8921	0.6979	0.5121	0.64010E+03		23.1034
5	1.422	0.456	2.182		0.8921	0.7137	0.5121	0.71591E+03		23.2611
6	1.499	0.480	2.231		0.9000	0.7137	0.5033	0.75818E+03		23.5151
7	1.524	0.488	2.219		0.9008	0.7273	0.4971	0.77709E+03		23.6963
8	1.575	0.506	2.267		0.9089	0.7431	0.4930	0.77966E+03		23.7968
9	1.600	0.513	2.286		0.9122	0.7534	0.4890	0.80050E+03		23.9041
10	1.676	0.537	2.348		0.9235	0.7827	0.4757	0.84501E+03		24.2708
11	1.753	0.561	2.367		0.9264	0.7832	0.4716	0.86665E+03		24.3648
12	1.803	0.578	2.404		0.9327	0.7946	0.4638	0.91233E+03		24.5726
13	2.032	0.651	2.482		0.9451	0.8189	0.4481	0.94536E+03		24.9850
14	2.032	0.651	2.528		0.9522	0.8300	0.4393	0.10278E+04		25.2272
15	2.083	0.667	2.548		0.9552	0.8389	0.4351	0.10954E+04		25.3295
16	2.159	0.692	2.605		0.9639	0.8704	0.4242	0.10572E+04		25.6246
17	2.210	0.708	2.649		0.9702	0.8683	0.4151	0.1118E+04		25.8632
18	2.235	0.716	2.650		0.9702	0.8694	0.4159	0.1131E+04		25.8632
19	2.388	0.765	2.689		0.9757	0.9113	0.4083	0.12008E+04		26.0341
20	2.413	0.773	2.709		0.9783	0.9172	0.4051	0.1221E+04		26.1266
21	2.464	0.789	2.710		0.9781	0.9206	0.4051	0.12458E+04		26.1201
22	2.489	0.797	2.741		0.9821	0.9313	0.3996	0.1259E+04		26.2606
23	2.565	0.825	2.773		0.9864	0.9366	0.3940	0.12908E+04		26.4111
24	2.616	0.838	2.795		0.9890	0.9421	0.3903	0.13232E+04		26.5032
25	2.667	0.854	2.814		0.9914	0.9533	0.3871	0.1349E+04		26.5919
26	2.743	0.879	2.851		0.9963	0.9661	0.3808	0.1388E+04		26.7673

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Ref.: 13, 27

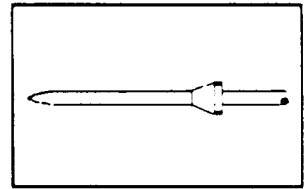
Author: Dunagan, S. E., Brown, J. L., *et al*

Geometry: Axisymmetric Ogive-Cylinder-Flare

Mach number: 3

Data:  $p_{wall}$ , flowfield surveys (LDV and holographic interferometry)

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Brown, J.L., Kussoy, M.I. and Coakley, T.J., "Turbulent Properties of Axisymmetric Shock-Wave/Boundary- Layer Interaction Flows," *Turbulent Shear-Layer/Shock-Wave Interactions*, edited by Delery, J., Springer-Verlag, Berlin, 1986, pp. 137-148.

Dunagan, S.E., Brown, J.L. and Miles, J.B., "Interferometric Data for a Shock-Wave/Boundary-Layer Interaction," *NASA TM 88227*, 1986.

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The data tabulated here and much of this discussion were provided by Dr. James L. Brown of NASA-Ames Research Center, who invites users to contact him if questions should arise. Supersonic flow over an axisymmetric ogive-cylinder-flare body was investigated by non-intrusive optical instrumentation as described below. Shock-induced separation on this model produced large-scale unsteady fluid motion.

#### INSTRUMENTS:

Instruments used included a 2D laser velocimeter using 0.55 micron polystyrene seed particles. The particle lag with this size particle is not entirely negligible but was felt to be acceptable with the shock and other velocity gradients being spread by about 0.5 cm in the *streamwise* direction. Other supersonic LV experiments have been reported with seed particles as large as 2 microns.

Another instrument used in these experiments was a Holographic Interferometer to obtain density measurements. With appropriate normal pressure gradient assumptions and a temperature-velocity relationship, density profiles may also be derived from the LV velocity data. Except for those regions, such as near the shock, where particle lag exists, these LV-derived density profiles agree quite well with the Holographic Interferometer-obtained density profiles.

#### EXPERIMENTAL CONDITIONS:

Freestream conditions are:

1.7 atmosphere	total pressure
270 degree K	total temperature
2.85	Mach number
1.928	$M^*$ , the freestream velocity normalized by the acoustic velocity at Mach 1
18,000,000	unit Reynolds number per meter



The oncoming boundary layer is developed on a 1 m long cylinder of 5.08 cm diameter (axis aligned with the free stream flow) and has (just prior to the shock/boundary layer interaction):

- 1.1 cm boundary layer thickness,
- 0.33 cm Compressible Displacement Thickness (includes density variation),
- 0.12 cm Incompressible Displacement Thickness (no density variation considered)

The shock/boundary layer interaction is generated by an axisymmetric cone placed on the 1 m long cylinder. Two cases are presented in the cited References:

- a. 12.5 degree axisymmetric cone on a 1 meter long cylinder.

(This case is unseparated and has a steady shock)

- b. 30 degree axisymmetric cone on a 1 meter long cylinder.

(This case is separated and has an unsteady shock)

Given the large quantity of data represented, the desire for reasonable brevity in this Report, and the greater interest in the separated-flow case, only the 30 degree case is tabulated below.

The velocity measurements were made and are reported in the cartesian XY coordinate system, where X is parallel both to the cylinder axis and to freestream velocity.  $X = 0$  at the cone/cylinder intersection. R is the radial distance from the cylinder axis. Y is the radial distance from the cylinder or cone surface.

The data tables include surface pressure and compressible displacement thickness distributions, as well as velocity-profile laser velocimeter measurements (vs. Y - distance from cylinder or cone surface) at the several X-stations where profiles were obtained for the 30 degree axisymmetric cone case.

For each X-station are given:

- X - X location of the profile in cm.
- R<sub>surface</sub> - Radius of the local surface in cm.
- N<sub>pts</sub> - number of points in the profile
- Date - Date data was taken (e.g., 5/31/85)
- P<sub>wall</sub>/PT1 - wall pressure/free stream total pressure data is interpolated from the surface pressure table if required.
- Theta - local slope of the Compressible Displacement Surface given by  $D_{star} + R_{surface}$ .  $D_{star}$  is obtainable from another table. The Compressible Displacement Thickness accounts for the velocity *and* density variation across the boundary layer.

Profiles at each X-station include:

- Y - distance from the local surface ( $R = Y + R_{surface}$ ) in cm.
- U<sub>MEAN</sub> - Mean X component of the velocity normalized by the acoustic velocity at Mach 1.
- V<sub>MEAN</sub> - Mean Y component of the velocity normalized by the acoustic velocity at Mach 1.
- U2 -  $(\text{Sum}(U - U_{MEAN})^2)/N$  and normalized by the square of the acoustic velocity at Mach 1.
- A Turbulent NORMAL stress in the XY coordinate system.

- V2 -  $(\text{Sum}(V - V_{\text{MEAN}})^2)/N$  and normalized by the square of the acoustic velocity at Mach 1.  
A Turbulent NORMAL stress in the XY coordinate system.
- UV -  $(\text{Sum}(U - U_{\text{MEAN}})(V - V_{\text{MEAN}}))/N$  and normalized by the square of the acoustic velocity at Mach 1.  
A Turbulent SHEAR stress in the XY coordinate system.
- Gminus - The fraction of the U velocity measurements that were negative.

Data tables are also given of density profile measurements (vs. R - distance from the cylinder or cone axis) at the several X-stations where profiles were obtained for the 30 degree axisymmetric cone case.

#### ERROR ASSESSMENT:

*Mean* Velocity measurements are subject to errors typically 0.01 times freestream velocity plus 0.1 times  $\sqrt{K}$ , where  $K = 0.5(U^2 + V^2 + W^2)$ , the turbulence kinetic energy, and where  $W^2$  is assumed to equal  $0.5(U^2 + V^2)$ .

Turbulence normal stress measurements are subject to errors typically 0.1 times  $K$ , the measured turbulence kinetic energy.

Turbulence *shear* stress measurements are subject to errors typically 0.05 times  $K$ , the measured turbulence kinetic energy.

Particle lag effects will always be present with laser velocimetry in supersonic flows, particularly with a shock present. Examinations of histograms obtained through a shock in the freestream demonstrate the 0.55 micron particles used were monodisperse. The primary effect of the particle lag is not entirely understood but appears to be to spread the velocity (mean and turbulence) gradients by about 0.5cm in the *streamwise* direction.

Errors in the density measurements are greatest at the cylinder or cone surface since an Abel transform inversion is required to deduce the density from the measured fringes. Typical errors in density appeared to be about 10 percent.

A *rational* theoretical error analysis on the various measurements invariably gives much smaller error bounds than given above. The above error bounds are based on *experience* with the data, guided by theoretical considerations and by appropriate cross-checks.

\*\*\*\*\*Dungan & Brown Axisymmetric Ogive-Cyl Flare Data\*\*\*\*\*  
 \*\*\*\*\*Surface Pressure Distribution\*\*\*\*\*  
 \*\*\*\*\*30 degree cone, Experiment\*\*\*\*\*

X cm	P/PI
-4.250, 0.0341	
-3.250, 0.0341	
-2.250, 0.0449	
-1.250, 0.0596	
-0.250, 0.0608	
0.750, 0.0694	
1.750, 0.0773	
2.750, 0.0873	
3.750, 0.0985	
4.750, 0.1091	
5.750, 0.1197	
6.750, 0.1287	
7.750, 0.1348	
8.750, 0.1409	
9.750, 0.1451	
10.750, 0.1495	
11.750, 0.1511	
12.750, 0.1533	
13.750, 0.1568	
14.750, 0.1609	
15.750, 0.1650	
16.750, 0.1691	
17.750, 0.1732	
18.750, 0.1773	
19.750, 0.1814	
20.750, 0.1855	
21.750, 0.1896	
22.750, 0.1937	
23.750, 0.1978	
24.750, 0.2019	
25.750, 0.2060	
26.750, 0.2101	
27.750, 0.2142	
28.750, 0.2183	
29.750, 0.2224	
30.750, 0.2265	
31.750, 0.2306	
32.750, 0.2347	
33.750, 0.2388	
34.750, 0.2429	
35.750, 0.2470	
36.750, 0.2511	
37.750, 0.2552	
38.750, 0.2593	
39.750, 0.2634	
40.750, 0.2675	
41.750, 0.2716	
42.750, 0.2757	
43.750, 0.2798	
44.750, 0.2839	
45.750, 0.2880	
46.750, 0.2921	
47.750, 0.2962	
48.750, 0.3003	
49.750, 0.3044	
50.750, 0.3085	
51.750, 0.3126	
52.750, 0.3167	
53.750, 0.3208	
54.750, 0.3249	
55.750, 0.3290	
56.750, 0.3331	
57.750, 0.3372	
58.750, 0.3413	
59.750, 0.3454	
60.750, 0.3495	
61.750, 0.3536	
62.750, 0.3577	
63.750, 0.3618	
64.750, 0.3659	
65.750, 0.3700	
66.750, 0.3741	
67.750, 0.3782	
68.750, 0.3823	
69.750, 0.3864	
70.750, 0.3905	
71.750, 0.3946	
72.750, 0.3987	
73.750, 0.4028	
74.750, 0.4069	
75.750, 0.4110	
76.750, 0.4151	
77.750, 0.4192	
78.750, 0.4233	
79.750, 0.4274	
80.750, 0.4315	
81.750, 0.4356	
82.750, 0.4397	
83.750, 0.4438	
84.750, 0.4479	
85.750, 0.4520	
86.750, 0.4561	
87.750, 0.4602	
88.750, 0.4643	
89.750, 0.4684	
90.750, 0.4725	
91.750, 0.4766	
92.750, 0.4807	
93.750, 0.4848	
94.750, 0.4889	
95.750, 0.4930	
96.750, 0.4971	
97.750, 0.5012	
98.750, 0.5053	
99.750, 0.5094	
100.750, 0.5135	
101.750, 0.5176	
102.750, 0.5217	
103.750, 0.5258	
104.750, 0.5299	
105.750, 0.5340	
106.750, 0.5381	
107.750, 0.5422	
108.750, 0.5463	
109.750, 0.5504	
110.750, 0.5545	
111.750, 0.5586	
112.750, 0.5627	
113.750, 0.5668	
114.750, 0.5709	
115.750, 0.5750	
116.750, 0.5791	
117.750, 0.5832	
118.750, 0.5873	
119.750, 0.5914	
120.750, 0.5955	
121.750, 0.5996	
122.750, 0.6037	
123.750, 0.6078	
124.750, 0.6119	
125.750, 0.6160	
126.750, 0.6201	
127.750, 0.6242	
128.750, 0.6283	
129.750, 0.6324	
130.750, 0.6365	
131.750, 0.6406	
132.750, 0.6447	
133.750, 0.6488	
134.750, 0.6529	
135.750, 0.6570	
136.750, 0.6611	
137.750, 0.6652	
138.750, 0.6693	
139.750, 0.6734	
140.750, 0.6775	
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142.750, 0.6857	
143.750, 0.6898	
144.750, 0.6939	
145.750, 0.6980	
146.750, 0.7021	
147.750, 0.7062	
148.750, 0.7103	
149.750, 0.7144	
150.750, 0.7185	
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152.750, 0.7267	
153.750, 0.7308	
154.750, 0.7349	
155.750, 0.7390	
156.750, 0.7431	
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159.750, 0.7554	
160.750, 0.7595	
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165.750, 0.7800	
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177.750, 0.8292	
178.750, 0.8333	
179.750, 0.8374	
180.750, 0.8415	
181.750, 0.8456	
182.750, 0.8497	
183.750, 0.8538	
184.750, 0.8579	
185.750, 0.8620	
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297.750, 1.3212	
298.750, 1.3253	
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300.750, 1.3335	
301.750, 1.3376	
302.750, 1.3417	
303.750, 1.3458	
304.750, 1.3499	
305.750, 1.3540	
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369.750, 1.6164	
370.750, 1.6205	
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372.750, 1.6287	
373.750, 1.6328	
374.750, 1.6369	
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376.750, 1.6451	
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382.750, 1.6697	
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384.750, 1.6779	
385.750, 1.6820	
386.750, 1.6861	
387.750, 1.6902	
388.750, 1.6943	
389.750, 1.6984	
390.750, 1.7025	
391.750, 1.7066	
392.750, 1.7107	
393.750, 1.7148	
394.750, 1.7189	
395.750, 1.7230	
396.750, 1.7271	
397.750, 1.7312	
398.750, 1.7353	
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400.750, 1.7435	
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402.750, 1.7517	
403.750, 1.7558	
404.750, 1.7599	
405.750, 1.7640	
406.750, 1.7681	
407.750, 1.7722	
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409.750, 1.7804	
410.750, 1.7845	
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414.750, 1.8009	
415.750, 1.8050	
416.750, 1.8091	
417.750, 1.8132	
418.750, 1.8173	
419.750, 1.8214	
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421.750, 1.8296	
422.750, 1.8337	
423.750, 1.8378	
424.750, 1.8419	
425.750, 1.8460	
426.750, 1.8501	
427.750, 1.8542	
428.750, 1.8583	
429.750, 1.8624	
430.750, 1.8665	
431.750, 1.8706	
432.750, 1.8747	
433.750, 1.8788	
434.750, 1.8829	
435.750, 1.8870	
436.750, 1.8911	
437.750, 1.8952	
438.750, 1.8993	
439.750, 1.9034	
440.750, 1.9075	
441.750, 1.9116	
442.750, 1.9157	
443.750, 1.9198	
444.750, 1.9239	
445.750, 1.9280	
446.750, 1.9321	
447.750, 1.9362	
448.750, 1.9403	
449.750, 1.9444	
450.750, 1.9485	
451.750, 1.9526	
452.750, 1.9567	
453.750, 1.9608	
454.750, 1.9649	
455.750, 1.9690	
456.750, 1.9731	
457.750, 1.9772	
458.750, 1.9813	
459.750, 1.9854	
460.750, 1.9895	
461.750, 1.9936	
462.750, 1.9977	
463.750, 2.0018	
464.750, 2.0059	
465.750, 2.0100	
466.750, 2.0141	
467.750, 2.0182	
468.750, 2.0223	
469.750, 2.0264	
470.750, 2.0305	
471.750, 2.0346	
472.750, 2.0387	
473.750, 2.0428	
474.750, 2.0469	
475.750, 2.0510	
476.750, 2.0551	
477.750, 2.0592	
478.750, 2.0633	
479.750, 2.0674	
480.750, 2.0715	
481.750, 2.0756	
482.750, 2.0797	
483.750, 2.0838	
484.750, 2.0879	
485.750, 2.0920	
486.750, 2.0961	
487.750, 2.1002	
488.750, 2.1043	
489.750,	

1.7301	.688E 02	.435E 02	.230E 02	.0000
1.7333	.619E 02	.402E 02	.207E 02	.0000
1.7377	.508E 02	.308E 02	.102E 02	.0000

Y	LINEAN	VMEAN	U2	V2	UV	CHINUS
0.051	1924	0.0051	121E-00	721E 01	380E 02	316E8
1.102	3990	0.1139	132E-00	794E 01	524E 02	1490
1.152	6451	0.1652	148E-00	831E 01	627E 04	0487
2.03	8273	0.6164	109E-00	672E 01	413E 02	0163
2.54	9701	0.8666	958E 01	556E 01	698E 02	0071
3.17	11184	1.3668	744E 01	435E 01	482E 02	0020
3.81	12468	1.6683	537E 01	333E 01	153E 02	0005
4.44	13676	1.9660	324E 01	244E 01	210E 02	0002
5.08	14482	2.091	226E 01	176E 01	433E 02	0001
5.72	15173	2.139	166E 01	141E 01	701E 02	0001
6.35	16081	2.015	225E 01	129E 01	998E 02	0000
6.98	16761	1.777	196E 01	120E 01	112E 01	0000
7.62	17315	1.525	183E 01	134E 01	125E 01	0000
8.25	18120	0.990	130E 01	123E 01	100E 01	0000
8.89	18570	0.680	841E 02	950E 02	755E 02	0000
0.016	10974	0.192	237E 02	343E 02	205E 02	0000
0.16	19237	0.019	718E 03	998E 03	126E 03	0000
0.270	19267	0.023	494E 03	558E 03	365E 03	0000
0.397	19281	0.073	429E 03	440E 03	873E 04	0000
0.524	19264	0.050	479E 03	499E 03	379E 04	0000
0.778	19258	0.058	461E 03	489E 03	769E 04	0000
0.932	19227	0.044	490E 03	511E 03	100E 03	0000

X(cm) = 2.5000 R_SURFACE = 2.5400 Mpts = 22 OBTAINED 6/19/85 Psalt/pl1 = 0.0524 Theta = 5.8										
Y	0.051	0.1331	0.0063	0.155E+00	V2	0.427E-01	UV	0.237E-02	GRINUS	-0.167
1	0.102	0.2504	0.0307	0.83E-01	2	0.298E-01	1	0.467E-02	0	-0.031
2	0.152	0.3663	0.0496	0.583E-01	3	0.215E-01	2	0.659E-02	1	0.008
3	0.203	0.4575	0.0577	0.368E-01	4	0.163E-01	3	0.904E-02	2	-0.003
4	0.254	0.5143	0.0565	0.297E-01	5	0.135E-01	4	0.101E-02	3	-0.001
5	0.305	0.5721	0.0551	0.247E-01	6	0.110E-01	5	0.137E-02	4	0.000
6	0.356	0.6309	0.0536	0.207E-01	7	0.089E-01	6	0.178E-02	5	0.000
7	0.407	0.6907	0.0521	0.177E-01	8	0.072E-01	7	0.228E-02	6	0.000
8	0.458	0.7515	0.0506	0.155E-01	9	0.058E-01	8	0.288E-02	7	0.000
9	0.509	0.8133	0.0491	0.139E-01	10	0.047E-01	9	0.360E-02	8	0.000
10	0.560	0.8761	0.0476	0.127E-01	11	0.038E-01	10	0.445E-02	9	0.000
11	0.611	0.9400	0.0461	0.118E-01	12	0.031E-01	11	0.546E-02	10	0.000
12	0.662	0.1005	0.0446	0.111E-01	13	0.025E-01	12	0.667E-02	11	0.000
13	0.713	0.1016	0.0431	0.105E-01	14	0.020E-01	13	0.809E-02	12	0.000
14	0.764	0.1027	0.0416	0.100E-01	15	0.016E-01	14	0.974E-02	13	0.000
15	0.815	0.1038	0.0401	0.096E-01	16	0.013E-01	15	0.116E-01	14	0.000
16	0.866	0.1049	0.0386	0.092E-01	17	0.010E-01	16	0.141E-01	15	0.000
17	0.917	0.1060	0.0371	0.089E-01	18	0.008E-01	17	0.168E-01	16	0.000
18	0.968	0.1071	0.0356	0.086E-01	19	0.006E-01	18	0.200E-01	17	0.000
19	1.019	0.1082	0.0341	0.083E-01	20	0.005E-01	19	0.237E-01	18	0.000
20	1.070	0.1093	0.0326	0.080E-01	21	0.004E-01	20	0.280E-01	19	0.000
21	1.121	0.1104	0.0311	0.077E-01	22	0.003E-01	21	0.329E-01	20	0.000
22	1.172	0.1115	0.0296	0.074E-01	23	0.002E-01	22	0.384E-01	21	0.000
23	1.223	0.1126	0.0281	0.071E-01	24	0.001E-01	23	0.445E-01	22	0.000
24	1.274	0.1137	0.0266	0.068E-01	25	0.000E-01	24	0.512E-01	23	0.000
25	1.325	0.1148	0.0251	0.065E-01	26	0.000E-01	25	0.586E-01	24	0.000
26	1.376	0.1159	0.0236	0.062E-01	27	0.000E-01	26	0.667E-01	25	0.000
27	1.427	0.1170	0.0221	0.059E-01	28	0.000E-01	27	0.754E-01	26	0.000
28	1.478	0.1181	0.0206	0.056E-01	29	0.000E-01	28	0.848E-01	27	0.000
29	1.529	0.1192	0.0191	0.053E-01	30	0.000E-01	29	0.950E-01	28	0.000
30	1.580	0.1203	0.0176	0.050E-01	31	0.000E-01	30	0.106E-01	29	0.000
31</										

Variable	Mean	Standard deviation	Skewness	Kurtosis
1. Age	1.7021	.0252	.109E+01	.536E+02
2. Sex	1.7429	.0190	.897E+02	.409E+02

Y	LINEAM	VHEAM	U2	V2	UV	CHIRUS
0.051	1.057	0.061	1.876E-01	3.34E-01	2.66E-02	5887
0.102	1.507	0.137	1.60E-01	5.94E-01	1.06E-01	5983
0.152	2.395	0.029	1.39E-00	7.17E-01	8.23E-02	1910
0.203	4.982	0.059	1.36E-00	7.13E-01	7.05E-02	1064
0.254	6.880	0.419	1.36E-00	6.55E-01	7.89E-03	0455
0.317	8.759	0.863	1.04E+00	6.02E-01	1.65E-02	0130
0.381	1.1234	1.234	9.20E-01	4.97E-01	3.06E-02	0054
0.444	1.1468	1.602	7.08E-01	4.04E-01	5.18E-02	0016
0.508	1.2512	1.910	5.26E-01	3.19E-01	3.57E-02	0003
0.572	1.3438	2.059	4.19E-01	2.24E-01	2.90E-02	0001
0.638	1.4391	2.330	3.21E-01	1.79E-01	6.59E-03	0000
0.698	1.4923	2.591	2.43E-01	1.34E-01	1.24E-02	0000
0.762	1.5601	2.581	2.05E-01	1.06E-01	3.11E-02	0000
0.825	1.6298	2.610	1.60E-01	8.04E-02	4.55E-02	0000
0.889	1.7051	2.396	1.31E-01	7.77E-02	6.18E-02	0000
0.916	1.7981	1.847	8.68E-02	1.07E-01	8.04E-02	0000
1.070	1.7981	1.847	5.37E-02	1.13E-01	6.78E-02	0000
1.273	1.8650	1.008	3.15E-02	3.26E-02	1.64E-02	0000
1.919	1.9179	0.160	1.55E-02	3.75E-03	1.18E-03	0000
3.397	1.9232	0.000	5.06E-03	7.75E-03	4.78E-04	0000
5.524	1.9256	0.023	4.94E-03	6.64E-03	6.54E-04	0000
7.778	1.9246	0.026	4.78E-03	4.47E-03	6.54E-04	0000
9.332	1.9236	0.052	4.18E-03	4.30E-03	4.97E-04	0000

[illegible]

1.666	1.5467	.1629	.319E-01	.171E-01	.129E-01	.0000
.508	1.6160	.1312	.275E-01	.155E-01	.138E-01	.0000

Y	LINEAM	WMEAN	U2	V2	UV	Chi-square
051	0421	0079	923E-01	561E-01	-480E-03	626E-02
102	0729	0196	109E+00	638E-01	954E-03	475E-02
152	1942	0136	123E+00	673E-01	-1490E-02	378E-02
203	3335	0186	137E+00	669E-01	624E-02	266E-02
254	4646	0188	140E+00	664E-01	-777E-02	121E-02
317	6219	0326	136E+00	602E-01	444E-02	056E-02
381	7984	0603	120E+00	570E-01	760E-04	019E-02
444	9822	1074	984E-01	507E-01	153E-02	006E-02
508	12330	1677	744E-01	406E-01	565E-02	0015
572	12138	1763	602E-01	342E-01	475E-02	0013
635	2269	2034	688E-01	275E-01	374E-02	0003
698	3674	2229	555E-01	202E-01	254E-02	0000
762	4475	2508	287E-01	152E-01	-499E-03	0000
825	5187	2674	223E-01	115E-01	355E-03	0000
889	5830	2814	160E-01	859E-02	185E-02	0000
916	16079	2750	937E-02	516E-02	316E-02	0000

1.6915	1.006	.222E-01	.142E-01	.135E-01	.0000
.572	.0762	.158E-01	.122E-01	.109E-01	.0000
.635	.17537	.158E-01	.122E-01	.109E-01	.0000
.698	1.8209	.0360	.1790E-02	.676E-02	.0000
.762	1.8517	.0256	.688E-02	.459E-02	.0000
.825	1.8811	.0161	.362E-02	.227E-02	.0000
.889	1.9024	.0090	.191E-02	.165E-02	.0000
.952	1.9225	.0074	.794E-03	.757E-03	.0000
.016	1.9273	.0024	.536E-03	.562E-03	.0000
.143	1.9273	.0024	.536E-03	.562E-03	.0000
.270	1.9260	.0013	.476E-03	.508E-03	.0000
.307	1.9252	.0028	.440E-03	.461E-03	.0000
.374	1.9253	.0039	.432E-03	.453E-03	.0000
.424	1.9238	.0061	.432E-03	.451E-03	.0000
.478	1.9216	.0033	.451E-03	.460E-04	.0000
.532					

X(cm) = 0.8640 R SURFACE = 3.0400 Mpts = 25 OBTAINED 6/20/85									
PsiII/PTI = 0.0873 Theta = 17.9									
Y	U	V	W	X	Y	U	V	W	X
0.051	0.4405	0.0855	0.139E-00	0.604E-01	0.833E-02	0.1279	0.0000	0.0000	0.0000
0.102	0.5935	0.0844	0.143E-00	0.600E-01	0.726E-03	0.0669	0.0000	0.0000	0.0000
0.152	0.6965	0.0708	0.145E-00	0.704E-01	0.232E-03	0.0410	0.0000	0.0000	0.0000
0.203	0.7865	0.0955	0.139E-00	0.670E-01	0.140E-02	0.0288	0.0000	0.0000	0.0000
0.254	0.8983	0.1215	0.123E-00	0.681E-01	0.167E-02	0.0140	0.0000	0.0000	0.0000
0.317	0.9870	0.1488	0.115E-00	0.611E-01	0.501E-02	0.0097	0.0000	0.0000	0.0000
0.381	1.0909	0.1840	0.959E-01	0.547E-01	0.562E-02	0.0043	0.0000	0.0000	0.0000
0.444	1.1633	0.2109	0.853E-01	0.512E-01	0.593E-02	0.0020	0.0000	0.0000	0.0000
0.508	1.2620	0.2409	0.688E-01	0.425E-01	0.612E-02	0.0013	0.0000	0.0000	0.0000
0.572	1.3563	0.2768	0.525E-01	0.343E-01	0.391E-02	0.0005	0.0000	0.0000	0.0000
0.635	1.4241	0.3013	0.450E-01	0.273E-01	0.232E-02	0.0002	0.0000	0.0000	0.0000
0.698	1.4961	0.3189	0.324E-01	0.210E-01	0.365E-02	0.0000	0.0000	0.0000	0.0000
0.762	1.5683	0.3433	0.235E-01	0.139E-01	0.455E-03	0.0000	0.0000	0.0000	0.0000
0.825	1.6296	0.3593	0.155E-01	0.905E-02	0.455E-03	0.0000	0.0000	0.0000	0.0000
0.889	1.6729	0.3643	0.907E-02	0.552E-02	0.254E-02	0.0000	0.0000	0.0000	0.0000
0.952	1.7239	0.3564	0.325E-02	0.177E-02	0.834E-03	0.0000	0.0000	0.0000	0.0000
1.016	1.7503	0.3262	0.163E-02	0.254E-02	0.135E-02	0.0000	0.0000	0.0000	0.0000
1.079	1.7692	0.2926	0.177E-02	0.288E-02	0.258E-02	0.0000	0.0000	0.0000	0.0000
1.143	1.7988	0.2439	0.226E-02	0.516E-02	0.258E-02	0.0000	0.0000	0.0000	0.0000
1.207	1.8384	0.1637	0.358E-02	0.104E-01	0.529E-02	0.0000	0.0000	0.0000	0.0000
1.270	1.9098	0.0275	0.199E-02	0.537E-02	0.261E-02	0.0000	0.0000	0.0000	0.0000
1.334	1.9785	0.0010	0.799E-03	0.150E-02	0.457E-03	0.0000	0.0000	0.0000	0.0000
1.397	1.9283	0.0076	0.505E-03	0.567E-03	0.727E-04	0.0000	0.0000	0.0000	0.0000
1.461	1.9290	0.0054	0.696E-03	0.518E-03	0.968E-04	0.0000	0.0000	0.0000	0.0000
1.524	1.9316	0.0053	0.677E-03	0.504E-03	0.993E-04	0.0000	0.0000	0.0000	0.0000
X(cm) = 1.7321 R SURFACE = 3.5400 Mpts = 25 OBTAINED 6/20/85									
PsiII/PTI = 0.1091 Theta = 23.9									
Y	U	V	W	X	Y	U	V	W	X
0.051	0.7397	0.2366	0.117E-00	0.461E-01	0.361E-03	0.0079	0.0000	0.0000	0.0000
0.102	0.8485	0.2355	0.115E-00	0.506E-01	0.289E-02	0.0059	0.0000	0.0000	0.0000
0.152	0.9305	0.2288	0.107E-00	0.563E-01	0.248E-02	0.0035	0.0000	0.0000	0.0000
0.203	0.9898	0.2242	0.101E-00	0.593E-01	0.801E-03	0.0025	0.0000	0.0000	0.0000
0.254	1.0666	0.2432	0.897E-01	0.582E-01	0.426E-02	0.0021	0.0000	0.0000	0.0000
0.317	1.1500	0.2742	0.790E-01	0.516E-01	0.720E-02	0.0010	0.0000	0.0000	0.0000
0.381	1.2290	0.3075	0.638E-01	0.448E-01	0.813E-02	0.0005	0.0000	0.0000	0.0000
0.444	1.3006	0.3503	0.561E-01	0.376E-01	0.922E-02	0.0004	0.0000	0.0000	0.0000
0.508	1.3837	0.3934	0.435E-01	0.297E-01	0.932E-02	0.0003	0.0000	0.0000	0.0000
0.572	1.4531	0.4219	0.326E-01	0.229E-01	0.762E-02	0.0001	0.0000	0.0000	0.0000
0.635	1.5058	0.4563	0.216E-01	0.153E-01	0.364E-02	0.0000	0.0000	0.0000	0.0000
0.698	1.5594	0.4721	0.141E-01	0.941E-02	0.173E-02	0.0000	0.0000	0.0000	0.0000
0.762	1.5928	0.4774	0.891E-02	0.571E-02	0.814E-03	0.0000	0.0000	0.0000	0.0000
0.825	1.6219	0.4671	0.623E-02	0.611E-02	0.204E-02	0.0000	0.0000	0.0000	0.0000
0.889	1.6484	0.4494	0.503E-02	0.364E-02	0.240E-02	0.0000	0.0000	0.0000	0.0000
0.952	1.6929	0.3973	0.387E-02	0.364E-02	0.268E-02	0.0000	0.0000	0.0000	0.0000
1.016	1.7383	0.3371	0.273E-02	0.329E-02	0.216E-02	0.0000	0.0000	0.0000	0.0000
1.079	1.7713	0.2854	0.230E-02	0.391E-02	0.211E-02	0.0000	0.0000	0.0000	0.0000
1.143	1.8193	0.1989	0.319E-02	0.797E-02	0.409E-02	0.0000	0.0000	0.0000	0.0000
1.207	1.8679	0.0727	0.106E-02	0.928E-02	0.471E-02	0.0000	0.0000	0.0000	0.0000
1.270	1.9225	0.0050	0.557E-03	0.218E-02	0.899E-03	0.0000	0.0000	0.0000	0.0000
1.334	1.9282	0.0060	0.577E-03	0.796E-03	0.108E-03	0.0000	0.0000	0.0000	0.0000
1.397	1.9308	0.0084	0.487E-03	0.520E-03	0.449E-04	0.0000	0.0000	0.0000	0.0000
1.461	1.9283	0.0087	0.483E-03	0.495E-03	0.363E-04	0.0000	0.0000	0.0000	0.0000
1.524	1.9276	0.0095	0.487E-03	0.504E-03	0.482E-04	0.0000	0.0000	0.0000	0.0000
X(cm) = 2.5931 R SURFACE = 4.0400 Mpts = 25 OBTAINED 6/20/85									
PsiII/PTI = 0.1287 Theta = 24.5									
Y	U	V	W	X	Y	U	V	W	X
0.051	0.8361	0.3491	0.370E-01	0.476E-02	0.0003	0.0000	0.0000	0.0000	0.0000
0.102	0.9463	0.3672	0.875E-01	0.408E-01	0.286E-02	0.0001	0.0000	0.0000	0.0000
0.152	1.0125	0.3382	0.837E-01	0.420E-01	0.176E-02	0.0002	0.0000	0.0000	0.0000
0.203	1.0708	0.3374	0.787E-01	0.481E-01	0.115E-02	0.0002	0.0000	0.0000	0.0000
0.254	1.1387	0.3643	0.713E-01	0.444E-01	0.204E-02	0.0001	0.0000	0.0000	0.0000
0.317	1.1987	0.3945	0.586E-01	0.401E-01	0.427E-02	0.0002	0.0000	0.0000	0.0000
0.381	1.2815	0.4101	0.484E-01	0.352E-01	0.989E-02	0.0000	0.0000	0.0000	0.0000
0.444	1.3483	0.5574	0.360E-01	0.247E-01	0.983E-02	0.0000	0.0000	0.0000	0.0000
0.508	1.3987	0.6941	0.263E-01	0.205E-01	0.775E-02	0.0000	0.0000	0.0000	0.0000
0.572	1.4408	0.5352	0.174E-01	0.135E-01	0.445E-02	0.0000	0.0000	0.0000	0.0000





```

203 1.7549 0.262 900E 02 527E 02 123E 03 .0000
254 1.7700 0.126 911E 02 555E 02 404E 03 .0000
317 1.7808 0.374 875E 02 580E 02 674E 03 .0000
381 1.7944 0.431 820E 02 620E 03 814E 02 .0000
444 1.8080 0.471 800E 02 599E 03 118E 02 .0000
508 1.8182 0.509 801E 02 594E 03 168E 02 .0000
572 1.8277 0.480 864E 03 574E 02 168E 02 .0000
635 1.8342 0.500 804E 02 519E 02 162E 02 .0000
698 1.8371 0.512 807E 02 485E 02 134E 02 .0000
762 1.8470 0.510 764E 02 444E 02 119E 02 .0000
825 1.8593 0.515 708E 02 409E 02 103E 02 .0000
889 1.8613 0.549 659E 02 359E 02 804E 03 .0000
1.016 1.8727 0.575 617E 02 324E 02 592E 03 .0000
1.143 1.8837 0.592 437E 02 284E 02 392E 03 .0000
1.270 1.8997 0.604 332E 02 231E 02 194E 03 .0000
1.397 1.8912 0.634 249E 02 168E 02 944E 04 .0000
1.524 1.8897 0.720 210E 02 167E 02 328E 04 .0000
1.778 1.8751 1.103 124E 02 113E 02 729E 04 .0000
2.032 1.8455 1.589 894E 03 841E 03 156E 03 .0000
2.540 1.7589 2.530 742E 03 651E 03 690E 05 .0000
3.048 1.6940 3.274 550E 03 483E 03 308E 04 .0000
3.556 1.6359 3.819 451E 03 423E 03 856E 05 .0000
4.064 1.5804 4.268 371E 03 356E 03 251E 04 .0000
4.572 1.5597 4.684 401E 03 392E 03 337E 04 .0000
5.080 1.5540 4.831 524E 03 523E 03 159E 03 .0000
5.588 1.5820 4.647 147E 02 183E 02 126E 02 .0000
5.842 1.9143 .0024 451E 03 408E 02 333E 04 .0000
420E 03 139E 05 .0000

```

# DENSITY DATA FROM HOLOGRAPHIC INTERFEROMETRY

```

MODEL CONE ANGLE = 30.0 Degrees
TOTAL PRESSURE = 1.7 Atmospheres
TOTAL TEMPERATURE = 270 K

```

```

X(cm) = -5.003 Mpts = 42
R SURFACE = 2.540
R_SHOCK = 0.000 SHOCK ANGLE = 0.0 Degrees

```

R(cm)	RHO/RHOT
2.542	0.04156
2.567	0.04372
2.591	0.04584
2.616	0.04791
2.640	0.04993
2.664	0.05191
2.689	0.05384
2.713	0.05571
2.738	0.05754
2.762	0.05931
2.787	0.06104
2.811	0.06271
2.836	0.06433
2.860	0.06590
2.884	0.06742
2.909	0.06889
2.933	0.07031
2.958	0.07168
2.982	0.07299
3.007	0.07426
3.031	0.07548
3.056	0.07665
3.080	0.07777
3.104	0.07885
3.129	0.07987
3.153	0.08085
3.178	0.08178
3.202	0.08266
3.227	0.08350
3.251	0.08427
3.276	0.08499

```

X(cm) = -3.009 Mpts = 41
R SURFACE = 2.540
R_SHOCK = 0.000 SHOCK ANGLE = 0.0 Degrees

```

R(cm)	RHO/RHOT
2.542	0.05583
2.567	0.05736
2.591	0.05789
2.616	0.05841
2.640	0.05895
2.664	0.05952
2.689	0.06012
2.713	0.06075
2.738	0.06143
2.762	0.06215

```

X(cm) = -4.013 Mpts = 41
R SURFACE = 2.540
R_SHOCK = 0.000 SHOCK ANGLE = 0.0 Degrees

```

R(cm)	RHO/RHOT
2.542	0.04138
2.567	0.04314
2.591	0.04473
2.616	0.04618
2.640	0.04752
2.664	0.04878
2.689	0.04998
2.713	0.05113
2.738	0.05226
2.762	0.05337
2.787	0.05448
2.811	0.05561
2.836	0.05675
2.860	0.05791
2.884	0.05910
2.909	0.06033
2.933	0.06158
2.958	0.06287
2.982	0.06420
3.007	0.06555
3.031	0.06693
3.056	0.06834
3.080	0.06977
3.104	0.07121
3.129	0.07266
3.153	0.07411
3.178	0.07556
3.202	0.07700
3.227	0.07841
3.251	0.07980
3.276	0.08116
3.300	0.08247
3.324	0.08374
3.349	0.08494
3.373	0.08609
3.398	0.08718
3.422	0.08829
3.447	0.08942
3.471	0.09054
3.496	0.09163
3.520	0.09261



2.787 0.06291  
2.811 0.06372  
2.836 0.06458  
2.860 0.06548  
2.884 0.06643  
2.909 0.06742  
2.933 0.06845  
2.958 0.06952  
2.982 0.07062  
3.007 0.07174  
3.031 0.07289  
3.056 0.07405  
3.080 0.07523  
3.104 0.07640  
3.129 0.07758  
3.153 0.07875  
3.178 0.07991  
3.202 0.08104  
3.227 0.08214  
3.251 0.08320  
3.276 0.08422  
3.300 0.08518  
3.324 0.08607  
3.349 0.08699  
3.373 0.08760  
3.398 0.08817  
3.422 0.08863  
3.447 0.08902  
3.471 0.08934  
3.496 0.08957  
3.520 0.08961

X(cm) = -2.007 Mpts = 42  
R\_SURFACE = 2.540  
R\_SHOCK = 3.373 SHOCK ANGLE = 3.4 Degrees

RHO/RHOT  
2.543 0.04549  
2.569 0.05780  
2.594 0.06534  
2.620 0.07062  
2.645 0.07484  
2.670 0.07845  
2.698 0.08156  
2.721 0.08412  
2.747 0.08610  
2.772 0.08747  
2.798 0.08831  
2.823 0.08871  
2.848 0.08879  
2.874 0.08872  
2.899 0.08840  
2.925 0.08855  
2.950 0.08843  
2.976 0.08846  
3.001 0.08823  
3.027 0.08793  
3.052 0.08729  
3.077 0.08686  
3.103 0.086137  
3.128 0.085179  
3.153 0.084208  
3.179 0.082241  
3.205 0.08219  
3.230 0.08200  
3.255 0.08189  
3.281 0.08176  
3.306 0.08162  
3.332 0.08147  
3.357 0.08130  
3.382 0.08112  
3.408 0.08093  
3.434 0.08074  
3.459 0.08050  
3.485 0.08024

3.510 0.09002  
3.536 0.08973  
3.561 0.08938  
3.586 0.08961

X(cm) = -1.007 Mpts = 42  
R\_SURFACE = 2.540  
R\_SHOCK = 3.769 SHOCK ANGLE = 25.7 Degrees

RHO/RHOT  
2.547 0.06940  
2.579 0.07055  
2.611 0.07158  
2.642 0.07249  
2.674 0.07331  
2.706 0.07404  
2.738 0.07471  
2.770 0.07534  
2.802 0.07595  
2.834 0.07656  
2.866 0.07722  
2.897 0.07793  
2.929 0.07874  
2.961 0.07966  
2.993 0.08074  
3.025 0.08198  
3.057 0.08343  
3.088 0.08509  
3.120 0.08700  
3.152 0.08916  
3.184 0.09158  
3.216 0.09426  
3.248 0.09721  
3.280 0.10041  
3.312 0.10384  
3.343 0.10746  
3.375 0.11122  
3.407 0.11505  
3.439 0.11885  
3.471 0.12268  
3.503 0.12577  
3.534 0.12845  
3.566 0.13014  
3.598 0.13018  
3.630 0.12727  
3.662 0.11733  
3.694 0.11649  
3.726 0.11426  
3.757 0.10960  
3.789 0.10400  
3.821 0.10412  
3.853 0.08961

X(cm) = -0.012 Mpts = 42  
R\_SURFACE = 2.541  
R\_SHOCK = 4.282 SHOCK ANGLE = 27.5 Degrees

RHO/RHOT  
2.551 0.07381  
2.594 0.07592  
2.636 0.07732  
2.679 0.07870  
2.721 0.07953  
2.763 0.08014  
2.804 0.08057  
2.848 0.08089  
2.890 0.08119  
2.933 0.08153  
2.975 0.08195  
3.018 0.08253  
3.060 0.08330  
3.103 0.08429  
3.145 0.08555  
3.187 0.08708

X(cm) = 0.996 Mpts = 42  
R\_SURFACE = 3.115  
R\_SHOCK = 4.795 SHOCK ANGLE = 27.5 Degrees

RHO/RHOT  
3.124 0.10106  
3.165 0.10254  
3.206 0.10160  
3.246 0.10058  
3.287 0.09976  
3.329 0.09933  
3.369 0.09943  
3.410 0.10013  
3.451 0.10147  
3.492 0.10344  
3.533 0.10600  
3.574 0.10909  
3.615 0.11263  
3.656 0.11652  
3.697 0.12065  
3.738 0.12491  
3.779 0.12919  
3.820 0.13339  
3.861 0.13740  
3.902 0.14112  
3.943 0.14450  
3.984 0.14745  
4.025 0.14993  
4.066 0.15191  
4.107 0.15338  
4.148 0.15434  
4.189 0.15482  
4.230 0.15485  
4.271 0.15448  
4.312 0.15378  
4.353 0.15282  
4.394 0.15167  
4.435 0.15041  
4.476 0.14907  
4.517 0.14769  
4.558 0.14624  
4.599 0.14462  
4.639 0.14262  
4.680 0.13987  
4.721 0.13589  
4.762 0.13347  
4.803 0.08961

```

X(cm) = 1.908 Mpts = 42
R_SURFACE = 3.694
R_SHOCK = 5.317 SHOCK ANGLE = 27.5 Degrees

R(cm) RHO/RHOT
3.699 0.13825
3.739 0.13974
3.778 0.14013
3.818 0.13989
3.858 0.13947
3.897 0.13922
3.937 0.13943
3.976 0.14031
4.016 0.14198
4.056 0.14451
4.095 0.14787
4.135 0.15199
4.175 0.15674
4.214 0.16196
4.254 0.16744
4.293 0.17298
4.333 0.17834
4.373 0.18331
4.413 0.18769
4.452 0.19130
4.492 0.19400
4.532 0.19568
4.571 0.19628
4.611 0.19581
4.650 0.19429
4.690 0.19183
4.730 0.18855
4.769 0.18465
4.809 0.18034
4.848 0.17585
4.888 0.17144
4.928 0.16735
4.968 0.16378
5.007 0.16089
5.047 0.15874
5.086 0.15723
5.126 0.15607
5.166 0.15462
5.205 0.15167
5.245 0.14492
5.285 0.13171
5.324 0.08961

X(cm) = 2.996 Mpts = 42
R_SURFACE = 4.270
R_SHOCK = 5.868 SHOCK ANGLE = 27.5 Degrees

R(cm) RHO/RHOT
4.276 0.16436
4.315 0.16485
4.354 0.16527
4.393 0.16585
4.432 0.16678
4.471 0.16823
4.510 0.17030
4.549 0.17305
4.588 0.17648
4.627 0.18057
4.666 0.18522
4.705 0.19033
4.744 0.19574
4.783 0.20131
4.822 0.20685
4.861 0.21219
4.900 0.21715
4.939 0.22158
4.978 0.22533
5.017 0.22830
5.056 0.23038
5.095 0.23153

X(cm) = 4.980 Mpts = 42
R_SURFACE = 5.415
R_SHOCK = 7.405 SHOCK ANGLE = 40.2 Degrees

R(cm) RHO/RHOT
5.428 0.19142
5.477 0.19362
5.525 0.19533
5.574 0.19830
5.622 0.20227
5.671 0.20701
5.719 0.21229
5.767 0.21791
5.816 0.22368
5.864 0.22944
5.913 0.23503
5.961 0.24032
6.010 0.24519
6.058 0.24956
6.107 0.25333
6.155 0.25644
6.204 0.25886
6.252 0.26055
6.301 0.26151
6.349 0.26174
6.397 0.26126
6.446 0.26011
6.494 0.25834
6.543 0.25601
6.591 0.25319
6.640 0.24997
6.688 0.24643
6.737 0.24267
6.785 0.23878
6.834 0.23487
6.882 0.23103
6.931 0.22734
6.979 0.22388
7.028 0.22071
7.076 0.21786
7.124 0.21533
7.173 0.21308
7.221 0.21103
7.270 0.20905
7.318 0.20699
7.367 0.20505
7.415 0.08961

X(cm) = 5.183 Mpts = 42
R_SURFACE = 5.528
R_SHOCK = 7.557 SHOCK ANGLE = 40.3 Degrees

R(cm) RHO/RHOT
5.534 0.17265
5.584 0.17647
5.633 0.18063
5.683 0.18517
5.732 0.19008
5.782 0.19533
5.832 0.20084
5.881 0.20652
5.931 0.21227
5.980 0.21797
6.030 0.22352
6.079 0.22880
6.129 0.23373
6.178 0.23822
6.228 0.24221
6.278 0.24563
6.327 0.24845
6.377 0.25065
6.426 0.25223
6.476 0.25319
6.525 0.25356
6.575 0.25338

```

```

6.675 0.25269
6.674 0.25154
6.724 0.25000
6.773 0.24812
6.823 0.24595
6.872 0.24355
6.922 0.24097
6.971 0.23824
7.021 0.23537
7.070 0.23237
7.120 0.22924
7.169 0.22592
7.219 0.22238
7.269 0.21855
7.318 0.21434
7.368 0.20970
7.417 0.20462
7.467 0.19936
7.516 0.19555
7.566 0.08961

X(cm) = 6.173 Mpts = 42
R_SURFACE = 5.540
R_SHOCK = 8.433 SHOCK ANGLE = 40.4 Degrees

R(cm) RHO/RHOT
5.554 0.06516
5.625 0.06799
5.695 0.06949
5.766 0.07142
5.836 0.07425
5.907 0.07820
5.978 0.08333
6.048 0.08959
6.119 0.09682
6.189 0.10485
6.260 0.11345
6.330 0.12260
6.401 0.13151
6.472 0.14057
6.542 0.14943
6.613 0.15794
6.683 0.16605
6.754 0.17363
6.824 0.18064
6.895 0.18711
6.965 0.19299
7.036 0.19827
7.107 0.20300
7.177 0.20717
7.248 0.21082
7.318 0.21395
7.389 0.21656
7.459 0.21866
7.530 0.22022
7.601 0.22123
7.671 0.22166
7.742 0.22147
7.812 0.22063
7.883 0.21911
7.953 0.21688
8.024 0.21395
8.094 0.21027
8.165 0.20593
8.235 0.20101
8.306 0.19583
8.376 0.19180
8.447 0.08961

```

```

X(cm) = 7.173 Mpts = 42
R_SURFACE = 5.540
R_SHOCK = 9.297 SHOCK ANGLE = 40.4 Degrees

```

```

R(cm) RHO/RHOT
5.562 0.06549
5.632 0.07051
5.703 0.07107
5.773 0.07135
5.843 0.07162
5.913 0.07220
6.020 0.07336
6.112 0.07333
6.203 0.07333
6.293 0.07827
6.386 0.08223
6.478 0.08722
6.569 0.09315
6.661 0.09986
6.752 0.10716
6.844 0.11483
6.936 0.12262
7.027 0.13029
7.119 0.13763
7.210 0.14446
7.302 0.15064
7.393 0.15610
7.485 0.16081
7.576 0.16481
7.668 0.16819
7.760 0.17110
7.851 0.17370
7.943 0.17620
8.035 0.17877
8.126 0.18159
8.218 0.18477
8.309 0.18835
8.401 0.19227
8.492 0.19636
8.584 0.20030
8.676 0.20363
8.767 0.20577
8.859 0.20601
8.950 0.20367
9.042 0.19822
9.133 0.18996
9.225 0.18441
9.317 0.08961

```

```

X(cm) = 8.194 Mpts = 41
R_SURFACE = 5.540
R_SHOCK = 10.090 SHOCK ANGLE = 40.5 Degrees

```

```

R(cm) RHO/RHOT
5.571 0.06075
5.792 0.06737
5.903 0.07010
6.014 0.07233
6.125 0.07420
6.235 0.07597
6.346 0.07786
6.457 0.08011
6.568 0.08286
6.679 0.08620
6.790 0.09015
6.900 0.09467
7.011 0.09966
7.122 0.10501
7.233 0.11057
7.343 0.11619
7.454 0.12175
7.565 0.12713
7.676 0.13226
7.786 0.13708
7.898 0.14160
8.008 0.14582

```

```

8.119 0.14981
8.230 0.15365
8.341 0.15741
8.451 0.16119
8.562 0.16506
8.673 0.16900
8.784 0.17325
8.894 0.17753
9.005 0.18182
9.116 0.18596
9.227 0.18967
9.338 0.19260
9.449 0.19427
9.559 0.19404
9.670 0.19101
9.781 0.18379
9.892 0.16962
10.002 0.14010
10.113 0.08961

```

.....END OF FILE.....

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Ref.: 12, 52

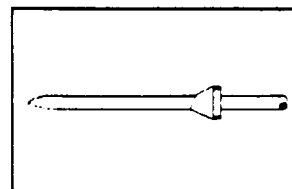
Author: Brown, J. D., *et al*

Geometry: Axisymmetric Ogive-Cylinder with Skewed Flare

Mach number: 3

Data:  $p_{wall}$ , flowfield surveys (LDV)

---



Brown, J.D., Brown, J.L. and Kussoy, M.I., "A Documentation of Two- and Three-Dimensional Shock-Separated Turbulent Boundary Layers," *NASA TM 101008*, July 1988.

Kussoy, M.I., Brown, J.D., Brown, J.L., Lockman, W.K. and Horstman, C.C., "Fluctuations and Massive Separation in Three-Dimensional Shock-Wave/Boundary-Layer Interactions," *Intl. Symp. on Transport Phenomena in Turbulent Flows, 2nd*, 1988, pp. 875-887.

---

Data are reported from a supersonic flow over an axisymmetric body of 5.02 cm diameter, aligned with the wind tunnel axis and supporting a turbulent boundary layer. A 3-D shock wave was generated by a 30° half-angle conical flare mounted on the cylinder with the flare axis inclined at an angle  $\alpha$  to the cylinder axis. Measurements were made in the upper symmetry plane of the test body, and consisted of mean surface pressures and both mean and turbulence data from LDV surveys of the flowfield for  $\alpha = 0, 5$ , and 10 degrees.

Unfortunately, this extensive dataset was not available in machine-readable form at the time of this writing. We have manually entered a subset of the data, namely surface pressures and every other flowfield survey for the  $\alpha = 10^\circ$  case only, and it is that which appears in the following tabulation. Users of these data are urged to consult the cited reference for a full data tabulation and discussion of the experiment.

Some nomenclature is defined at the beginning of the data tabulation. Additional definitions of terms used in the tables are given below:

CP  $\equiv$  pressure coefficient

UMEAN  $\equiv u/u_\infty$

VMEAN  $\equiv v/v_\infty$

$$U2 \equiv \overline{u'^2}$$

$$V2 \equiv \overline{v'^2}$$

$$UV \equiv \overline{u'v'}$$

$$U2V2 \equiv 0.75(\overline{u'^2} + \overline{v'^2})/u_\infty^2$$

GMINUS  $\equiv$  reverse-flow intermittency, *ie* the percentage of time that the flow is reversed at a given point

\*\*\*\*\*Brown, Brown, and Kussoy Mach 3 Ogive-Cyl-Skewed Flare\*\*\*\*\*

\*\*\*\*\*NOMINAL TEST CONDITIONS AND MEASUREMENT UNCERTAINTIES\*\*\*\*\*

Quantity	Value	Uncertainty
PT (stagnation pressure)	1.7 atm	±0.4%
TT (total temperature)	265K	±7.0%
M <sub>∞</sub> (freestream Mach number)	2.85	±0.9%
U <sub>∞</sub> (freestream velocity)	577 m/s	±1.9%
Re <sub>∞</sub> (unit Reynolds number)	16 x 10 <sup>6</sup> m <sup>-1</sup>	---
δ <sub>o</sub> (incoming b.l. thickness)	1.10cm	±4.5%
C <sub>fo</sub> (incoming skin fric. coeff.)	0.00175	±5%
p(x) (mean wall pressure)	---	±5%
mean velocity components	---	±5%
turbulent stresses	---	±15%

\*\*\*\*\*DISTRIBUTION OF SURFACE PRESSURE VS. STREAMWISE COORDINATE (x)\*\*\*\*\*

α = 10deg; Test 47; Runs 829 and 830; All Frames  
PT = 1.7006 atm; TT = 259.6 K; Re/m = 0.1690E+08; M<sub>∞</sub> = 2.8266

X (CM)	P/PT	CP
-8.500	0.3427E-01	-0.5294E-02
-8.000	0.3523E-01	-0.1163E-02
-6.500	0.3387E-01	-0.7282E-02
-6.000	0.3473E-01	-0.3334E-02
-5.500	0.3483E-01	-0.2808E-02
-5.000	0.3456E-01	-0.4199E-02
-4.500	0.3750E-01	0.1032E-02
-3.500	0.5842E-01	0.1169E+00
-3.000	0.6236E-01	0.1363E+00
-2.500	0.6528E-01	0.1510E+00
-2.000	0.6549E-01	0.1516E+00
-1.500	0.6578E-01	0.1541E+00
-1.000	0.6695E-01	0.1589E+00
-0.500	0.6792E-01	0.1649E+00
0.000	0.7589E-01	0.2041E+00
0.766	0.9148E-01	0.2834E+00
1.379	0.1248E+00	0.4515E+00
1.915	0.1047E+00	0.3497E+00
2.451	0.1938E+00	0.8003E+00
2.988	0.2165E+00	0.9150E+00
3.983	0.2394E+00	0.1031E+01

\*\*\*\*\*LDV FLOWFIELD DATA\*\*\*\*\*

α = 10deg; X = -6.500cm; Obtained 10/1/85-17:53:38  
PT = 1.7 atm; TT = 271.0 K; U<sub>∞</sub> = 578.0 m/s

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	0.6867	5.1261E-03	5.232E-03	2.335E-03	-1.190E-03	5.676E-03	0.000E+00
0.102	0.7421	6.0740E-03	4.160E-03	1.960E-03	-1.075E-03	4.590E-03	0.000E+00
0.152	0.7733	5.5033E-03	3.778E-03	1.772E-03	-1.077E-03	4.163E-03	0.000E+00
0.203	0.7992	5.2813E-03	3.355E-03	1.653E-03	-1.000E-03	3.756E-03	0.000E+00
0.254	0.8218	4.5980E-03	3.113E-03	1.559E-03	-9.547E-04	3.504E-03	0.000E+00
0.317	0.8443	4.1875E-03	2.751E-03	1.460E-03	-8.826E-04	3.159E-03	0.000E+00
0.381	0.8670	5.0013E-03	2.417E-03	1.326E-03	-7.878E-04	2.807E-03	0.000E+00
0.444	0.8889	3.1890E-03	2.170E-03	1.166E-03	-6.895E-04	2.502E-03	0.000E+00
0.508	0.9095	3.1548E-03	1.860E-03	1.036E-03	-5.826E-04	2.172E-03	0.000E+00
0.572	0.9274	2.8103E-03	1.602E-03	8.786E-04	-4.732E-04	1.860E-03	0.000E+00
0.635	0.9409	3.6600E-03	1.378E-03	7.499E-04	-3.931E-04	1.596E-03	0.000E+00
0.698	0.9592	6.1142E-03	1.036E-03	6.379E-04	-2.668E-04	1.255E-03	0.000E+00
0.762	0.9701	6.6672E-03	8.437E-04	5.164E-04	-1.967E-04	1.020E-03	0.000E+00
0.825	0.9810	7.2850E-03	6.430E-04	3.956E-04	-1.406E-04	7.790E-04	0.000E+00
0.889	0.9897	7.9275E-03	4.208E-04	2.904E-04	-8.006E-05	5.335E-04	0.000E+00
1.016	0.9992	8.5083E-03	2.207E-04	1.973E-04	-3.930E-05	3.135E-04	0.000E+00
1.143	1.0024	1.0894E-02	1.588E-04	1.572E-04	-1.346E-05	2.369E-04	0.000E+00
1.270	1.0035	1.0887E-02	1.637E-04	1.650E-04	-9.350E-06	2.465E-04	0.000E+00
1.397	1.0008	1.0484E-02	1.270E-04	1.296E-04	1.187E-06	1.924E-04	0.000E+00
1.524	1.0013	9.8296E-03	1.196E-04	1.225E-04	-2.217E-06	1.816E-04	0.000E+00

1.778	1.0006	9.3038E-03	1.174E-04	1.160E-04	1.944E-06	1.751E-04	0.000E+00
2.032	0.9999	1.1037E-02	1.237E-04	1.252E-04	4.832E-06	1.867E-04	0.000E+00

$\alpha = 10\text{deg}$ ;  $X = -4.500\text{cm}$ ; Obtained 10/1/85-21:18:59  
PT = 1.7 atm; TT = 271.0 K;  $u_\infty = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	0.6783	1.6354E-02	1.541E-02	3.561E-03	-1.674E-03	1.423E-02	2.259E-03
0.102	0.7378	1.7058E-02	1.014E-02	2.930E-03	-1.956E-03	9.800E-03	5.745E-04
0.152	0.7734	1.6377E-02	7.325E-03	2.622E-03	-1.966E-03	7.461E-03	7.659E-05
0.203	0.8037	1.4286E-02	5.386E-03	2.292E-03	-1.798E-03	5.759E-03	3.833E-05
0.254	0.8286	1.1434E-02	4.400E-03	2.145E-03	-1.632E-03	4.908E-03	0.000E+00
0.317	0.8538	9.8650E-03	3.795E-03	1.981E-03	-1.485E-03	4.332E-03	0.000E+00
0.381	0.8773	8.5000E-03	2.822E-03	1.692E-03	-1.111E-03	3.386E-03	0.000E+00
0.444	0.8994	6.8690E-03	2.283E-03	1.433E-03	-7.983E-04	2.787E-03	0.000E+00
0.508	0.9146	4.2111E-03	1.913E-03	1.158E-03	-6.643E-04	2.304E-03	0.000E+00
0.572	0.9335	3.6193E-03	1.671E-03	9.673E-04	-5.374E-04	1.978E-03	0.000E+00
0.635	0.9489	3.8295E-03	1.274E-03	7.842E-04	-3.538E-04	1.543E-03	0.000E+00
0.698	0.9652	4.3528E-03	9.733E-04	6.218E-04	-2.044E-04	1.196E-03	0.000E+00
0.762	0.9738	4.2056E-03	8.657E-04	5.473E-04	-1.901E-04	1.060E-03	0.000E+00
0.825	0.9829	5.4869E-03	6.463E-04	4.324E-04	-1.289E-04	8.090E-04	0.000E+00
0.889	0.9894	5.9883E-03	4.726E-04	3.528E-04	-8.015E-05	6.191E-04	0.000E+00
1.016	1.0019	8.1437E-03	2.671E-04	2.331E-04	-2.167E-05	3.752E-04	0.000E+00
1.143	1.0036	7.5632E-03	2.235E-04	2.191E-04	-9.908E-06	3.319E-04	0.000E+00
1.270	1.0049	7.0466E-03	1.820E-04	1.809E-04	3.831E-05	2.722E-04	0.000E+00
1.397	1.0030	5.2312E-03	1.808E-04	1.841E-04	3.950E-05	2.736E-04	0.000E+00
1.524	1.0037	6.7394E-03	1.551E-04	1.535E-04	1.695E-05	2.314E-04	0.000E+00
1.778	1.0012	5.3627E-03	1.477E-04	1.463E-04	-3.493E-07	2.205E-04	0.000E+00
2.032	1.0000	3.4085E-03	1.449E-04	1.450E-04	2.271E-06	2.174E-04	0.000E+00

$\alpha = 10\text{deg}$ ;  $X = -2.500\text{cm}$ ; Obtained 10/1/85-22:55:33  
PT = 1.7 atm; TT = 271.0 K;  $u_\infty = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	0.0554	1.0379E-02	3.204E-02	1.122E-02	2.179E-03	3.244E-02	4.059E-01
0.102	0.1574	1.9333E-02	4.038E-02	1.129E-02	1.721E-03	3.876E-02	2.268E-01
0.152	0.2595	3.1679E-02	4.189E-02	1.152E-02	1.278E-03	4.006E-02	1.077E-01
0.203	0.3480	4.3897E-02	4.484E-02	9.930E-03	1.972E-03	4.108E-02	5.902E-02
0.254	0.4341	6.3681E-02	4.008E-02	1.013E-02	1.958E-03	3.766E-02	2.137E-02
0.317	0.5176	8.6482E-02	3.477E-02	8.763E-03	1.553E-03	3.265E-02	7.774E-03
0.381	0.5799	1.0075E-01	2.928E-02	7.752E-03	1.092E-03	2.777E-02	3.064E-03
0.444	0.6399	1.1602E-01	2.325E-02	6.678E-03	-4.716E-05	2.245E-02	1.264E-03
0.508	0.6921	1.2262E-01	1.731E-02	5.421E-03	-9.957E-04	1.705E-02	2.298E-04
0.572	0.7251	1.2879E-01	1.459E-02	4.771E-03	-1.430E-03	1.452E-02	3.064E-04
0.635	0.7516	1.3559E-01	1.235E-02	4.351E-03	-2.254E-03	1.253E-02	3.833E-05
0.698	0.7889	1.3541E-01	1.001E-02	4.089E-03	-2.614E-03	1.057E-02	3.833E-05
0.762	0.8181	1.3326E-01	9.162E-03	4.132E-03	-3.292E-03	9.070E-03	3.833E-05
0.825	0.8567	1.2344E-01	7.481E-03	4.042E-03	-3.784E-03	8.642E-03	0.000E+00
0.889	0.8863	1.1005E-01	6.543E-03	4.455E-03	-4.188E-03	8.248E-03	0.000E+00
1.016	0.9365	7.8938E-02	4.001E-03	4.802E-03	-3.816E-03	6.602E-03	0.000E+00
1.143	0.9701	4.8226E-02	2.095E-03	4.044E-03	-2.595E-03	4.605E-03	0.000E+00
1.270	0.9870	2.6826E-02	1.047E-03	2.624E-03	-1.444E-03	2.753E-03	0.000E+00
1.397	0.9972	1.1309E-02	4.720E-04	1.187E-03	-5.671E-04	1.244E-03	0.000E+00
1.524	1.0018	2.5373E-03	2.120E-04	4.123E-04	-1.418E-04	4.682E-04	0.000E+00
1.778	1.0015	6.9572E-04	1.345E-04	1.424E-04	3.416E-06	2.077E-04	0.000E+00
2.032	1.0000	3.0658E-04	1.327E-04	1.381E-04	2.413E-05	2.030E-04	0.000E+00

$\alpha = 10\text{deg}$ ;  $X = -0.500\text{cm}$ ; Obtained 10/2/85-20:30:21  
PT = 1.7 atm; TT = 271.0 K;  $u_\infty = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	-0.1389	-7.8508E-03	1.480E-02	8.719E-03	2.534E-03	1.764E-02	8.796E-01
0.102	-0.1056	-1.5967E-02	1.879E-02	1.050E-02	2.450E-03	2.197E-02	8.130E-01
0.152	-0.0656	-2.0530E-02	2.331E-02	1.228E-02	2.036E-03	2.669E-02	7.221E-01
0.203	-0.0141	-1.9467E-02	3.095E-02	1.489E-02	2.345E-03	3.438E-02	5.995E-01
0.254	0.0400	-2.1900E-02	3.708E-02	1.563E-02	1.755E-03	3.953E-02	4.761E-01
0.317	0.1324	-1.9023E-02	4.394E-02	1.539E-02	1.037E-03	4.449E-02	3.006E-01
0.381	0.2066	-1.2768E-03	4.663E-02	1.6780E-02	1.507E-03	4.755E-02	1.947E-01

0.444	0.2824	8.9795E-03	4.569E-02	1.615E-02	1.262E-03	4.637E-02	1.160E-01
0.508	0.3572	2.6433E-02	4.245E-02	1.609E-02	1.387E-03	4.390E-02	5.843E-02
0.572	0.4144	3.6887E-02	3.949E-02	1.570E-02	1.633E-03	4.139E-02	3.174E-02
0.635	0.4772	5.4385E-02	3.520E-02	1.473E-02	1.966E-03	3.745E-02	1.457E-02
0.698	0.5253	6.9559E-02	2.941E-02	1.390E-02	1.568E-03	3.248E-02	7.324E-03
0.762	0.5739	8.1608E-02	2.773E-02	1.229E-02	2.381E-03	3.002E-02	3.743E-03
0.825	0.6142	9.6120E-02	2.233E-02	1.147E-02	1.494E-03	2.535E-02	1.383E-03
0.889	0.6738	1.1551E-01	1.906E-02	8.730E-03	1.074E-03	2.084E-02	8.138E-04
1.016	0.7285	1.3486E-01	1.295E-02	6.877E-03	1.049E-04	1.487E-02	2.441E-04
1.143	0.8122	1.5609E-01	7.481E-03	3.514E-03	-2.419E-04	8.246E-03	0.000E+00
1.270	0.8554	1.6503E-01	4.650E-03	2.149E-03	-2.047E-04	5.099E-03	0.000E+00
1.397	0.8959	1.7128E-01	1.507E-03	8.701E-04	-3.639E-04	1.783E-03	0.000E+00
1.524	0.9073	1.6842E-01	8.890E-04	7.521E-04	-3.510E-04	1.231E-03	0.000E+00
1.778	0.9242	1.4639E-01	6.027E-04	1.554E-03	-7.333E-04	1.617E-03	0.000E+00
2.032	0.9608	6.9851E-02	1.154E-03	3.881E-03	-1.930E-03	3.776E-03	0.000E+00
2.286	9.9909	1.4258E-02	4.583E-04	1.394E-03	-6.493E-04	1.389E-03	0.000E+00
2.540	0.9992	2.3703E-03	1.318E-04	1.761E-04	-2.875E-05	2.309E-04	0.000E+00
2.794	1.0002	-1.5291E-04	1.048E-04	1.141E-04	8.940E-06	1.641E-04	0.000E+00
3.048	1.0000	4.1733E-05	1.335E-04	1.348E-04	1.287E-05	2.012E-04	0.000E+00

$\alpha = 10\text{deg}$ ;  $X = 0.766\text{cm}$ ; Obtained 10/4/85-20:36:58  
PT = 1.7 atm; TT = 271.0 K;  $u_\infty = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	0.0938	6.9128E-03	3.605E-02	2.202E-02	1.036E-02	4.355E-02	3.508E-01
0.102	0.1731	3.6921E-03	4.779E-02	2.037E-02	7.392E-03	5.112E-02	2.482E-01
0.152	0.2508	5.3503E-03	5.199E-02	2.122E-02	6.623E-03	5.491E-02	1.682E-01
0.203	0.3154	1.5617E-02	5.345E-02	2.103E-02	6.917E-03	5.586E-02	1.1153E-01
0.254	0.3714	3.2674E-02	5.045E-02	2.108E-02	7.631E-03	5.365E-02	7.389E-02
0.317	0.4365	5.6914E-02	4.482E-02	2.010E-02	7.549E-03	4.869E-02	4.141E-02
0.381	0.4797	7.6382E-02	4.022E-02	1.892E-02	6.817E-03	4.435E-02	2.461E-02
0.444	0.5243	1.0017E-01	3.609E-02	1.750E-02	6.044E-03	4.019E-02	1.550E-02
0.508	0.5862	1.3067E-01	2.857E-02	1.502E-02	4.498E-03	3.269E-02	6.901E-03
0.572	0.6147	1.4552E-01	2.518E-02	1.348E-02	3.592E-03	2.900E-02	3.841E-03
0.635	0.6511	1.6168E-01	2.238E-02	1.216E-02	2.984E-03	2.590E-02	3.776E-03
0.698	0.6941	1.8118E-01	1.788E-02	9.740E-03	1.781E-03	2.071E-02	1.042E-03
0.762	0.7335	1.9219E-01	1.499E-02	7.432E-03	1.112E-03	1.682E-02	5.859E-04
0.825	0.7581	1.9709E-01	1.274E-02	6.437E-03	4.142E-04	1.438E-02	3.906E-04
0.889	0.8024	2.0503E-01	9.698E-02	4.419E-03	-5.552E-04	1.059E-02	0.000E+00
1.016	0.8479	2.0938E-01	5.271E-03	2.522E-03	-1.197E-03	5.845E-03	0.000E+00
1.143	0.8806	2.0122E-01	2.628E-03	1.599E-03	-1.070E-03	3.171E-03	0.000E+00
1.270	0.8965	1.9157E-01	1.244E-03	1.019E-03	-7.242E-04	1.698E-03	0.000E+00
1.397	0.9051	1.8215E-01	6.869E-04	7.159E-04	-4.845E-04	1.052E-03	0.000E+00
1.524	0.9157	1.6551E-01	4.170E-04	7.050E-04	-3.357E-04	8.415E-04	0.000E+00
1.778	0.9328	1.3095E-01	6.554E-04	2.088E-03	-9.597E-04	2.057E-03	0.000E+00
2.032	0.9607	7.3758E-02	1.027E-03	3.614E-03	-1.739E-03	3.481E-03	0.000E+00
2.286	0.9936	1.0193E-02	3.960E-04	1.188E-03	-5.490E-04	1.189E-03	0.000E+00
2.540	1.0000	3.7215E-04	1.196E-04	1.715E-04	-2.448E-05	2.183E-04	0.000E+00
2.794	0.9997	5.7221E-05	1.103E-04	1.176E-04	1.175E-05	1.710E-04	0.000E+00
3.048	1.0000	-9.8584E-04	1.099E-04	1.162E-04	1.306E-05	1.696E-04	0.000E+00

$\alpha = 10\text{deg}$ ;  $X = 1.915\text{cm}$ ; Obtained 10/3/85-21:09:35  
PT = 1.7 atm; TT = 271.0 K;  $u_\infty = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	UV	U <sub>2</sub> V <sub>2</sub>	GMINUS
0.051	0.3751	1.9584E-01	2.608E-02	2.048E-02	4.301E-03	3.492E-02	1.224E-02
0.102	0.4454	1.8995E-01	2.748E-02	1.869E-02	3.497E-03	3.463E-02	8.919E-03
0.152	0.5009	1.9573E-01	2.619E-02	1.675E-02	3.989E-03	3.221E-02	7.031E-03
0.203	0.5396	2.0072E-01	2.337E-02	1.579E-02	4.008E-03	2.937E-02	3.906E-03
0.254	0.5773	2.1769E-01	2.078E-02	1.413E-02	4.228E-03	2.619E-02	4.362E-03
0.317	0.6160	2.4744E-01	1.473E-02	1.173E-02	2.975E-03	1.985E-02	1.302E-03
0.381	0.6423	2.7257E-01	1.168E-02	9.636E-03	2.375E-03	1.599E-02	1.107E-03
0.444	0.6650	2.8790E-01	9.628E-03	8.027E-03	2.218E-03	1.324E-02	1.302E-04
0.508	0.6880	3.1036E-01	6.951E-03	5.829E-03	9.858E-04	9.585E-03	2.604E-04
0.572	0.7074	3.1784E-01	6.062E-03	4.837E-03	5.651E-04	8.174E-03	1.302E-04
0.635	0.7221	3.2509E-01	4.830E-03	3.368E-03	-8.414E-05	6.148E-03	6.515E-05
0.698	0.7378	3.2500E-01	4.401E-03	2.846E-03	-4.427E-04	5.436E-03	0.000E+00
0.762	0.7524	3.2185E-01	3.953E-03	2.187E-03	-9.958E-04	4.605E-03	0.000E+00
0.825	0.7703	3.1118E-01	4.160E-03	2.313E-03	-1.450E-03	4.855E-03	0.000E+00
0.889	0.7891	2.9734E-01	4.236E-03	2.480E-03	-2.101E-03	5.037E-03	0.000E+00

1.016	0.8282	2.6234E-01	4.756E-03	3.522E-03	-3.424E-03	6.209E-03	0.000E+00
1.143	0.8727	2.1651E-01	4.068E-03	3.654E-03	-3.444E-03	5.792E-03	0.000E+00
1.270	0.9037	1.7947E-01	2.447E-03	2.846E-03	-2.357E-03	3.969E-03	0.000E+00
1.397	0.9307	1.4379E-01	1.137E-03	2.272E-03	-1.352E-03	2.557E-03	0.000E+00
1.524	0.9453	1.1604E-01	9.085E-04	2.571E-03	-1.311E-03	2.609E-03	0.000E+00
1.778	0.9832	4.1551E-02	8.782E-04	2.970E-03	-1.460E-03	2.886E-03	0.000E+00
2.032	1.0012	3.3496E-03	1.983E-04	4.785E-04	-1.889E-04	5.077E-04	0.000E+00
2.286	1.0006	-1.4544E-03	9.893E-05	1.234E-04	-1.208E-05	1.667E-04	0.000E+00
2.540	1.0001	-5.0309E-03	8.689E-05	9.737E-05	3.415E-06	1.382E-04	0.000E+00

$\alpha = 10\text{deg}$ ; X = 3.447cm; Obtained 10/3/85-22:49:02  
PT = 1.7 atm; TT = 271.0 K;  $u_{\infty} = 578.0 \text{ m/s}$

Y (CM)	U <sub>MEAN</sub>	V <sub>MEAN</sub>	U <sub>2</sub>	V <sub>2</sub>	U <sub>V</sub>	U <sub>2V2</sub>	GMINUS
0.051	0.4640	3.6066E-01	9.306E-03	8.936E-03	3.745E-03	1.368E-02	0.000E+00
0.102	0.5074	3.5398E-01	8.700E-03	7.810E-03	1.681E-03	1.238E-02	6.509E-05
0.152	0.5382	3.5790E-01	7.272E-03	6.454E-03	1.054E-03	1.029E-02	6.509E-05
0.203	0.5554	3.5421E-01	6.957E-03	6.376E-03	7.977E-04	1.000E-02	0.000E+00
0.254	0.5697	3.5839E-01	5.754E-03	5.333E-03	6.779E-04	8.316E-03	6.509E-05
0.317	0.5845	3.7119E-01	4.310E-03	4.085E-03	1.286E-04	6.296E-03	1.953E-04
0.381	0.5902	3.7947E-01	3.481E-03	3.466E-03	6.452E-05	5.210E-03	6.509E-05
0.444	0.5979	3.8270E-01	3.290E-03	2.997E-03	-1.984E-05	4.716E-09	0.000E+00
0.508	0.6019	3.9052E-01	2.768E-03	2.481E-03	-2.796E-05	3.937E-03	0.000E+00
0.572	0.6072	3.9539E-01	2.340E-03	1.913E-03	-1.907E-04	3.190E-03	0.000E+00
0.635	0.6057	3.9967E-01	2.147E-03	1.610E-03	-3.041E-04	2.818E-03	0.000E+00
0.698	0.6083	4.0180E-01	1.886E-03	1.366E-03	-2.112E-04	2.439E-03	0.000E+00
0.762	0.6094	4.0137E-01	1.954E-03	1.292E-03	-1.995E-04	2.435E-03	0.000E+00
0.825	0.6160	4.0039E-01	1.863E-03	1.111E-03	-1.816E-04	2.231E-03	0.000E+00
0.889	0.6154	3.9906E-01	1.816E-03	1.021E-03	-1.754E-05	2.127E-03	0.000E+00
1.016	0.6070	3.9207E-01	2.308E-03	1.123E-03	1.392E-04	2.573E-03	0.00E+000
1.143	0.6061	3.7149E-01	3.948E-03	1.419E-03	5.750E-04	4.025E-03	0.000E+00
1.270	0.5808	3.3562E-01	5.642E-03	1.646E-03	1.909E-04	5.466E-03	0.000E+00
1.397	0.5965	2.9490E-01	6.199E-03	2.925E-03	-2.499E-03	6.842E-03	0.000E+00
1.524	0.6548	2.4484E-01	8.903E-03	5.325E-03	-5.773E-03	1.067E-02	0.000E+00
2.032	1.0025	-1.0258E-02	1.371E-04	1.424E-04	-4.459E-05	2.096E-04	0.000E+00
2.286	1.0018	-0.5308E-03	7.891E-05	8.937E-05	5.628E-06	1.262E-04	0.000E+00
2.540	1.0001	-8.1080E-03	7.925E-05	8.752E-05	8.536E-06	1.251E-04	0.000E+00

\*\*\*\*\*END OF FILE\*\*\*\*\*



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Ref.: 85, 48

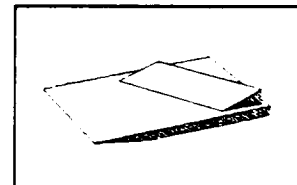
Author: McKenzie, T. M., *et al*

Geometry: 3-D Swept Compression Corner

Mach number: 3

Data:  $p_{\text{wall}}$ , mean flowfield surveys ("cobra" probe)

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Settles, G.S., Horstman, C.C. and McKenzie, T.M., "Flowfield Scaling of a Swept Compression Corner Interaction - A Comparison of Experiment and Computation," AIAA Paper 84-0096, 1984, and *AIAA Journal*, Vol. 24, May 1986, pp. 744-752.

Knight, D.D., Horstman, C.C., Bogdonoff, S.M., Raufer, D. and Ketchum, A., "Supersonic Turbulent Flow Past a Swept Compression Corner at Mach 3, Part II," *AIAA Paper 88-0310*, 1988.

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The data include surface pressures and "cobra" yaw-probe surveys of the swept interaction generated by a swept compression corner mounted on the floor of a Mach 2.95 wind tunnel. The corner was swept back by 40 degrees from the normal to the freestream direction, and had a streamwise compression angle of 24 degrees. The freestream unit Reynolds number was 63 million/meter and the wall temperature was nearly adiabatic.

The compression corner was mounted with its apex 30 cm downstream from the exit of the wind tunnel nozzle. Cobra-probe data were taken at 14 stations in a vertical streamwise (x-y) plane located 8.89 cm (3.5 in) spanwise from the apex of the swept corner. The x-dimension given in the profile tables is defined such that its origin coincides with the compression corner location, values being positive downstream. The vertical traverse data are indexed by coordinate y.

The first three columns of the profile tables list y, pitot pressure, and yaw angle, respectively. The final two columns denoted DELP and SENSITCOEF convey raw cobra-probe information which should be ignored by most users of this dataset.

The mean surface pressure data were taken on 4 streamwise "cuts" located at 2, 3, 4, and 5 inches from the apex of the swept corner. Here (x-xcorner) in the tables has the same meaning as x defined above for the surveys.

A similar set of measurements was also carried out on a flat plate with a thinner initial boundary layer. The comparison of these two datasets was used in Ref. 85 to confirm a scaling law for incoming boundary-layer thickness effects on such interactions. Unfortunately, incomplete data were available for the thin-boundary-layer case, so it is not tabulated here.

The incoming turbulent boundary layer for the tabulated data had an overall thickness of 1.54 cm, a displacement thickness of 0.408 cm, and a momentum thickness of 0.0807 cm. The incoming skin friction coefficient was 0.00116, and the wake-strength parameter of the boundary layer was 0.7.

The uncertainty of probe tip position was  $\pm 0.02$  cm in x and y. The experimenters found their data repeatable to  $\pm 1\%$ , and claimed an overall accuracy of  $\pm 1\%$  in yaw angle, though in retrospect that seems to have been wishful thinking.

McKenzie et al. (24,40) Supt Compression Corner

Thick Boundary Layer

Wall Pressure

Experimental Surface Pressure Data

pressure.2 Experimental p/pinfinity at z = 2 inch  
pressure.3 Experimental p/pinfinity at z = 3 inch  
pressure.4 Experimental p/pinfinity at z = 4 inch  
pressure.5 Experimental p/pinfinity at z = 5 inch

xcor.2 Experimental (x - scorer)/deltainf at z = 2 inch  
xcor.3 Experimental (x - scorer)/deltainf at z = 3 inch  
xcor.4 Experimental (x - scorer)/deltainf at z = 4 inch  
xcor.5 Experimental (x - scorer)/deltainf at z = 5 inch

where pinfinity = 0.209665 Pa, acorner = z tan lambda0, lambda0 = 40 deg,  
and deltainf set equal to 0.5 inch.

pressure.2	xcor.2	pressure.3	xcor.3	pressure.4	xcor.4	pressure.5	xcor.5
0.1014E+01	-3818E+01	0.9833E+00	-5898E+01	0.1052E+01	-4325E+01	0.1114E+01	-4452E+01
0.1000E+01	-3418E+01	0.1007E+01	-5598E+01	0.1052E+01	-4125E+01	0.1041E+01	-4052E+01
0.9948E+00	-3018E+01	0.1003E+01	-5098E+01	0.1050E+01	-3925E+01	0.1172E+01	-3652E+01
0.9919E+00	-2818E+01	0.1001E+01	-4498E+01	0.1045E+01	-3725E+01	0.1653E+01	-3252E+01
0.9876E+00	-2618E+01	0.1008E+01	-3898E+01	0.1043E+01	-3525E+01	0.1828E+01	-2852E+01
0.9957E+00	-2418E+01	0.9671E+00	-3198E+01	0.1097E+01	-3125E+01	0.1914E+01	-2452E+01
0.9919E+00	-2218E+01	0.9990E+00	-2598E+01	0.1349E+01	-2725E+01	0.2002E+01	-2052E+01
0.9847E+00	-2018E+01	0.9948E+00	-1998E+01	0.1613E+01	-2325E+01	0.2059E+01	-1652E+01
0.1203E+01	-1820E+01	0.9981E+00	-1398E+01	0.1744E+01	-1925E+01	0.2079E+01	-1252E+01
0.1547E+01	-1620E+01	0.1107E+01	-798E+01	0.1818E+01	-1525E+01	0.2096E+01	-852E+01
0.1674E+01	-1420E+01	0.1399E+01	-2198E+01	0.1870E+01	-1125E+01	0.2105E+01	-452E+01
0.1771E+01	-1220E+01	0.1617E+01	-1598E+01	0.1920E+01	-725E+01	0.2121E+01	-52E+01
0.1837E+01	-1020E+01	0.1723E+01	-998E+01	0.1967E+01	-325E+01	0.2142E+01	-452E+01
0.1842E+01	-820E+01	0.1797E+01	-398E+01	0.1986E+01	-125E+01	0.2170E+01	-252E+01
0.1906E+01	-620E+01	0.1839E+01	1598E+01	0.2021E+01	1527E+01	0.2220E+01	1520E+01
0.1959E+01	-420E+01	0.1875E+01	1398E+01	0.2037E+01	1127E+01	0.2220E+01	1520E+01
0.1959E+01	-220E+01	0.1924E+01	1198E+01	0.2051E+01	726E+01	0.2220E+01	1520E+01
0.2140E+01	-200E+01	0.1949E+01	998E+01	0.2040E+01	326E+01	0.2220E+01	1520E+01
		0.1958E+01	798E+01	0.2041E+01	-726E+01	0.2220E+01	1520E+01
		0.1970E+01	598E+01	0.2051E+01	-326E+01	0.2220E+01	1520E+01
		0.1994E+01	398E+01	0.2066E+01	-126E+01	0.2220E+01	1520E+01
		0.2013E+01	198E+01	0.2151E+01	126E+01	0.2220E+01	1520E+01
		0.2033E+01	98E+01	0.2217E+01	4032E+01	0.2220E+01	1520E+01
		0.2181E+01	0.000E+00	0.2335E+01	0.4032E+01	0.2220E+01	1520E+01
		0.2433E+01	0.8020E+00	0.2450E+01	0.8032E+01	0.2220E+01	1520E+01
		0.2799E+01	0.1202E+01	0.2594E+01	0.1003E+01	0.2220E+01	1520E+01
		0.3074E+01	0.1602E+01	0.2678E+01	0.1203E+01	0.2220E+01	1520E+01
		0.3294E+01	0.2002E+01	0.2792E+01	0.1403E+01	0.2220E+01	1520E+01
		0.3495E+01	0.2402E+01	0.2902E+01	0.1603E+01	0.2220E+01	1520E+01
		0.3677E+01	0.2802E+01	0.3112E+01	0.2003E+01	0.2220E+01	1520E+01
				0.3312E+01	0.2403E+01	0.2220E+01	1520E+01
				0.3476E+01	0.2803E+01	0.2220E+01	1520E+01

(24,40) Supt Compression Corner  
Thick Boundary Layer  
Cobra Probe Surveys

P0=100 PSIA  
TO IS IN DATA FILES  
PRESTREAM MACH NUMBER=2.93

DATA FORMAT:

RUN TEST TYPE NUM OF PTS STATION P0 TO X  
DATE TITLE  
Y(METERS) PT(EPAS) TAN(DEG) DELP(EPAS) SENSITCOEF

DATA FILES FOR COBRA SURVEYS:

FILE STATION X(INCHES)

SRF23.DAT 2 -1.6

SRF25.DAT 3 -1.4

STAGM.DAT 4 -1.2

SRF20.DAT 5 -1.0

SRF30.DAT 6 -0.7

SRF33.DAT 7 -0.4

STAGM.DAT 8 -0.1

SRP12.DAT 10 0.7

SRP10.DAT 13 1.3

SRP11.DAT 14 1.5

STAGM.DAT 15 1.8

STAGM.DAT 16 2.1

STAGM.DAT 17 2.4

SRP24.DAT 18 3.0

Z=3.5 INCHES FOR ALL STATIONS

X = STREAMWISE DISTANCE FROM WEDGE CORNER

Z = SPANWISE DISTANCE FROM WEDGE APEX.

File SRF23.DAT

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2203 2 0 27 5 .6902E+06 258.5 .2196E+05-1.600  
1982 265 COBRA PROBE SURVEYS 24,40;2-3.5INS. ON FLOOR BY W.MCK.

.2794E-03 .3807E+05 .5037E-01-842.7 .1520E-02  
.1193E-02 .8146E+05 .4276E-01-1871. .1520E-02  
.2189E-02 .9668E+05 .4885E-01-1627. .1520E-02  
.3198E-02 .1107E+06 .5608E-01-20.86 .1520E-02  
.4189E-02 .1290E+06 .5265E-01-851.1 .1520E-02  
.5171E-02 .1403E+06 .5950E-01-163.1 .1520E-02  
.6198E-02 .1496E+06 .4580E-01-3368. .1520E-02  
.7232E-02 .1585E+06 .5456E-01-2256. .1520E-02  
.8212E-02 .1687E+06 .6542E-01-4280. .1520E-02  
.9232E-02 .1810E+06 .6314E-01-3727. .1520E-02  
.1024E-01 .1971E+06 .3342E-01-3199. .1520E-02  
.1124E-01 .2060E+06 .3667E-01-2594. .1520E-02  
.1221E-01 .2078E+06 .2373E-01-4221. .1520E-02  
.1328E-01 .2213E+06 .1840E-01-3451. .1520E-02  
.1425E-01 .2279E+06 .1573E-01-2963. .1520E-02  
.1525E-01 .2300E+06 .1992E-01-2145. .1520E-02  
.1626E-01 .2328E+06 .2563E-01-1430. .1520E-02  
.1723E-01 .2343E+06 .1878E-01-420.9 .1520E-02  
.1821E-01 .2345E+06 .1535E-01-1319. .1520E-02  
.1924E-01 .2339E+06 .1383E-01-1182. .1520E-02  
.2023E-01 .2333E+06 .1078E-01-1303. .1520E-02  
.2121E-01 .2344E+06 .7359E-02-442.2 .1520E-02  
.2224E-01 .2350E+06 .9189E-02-456.7 .1520E-02  
.2324E-01 .2352E+06 .1890E-01-572.7 .1520E-02  
.2422E-01 .2349E+06 .1358E-01-301.6 .1520E-02  
.2522E-01 .2344E+06 .4555E-01-181.1 .1520E-02  
.2617E-01 .2344E+06 .1392 1236. .1520E-02

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File SR125.DAT
2203 4 0 26 5 6002E+06 258.6 2059E+05 1.400 2148E+05 1.200
1982 265 COBRA PROBE SURVEYS 24,40;2-3.51MS. ON FLOOR BY M.MCK.
-2794E-03 3570E+05 2.555 -4018. -1520E-02
-1279E-02 7398E+05 2.554 -4284. -1520E-02
-2258E-02 1028E+06 2.559 3694. -1520E-02
-3276E-02 1208E+06 2.541 2523. -1520E-02
-3760E-02 1228E+06 2.201 4216. -1520E-02
-4770E-02 1286E+06 7352 -3558. -1520E-02
-5786E-02 1402E+06 5699 -291.2 -1520E-02
-6378E-02 1543E+06 2610 1485. -1520E-02
-7802E-02 1647E+06 2220E 01 2787. -1520E-02
-8790E-02 1722E+06 1459E 01 3650. -1520E-02
-9195E-02 1813E+06 6801E 01 3505. -1520E-02
-1081E-01 1950E+06 5544E 01 2882. -1520E-02
-1181E-01 2088E+06 6344E 01 1840. -1520E-02
-1283E-01 2176E+06 6115E 01 1650. -1520E-02
-1366E-01 2224E+06 7257E 01 1944. -1520E-02
-1464E-01 2274E+06 7181E 01 1623. -1520E-02
-1584E-01 2299E+06 7295E 01 1714. -1520E-02
-1685E-01 2339E+06 6953E 01 612.5 -1520E-02
-1781E-01 2321E+06 6724E 01 804.1 -1520E-02
-1878E-01 2328E+06 6458E 01 116.8 -1520E-02
-1980E-01 2328E+06 6344E 01 155.4 -1520E-02
-2080E-01 2339E+06 5089E 01 77.54 -1520E-02
-2285E-01 2339E+06 4935E 01 105.6 -1520E-02
-2384E-01 2339E+06 2403 -2405. -1520E-02
-2586E-01 2337E+06 4367 -4317. -1520E-02
-2840E-01 2337E+06 4362 -3814. -1520E-02

File SR126.DAT
2203 3 0 29 5 6920E+06 271.0 2148E+05 1.200
1982 267 COBRA PROBE SURVEYS 24,40;2-3.51MS. ON FLOOR BY M.MCK.
-2794E-03 3965E+05 49.15 -2209. -8825E-01
-7462E-03 4136E+05 16.13 -4343. -8825E-01
-1830E-02 7028E+05 3.782 3060. -1520E-02
-2816E-02 1038E+06 3.780 1154. -1520E-02
-3783E-02 1264E+06 3.775 2981. -1520E-02
-4794E-02 1640E+06 3.764 705.0 -1520E-02
-5788E-02 1610E+06 3.581 387.5 -1520E-02
-5995E-02 1469E+06 3.390 2210. -1520E-02
-6517E-02 1539E+06 2.161 3632. -1520E-02
-7015E-02 1614E+06 1.603 1201. -1520E-02
-8040E-02 1680E+06 4634E 01 3897. -1520E-02
-9037E-02 1793E+06 5319E 01 2567. -1520E-02
-1066E-01 1920E+06 5053E 01 2108. -1520E-02
-1105E-01 2051E+06 4824E 01 2853. -1520E-02
-1207E-01 2143E+06 4672E 01 2898. -1520E-02
-1307E-01 2227E+06 5015E 01 2303. -1520E-02
-1410E-01 2280E+06 4749E 01 2549. -1520E-02
-1508E-01 2323E+06 4254E 01 2698. -1520E-02
-1602E-01 2344E+06 3781E 01 2324. -1520E-02
-1700E-01 2367E+06 4161E 01 2184. -1520E-02
-1799E-01 2352E+06 4542E 01 2595. -1520E-02
-1901E-01 2335E+06 6293E 01 2309. -1520E-02
-2098E-01 2365E+06 4352E 01 2016. -1520E-02
-2298E-01 2367E+06 5532E 01 1520. -1520E-02
-2403E-01 2373E+06 6628E 01 1335. -1520E-02
-2503E-01 2378E+06 8628E 02 614.5 -1520E-02
-2599E-01 2365E+06 1571E 01 1142. -1520E-02
-2497E-01 2362E+06 1434E 01 1236. -1520E-02
-2592E-01 2378E+06 5202E 01 40.99 -1520E-02

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File SR129.DAT
2303 1 0 29 5 6900E+06 261.1 2289E+05 1.000
1982 266 COBRA PROBE SURVEYS 24,40;2-3.51MS. ON FLOOR BY M.MCK.
-1373E-02 5267E+05 29.26 116.5 -2259E-01
-1394E-02 5267E+05 29.27 103.8 -2259E-01
-1436E-02 5331E+05 21.08 4065. -2712E-01
-2486E-02 7808E+05 6.690 2641. -1520E-02
-2704E-02 8222E+05 6.490 2049. -1520E-02
-3719E-02 1168E+06 4.141 1933. -1520E-02
-4704E-02 1389E+06 4.135 1465. -1520E-02
-5720E-02 1597E+06 4.132 1381. -1520E-02
-6766E-02 1867E+06 4.124 1629. -1520E-02
-7731E-02 2077E+06 4.113 2658. -1520E-02
-8733E-02 2121E+06 3.683 2052. -1520E-02
-9778E-02 2015E+06 3.042 3364. -1520E-02
-10994E-02 1974E+06 2.862 2710. -1520E-02
-1100E-01 2064E+06 6852 2678. -1520E-02
-1201E-01 2125E+06 2606 3394. -1520E-02
-1305E-01 2238E+06 2610 2734. -1520E-02
-1405E-01 2284E+06 2580 1790. -1520E-02
-1503E-01 2331E+06 2610 1229. -1520E-02
-1607E-01 2382E+06 2644 1227. -1520E-02
-1703E-01 2355E+06 2538 671.3 -1520E-02
-1804E-01 2368E+06 2614 700.2 -1520E-02
-1903E-01 2372E+06 2587 378.8 -1520E-02
-2003E-01 2378E+06 2606 455.9 -1520E-02
-2105E-01 2377E+06 2584 103.2 -1520E-02
-2206E-01 2369E+06 2610 114.9 -1520E-02
-2306E-01 2377E+06 2599 552.8 -1520E-02
-2403E-01 2402E+06 2637 66.69 -1520E-02
-2504E-01 2390E+06 2644 459.6 -1520E-02
-2595E-01 2392E+06 3947 964.5 -1520E-02

File SR130.DAT
2303 1 0 45 5 6911E+06 270.0 2341E+05 0.700
1982 266 COBRA PROBE SURVEYS 24,40;2-3.51MS. ON FLOOR BY M.MCK.
-4362E-03 5921E+05 99.09 44.60 -7755E-01
-9309E-03 6112E+05 91.08 35.37 -4808E-01
-9672E-03 6149E+05 91.11 6.650 -4808E-01
-1979E-02 6321E+05 51.28 79.95 -2211E-01
-2005E-02 6362E+05 51.29 149.4 -2211E-01
-2024E-02 6390E+05 49.68 272.3 -2211E-01
-3029E-02 7007E+05 25.96 170.0 -1384E-01
-3054E-02 6950E+05 25.97 38.38 -1384E-01
-3078E-02 6951E+05 24.22 194.3 -1384E-01
-3118E-02 7264E+05 23.20 56.31 -1384E-01
-3150E-02 6840E+05 22.75 35.77 -1384E-01
-3170E-02 7273E+05 21.63 194.6 -1384E-01
-3252E-02 7329E+05 21.24 34.52 -1384E-01
-3343E-02 7401E+05 19.58 206.8 -1384E-01
-3392E-02 7271E+05 19.58 314.2 -1384E-01
-3424E-02 7467E+05 14.90 261.1 -1384E-01
-3483E-02 7438E+05 14.90 3558. -1394E-01
-3992E-02 8603E+05 9.636 3205. -1520E-02
-4076E-02 8610E+05 9.509 3568. -1520E-02
-4326E-02 9707E+05 7.232 2280. -1520E-02
-4820E-02 1028E+06 5.562 2924. -1520E-02
-5034E-02 1316E+06 4.199 1536. -1520E-02
-5879E-02 1316E+06 4.083 645.8 -1520E-02
-7849E-02 1842E+06 3.642 2566. -1520E-02
-8874E-02 2131E+06 3.153 3213. -1520E-02
-9893E-02 2302E+06 3.151 1304. -1520E-02
-1089E-01 2593E+06 3.142 2340. -1520E-02

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File SR133.DAT
2401 2 0 45 5 6922E+06 270.4 2213E+05 4.000
1982 267 COBRA PROBE SURVEYS 24,40;2-3.51MS. ON FLOOR BY M.MCK.
-2794E-03 5968E+05 105.1 3390. -4310E-01
-1323E-02 6804E+05 82.14 95.78 -4310E-01
-1350E-02 6809E+05 82.16 6091. -4310E-01
-1369E-02 6829E+05 82.80 116.3 -4310E-01
-1418E-02 6817E+05 80.94 105.8 -4310E-01
-1440E-02 6844E+05 81.25 57.94 -4310E-01
-1666E-02 6827E+05 80.19 252.3 -4310E-01
-1962E-02 6850E+05 70.16 150.5 -2270E-01
-1995E-02 6859E+05 69.13 307.8 -2270E-01
-3016E-02 6791E+05 49.12 53.52 -1562E-01
-3039E-02 6791E+05 49.12 228.1 -1562E-01
-3119E-02 6710E+05 47.47 342.4 -1562E-01
-3176E-02 6784E+05 46.71 258.7 -1562E-01
-3212E-02 6844E+05 46.21 32.59 -1562E-01
-4248E-02 7071E+05 29.24 111.0 -1334E-01
-4282E-02 7098E+05 29.24 324.7 -1334E-01
-4297E-02 7061E+05 23.81 3977. -1334E-01
-5292E-02 8326E+05 12.77 3618. -1520E-02
-5344E-02 8184E+05 12.74 3755. -1520E-02
-5604E-02 9182E+05 10.05 3975. -1520E-02
-5860E-02 8808E+05 9.296 3877. -1520E-02
-6374E-02 1007E+06 7.296 2817. -1520E-02
-6492E-02 1083E+06 6.964 3334. -1520E-02
-7632E-02 1290E+06 4.078 6142. -1520E-02
-8645E-02 1662E+06 3.808 4135. -1520E-02
-9662E-02 1861E+06 3.457 2670. -1520E-02
-1065E-01 2129E+06 3.270 649.5 -1520E-02
-1163E-01 2407E+06 2.932 1320. -1520E-02
-1264E-01 2674E+06 2.829 454.3 -1520E-02
-1368E-01 2956E+06 2.838 379.5 -1520E-02
-1469E-01 3132E+06 2.829 747.7 -1520E-02
-1566E-01 3347E+06 2.822 2406. -1520E-02
-1670E-01 3350E+06 2.926 48.28 -1520E-02
-1765E-01 2918E+06 3.569 3706. -1520E-02
-1817E-01 2609E+06 3.532 843.1 -1520E-02
-1839E-01 2683E+06 3.268 2877. -1520E-02
-1941E-01 2417E+06 3.848 1815. -1520E-02
-204E-01 2395E+06 1096E 01 1866. -1520E-02
-2139E-01 2395E+06 1781E 01 1545. -1520E-02
-2242E-01 2403E+06 8678E 02 1366. -1520E-02
-2344E-01 2403E+06 8678E 02 1237. -1520E-02
-2437E-01 2394E+06 3352E 02 725.1 -1520E-02
-2540E-01 2394E+06 9059E 02 989.1 -1520E-02
-2635E-01 2398E+06 2070 1064. -1520E-02

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file SR10M.DAT
2401 4 0 34 5 .6912E+06 269.8 .2107E+05 .1000
1982 267 COBRA PROBE SURVEYS 24,40,2-3,51MS, ON FLOOR BY M. MCK.

.8645E-03 .7241E+05 50.00 .103.9 .2231E-01
.2958E-02 .6964E+05 62.16 .236.2 .1441E-01
.3007E-02 .6975E+05 62.17 .61.97 .1441E-01
.3081E-02 .6802E+05 48.96 .339.2 .1304E-01
.4035E-02 .6854E+05 48.97 .247.1 .1304E-01
.5032E-02 .7042E+05 35.53 .541.2 .1179E-01
.5090E-02 .7115E+05 35.53 .64.91 .1179E-01
.5131E-02 .7034E+05 34.56 .634.6 .1179E-01
.6439E-02 .7628E+05 23.22 .408.0 .9129E-02
.7279E-02 .9551E+05 14.19 .594.0 .6997E-02
.7355E-02 .9649E+05 10.48 .4329.9 .6180E-02
.8582E-02 .1219E+06 6.738 .3662.2 .1520E-02
.9077E-02 .1411E+06 6.272 .3524.2 .1520E-02
.1010E-01 .1572E+06 4.837 .2930.0 .1520E-02
.1109E-01 .1853E+06 4.816 .193.5 .1520E-02
.1158E-01 .1988E+06 4.419 .1872.2 .1520E-02
.1259E-01 .2252E+06 2.937 .3393.7 .1520E-02
.1362E-01 .2597E+06 2.938 .3307.7 .1520E-02
.1453E-01 .3029E+06 2.928 .4137.7 .1520E-02
.1559E-01 .3209E+06 2.574 .2348.0 .1520E-02
.1644E-01 .3184E+06 2.310 .2308.0 .1520E-02
.1788E-01 .3353E+06 2.415 .3627.7 .1520E-02
.1861E-01 .3433E+06 2.905 .4302.2 .1520E-02
.1959E-01 .3433E+06 2.884 .4265.2 .1520E-02
.2060E-01 .3465E+06 2.875 .3217.7 .1520E-02
.2108E-01 .3444E+06 2.642 .2702.2 .1520E-02
.2210E-01 .3003E+06 4.581 .928.4 .1520E-02
.2233E-01 .2742E+06 4.655 .757.7 .1520E-02
.2334E-01 .2417E+06 1.673 .3659.0 .1520E-02
.2431E-01 .2379E+06 .6833 .901.8 .1520E-02
.2532E-01 .2383E+06 .6807 .1045.5 .1520E-02
.2628E-01 .2388E+06 .474.9 .543.6 .1520E-02

file SRP12.DAT
2701 4 0 48 5 .6888E+06 267.0 .2121E+05 .7000
1982 270 COBRA PROBE SURVEYS 24,40,2-3,51MS, ON FLOOR BY M. MCK.

.2794E-03 .9533E+05 85.72 .6728.0 .4727E-01
.1163E-02 .9692E+05 84.72 .11.99 .4727E-01
.1197E-02 .9688E+05 83.56 .152.0 .4727E-01
.2259E-02 .1090E+06 54.99 .442.3 .7433E-02
.2301E-02 .1078E+06 55.00 .511.0 .7433E-02
.2368E-02 .1083E+06 54.33 .376.9 .7433E-02
.3428E-02 .1108E+06 44.30 .70.25 .6968E-02
.3508E-02 .1103E+06 44.32 .138.9 .6968E-02
.4015E-02 .1122E+06 39.68 .878.5 .7092E-02
.4116E-02 .1106E+06 39.87 .427.5 .7092E-02
.5126E-02 .1205E+06 30.65 .333.2 .5729E-02
.5161E-02 .1183E+06 30.64 .25.29 .5729E-02
.5184E-02 .1212E+06 29.61 .394.2 .5729E-02
.5279E-02 .1252E+06 28.04 .201.9 .5729E-02
.6283E-02 .1379E+06 20.72 .395.4 .5150E-02
.6385E-02 .1449E+06 18.20 .1969.9 .5150E-02
.7366E-02 .1781E+06 12.92 .1945.5 .1520E-02
.7628E-02 .1685E+06 12.80 .2393.1 .1520E-02
.8635E-02 .2107E+06 10.21 .217.9 .1520E-02
.8920E-02 .2038E+06 10.29 .3106.7 .1520E-02
.9198E-02 .2279E+06 8.752 .544.7 .1520E-02
.1042E-01 .2625E+06 7.537 .1134.0 .1520E-02

file SRP11.DAT
2701 3 0 49 5 .6885E+06 266.4 .2242E+05 .1500
1982 270 COBRA PROBE SURVEYS 24,40,2-3,51MS, ON FLOOR BY M. MCK.

.2794E-03 .1597E+06 34.58 .5082.2 .3568E-02
.1260E-02 .1835E+06 34.59 .1877.7 .3568E-02
.1356E-02 .1874E+06 34.57 .355.2 .3568E-02
.1605E-02 .1903E+06 33.45 .496.5 .3568E-02
.1848E-02 .1907E+06 32.58 .1733.0 .3568E-02
.2349E-02 .1952E+06 30.73 .1638.6 .3568E-02
.2462E-02 .1937E+06 30.89 .948.6 .3568E-02
.3477E-02 .2084E+06 25.10 .190.2 .2981E-02
.3572E-02 .2093E+06 25.11 .1987.2 .2981E-02
.3678E-02 .2097E+06 25.52 .2004.6 .2981E-02
.3894E-02 .2115E+06 23.16 .2829.0 .2981E-02
.4900E-02 .2331E+06 19.14 .861.9 .1912E-02
.5005E-02 .2387E+06 19.10 .218.3 .1912E-02
.5515E-02 .2504E+06 16.33 .3374.0 .1912E-02
.5750E-02 .2581E+06 16.05 .1273.0 .1912E-02
.6284E-02 .2651E+06 14.67 .2069.9 .1912E-02
.6331E-02 .2722E+06 14.61 .2920.0 .1912E-02
.7522E-02 .2989E+06 11.87 .1831.0 .1912E-02
.7993E-02 .3190E+06 11.41 .416.3 .1912E-02
.9017E-02 .3582E+06 10.00 .2919.9 .1912E-02
.1002E-01 .3840E+06 9.087 .973.5 .1912E-02
.1099E-01 .4131E+06 8.655 .2276.6 .1912E-02
.1196E-01 .4264E+06 8.278 .284.9 .1912E-02
.1299E-01 .4346E+06 8.129 .3477.7 .1912E-02
.1396E-01 .4344E+06 8.337 .1609.0 .1912E-02
.1497E-01 .4341E+06 8.056 .759.1 .1912E-02
.1597E-01 .4309E+06 8.150 .394.2 .1912E-02
.1697E-01 .4285E+06 7.717 .3477.7 .1912E-02
.1793E-01 .4277E+06 7.804 .177.8 .1912E-02
.1891E-01 .4186E+06 8.128 .3354.0 .1912E-02
.1993E-01 .3991E+06 8.449 .3397.7 .1912E-02
.2044E-01 .3867E+06 8.140 .1026.6 .1912E-02
.2138E-01 .3511E+06 6.720 .2997.7 .1912E-02
.2163E-01 .3475E+06 6.792 .1313.0 .1912E-02
.2214E-01 .3439E+06 5.342 .1703.0 .1912E-02
.2315E-01 .3352E+06 4.155 .3110.0 .1912E-02
.2336E-01 .3225E+06 4.298 .1276.6 .1912E-02
.2394E-01 .2643E+06 5.420 .1465.5 .1912E-02
.2394E-01 .2691E+06 5.291 .734.6 .1912E-02
.2504E-01 .2414E+06 4.670 .322.7 .1912E-02
.2504E-01 .2253E+06 .2375.0 .2659.0 .1520E-02
.2598E-01 .2265E+06 .2131.0 .2297.7 .1520E-02
.2696E-01 .2261E+06 .1466.0 .1244.0 .1520E-02
.2794E-01 .2262E+06 .1335.0 .1983.0 .1520E-02
.2895E-01 .2270E+06 .1259.0 .1434.0 .1520E-02
.2985E-01 .2270E+06 .1282.0 .264.3 .1520E-02
.3081E-01 .2268E+06 .1350.0 .3047.7 .1520E-02
.3174E-01 .2276E+06 .1206.0 .2980.0 .1520E-02
.3272E-01 .2279E+06 .1175.0 .3351.0 .1520E-02

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File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

File STATSM.DAT

2003 1 0 46 5 .6920E+06 263.1 .2361E+05 1.000  
1982 272 CORBA PROBE SURVEYS 24,40,2-3.5IMS. ON FLOOR BY M.MCK.

# Report Documentation Page

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15. Supplementary Notes Point of Contact: C. C. Horstman, Ames Research Center, MS 229-1, Moffett Field, CA 94035-1000 (415) 604-6255 or FTS 464-6255					
16. Abstract  The present database collection and assessment effort with respect to shock/boundary layer interactions include both supersonic (M 3 and above) and hypersonic data, both two-dimensional (2-D) and three-dimensional (3-D) data, and both unseparated and separated turbulent boundary layer cases (though the emphasis is on the latter). Consideration also includes not only perfect-gas behavior, but real gases and (where appropriate) chemically-reacting flows as well. It is recognized, however, that very little data of the latter two types exist within the chosen subject area.					
17. Key Words (Suggested by Author(s)) Hypersonic Shock-wave Turbulent boundary layer Shock-wave/boundary-layer interaction				18. Distribution Statement Unclassified-Unlimited  Subject Category - 34	
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